

Finite element formulation for compressible multiphase flows and its application to pyroclastic gravity currents

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Abstract

The numerical modeling of compressible multiphase flows is of high interest for several engineering applications. In this work, we focus on the study of pyroclastic flows arising from volcanic eruptive events. An accurate evaluation of the effects of this multiphase flow is of crucial importance for the Civil Protection for the preservation of urban settlements in volcanic areas. In this work, we propose a Finite Element formulation for the simulation of pyroclastic flows at conditions of thermal and kinetic equilibrium. This analysis belongs to the class of advection dominated problems, which are known to suffer from numerical instabilities. The required stabilization is provided by using a Variational Multiscale Method, typically used for monophasic compressible flows and extended here for the first time to multiphase flows. The stabilized formulation is validated with benchmark problems for compressible flows, which are solved for both monophasic and multiphase cases. A throughout comparison of the numerical results of the two different flows is also presented. Moreover, the numerical formulation is applied to the simulation of representative cases of pyroclastic flows considering large-scale computational domains and realistic material properties and initial thermal-kinematic conditions. The numerical analyses presented show the accuracy

of the proposed method for the simulation of compressible multiphase flows and its suitability for risk assessment studies of urban settlements prone to be affected by pyroclastic gravity currents.

Keywords: Multiphase flows, pyroclastic flows, Variational Multiscale stabilization

1. Introduction

The numerical simulation of compressible multiphase flows is a topic of great relevance for multiple engineering applications, such as air pollution dispersion, transport phenomena, particulate flows, and the dynamic of non-mixable fluids.

Another application of interest is the simulation of pyroclastic flows originated from explosive eruptive events. Pyroclastic flows consist of a dilute mixture of gases and solid particulate [1, 2] and their study is of particular relevance for institutions of Civil Protection, which need to quantify the risk associated with volcanic eruptive events, assess the vulnerability of the built environment [3, 4], design the emergency plans [5], and implement the best engineering countermeasures (including the design of *ad hoc* systems for building protection, such as panels and resistant fixtures [6]). For a correct realization of these tasks, it is of paramount importance to value accurately the potential effects of pyroclastic flows on the urban environment and estimate quantities of interest, such as the pressure and temperature values acting on civil structures and infrastructures [7].

The numerical simulation of pyroclastic volcanic eruptions has been generally focused on the phenomena occurring at the volcano level. Different numerical approaches and formulations have been proposed in the literature for this specific objective. Fundamentals about the description of multiphase flows can be found in [8]. Following this work, in [9, 10] the conservation laws are written in terms of densities, velocity, enthalpy, and pressure, using implicit solvers to couple the different variables through the equation of state. On the other hand, in [11] the same problem is approached in terms of conservative variables (density, momentum and total energy).

Differently from the mentioned works, here we focus on the study of pyroclastic flows at the scale of the urban settlements [12]. In particular, we analyze the effect of the pyroclastic flow at a distance from the volcano such that the condition of thermal and kinetic equilibrium can be assumed

31 [13, 14, 15]. This hypothesis allows us to reduce the number of unknown
32 variables and to neglect the terms related to the drag and heat transfer
33 between different phases of the mixture.

34 The resulting set of governing equations has formal similarity with the
35 case of the compressible monophase flow. It is well known that the finite
36 element solution of this kind of problem leads to numerical instabilities due
37 to the dominance of the convective term [16, 17, 18]. In the literature, several
38 stabilization methods for advection dominant problems have been proposed,
39 basing either on algebraic arguments, such as in [19, 20, 21], or on the in-
40 capacity of the Finite Element Method (FEM) to represent phenomena at
41 a smaller scale than the discretization size. The latter is the case of the
42 Variational Multiscale (VMS) technique [22], which separates variables and
43 test functions on the part that can be represented at the finite elements
44 scale and the remainder that has to be suitably approximated. The VMS
45 stabilization method has been applied to different kinds of analyses, such as
46 advection-diffusion-reaction problems [23], incompressible [24, 25] and com-
47 pressible Navier-Stokes equations [26, 27, 28], and Burger’s equation [29].
48 Nevertheless, to the best of the Authors’ knowledge, so far the VMS method
49 has been applied only to compressible monophase flows. In this work, we
50 extend its application to compressible multiphase flows, *i.e.* dilute gas-solid
51 mixtures.

52 Among the large variety of stabilization methods introduced in the frame-
53 work of the VMS techniques, we use here the quasi-static Algebraic-Sub Grid
54 Scales (ASGS, see [30]), leaving the development of different VMS stabiliza-
55 tion methods for future work.

56 In order to deal with the shock waves that may appear in compressible
57 flows at the supersonic regime and to avoid undesired nonphysical oscilla-
58 tions of the FEM solution across discontinuities, the numerical method is
59 also provided with a shock-capturing technique. To this end, the method
60 proposed by [31] for monophase problems and also used in [32, 33], is here
61 extended to the multiphase case.

62 Concerning the time marching scheme, considering the high non-linearity
63 of the problem and the need for a fast and robust algorithm for large-scale
64 computations, an explicit approach was considered preferable to an implicit
65 one. This allows to avoid problems of convergence, to limit the memory
66 occupation, and to ease the code parallelization. These features are of crucial
67 importance to enable the solution of large-scale problems, as those addressed
68 in this work.

69 Several numerical examples are presented to show the validity of the
70 method and its suitability for risk assessment studies of urban settlements
71 potentially affected by pyroclastic flows. In particular, first, we analyze accu-
72 racy, convergence and robustness of the stabilized method in the solution of
73 benchmark problems for compressible flows, both in supersonic and subsonic
74 regimes and for both monophase and multiphase flows. Then, we apply the
75 proposed method to the simulation of two representative problems of pyro-
76 clastic gravity currents impacting against civil constructions. In one of these
77 tests, we consider a large scale three-dimensional geometry and realistic ma-
78 terial properties and boundary conditions, mimicking an actual pyroclastic
79 flow scenario. A quantitative estimation of the damages produced by the py-
80 roclastic flow over the civil constructions is also given, basing on the criteria
81 for damages estimation provided in [4].

82 We also remark that, given the lack of reference solutions for compress-
83 ible multiphase analysis, the present work also provides benchmark cases
84 solutions that can be useful for comparison purposes with other numerical
85 strategies.

86 The paper is organized as follows. In Section 2, we introduce the gov-
87 erning equations, specifying the hypotheses at the base of the formulation.
88 In Section 3, we derive the stabilized finite element method used for the
89 multiphase compressible flows. Specifically, in Section 3.1, we derive the
90 weak form, in Section 3.2, we extend the application of the VMS stabiliza-
91 tion method to the case of multiphase flows, in Section 3.3, we present the
92 adopted shock-capturing technique, and finally, in Section 3.4, we derive the
93 fully discretized form of the problem. Section 4 is fully devoted to present the
94 numerical examples solved with the proposed formulation. Finally, in Sec-
95 tion 5, we discuss the results obtained and highlight the main contributions
96 of the work.

97 2. Governing equations

98 The dynamic of multiphase flows is described by the equations of conser-
99 vation of mass, linear momentum, and total energy. In this work, we focus
100 on the study of a two-phase mixture with one gas and one solid phase.

101 Following [11], the dynamic of the gas phase is described as

$$\frac{\partial}{\partial t}(\varepsilon_g \rho_g) + \frac{\partial}{\partial x_i}(\varepsilon_g \rho_g u_{g,i}) = 0 \quad (1a)$$

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$$\frac{\partial}{\partial t}(\varepsilon_g \rho_g u_{g,i}) + \frac{\partial}{\partial x_j}(\varepsilon_g \rho_g u_{g,i} u_{g,j} + \varepsilon_g p_g) = p_g \frac{\partial \varepsilon_g}{\partial x_i} + \frac{\partial \tau_{g,ij}}{\partial x_j} + \varepsilon_g \rho_g b_{g,i} - D(u_{g,i} - u_{s,i}) \quad (1b)$$

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$$\begin{aligned} \frac{\partial}{\partial t}(\varepsilon_g \rho_g e_g) + \frac{\partial}{\partial x_i}((\varepsilon_g \rho_g e_g + \varepsilon_g p_g) u_{g,i}) &= -p_g \frac{\partial}{\partial x_i}(\varepsilon_s u_{s,i}) + u_{g,i} \frac{\partial}{\partial x_j} \tau_{g,ij} \\ &- \frac{\partial}{\partial x_i} q_{g,i} + \varepsilon_g \rho_g b_{g,i} u_{g,i} + \varepsilon_g \rho_g r_g - D(u_{g,i} - u_{s,i}) u_{g,i} - Q(T_g - T_s) \end{aligned} \quad (1c)$$

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where t is the time, x_i are the spatial coordinates, ε_g and ε_s are the phase concentrations for the gas and the solid phase, respectively, being $\varepsilon_g + \varepsilon_s = 1$, ρ_g is the gas density, \mathbf{u}_g and \mathbf{u}_s are the velocities of the gas and solid phases, respectively, p_g is the thermodynamic pressure of the gas phase, e_g is the specific energy of the gas phase, T_g and T_s are the temperatures of the two phases, $\boldsymbol{\tau}_g$ and \mathbf{q}_g are the stress tensor and the heat flow vector in the gas phase, respectively, \mathbf{b}_g and r_g are the body force and the heat source for the gas phase, respectively. Finally, $D(u_{g,i} - u_{s,i})$ and $Q(T_g - T_s)$ are dissipative terms that account for momentum and energy exchanges between the two phases.

Similarly, the equations of conservation for the solid phase read

$$\frac{\partial}{\partial t}(\varepsilon_s \rho_s) + \frac{\partial}{\partial x_i}(\varepsilon_s \rho_s u_{s,i}) = 0 \quad (2a)$$

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$$\frac{\partial}{\partial t}(\varepsilon_s \rho_s u_{s,i}) + \frac{\partial}{\partial x_j}(\varepsilon_s \rho_s u_{s,i} u_{s,j}) = -\varepsilon_s \frac{\partial p_g}{\partial x_i} + \frac{\partial \tau_{s,ij}}{\partial x_j} + \varepsilon_s \rho_s b_{s,i} + D(u_{g,i} - u_{s,i}) \quad (2b)$$

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$$\begin{aligned} \frac{\partial}{\partial t}(\varepsilon_s \rho_s e_s) + \frac{\partial}{\partial x_i}(\varepsilon_s \rho_s e_s u_{s,i}) &= -\varepsilon_s u_{s,i} \frac{\partial p_g}{\partial x_i} + u_{s,i} \frac{\partial}{\partial x_j} \tau_{s,ij} \\ &- \frac{\partial}{\partial x_i} q_{s,i} + \varepsilon_s \rho_s b_{s,i} u_{s,i} + \varepsilon_s \rho_s r_s + D(u_{g,i} - u_{s,i}) u_{s,i} + Q(T_g - T_s) \end{aligned} \quad (2c)$$

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In Equations (2), ρ_s is the density of the solid phase, e_s is the specific energy of the solid phase, $\boldsymbol{\tau}_s$ and \mathbf{q}_s are the stress tensor and the heat flow vector in the solid phase, respectively, \mathbf{b}_s is the vector of body forces applied on the solid phase, and r_s are energy sources due to the solid phase.

The energy terms e_g and e_s comprise a thermal and a kinetic contribution and are computed as

$$e_g = c_v T_g + \frac{1}{2} u_{g,i} u_{g,i} \quad (3a)$$

$$e_s = c_s T_s + \frac{1}{2} u_{s,i} u_{s,i} \quad (3b)$$

where c_v is the specific heat at constant volume of the gas, and c_s is the specific heat of the solid material.

