# IMPACT OF USING WIDE- OR NARROW-BODY AIRCRAFT ON SHORT-HAUL FLIGHTS 

TITLE OF THE TFG: Impact of using wide- or narrow-body aircraft on short-haul flights

BACHELOR'S DEGREE: Grau en Enginyeria d'Aeronavegació
AUTHOR: Jessica Núñez Maturano
DIRECTOR: Roger Mulet Morato
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## OVERVIEW

The aim of this project is to study whether is better to use wide- or narrow-body aircrafts in short-haul flights. Research into short-haul flights shows that there are multiple flights a day travelling the same journey, some of them within less than onehour difference.

As a passenger, having a range of choices to plan a trip is a positive aspect, as any option would be suitable for their schedule. But, knowing that the airspace is reaching its maximum capacity and aviation is not environment friendly, the decision to make a study to determine if these problems could be diminished seemed mandatory.

The project will follow the next structure. First, an analysis of the evolution of the air traffic in Europe until the current moment and some predictions for the future will be made. Then, a research of what type of planes are preferably used for the different existing routes will be done. Later, the journey between Josep Tarradellas Barcelona-El Prat airport and Gran Canaria airport will be studied and compared if it would be better performed with an A320 or an A330. This study will consist in stablishing the route travelled, examine the weight and balance of both planes and conduct an economical and environmental research. Finally, the results will be revealed and a final conclusion will be made.

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## RESUMEN

El objetivo de este proyecto es estudiar la mejor opción entre usar aviones de fuselaje estrecho o de fuselaje ancho en vuelos de corto recorrido. Estudiando los vuelos de corto recorrido, se ha encontrado que hay múltiples vuelos al día realizando el mismo trayecto, muchos de ellos con diferencia de salida inferior a una hora.

Como pasajero, tener varias alternativas para planear un viaje es un aspecto positivo, ya que alguna de ellas encajara en sus horarios. Pero, sabiendo que el espacio aéreo está llegando a su límite de capacidad y la aviación es desfavorable para el medio ambiente, la decisión de hacer este estudio para saber si se pueden mitigar estos problemas parecía casi obligatoria.

El proyecto seguirá la siguiente estructura. Primero se realizará un análisis sobre la evolución del tráfico aéreo en Europa, donde también se incluirán predicciones de futuro. Además, se investigará que tipo de aviones se usan en las diferentes rutas existentes. Después, se estudiará el trayecto entre los aeropuertos de Josep Tarradellas Barcelona-El Prat y Gran Canaria con un avión A320 y un A330. El estudio consistirá en establecer una ruta, examinar la masa y centrado de cada avión y realizar un estudio económico y medioambiental de cada opción. Por último, se analizarán los resultados para ver que opción es preferible y se expondrán unas conclusiones finales.

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## INTRODUCTION

Flying has experienced an immense change since the first commercial flight started. In the $20^{\text {th }}$ century the first planes able to take off by their own means, sustainable and controlled by a pilot were constructed. Aviation became a focus point and it started to be more studied and advanced. Sadly, the arrival of the world wars helped and accelerated this development, although mainly for military reasons.

Shortly after, aviation converted into a very competitive means of transport. Travelling longer distances was required and transporting as many passengers as possible created a new goal. The construction of wide-body aircrafts became a reality. More targets as make travelling at supersonic speed reliable and affordable or even travelling to space are future aspirations. Nonetheless aviation is in constant research and evolution.

In the late 1900s with the creation of low-cost companies flying became affordable for everyone. In fact, the outcome was an increase of the demand, which created more competition among companies. Also, higher demand meant the expansion of the fleet of some airlines, the opening of new routes and the increment of flights in general.

In recent years, flying has changed our way of life. Business evolved into a way that business meetings can be anywhere, with people from different locations, who take a flight only to attend to that specific meeting. In personal terms, there has also been a behaviour modification. People travel to spend the day in another city or they just move out to another country without fearing to be far from their families as they are only one flight away. In the past, all of this was unthinkable but now it is a fact.

All of these behaviour patterns lead to a continuous increase of the air traffic. As a matter of fact, at Josep Tarradellas Barcelona El-Prat airport, the number of passengers transferred beats a new record year after year, getting its maximum in August of 2019 with 187.752 passengers in one day. [1] The same happens in many other airports in Europe.

The problem of the non-stop increase of passengers and flights can create a problem if airports and airspace reach their maximum capacity. At Josep Tarradellas Barcelona El-Prat airport the capacity limitation is due to the limitations of the runways. The limitations in the airspace are because of the navigation system used. New ways of organization and navigation are being studied to increase the capacity.