Problems (1)-(2) are closed by the equation of state for ideal gases, that reads

$$p_g = \rho_g R T_g \quad (4)$$

being R the specific constant of the gas, and by the incompressibility of the solid particles, expressed by $\rho_s = \text{const}$.

We highlight that system (1)-(2) can be considered as a particular case of the Baer-Nunziato [34, 35, 36, 37, 38, 39, 40] (or Saurel-Abgrall [41]) model, which holds also for high values of phase concentrations or concentration gradients. On the other hand, our approach considers extremely dilute mixtures, where the solid concentration is limited, *i.e.*, $\varepsilon_s < 10^{-2}$. Under these conditions, the Baer-Nunziato model can be simplified as system (1)-(2). Furthermore, for such dilute mixtures, the terms $p_g \partial \varepsilon_g / \partial x_i$ and $-p_g \partial (\varepsilon_s u_{s,i}) / \partial x_i$ of Equations (1), and the terms $-\varepsilon_s \partial p_g / \partial x_i$ and $-\varepsilon_s u_{s,i} \partial p_g / \partial x_i$ of Equations (2) can be dropped, and the total pressure can be defined as $p = \varepsilon_g p_g$ [11]. Therefore, Equations (1) and (2) can be simplified into

$$\frac{\partial}{\partial t}(\varepsilon_g \rho_g) + \frac{\partial}{\partial x_i}(\varepsilon_g \rho_g u_{g,i}) = 0 \quad (5a)$$

$$\frac{\partial}{\partial t}(\varepsilon_g \rho_g u_{g,i}) + \frac{\partial}{\partial x_j}(\varepsilon_g \rho_g u_{g,i} u_{g,j} + p) = \frac{\partial \tau_{g,ij}}{\partial x_j} + \varepsilon_g \rho_g b_{g,i} - D(u_{g,i} - u_{s,i}) \quad (5b)$$

$$\frac{\partial}{\partial t}(\varepsilon_g \rho_g e_g) + \frac{\partial}{\partial x_i}((\varepsilon_g \rho_g e_g + p)u_{g,i}) = u_{g,i} \frac{\partial}{\partial x_j} \tau_{g,ij} - \frac{\partial}{\partial x_i} q_{g,i} + \varepsilon_g \rho_g b_{g,i} u_{g,i} + \varepsilon_g \rho_g r_g - D(u_{g,i} - u_{s,i})u_{g,i} - Q(T_g - T_s) \quad (5c)$$

$$\frac{\partial}{\partial t}(\varepsilon_s \rho_s) + \frac{\partial}{\partial x_i}(\varepsilon_s \rho_s u_{s,i}) = 0 \quad (5d)$$

$$\frac{\partial}{\partial t}(\varepsilon_s \rho_s u_{s,i}) + \frac{\partial}{\partial x_j}(\varepsilon_s \rho_s u_{s,i} u_{s,j}) = \frac{\partial \tau_{s,ij}}{\partial x_j} + \varepsilon_s \rho_s b_{s,i} + D(u_{g,i} - u_{s,i}) \quad (5e)$$

$$\begin{aligned} \frac{\partial}{\partial t}(\varepsilon_s \rho_s e_s) + \frac{\partial}{\partial x_i}(\varepsilon_s \rho_s e_s u_{s,i}) &= u_{s,i} \frac{\partial}{\partial x_j} \tau_{s,ij} - \frac{\partial}{\partial x_i} q_{s,i} + \\ \varepsilon_s \rho_s b_{s,i} u_{s,i} + \varepsilon_s \rho_s r_s + D(u_{g,i} - u_{s,i}) u_{s,i} + Q(T_g - T_s) \end{aligned} \quad (5f)$$

145 *2.1. Multiphase flows in thermal and kinetic equilibrium*

146 Equations (5) are explicitly coupled by the dissipative terms $D(u_{g,i} - u_{s,i})$
 147 and $Q(T_g - T_s)$ that vanish when the velocities and the temperatures of
 148 the two phases are the same. The effect of both dissipative terms is to
 149 homogenize the temperature and velocity fields: the time and space needed
 150 for reaching this effect depend on the characteristics of the flow and the size
 151 of the particles [9, 42, 43]. In particular, a particle of mass m and radius σ ,
 152 inserted in a flow with velocity v , adapts to the fluid velocity in a time

$$\tau_u = \frac{m}{6\pi\sigma\mu_g} \quad (6)$$

153 where μ_g is the viscosity of the gas [42]. Similarly, the time needed for a
 154 particle at a temperature T_p to adapt to the flow temperature T_f is

$$\tau_T = \frac{mc_p}{4\pi\sigma\lambda_g} \quad (7)$$

155 being c_p the specific heat at constant pressure of the gas, and λ_g is the thermal
 156 diffusivity [42].

157 When $t > \tau_u$ and $t > \tau_T$, we can consider that the condition of thermal
 158 and kinetic equilibrium is fulfilled [15, 42]. In this situation, the solid and
 159 the gas phases are assumed to have the same velocity and temperature fields.
 160 This hypothesis allows us to set $\mathbf{u}_g = \mathbf{u}_s = \mathbf{u}$ and $T_g = T_s = T$ in Equations
 161 (5). Using this and summing up the equations of conservation of mass, mo-
 162 mentum, and total energy of the two phases (Eq. 5), we obtain the following
 163 compact form of the conservation equations for the solid-gas mixture

$$\frac{\partial}{\partial t} \rho + \frac{\partial}{\partial x_i} (\rho u_i) = 0 \quad (8a)$$

$$\frac{\partial}{\partial t} (\rho u_i) + \frac{\partial}{\partial x_j} (\rho u_i u_j + p) = \frac{\partial \tau_{ij}}{\partial x_j} + \rho b_i \quad (8b)$$

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$$\frac{\partial}{\partial t}E + \frac{\partial}{\partial x_i}((E + p)u_i) = u_i \frac{\partial}{\partial x_j} \tau_{ij} - \frac{\partial}{\partial x_i} q_i + \rho b_i u_i + \rho r \quad (8c)$$

166 where we can recognize the mixture mass conservation equation (Eq. (8a)),
 167 the mixture momentum conservation equation (Eq. (8b)), and the the mix-
 168 ture energy conservation equation (Eq. (8c)). Finally, the mass conservation
 169 of the solid phase is guaranteed by retaining Equation (5d), which is rewritten
 170 in terms of the mixture velocity as

$$\frac{\partial}{\partial t}(\varepsilon_s \rho_s) + \frac{\partial}{\partial x_i}(\varepsilon_s \rho_s u_i) = 0 \quad (9)$$

171 In Equations (8)-(9), $\rho = \varepsilon_g \rho_g + \varepsilon_s \rho_s$ is the total density of the mixture,
 172 $E = \varepsilon_g \rho_g e_g + \varepsilon_s \rho_s e_s$ is the total energy of the mixture, $\boldsymbol{\tau}$ and \mathbf{q} are suitable
 173 compositions of the stress tensors and the heat flows of the components of
 174 the mixture, and \mathbf{b} and r are source terms.

175 The viscous term of the mixture is assumed as the usual one adopted for
 176 Newtonian fluids [42], that is

$$\tau_{ij} = \mu_m \left[\left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) - \frac{2}{3} \frac{\partial u_k}{\partial x_k} \delta_{ij} \right] \quad (10)$$

177 where the mixture dynamic viscosity μ_m is computed as [42]

$$\mu_m = \frac{\mu_g}{1 + \kappa} \quad (11)$$

178 being μ_g the dynamic viscosity of the gas, and

$$\kappa = \frac{\varepsilon_s \rho_s}{\varepsilon_g \rho_g} \quad (12)$$

179 It is worth remarking that, since $\kappa > 0$, the equivalent viscosity of a mul-
 180 tiphase flow is lower than the one in a monophasic case. As a consequence,
 181 for the same velocity field, a multiphase analysis is expected to have a higher
 182 Reynolds number than the respective monophasic problem and turbulent phe-
 183 nomena are expected to appear earlier.

184 Similarly, the heat flux for the mixture is written according to the Fourier
 185 law, as $q_i = -\lambda_m \partial T / \partial x_i$, being λ_m the mixture thermal diffusivity expressed
 186 as

$$\lambda_m = \frac{\lambda_g}{1 + \kappa} \quad (13)$$

187 where λ_g is the thermal diffusivity of the gas.

188 2.2. Conservative form

189 For the explicit solution of problem (8)-(9) it is convenient to write all
190 equations in terms of a set of conservative variables as

$$\frac{\partial}{\partial t} \mathbf{U} + \frac{\partial}{\partial x_i} \mathbf{F}_i = \frac{\partial}{\partial x_i} \mathbf{G}_i + \mathbf{S} \quad (14)$$

191 where \mathbf{U} is the set of conservative variables, \mathbf{F} is the matrix of the convective
192 flux, \mathbf{G} is the dissipative flux, and \mathbf{S} is the source term.

193 For a two-phase mixture, the vector of conservative variables is

$$\mathbf{U} = [\rho, P_s, M_i, E]^T \quad (15)$$

194 where $P_s = \varepsilon_s \rho_s$ is the contribution of the solid part to the total density of
195 the mixture, and $M_i = \rho u_i$ is the linear momentum of the mixture. The
196 corresponding convective and dissipative fluxes are then

$$\mathbf{F} = \begin{bmatrix} M_j \\ \frac{P_s}{\rho} M_j \\ M_i \frac{M_j}{\rho} + p \delta_{ij} \\ \frac{M_j}{\rho} (E + p) \end{bmatrix}, \quad \mathbf{G} = \begin{bmatrix} 0_j \\ 0_j \\ \tau_{ij} \\ u_i \tau_{ij} + q_j \end{bmatrix} \quad (16)$$

197 where δ_{ij} is the Kronecker delta. The variable p can be expressed in terms
198 of the conservative variables as

$$p = \frac{R(\rho - P_s)}{c_m} \left(E - \frac{1}{2} \frac{M_i M_i}{\rho} \right) \quad (17)$$

199 where

$$c_m = P_s c_s + (\rho - P_s) c_v \quad (18)$$

200 is the specific heat of the mixture. Finally, the source vector is

$$\mathbf{S} = [0, 0, \rho b_i, M_j b_j + \rho r]^T \quad (19)$$

201 that can be conveniently written as $\mathbf{S} = \mathbf{T}\mathbf{U}$, where

$$\mathbf{T} = \begin{bmatrix} 0 & 0 & 0_j & 0 \\ 0 & 0 & 0_j & 0 \\ b_i & 0_i & 0_{ij} & 0_i \\ r & 0 & b_j & 0 \end{bmatrix} \quad (20)$$

202 Other fields of interest can be retrieved from the conservative variables.
 203 Among them we mention the temperature

$$T = \frac{p}{\rho_g R} = \frac{1}{c_m} \left(E - \frac{1}{2} \frac{M_i M_i}{\rho} \right) \quad (21)$$

204 and the speed of sound (expressed as [11])

$$a_m = a_g \sqrt{\frac{\rho_g}{\gamma \varepsilon_g \rho}} \quad (22)$$

205 being a_g the speed of sound in the monophasic gas, and γ the heat capacity
 206 ratio of the gas. For extremely dilute mixtures (*i.e.* $\varepsilon_g \simeq 1$), Equation (22)
 207 reduces to

$$a_m = a_g \sqrt{\frac{\rho - P_s}{\gamma \rho}} \quad (23)$$

208 It is interesting to notice that, since $\sqrt{(\rho - P_s)/\gamma \rho} < 1$, in dilute mixtures
 209 the speed of sound is lower than in monophasic flows.

210 Another variable of interest is the Mach number $Ma = u/a$, which gives
 211 information about the compressibility regime of the mixture. It is worth
 212 noting that, for the same velocity field, a mixture flow has a higher Mach
 213 number with respect to the monophasic flow, since the speed of sound is lower
 214 in mixture than in the monophasic flow.

215 3. Stabilized FEM formulation

216 3.1. Weak form

217 Using the Euler-Jacobian matrix $A_{ijk} = A_{ijk}(\mathbf{U}) = \frac{\partial F_{ij}}{\partial U_k}$, problem (14) is
 218 rewritten as

$$\frac{\partial U_i}{\partial t} + A_{ijk} \frac{\partial U_k}{\partial x_j} = \frac{\partial G_{ij}}{\partial x_j} + T_{ik} U_k \quad (24)$$

219 Let us consider a domain $\Omega \in \mathbb{R}^3$ and a discretization in a finite number
 220 of elements Ω_h . Premultiplying Equation (24) by the test function vector \mathbf{V}
 221 and integrating over Ω_h , we get the weak form as

$$\left(V_i, \frac{\partial U_i}{\partial t} \right) + (V_i, L_{ik} U_k) - \left(V_i, \frac{\partial G_{ij}}{\partial x_j} \right) = 0 \quad \forall V_i \in \mathbb{W} \quad (25)$$

222 where \mathbb{W} is the space of the variables and the test functions. The non-linear
 223 operator L_{ik} is defined as

$$L_{ik}(\mathbf{U}) = A_{ijk}(\mathbf{U}) \frac{\partial}{\partial x_j} - T_{ik} \quad (26)$$

224 We remark that in Eq. 25 we have also introduced the notation $(a, b) =$
 225 $\int_{\Omega_h} abd\Omega_h$.

226 3.2. Variational Multiscale stabilization

227 It is well known that a standard Galerkin formulation for advection dom-
 228 inated problems suffers from instabilities that result in spurious numerical
 229 results [16, 17, 18].

230 In this work, we stabilize the numerical formulation using the Variational
 231 Multiscale Method [21, 22]. This stabilization procedure is designed to ac-
 232 count for the effects of the subgrid scale, *i.e.* those effects that cannot be
 233 captured at the scale of the finite elements. This is achieved by splitting the
 234 space of the conservative variables and test functions as $\mathbb{W} = \mathbb{W}^h + \tilde{\mathbb{W}}$, being
 235 \mathbb{W}^h the space of the functions that can be represented by standard finite
 236 elements, and $\tilde{\mathbb{W}}$ the space of the subscale.

237 Following the standards VMS technique, Equation (25) is therefore pro-
 238 jected onto the spaces W^h and $\tilde{\mathbb{W}}$, getting two different sets of equations

$$\left(V_i^h, \frac{\partial(U_i^h + \tilde{U}_i)}{\partial t} \right) + \left(V_i^h, L_{ik}(U_k^h + \tilde{U}_k) \right) - \left(V_i^h, \frac{\partial}{\partial x_j} G_{ij}(\mathbf{U}^h + \tilde{\mathbf{U}}) \right) = 0, \quad \forall V^h \in \mathbb{W}^h \quad (27a)$$

$$\left(\tilde{V}_i, \frac{\partial(U_i^h + \tilde{U}_i)}{\partial t} \right) + \left(\tilde{V}_i, L_{ik}(U_k^h + \tilde{U}_k) \right) - \left(\tilde{V}_i, \frac{\partial}{\partial x_j} G_{ij}(\mathbf{U}^h + \tilde{\mathbf{U}}) \right) = 0, \quad \forall \tilde{V} \in \tilde{W} \quad (27b)$$

239 *Finite element scale.* Let us focus before on Equation (27a) referring to the
 240 finite element scale. We separate the terms computed at the finite element
 241 scale from the ones computed at the subgrid scale, neglect the contribution
 242 of the time derivative of the subgrid scale $\partial\tilde{\mathbf{U}}/\partial t$ (hypothesis of quasi-static
 243 subscales [27]) as well as the contribution of the subscale to the dissipative
 244 term $\partial G_{ij}(\tilde{\mathbf{U}})/\partial x_j$, since it involves the calculation of higher order derivatives
 245 that vanish for linear elements. Therefore, we obtain

$$\left(V_i^h, \frac{\partial U_i^h}{\partial t} \right) + (V_i^h, L_{ik}U_k^h) + (V_i^h, L_{ik}\tilde{U}_k) - \left(V_i^h, \frac{\partial G_{ij}(\mathbf{U}^h)}{\partial x_j} \right) = 0, \quad \forall V^h \in \mathbb{W}^h \quad (28)$$

246 After integrating by parts the third term of Equation (28) and some
 247 algebra, we obtain

$$\left(V_i^h, \frac{\partial U_i^h}{\partial t} + L_{ik}U_k^h \right) - (L_k^*, \tilde{U}_k) + \left(\frac{\partial V_i^h}{\partial x_j}, G_{ij}(\mathbf{U}^h) \right) = 0, \quad \forall V^h \in \mathbb{W}^h \quad (29)$$

248 where

$$L_k^* = \frac{\partial V_i^h}{\partial x_j} A_{ijk} + V_i B_{ijkm} \frac{\partial U_m^h}{\partial x_j} - V_i T_{ik} \quad (30)$$

249 is the adjoint term, and

$$B_{ijkm} = \frac{\partial A_{ijk}}{\partial U_m^h} \quad (31)$$

250 Equation (29) contains all terms that can be computed at the finite ele-
 251 ment scale, with exception of the subgrid scale term \tilde{U} , that is obtained from
 252 Equation (27b).

253 *Subgrid scale.* Focusing now on the subgrid scale, Equation (27b) is rewritten
 254 neglecting the time variation of the subscale and its contribution to dissipa-
 255 tion as

$$\left(\tilde{V}_i, L_{ik}\tilde{U}_k\right) = - \left(\tilde{V}_i, \frac{\partial U_i^h}{\partial t} + L_{ik}U_k^h\right), \quad \forall \tilde{V} \in \tilde{W} \quad (32)$$

256 One of the key points of the VMS technique is the choice of the approxi-
 257 mation of the non-linear operator applied to $\tilde{\mathbf{U}}$. Following the standard VMS
 258 practice, we used

$$L_{ik}\tilde{U}_k = \tau_{ik}^{-1}(\mathbf{U})\tilde{U}_k \quad (33)$$

259 where $\boldsymbol{\tau}$ is a diagonal matrix that depend on the characteristics of the flow

$$\boldsymbol{\tau}^{-1} = \begin{bmatrix} \tau_\rho^{-1} & 0 & 0 & 0 \\ 0 & \tau_\rho^{-1} & 0 & 0 \\ 0 & 0 & \tau_M^{-1} & 0 \\ 0 & 0 & 0 & \tau_E^{-1} \end{bmatrix} \quad (34)$$

260 with

$$\tau_\rho^{-1} = c_2 \frac{\|\mathbf{u}\| + c_m}{h}, \quad \tau_M^{-1} = c_1 \frac{\nu}{h^2} + c_2 \frac{\|\mathbf{u}\| + c_m}{h}, \quad \tau_E^{-1} = c_1 \frac{\lambda_m}{\rho c_p h^2} + c_2 \frac{\|\mathbf{u}\| + c}{h} \quad (35)$$

261 being c_1 and c_2 algorithmic constants (in this work, $c_1 = 4$ and $c_2 = 2$) and
 262 h the characteristic length of the element.

263 The coefficients of $\boldsymbol{\tau}^{-1}$, proposed in [27] for a monophasic compressible
 264 flow, are here adapted for multiphase flows replacing the gas constants with
 265 the parameters of the mixture. In particular, the speed of the sound c is
 266 replaced with the speed of the sound of the mixture c_m , and the specific heat
 267 at constant pressure c_p is replaced by the corresponding value of the mixture.
 268 Basing on [42], this leads to

$$c_{p,m} = \frac{c_{p,g} + \kappa c_s}{1 + \kappa} \quad (36)$$

269 To solve Equation (32) we also need to choose a proper space for the
 270 projection of the residuals. Following the algebraic subgrid scale projection
 271 (ASGS) [30], we project the adjoint term L_k^* onto the space of the residuals,
 272 obtaining

$$\tau_{ij}^{-1}\tilde{U}_j = R_i^h \quad (37)$$

273 where \mathbf{R}^h is the residual at the scale of the finite elements.

274 Introducing Equation (37) into Equation (29), we obtain

$$\left(V_i^h, \frac{\partial U_i}{\partial t} + L_{ik} U_k \right) - (L_k^*, \tau_{kj} R_j^h) + \left(\frac{\partial V_i^h}{\partial x_j}, G_{ij}(\mathbf{U}_h) \right) = 0 \quad (38)$$

275 where all terms are evaluated at the scale of the finite element.

276 3.3. Discontinuity capturing technique

277 In presence of strong discontinuities in the unknown variable fields, the
278 finite element solution can give rise to spurious oscillations across the dis-
279 continuity.

280 A commonly adopted technique for this issue is the discontinuity cap-
281 turing technique [31, 32, 33]. This procedure is generally adopted for com-
282 pressible flows in a supersonic regime, where shock waves are admitted. The
283 method consists of increasing the viscosity and thermal diffusivity affecting
284 the conservation equations of momentum and total energy, as follows

$$\boldsymbol{\tau}_{DC} = \left(1 + \frac{\rho \nu_{DC}}{\mu} \right) \mathbb{D} \boldsymbol{\tau} \quad (39a)$$

285

$$\mathbf{q}_{DC} = \left(1 + \frac{\rho c_v k_{DC}}{\lambda} \right) \mathbf{K} \mathbf{q} \quad (39b)$$

286 where $\boldsymbol{\tau}_{DC}$ and \mathbf{q}_{DC} are the stress tensor and the heat flux after the inclusion
287 of the numerical diffusivity, respectively, and \mathbb{D} and \mathbf{K} are fourth- and second-
288 order tensors defining the eventual anisotropy of the residual-based shock
289 capturing on the momentum and energy equations, respectively. Finally, the
290 parameters ν_{DC} and k_{DC} are computed as

$$\nu_{DC} = \frac{1}{2} h C_a \frac{\|R_M\|}{\|\nabla M_h\|} \quad \text{if } \|\nabla M_h\| \neq 0, \quad 0 \quad \text{otherwise} \quad (40a)$$

291

$$k_{DC} = \frac{1}{2} h C_a \frac{\|R_E\|}{\|\nabla E_h\|} \quad \text{if } \|\nabla E_h\| \neq 0, \quad 0 \quad \text{otherwise} \quad (40b)$$

292 where R_M and R_E are the residuals of the equations of the momentum and
293 total energy, and C_a is an algorithmic constant that regulates the inten-
294 sity of the added numerical viscosity. More details about the isotropic and
295 anisotropic shock capturing models are given in [29].

296 For multiphase flows, a discontinuity capturing technique is needed also
 297 for the equation of conservation of the mass of the solid phase, to prevent
 298 spurious oscillations that could give nonphysical values of P_s . Thus, following
 299 [44], we include the following diffusive term on the right-hand side of Equation
 300 (9)

$$P_{s,diff} = \nabla \cdot \left(\frac{1}{2} h C_P \frac{R_P}{\|\nabla P_{s,h}\|} \mathbf{D} \nabla P_s \right) \quad \text{if } \|\nabla P_{s,h}\| \neq 0, \quad 0 \quad \text{otherwise} \quad (41)$$

301 where C_P is the algorithmic constant and, in analogy with Equations (40),
 302 R_P is the residual of the equation of conservation of the mass of the solid
 303 phase at the finite element scale. Here we have also introduced the second-
 304 order tensor

$$\mathbf{D} = \alpha_a \mathbf{a} \otimes \mathbf{a} + \alpha_c (\mathbf{I} - \mathbf{a} \otimes \mathbf{a}) \quad (42)$$

305 that regulates the anisotropy of the diffusivity on the solid phase transport
 306 equation, being \mathbf{a} the unit vector in the direction of the flow. α_a and α_c
 307 are algorithmic constants. When $\alpha_a = \alpha_c$ a isotropic diffusivity is obtained,
 308 whereas the anisotropic one arises for $\alpha_a \neq \alpha_c$.

309 In the numerical tests presented in this work, we always consider isotropic
 310 shock-capturing. For all numerical examples, we set $C_a = 0.5$; for monophasic
 311 cases, the constant C_p is set to 0.0, whereas for multiphase cases we set
 312 $C_p = 0.5$.

313 3.4. Space and time discretization

314 Equation (38) is discretized over a simplicial finite element mesh. On
 315 each element, the test functions and the unknowns are computed as

$$V_i^h = N_{ip} \hat{V}_p^h, \quad U_k^h = N_{kp} \hat{U}_p^h \quad (43)$$

316 being \mathbf{N} the matrix of the linear shape functions.

317 After some algebra (whose details are illustrated in Appendix A), we get
 318 to the semi-discrete form

$$\frac{\partial}{\partial t} \hat{U}_i^h = \frac{\hat{F}_i(\mathbf{U}, t)}{m_i} \quad (44)$$

319 where \hat{U}_i^h is the vector of nodal values of the conservative variables, $\hat{F}_i(\mathbf{U}, t)$ is
 320 the vector of nodal forces, and m_i is the nodal mass, computed after lumping
 321 of the consistent mass matrix. The expression of these terms is reported in
 322 the Appendix A.

323 Using an explicit first-order Euler algorithm for the time integration of
 324 Equation (44), we obtain

$$\hat{U}_i^{h,n+1} = \hat{U}_i^{h,n} + \frac{\Delta t}{m_i} \hat{F}_i(\mathbf{U}^{h,n}, t^n) \quad (45)$$

325 where n is the time step number and Δt is the time step increment that is
 326 limited by the following CFL condition

$$\Delta t < \frac{h}{u + c} \quad (46)$$

327 In this work, we adopted an explicit method for its better performances for
 328 large scale computations. We remark that this a key skill for the analysis of
 329 real-world problems of pyroclastic flows. Computational cost reasons are also
 330 behind our choice of a first order Euler scheme instead of other linear multi-
 331 step explicit methods of higher order accuracy (*e.g.* Runge-Kutta methods)
 332 that would require multiple computations of the nodal residuals at each time
 333 step.

334 The implementation of the whole stabilized finite element method has
 335 been carried on in Kratos Multiphysics [45], an open-source coding environ-
 336 ment for multiphysics applications.

337 4. Applications

338 In this section, we present several numerical tests to validate the proposed
 339 formulation for compressible multiphase flows. Before approaching multi-
 340 phase analyses, we assess the accuracy and convergence of the formulation
 341 for the monophasic case by solving some well-known benchmark problems,
 342 such as the Taylor-Green vortex problem [46], the monophasic Sod tube [47],
 343 and a subsonic flow around a cylinder [27]. Some of these tests are then solved
 344 again considering multiphase flows to prove the robustness of the algorithm,
 345 confirm its convergent behavior, and assess its validity for multiphase anal-
 346 yses. In all multiphase examples, we assume that the hypothesis of kinetic
 347 and thermal equilibrium is already fulfilled at the beginning of the simulation.
 348 Finally, we analyze two cases of pyroclastic flows impacting civil buildings.

349 Again, the urban settlement is considered to be placed at a distance from the
 350 eruptive source such that the hypothesis of thermal and kinetic equilibrium
 351 is admissible. The first analysis is performed over a two dimensional domain
 352 and is presented to give an example of parametric study in the framework of
 353 pyroclastic gravity current, the second one is realized on a large-scale three-
 354 dimensional domain and is aimed at showing the applicability of the method
 355 to real-world pyroclastic flow problems.

356 In all the numerical tests, if not differently specified, the material param-
 357 eters of the gas are $c_v = 722$ W/kg K, $c_p = 1010$ W/kg K, while, for the solid
 358 phase, we considered $\rho_s = 1800$ kg/m³, and $c_s = 1300$ W/kg K.

359 4.1. The Taylor Green Vortex

360 The Taylor Green vortex [46] is a well-assessed test in the framework of
 361 fluid dynamics analysis and is here studied to verify accuracy and robustness
 362 of the proposed numerical method. The problem consists in a square domain
 363 $(x, y) \in [0 \ 2\pi] \times [0 \ 2\pi]$ over which a smooth flow is induced by the following
 364 force term

$$\begin{aligned}
 b_x &= \cos x \sin x (F(t))^2 - \cos y \sin x (2\mu F(t) + F'(t)) \\
 b_y &= \sin y \cos y (F(t))^2 + \cos x (2\mu F(t) + F'(t)) \\
 r &= 0
 \end{aligned}
 \tag{47}$$

365 with $F(t) = 0.1 t e^{-2\mu t}$ and $\mu = 0.1 \text{ kg/ms}$.

366 A representation of the body forces at $t = 5$ s is given in Figure 1.

367 Slip conditions are imposed at the whole boundary of the domain and the
 368 initial conditions of the problem are

$$\begin{aligned}
 \rho(x, y, z, t) &= 1 \\
 u_1(x, y, z, t) &= 0 \\
 u_2(x, y, z, t) &= 0 \\
 T(x, y, z, t) &= 0.1
 \end{aligned}
 \tag{48}$$

369 The analytical solution of the problem is

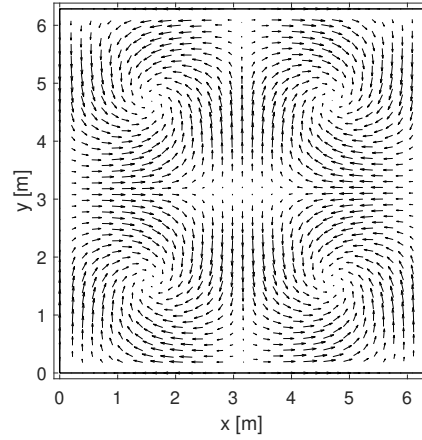


Figure 1: Taylor Green vortex problem: representation of the body forces

$$\begin{aligned}
 \rho(x, y, z, t) &= 1 \\
 u_1(x, y, z, t) &= -\sin x \cos y F(t) \\
 u_2(x, y, z, t) &= \cos x \sin y F(t) \\
 T(x, y, z, t) &= 0.1
 \end{aligned}
 \tag{49}$$

370 In Figure 2 we show the streamlines and velocity contours of the numerical
 371 solution.

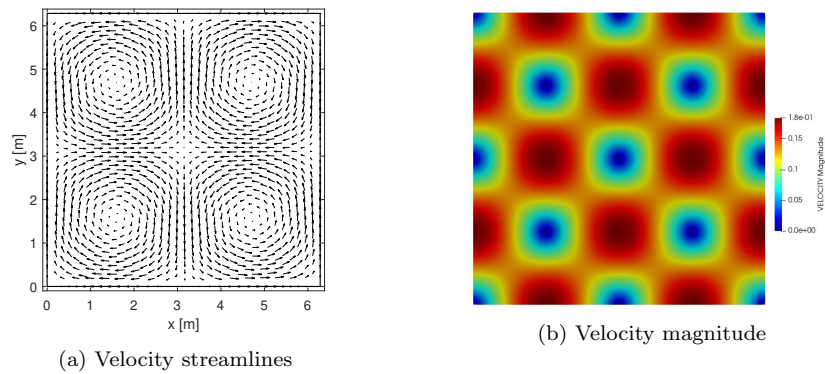


Figure 2: Taylor Green vortex problem. Streamlines and velocity contours.

372 The problem has been solved for five different finite element discretiza-

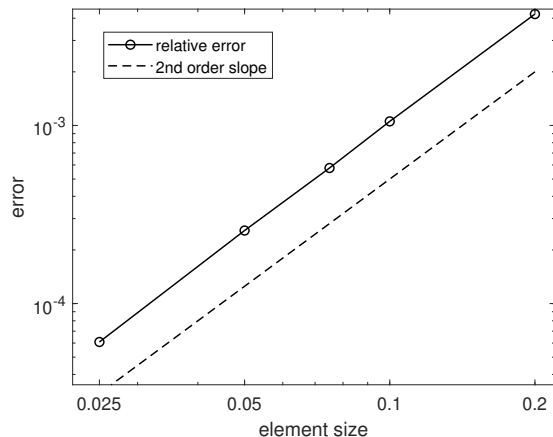


Figure 3: Taylor Green vortex problem: convergence diagram of the relative error for space discretization

373 tions ($h = 0.025m, 0.05m, 0.075m, 0.1m, 0.2m$) and time step sizes ($\Delta t =$
 374 $2.5 \cdot 10^{-4}s, 5 \cdot 10^{-4}s, 10^{-3}s, 2.5 \cdot 10^{-3}s, 5 \cdot 10^{-3}s$). The L2 norm of the
 375 error with respect to the analytical solution has been evaluated at $t = 5s$,
 376 that is when $F(t)$ reaches its maximum value. The error measure has been
 377 computed as follows

$$err = \sqrt{\int_{\Omega} (sol_{an} - sol_{num})^2 d\Omega} \quad (50)$$

378 In Figure 3, we show the convergence diagram of the relative error versus
 379 the element characteristic length. As expected, the slope of the error curve
 380 is 2 in a logarithmic diagram.

381 In Figure 4, we show the convergence diagram of the relative error versus
 382 the time discretization. In accordance with forward Euler method theory,
 383 the error curve in a logarithmic diagram has a first order slope.

384 4.2. Sod shock tube

385 The Sod shock tube [47] is another widely used benchmark for assess-
 386 ing the accuracy and robustness of numerical methods for the simulation of
 387 compressible flows in supersonic regime. Besides classical stabilized Eule-
 388 rian finite elements [48], the Sod shock tube has been tackled using different
 389 numerical techniques, among which the Finite Volume Method [49, 50] and

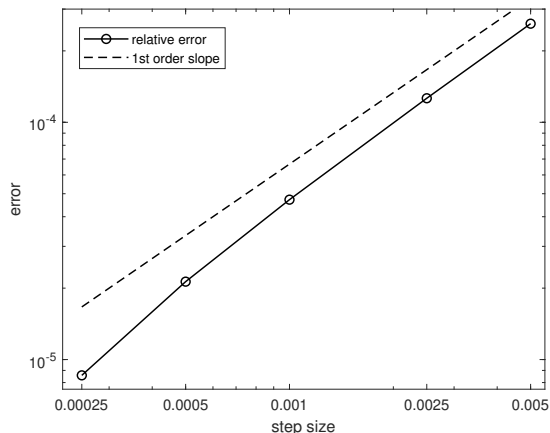


Figure 4: Taylor Green vortex problem: convergence diagram of the relative error for time discretization

390 the Particle Finite Element Method (PFEM) [51]. The problem consists in
 391 a one-dimensional domain of length $L = 1$ m, divided in two parts with dif-
 392 ferent initial conditions: on the left side, $\rho_g = 1$ kg/m³, $p = 1$ Pa, and $u = 0$
 393 m/s; on the right side, $\rho_g = 0.125$ kg/m³, $p = 0.1$ Pa, and $u = 0$ m/s.

394 The difference in pressure and density between the two parts of the do-
 395 main gives rise to a shock wave. The analytical solution of this problem can
 396 be obtained via the iterative procedure detailed in [52].

397 The problem is here solved considering both a monophasic flow, like in
 398 the original reference, and a two-phase one.

399 *Monophasic case.* Figure 5 shows the pressure field obtained numerically at
 400 $t = 0.2$ s assuming $\varepsilon_s = 0$. The results are given for three different FEM
 401 meshes (*i.e.* 2168, 9062, and 56708 triangular elements), and are compared
 402 to the analytical solution. Also in this case, a very good agreement with the
 403 analytical solution and a clear convergent behavior are obtained. We also
 404 remark that, also for the coarser discretizations, the numerical solution of
 405 the three pressure plateaux is very well captured.

406 *Biphase case.* The same problem has been solved considering different initial
 407 solid particle concentrations in the domain. The initial gas density and pres-
 408 sure distributions are the same of the monophasic case, whereas the initial
 409 solid concentration are $\varepsilon_s = 10^{-5}$, $\varepsilon_s = 10^{-4}$, and $\varepsilon_s = 10^{-3}$ for the three
 410 different examples.

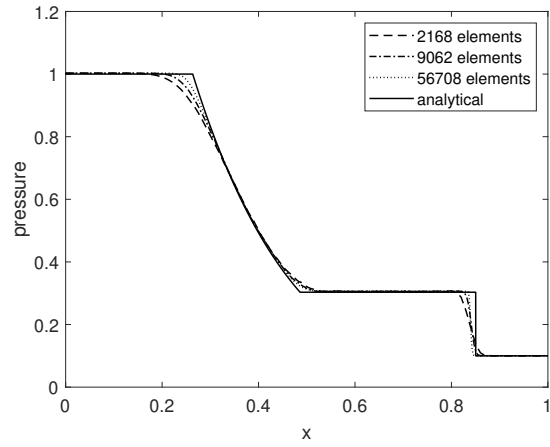


Figure 5: Monophase Sod shock tube problem: numerical and analytical solutions for different discretizations

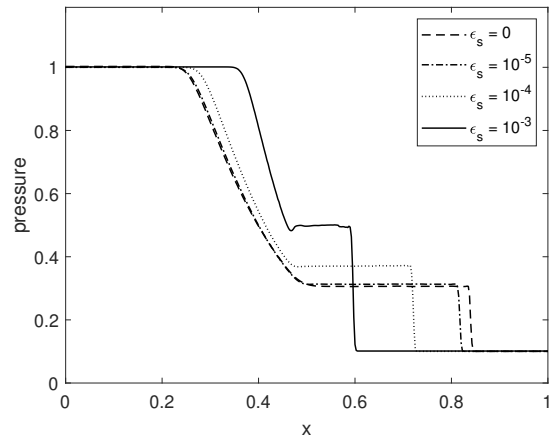


Figure 6: Biphasic Sod shock tube problem: pressure diagram for different values of ϵ_s with 56708 finite elements

411 Figure 6 plots the pressure solution obtained at $t = 0.2$ s for the three
 412 different solid concentrations.

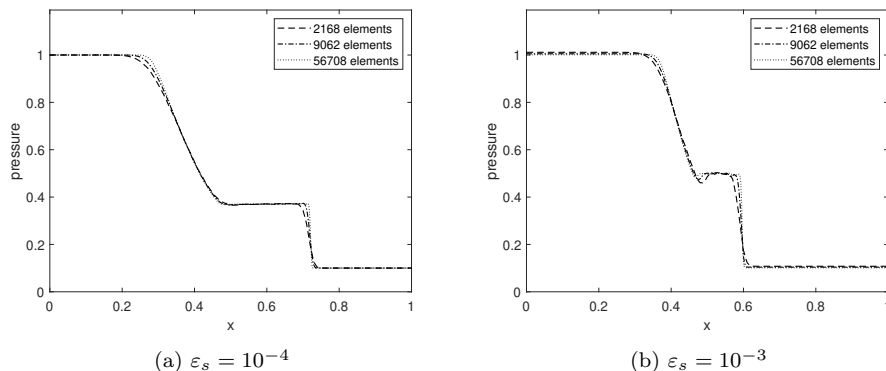


Figure 7: Biphase Sod shock tube problem: pressure diagram for different values of ε_s and different discretizations

413 It is interesting to note that, for higher values of the solid concentration
 414 ε_s , due to the consequent decreasing of the speed of sound c_m (Equation
 415 (22)), the pressure plateaux are preserved for a larger part. Furthermore, the
 416 graphs also show that, increasing ε_s , also the pressure value in the central
 417 part of the domain increases.

418 In Figures 7a and 7b, we show the pressure diagrams for the cases $\varepsilon_s =$
 419 10^{-4} and $\varepsilon_s = 10^{-3}$ for different discretizations. The graphs confirm the
 420 convergent behavior of the algorithm also for these multiphase analyses.

421 In Figure 8, we show the time evolution of the pressure for the cases
 422 $\varepsilon_s = 10^{-3}$ and $\varepsilon_s = 0$ from the beginning of the simulation until $t = 0.2$ s;
 423 it can be observed that, as expected, for the case $\varepsilon_s = 10^{-3}$ the speed of the
 424 sound is lower with respect to the monophasic case.

425 4.3. Compressible flow around a cylinder

426 This benchmark test consists of a compressible flow investing a cylinder of
 427 radius $r = 0.1$ m with a velocity of 1 m/s. When the flow is fully developed,
 428 an instability occurs behind the cylinder, generating a pattern of swirling
 429 vortices.

430 The domain and the dimensions of the problem are represented in Figure
 431 9. As for the previous test, the problem is here solved considering both
 432 monophasic and two-phase flows.

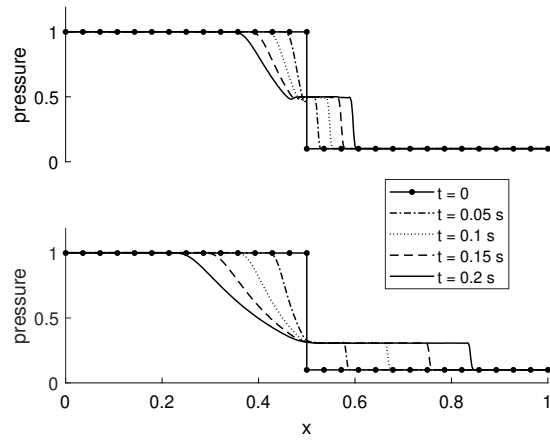


Figure 8: Biphase Sod shock tube problem: time evolution of the pressure for $\varepsilon_s = 10^{-3}$ (top) and $\varepsilon_s = 0$ (bottom) with 56708 finite elements

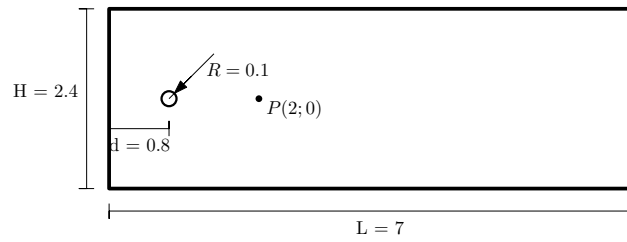


Figure 9: Flow around a cylinder: computational domain

433 *Monophase case.* Following [27], we considered $\rho_g = 1 \text{ kg/m}^3$, $T = 9.73 \cdot 10^{-3}$
434 K, and $u_1 = u_2 = 0 \text{ m/s}$ as the initial conditions of the whole domain. The
435 inlet condition is placed on the left side of the domain and it consists of a
436 velocity of $u_{in} = 1 \text{ m/s}$, a gas density $\rho_g = 1 \text{ kg/m}^3$, and a temperature
437 $T = 9.73 \cdot 10^{-3} \text{ K}$. Since the flow is at a subsonic regime, an output boundary
438 condition is required, therefore we set the gas density to $\rho_g = 1 \text{ kg/m}^3$ at the
439 right side of the domain. On the upper and lower sides of the domain, a slip
440 condition is imposed, whereas a no-slip condition is imposed on the boundary
441 of the cylinder. The gas viscosity is set to 0.002 kg/m s and the thermal
442 diffusivity is 2.8676 W/m K . These coefficients determine, together with the
443 dimensions of the problem and the values of the inlet boundary conditions, a
444 monophase compressible flow with a Reynolds number $Re = 100$ and a Mach
445 number $Ma = 0.5$.

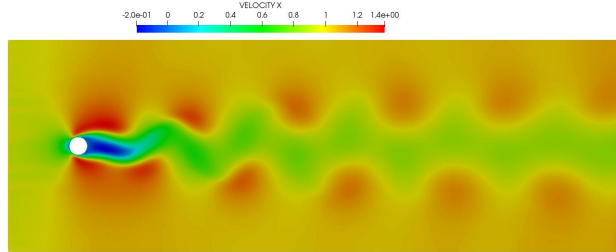


Figure 10: Monophase flow around a cylinder: horizontal velocity field at $t = 30 \text{ s}$.

446 In Figure 10, we show the velocity field at $t = 30 \text{ s}$ when the vortices
447 behind the cylinder are clearly visible.

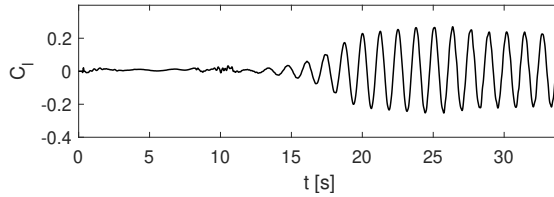


Figure 11: Monophase flow around a cylinder: lift coefficient versus time.

448 In Figure 11, we show the trend of the lift coefficient C_l , computed as

$$C_l = \frac{\int_{\Gamma} (-p\mathbf{I} + \boldsymbol{\tau}) \cdot \mathbf{n}d\Gamma}{\frac{1}{2}\rho_{in}u_{in}^2D} \quad (51)$$

449 versus time, being Γ the boundary of the cylinder. The lift coefficient begins
 450 to oscillate around zero after about 15 s, with an amplitude that, at regime,
 451 is of about 0.25, comparable with the results shown in [27]. The frequency of
 452 the oscillations f_f of the lift coefficient is also comparable to the one obtained
 453 in [27]. In the present work, we obtain $f_f \simeq 0.85$, whereas in [27] the reference
 454 value is $f_f = 0.91$. The slight discrepancy may be due to the use of different
 455 time integration methods and finite element meshes.

456 *Biphase case.* We repeat the simulation for the same initial conditions (ab-
 457 sence of solid particulate on the whole domain), and an impulsive inlet con-
 458 dition for the solid phase concentration, that is set to $\varepsilon_s = 10^{-4}$ after the
 459 first time step. In this simulation, the Reynolds number and the compress-
 460 ibility regime evolve during the analysis, according to Equations (11) and
 461 (22), therefore we cannot provide unique Reynolds and Mach numbers for
 462 this problem.

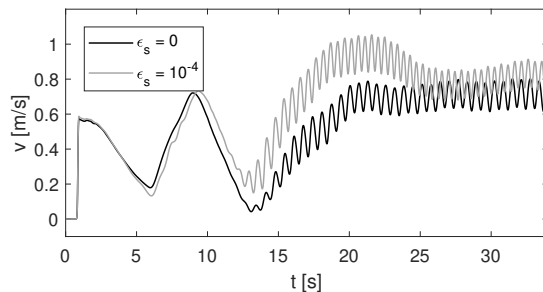


Figure 12: Flow around a cylinder: horizontal velocity field at the point $P(2;0)$ for the monophase case and the multiphase case with $\varepsilon_s = 10^{-4}$.

463 In Figure 12, we show the time evolution of the velocity field computed
 464 at the point $P(2,0)$ and obtained for the monophase and two-phase flows.
 465 We notice that the two plots are similar in the first part of the graph, be-
 466 cause, during the first phase of the analysis, the solid concentration has
 467 not yet reached the point P , and the flow is practically monophase. Later,
 468 once the particle wave reaches the sample point P , the two curves diverge.
 469 progressively.

470 In Figure 13, we show the horizontal velocity and the solid concentration
 471 fields at $t = 30s$. Comparing Figure 13b and Figure 10, we observe that the
 472 multiphase case gives higher velocities with respect to the monophase case.

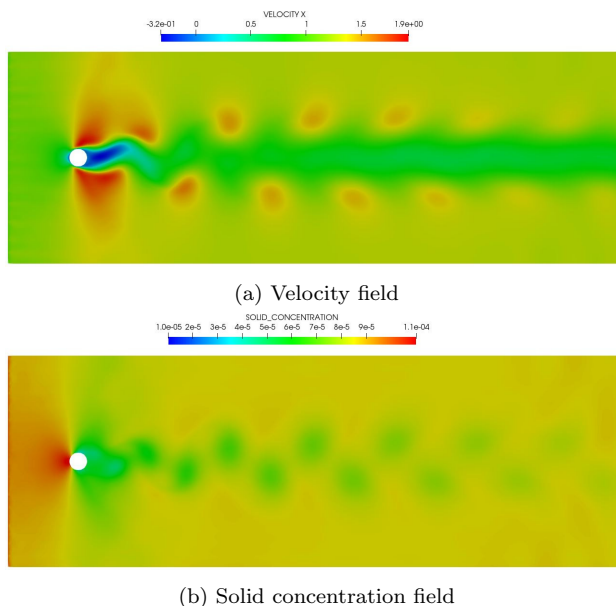


Figure 13: Biphase flow around a cylinder: velocity and solid concentration fields at $t = 30$ s

473 *4.4. A real case application: pyroclastic flow against buildings*

474 In this section, we present the application of the proposed formulation to
 475 the simulation of a pyroclastic gravity current, a multiphase flow composed
 476 by a mixture of gas and solid particulates, originated from the fragmentation
 477 of magma after a volcanic event. Assuming that the gas phase is composed
 478 by only air ($\mu_g = 2 \cdot 10^{-5}$ kg/m s, $c_p = 1010$ J/kg K, $\lambda_g = 0.026$ W/m K), and
 479 considering a solid phase composed by particles of diameter $190 \mu\text{m}$ [9] and
 480 density $\rho_s = 1800$ kg/m³, we obtain the characteristic times $\tau_u = 0.36$ s (see
 481 Equation (6)) and $\tau_T = 0.21$ s (see Equation (7)). We analyze the effects of
 482 the volcanic event over a urban settlement placed at a distance of 5 km from
 483 the crater of the volcano, thus at a distance such that the characteristic times
 484 are abundantly overcome. This allows us to consider that the hypothesis of
 485 thermal and kinetic equilibrium is fulfilled, and to use Equations (8)-(9) for
 486 the problem solution.

487 The computational domain, shown in Figure 14, consists of a rectangular
 488 domain of 1 km in length and 200 m in height, with two obstacles representing
 489 two buildings, the first one (20 m high) located at 150 m from the inlet (at
 490 the left side of the domain), and the second one (50 m high) placed at a

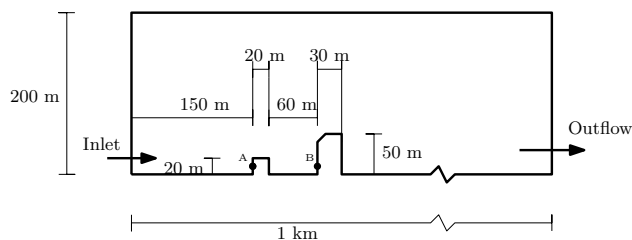


Figure 14: Pyroclastic flow simulation: computational domain.

491 distance of 60 m from the first building.

492 The length and the height of the computational domain have been chosen
 493 such that the pyroclastic flow does not alter the variables in correspondence
 494 of the outflow and the upper part of the domain.

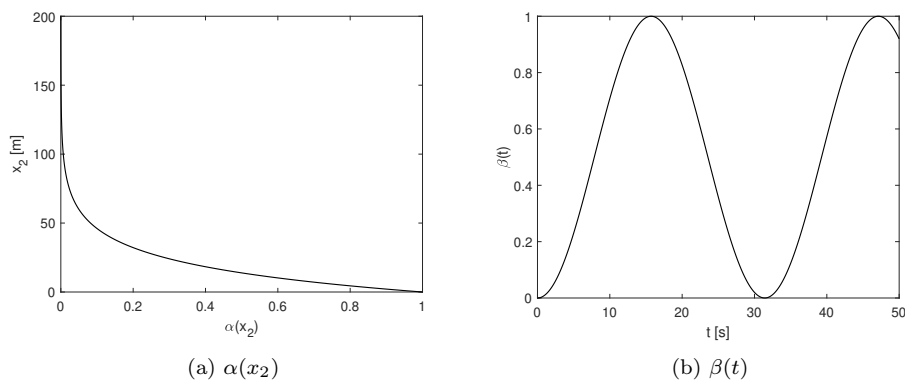


Figure 15: Plot of function $\alpha(x_2)$ and $\beta(t)$ used for the inlet of the pyroclastic flow simulation

495 The initial conditions are coherent with a situation of stationary air at
 496 300 K. The initial density of the gas is set to $\rho_g = 1.225 \text{ kg/m}^3$. At the inlet
 497 a time-varying boundary condition is imposed, that is

$$\begin{aligned}
 \varepsilon_s(x_1 = 0, x_2, t) &= \varepsilon_{s,max} \alpha(x_2) \beta(t) \\
 u_1(x_1 = 0, x_2, t) &= u_{1,max} \alpha(x_2) \beta(t) \\
 u_2(x_1 = 0, x_2, t) &= 0 \text{ m/s} \\
 T(x_1 = 0, x_2, t) &= 300 \text{ K}
 \end{aligned}
 \tag{52}$$

498 where $\varepsilon_{s,max} = 5 \cdot 10^{-3}$, $u_{1,max} = 60 \text{ m/s}$. $\alpha(x_2)$ and $\beta(t)$ are functions
 499 describing the spatial and temporal profiles of the inlet conditions, defined

$$\begin{aligned}\alpha(x_2) &= \exp\left(-\frac{x_2}{20}\right) \\ \beta(t) &= \frac{1}{2}\left(1 - \cos\left(\frac{t}{5}\right)\right)\end{aligned}\tag{53}$$

501 and represented in Figure 15.

502 The spatial exponential function $\alpha(x_2)$ models a distribution that is larger
503 in the proximity of the ground and vanishes at high altitudes, coherently with
504 the results presented in [9]. On the other hand, the time function $\beta(t)$ allows
505 us to reproduce subsequent waves of pyroclastic material. The gravitational
506 load is modeled as a field load $g = -9.81 \text{ m/s}^2$ in the direction of the vertical
507 axis.

508 The simulation is run for a total time duration of 50 s.

509 In Figure 16, we show the concentration of the solid phase at different
510 time instants.

511 Figure 17 shows the velocity vectors diagram obtained at $t = 24 \text{ s}$ in
512 the part of domain comprised between the two obstacles. The formation of a
513 vortex between the two obstacles can be clearly appreciated from the picture.
514 Figure 18 shows the pressure distribution on the domain for the same time
515 instant.

516 We repeat the simulation for a less dense mixture ($\varepsilon_{s,max} = 1 \cdot 10^{-3}$) with
517 the aim of comparing the pressure fields over the civil buildings given by the
518 two different pyroclastic flow scenarios.

519 In Figure 19, we show the diagram of the pressure versus time computed
520 at the two sample points A and B of Figure 14 for the two different values of
521 the inlet concentrations $\varepsilon_{s,max}$. Initially, the time evolution of the pressure
522 value at points A and B is the same for both solid concentrations.

523 As the gravity current reaches the first building (at around $t = 10 \text{ s}$), the
524 solutions start to diverge. In particular, the pressure at point A increases
525 sensibly for both $\varepsilon_{s,max} = 1 \cdot 10^{-3}$ and $\varepsilon_{s,max} = 5 \cdot 10^{-3}$ while it does not
526 vary significantly for point B (hydrostatic value). At about $t = 18 \text{ s}$, the
527 pressure in the point B increases for the case $\varepsilon_{s,max} = 5 \cdot 10^{-3}$ since the
528 mixture overcomes the first obstacle and hits the second one; for the case
529 $\varepsilon_{s,max} = 1 \cdot 10^{-3}$, on the contrary, the amount of material overcoming the
530 first obstacle is minor, and the increase of pressure in B happens later, and

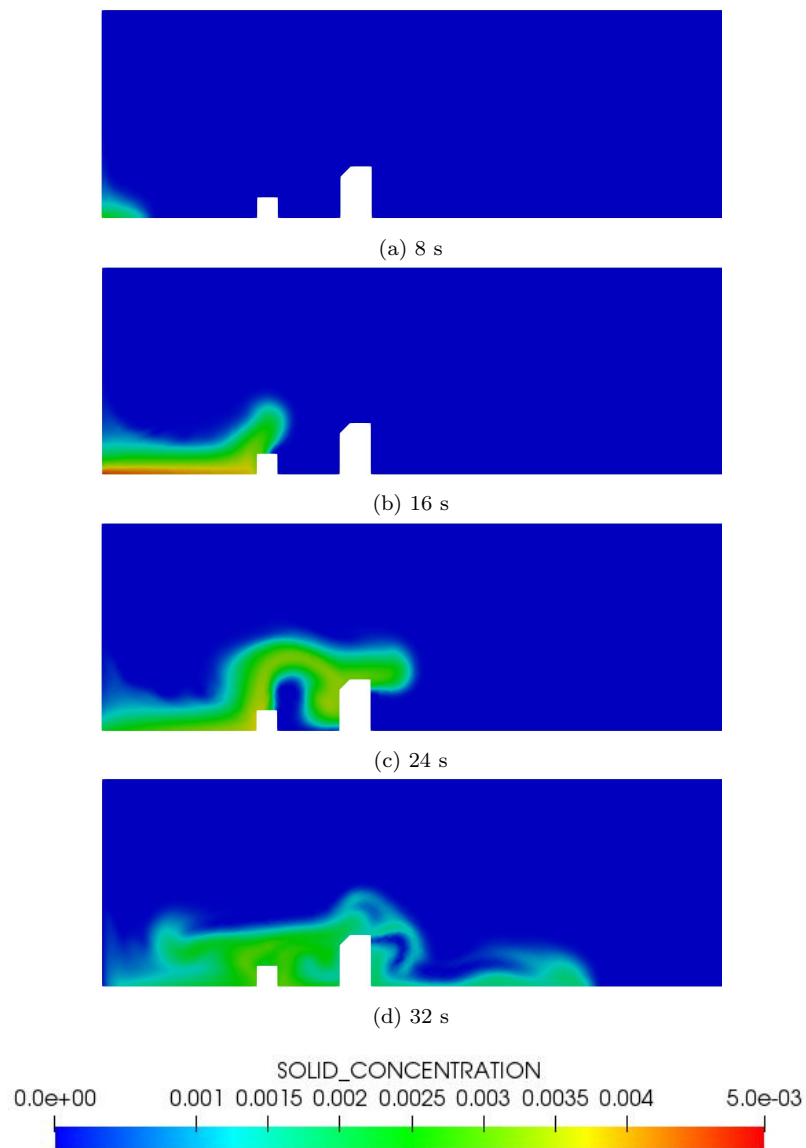


Figure 16: Pyroclastic flow simulation: solid concentration at different time instants

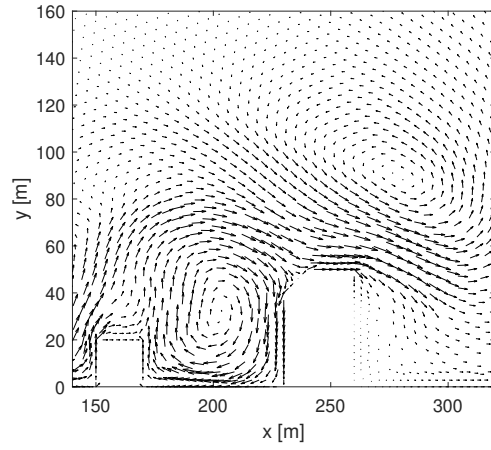


Figure 17: Pyroclastic flow simulation: velocity diagram at $t = 24$ s

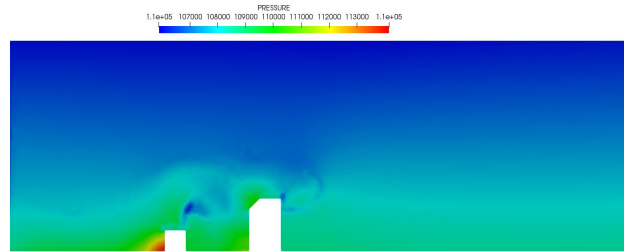


Figure 18: Pyroclastic flow simulation: pressure field at $t = 24$ s.

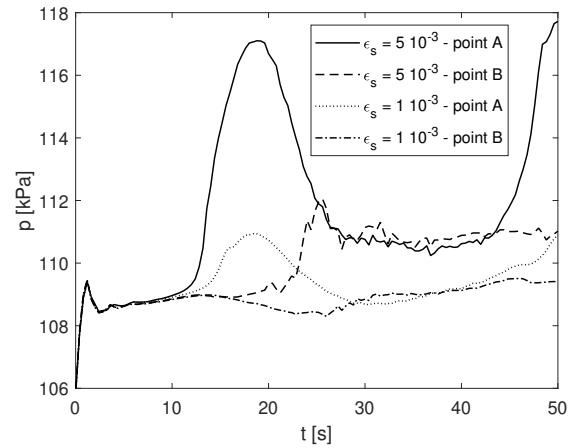


Figure 19: Pyroclastic flow simulation: pressure at points A and B for different values of the maximum concentration of the solid phase.

531 with less intensity. These pressure results show clearly the barrier effect
 532 produced by those buildings of the urban settlement that are most exposed
 533 to the pyroclastic gravity current.

534 Basing on the classification proposed in [4] to correlate different levels of
 535 damage on civil constructions with the dynamic pressure acting on them, we
 536 have estimated the damages produced by the pyroclastic flow over the two
 537 buildings considered in this test case. In particular, we have evaluated the
 538 overpressure values with respect to the hydrostatic ones from the pressure
 539 solution obtained at the façades of the two buildings. For the case of solid
 540 concentration $\varepsilon_{s,max} = 5 \cdot 10^{-3}$, the overpressure reaches 9 kPa in the first
 541 building, while it is only 3 kPa in the second building. The lower value
 542 of overpressure on the second building is attributable to the barrier effect
 543 produced by the first one. In terms of damages, these pressure values would
 544 lead to heavy damages for the first building and light/moderate ones for the
 545 second construction, according to [4]. On the other hand, for the case of solid
 546 concentration $\varepsilon_{s,max} = 1 \cdot 10^{-3}$, the overpressure is almost always below the
 547 value of 2 kPa, which corresponds to light damages.

548 4.5. Real-scale application

549 In this test, we analyze the impact of a realistic pyroclastic flow impacting
 550 a urban settlement considering a large-scale three-dimensional (3D) geome-
 551 try. Ten buildings of height of 30 m are distributed over a rectangular plan of
 552 sides $L = 600$ m and $W = 100$ m. A maximum height $H = 200$ m is consid-
 553 ered for the external environment. 3D and planar views of the computational
 554 domain are provided in Figure 20.

555 Slip boundary conditions are considered on the ground, on the lateral
 556 surfaces of the domain, and on the external boundaries of the buildings. The
 557 following inlet condition is applied on the rectangular surface placed at $x = 0$:

$$\begin{aligned}
 \varepsilon_s(x = 0, y, z, t) &= \varepsilon_{s,max} e^{-z/20} (1 - e^{-t}) \\
 u_1(x = 0, y, z, t) &= u_{1,max} e^{-z/100} (1 - e^{-t}) \\
 u_2(x = 0, y, z, t) &= 0 \text{ m/s} \\
 u_3(x = 0, y, z, t) &= 0 \text{ m/s} \\
 T(x = 0, y, z, t) &= T_0 (1 - k z)
 \end{aligned}
 \tag{54}$$

558 with $\varepsilon_{s,max} = 10^{-2}$, $u_{1,max} = 30$ m/s, $T_0 = 300$ K, and $k = 0.01$ m⁻¹. The
 559 solid density ρ_s is set to $2.8 \cdot 10^3$ kg/m³.

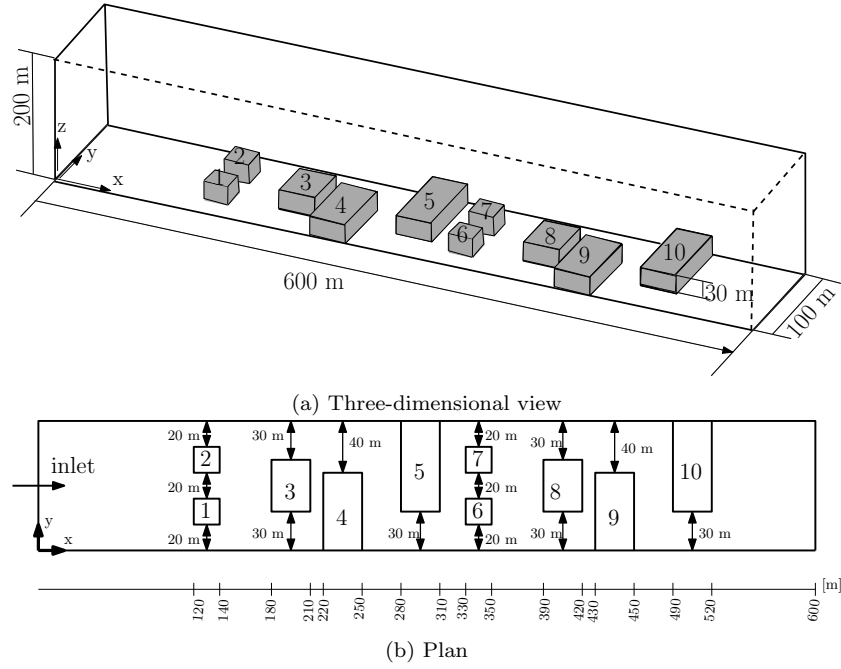


Figure 20: Computational domain of the 3d real scale application

560 As initial condition, we consider zero velocity and zero solid particle con-
 561 centration in the whole domain. In order to avoid unphysical velocities at the
 562 top of the computational domain, we also consider two functions of the sole
 563 variable z as initial conditions for the density and the temperature. These
 564 functions satisfy at the same time the stationary version of Equations (8b),
 565 (8c), and the equation of state for gas (4), under the constraints of zero
 566 velocity.

567 Assuming a linear variation along the z -axis for the initial temperature
 568 field [53], such that

$$T(z) = T_0(1 - kz), \quad (55)$$

569 the initial condition for the gas density is found by solving

$$\frac{\partial}{\partial z}[\rho RT_0(1 - kz)] + \rho g = 0 \quad (56a)$$

570

$$\rho(z = 0) = \rho_0 \quad (56b)$$

571 ρ_0 being the density at the ground level and at atmospheric standard condi-
 572 tions.

573 The domain has been discretized with a finite elements mesh of 940785
 574 linear tetrahedra with mean size of 5 m, giving an overall amount of 168841
 575 nodes.

576 Following the standard practice for this natural hazard scenario [12, 54],
 577 we measure the impact of the pyroclastic flow in terms of the dynamic pres-
 578 sure, which is computed as $p_{dyn} = 0.5\rho\|\mathbf{u}\|^2$. This value is measured for each
 579 building at the geometrical center of its façade closest to the inlet.

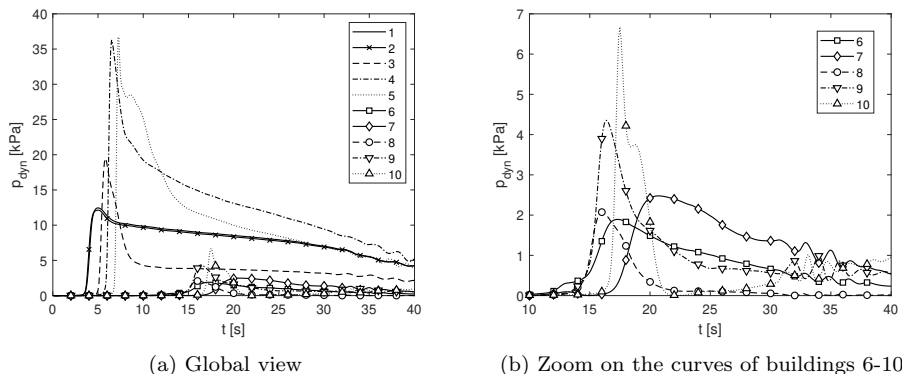


Figure 21: Real scale application: dynamic pressure on buildings

580 The graphs of Figure 21 show the time evolution of the dynamic pressure
 581 at the ten buildings. Despite the differences in magnitude, all the curves show
 582 an initial peak followed by a relatively sudden decrease. Due to symmetry,
 583 the curves of buildings 1 and 2 are practically superimposed. However, for the
 584 rest of buildings, this symmetrical behavior is lost due to their asymmetrical
 585 distribution. This is clearly shown by the response of buildings 6 and 7,
 586 which, although they have the same relative position as buildings 1 and 2,
 587 give rise to totally different pressure curves (see Figure 21b).

588 The most noticeable finding is that buildings 6-10 exhibit a first peak of
 589 dynamic pressure much lower than the one of buildings 1-5. In particular,
 590 the highest peaks in the first set of buildings is obtained on buildings 4 and
 591 5. These results show clearly the shielding effect produced by the buildings
 592 closest to the pyroclastic flow sources. Indeed, the first set of building retains
 593 and slows down the pyroclastic material and this makes reducing the flow
 594 density and velocity, and so the overall dangerousness of the pyroclastic flow.

595 Analogously to what done for the previous test, we have analysed the
 596 potential damages over the buildings basing on the criteria given in [4]. In

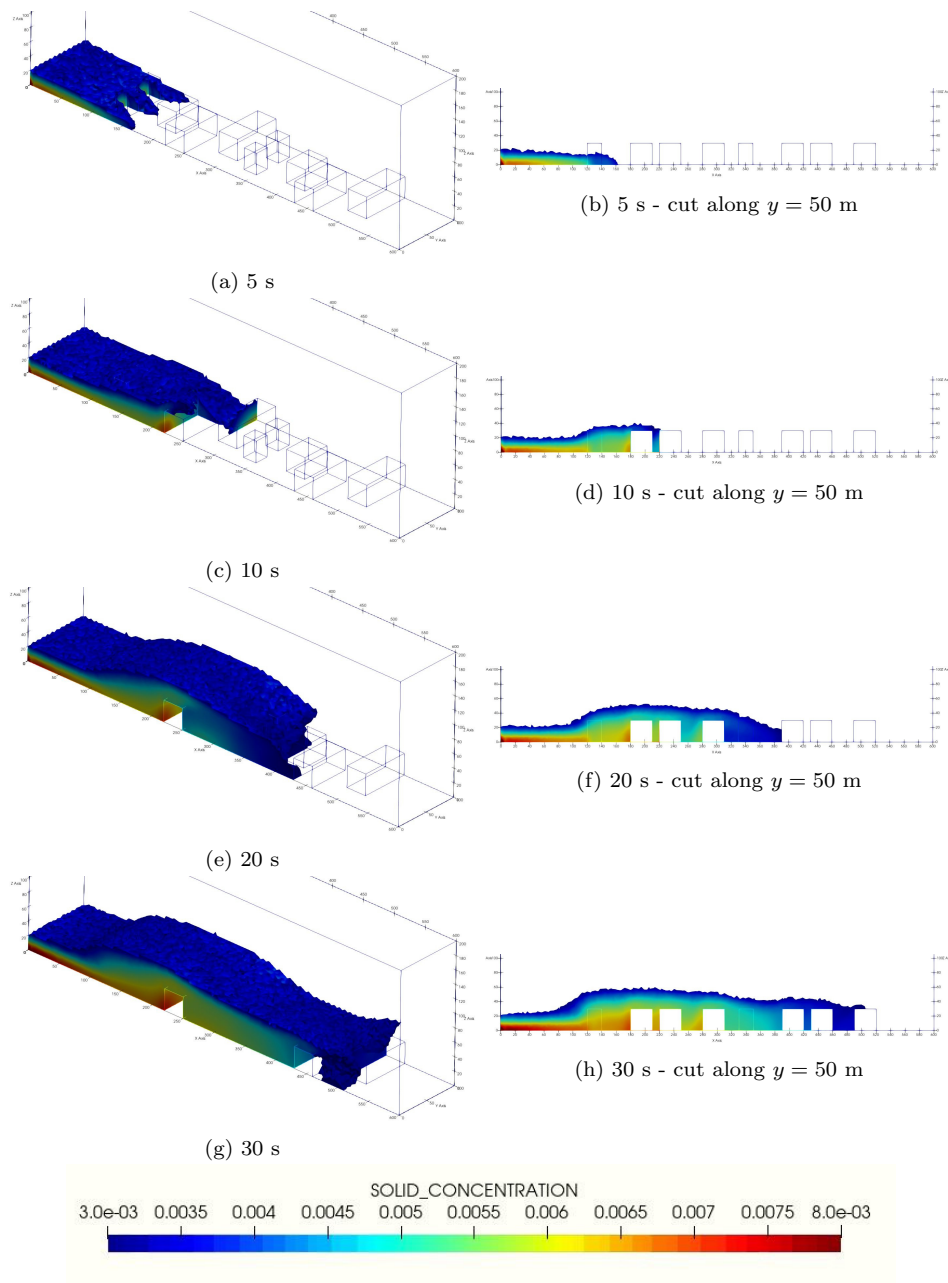


Figure 22: Three-dimensional simulation: solid concentration at different time instants (the values lower than $\varepsilon_s = 3 \cdot 10^{-3}$ are not represented). Left column: Three dimensional view; right column: cut view at $y = 50$ m.

597 this case, the peaks of dynamic pressure measured at buildings 4 and 5 would
 598 lead to a *total devastation* of the structures, meaning that walls would be
 599 completely destroyed and washed out. The other buildings of the first group
 600 (buildings 1, 2, 3) would also undergo severe damages (*partial devastation*
 601 according to [4]), while the buildings of the second group would suffer from
 602 moderate damages. A color map of the level of damage in accordance with
 603 this classification is reported in Figure 23.

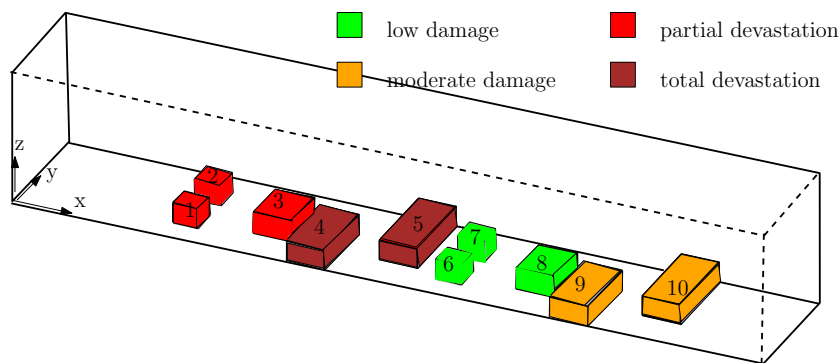


Figure 23: Three-dimensional simulation: simulated level of damage for each building.

604 The solution in terms of the solid material concentration obtained at
 605 four different time instants is plotted in Figure 22. Figures 22a-22b show
 606 the advancement of the flow at $t = 5s$. At this moment, the solid particu-
 607 late is flowing around buildings 1-2, following three different channels in a
 608 symmetrical way. At $t = 10s$, as shown in Figures 22c-22d, the flow has
 609 already overcome buildings 1-3, and, after surrounding building 4, has im-
 610 pacted against building 5. At $t = 20s$ (Figures 22e-22f), the pyroclastic
 611 current has already reached building 9, and finally, in Figures 22g-22h, the
 612 flow is overcoming also building 10.

613 Figure 22 also shows that the the highest concentration of pyroclastic
 614 material is retained by the first set of buildings, remarking again the above-
 615 mentioned shielding effects.

616 In order to assess the stability and smoothness of the pressure field, in
 617 Figure 24 we show the pressure distribution on the plan $x_2 = 50$ m. The
 618 pressure results confirm the validity of the proposed stabilized formulation,
 619 since no instabilities are exhibited.

620 To conclude, this numerical test shows the applicability of the proposed
 621 explicit multiphase formulation to demanding computations of pyroclastic

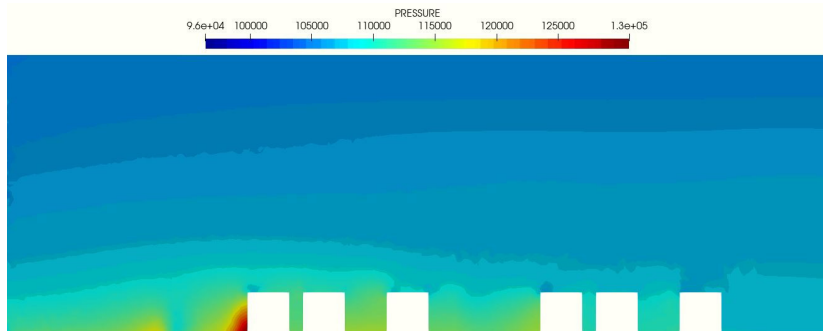


Figure 24: Three-dimensional simulation: pressure distribution at $t = 20$ s, for $y = 50$ m.

622 gravity currents on urban settlements and its suitability for risk assessment
 623 studies.

624 5. Conclusion

625 In this work, we extend the stabilized finite element formulation proposed
 626 for compressible monophasic flows to multiphase mixtures composed of a
 627 compressible fluid and a dilute solid phase. In particular, we investigate the
 628 case of compressible multiphase flows in kinetic and thermic equilibrium, for
 629 which a unique velocity and temperature field can be assumed for all the
 630 mixture components. This situation is representative of pyroclastic flows
 631 considered at a certain distance from a volcanic eruption source.

632 Given the formal similarities between the formulation for a compressible
 633 monophasic flow and a multiphase flow, we adopt the same stabilization
 634 strategies as those used in the literature for the monophasic compressible case.
 635 In particular, we use the Variational Multiscale Method for the stabilization
 636 of the convective part, and an isotropic discontinuity capturing technique for
 637 dealing with the eventual discontinuities.

638 The robustness of the approach has been tested on several benchmarks
 639 proposed in the literature, and here slightly modified to introduce multi-
 640 phase examples. The results confirm the accuracy and robustness of the
 641 formulation for monophasic compressible problems and show also its validity
 642 for multiphase problems.

643 Finally, we applied the proposed stabilized formulation to the study of
 644 the dynamics of pyroclastic flows impacting against buildings, an application
 645 of interest for the evaluation of the risk in case of volcanic eruptive events.

646 We first analyzed a two-dimensional case considering pyroclastic flows with
647 different solid concentrations. Then, we applied the numerical method to the
648 solution of a large-scale problem in three dimensions. Basing on the damages
649 classifications proposed by [4], we also quantified the potential damages over
650 the civil constructions produced by the impact of the pyroclastic flow. These
651 results proved the suitability of the proposed technique for the study of real-
652 world scenarios of pyroclastic flows.

653 Future developments of this work will include analyses of the sensitivity
654 of the model with respect to the involved physical parameters, as well as the
655 evaluation of the barrier effect of the buildings depending on the geometrical
656 characteristics of the urban settlement. From the numerical point of view,
657 future work will be the coupling of the present Eulerian formulation with
658 Lagrangian approaches, such as the Discrete Element Method (DEM), for
659 the simulation of particles in suspension for case in which the hypotheses of
660 kinetic and thermal equilibrium are not applicable.

661 Appendix A. Appendix

Here we report the steps to bring Equation (38) to its semidiscrete form (44). Introducing (43) we get

$$\begin{aligned}
& \hat{V}_s^h \int_{\Omega_h} N_{is} N_{ip} d\Omega \frac{\partial}{\partial t} \hat{U}_p^h + \hat{V}_s^h \int_{\Omega_h} A_{ijk}(\mathbf{U}^h) D_{kpj} d\Omega \hat{U}_p^h - \hat{V}_s^h \int_{\Omega_h} T_{ik} N_{kp} d\Omega \hat{U}_p^h \\
& - \hat{V}_s^h \int_{\Omega_h} \left(D_{isj} A_{ijk}(\mathbf{U}^h) + N_{is} B_{ijkm}(\mathbf{U}^h) \frac{\partial U_m^h}{\partial x_j} - N_{is} T_{sk} \tau_{kj} N_{jp} \right) d\Omega \hat{R}_p^h + \\
& \hat{V}_s^h \int_{\Omega_h} (D_{isj} G_{ij}(\mathbf{U}^h) d\Omega_h) d\Omega = 0
\end{aligned} \tag{A.1}$$

662 where Ω_h is the domain of the finite element and matrix \mathbf{D} containing the
663 derivatives of the shape functions is defined as

$$D_{ijk} = \frac{\partial N_{ij}}{x_k} \tag{A.2}$$

664 Since Equation (A.1) holds for every choice of the test function \hat{V}_s^h , then

$$M_{sp} \frac{\partial}{\partial t} \hat{U}_p^h = \hat{F}_s \tag{A.3}$$

also holds, where

$$\begin{aligned}
M_{sp} &= \int_{\Omega_h} N_{is} N_{ip} d\Omega \\
\hat{F}_s &= - \int_{\Omega_h} A_{ijk}(\mathbf{U}^h) D_{kpj} d\Omega \hat{U}_p^h + \int_{\Omega_h} T_{ik} N_{kp} d\Omega \hat{U}_p^h + \\
&+ \int_{\Omega_h} \left(D_{isj} A_{ijk}(\mathbf{U}^h) + N_{is} B_{ijkm}(\mathbf{U}^h) \frac{\partial U_m^h}{\partial x_j} - N_{is} T_{sk} \tau_{kj} N_{jp} \right) d\Omega \hat{R}_p^h + \\
&- \int_{\Omega_h} (D_{isj} G_{ij}(\mathbf{U}^h) d\Omega_h)
\end{aligned} \tag{A.4}$$

665 M_{sp} is the consistent mass matrix, and, in view of an explicit algorithm
666 for the time integration, can be lumped, bringing to the final semidiscrete
667 form (44).

668 Acknowledgements

669 This work is partially funded by the Italian Ministry of University and Re-
670 search (MUR) through the project PON-AIM "Attraction and International
671 Mobility" CUP: E66C19000230005, and partially funded by the Spanish Min-
672 istry of Economy and Competitiveness (Ministerio de Economía y Competi-
673 tividad, MINECO) through the project PRECISE (BIA2017- 83805-R).

674 Authors are also grateful to the CINECA consortium for the provision of
675 computational utilities through the IsCRA project "Vulnerability and Risk
676 Assessment" (IsC79-VRA).

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