Knowing that this situation of overcapacity exists and the number of flights and demand keeps increasing, the study to know whether it is better to use wide- or narrow-body aircrafts in short-haul flights seems important. If using wide-body instead of narrow-body aircrafts would be better, a new consideration of change in aviation could come up and it would represent a way to diminish the existing capacity
problems. Also, a study of the environmental effects that this would trigger will be done to see if it would be more environmentally friendly to do such a change.

The project will be divided into three chapters. The first one will show a research of the evolution of the air traffic in Europe and it will show future predictions. Then a comparison of what type of planes are preferably used for the different existing routes will be done. To conclude, a case of study between Josep Tarradellas Barcelona El-Prat airport and Gran Canaria airport will be accomplished. This study will compare if it would be better to perform such a route with A320s or A330s. For that, the route, weight and balance, the economical- and the environmental point of view will be studied. The results and conclusions will be shown at the end of the project.

## CHAPTER 1. EVOLUTION OF AIR TRAFFIC IN EUROPE

Air traffic is the number of passengers, planes and cargo carried by aircrafts and handled by airports. Since the beginning of aviation, air traffic has been evolving. The reasons of this evolution, come together with history.

After WWII, Europe was facing a terrible post-war period, but technology kept developing. This meant the growth of aviation and it brought down the cost of jet travel. The result of it, is reflected in the evolution of air traffic over the years, which kept growing and growing.

For this project it was possible to collect the data of the that evolution by the different countries of Europe over the period 1993-2019. The tables with such information are in annex 1 and annex 2. [2] Being aware that air traffic and economic growth are closely related, it is important to keep in mind the historical background of every country.

This chapter is divided into the evolution of the carried passengers and the freight and mail transported to have a more accurate result.

### 1.1. PASSENGER TRAFFIC EVOLUTION

Figure 1.1 shows the evolution of the passengers carried at every European country. The difference between countries is clear, but it is even more clear that there are five countries where the movement of people is much higher and growing faster than the rest.


Figure 1.1 Passengers air traffic evolution by country

Looking into the graphic of figure 1.1, the trendline of every country approaches a lineal behaviour. The slope of the five leading countries is steeper and the shape of the lines is less smooth than the others.

Focusing now at the five leading countries: United Kingdom, Germany, Spain, France and Italy, figure 1.2 shows the evolution of their air traffic. It is certain that every graph line is different and with a different slope, but they have some aspects in common. They are all ascending and they have some stagnation or even descending fragments, where sometimes it is shown at all five lines during the same period. For that, and as it is mentioned before, a look into the background history is needed.


Figure 1.2 Passengers air traffic of the top five European countries

In 1991, the Gulf war started, having a big impact in aviation. A dramatic slump in the number of people flying followed, but the recovery was very fast. That is the reason why this graph starts with a slow growth for all the countries. Furthermore, the implementation of the "Schengen Agreement" in 1995, turned out into a good tendency of carrying passengers. The next important bump for aviation is in 2001 with the $9 / 11$ attacks. Globally it took 1.5 years to recover to previous levels. Afterwards, it can be seen a very similar and lineal growing behaviour in the five countries until 2006-2007, depending on the country. Aviation already started noticing what was coming in 2008, the Great Financial Crisis. This generated a 3year recession and it took 8 years to recover to the levels prior to the crisis. After that, traffic kept growing until 2020 when the arrival of Covid-19 provoked the worse
crisis ever with more than 6 million flights lost and with a big uncertainty of the recovery time to previous levels.

The United Kingdom, on top of the graph, is the country that transports the biggest number of passengers all over the years, with a difference of more than 20.000 thousand passengers from the second one. In fact, London-Heathrow airport has been the airport with the most passenger traffic all over the years. This is somehow possible due to the high number of long-haul flights with wide-body planes where more passengers can be carried plus the so called "minute of Heathrow". By virtue of the logistics of the airport, flights can take off within 1-minute difference instead the 2 minutes of a normal procedure, so the capacity is incremented. The outcome of all this keeps London Heathrow Airport at the top busiest airports of Europe, and second of the world.

Second in the ranking is Germany, followed very closely by Spain. Germany is a powerful country with a strong and stable economy. Their citizens income and stability allow them to travel and spend money on holidays, which is an important factor of passenger exchange. In general, the people form the north of Europe are attracted to travel to the south, as the good weather and food are very attractive. This positions Spain, France and Italy on the top of the list. The arrival of tourists to these regions, rises the number of flights between the north and the south, and consequently the number of passengers carried.

A very interesting fact that is worth to look at, are the curves of the United Kingdom and Spain alone, showed in Figure 1.3. Their shape is practically identical. This supports the idea that tourism is an important part of the increasing number of passengers carried, as people travel between these countries the most.


Figure 1.3 Passengers carried in United Kingdom and Spain

### 1.2. FREIGHT TRAFFIC EVOLUTION

Another factor for the evolution of air traffic as important as the passengers carried, is the evolution of the cargo and mail traffic.

Following the same structure as the one for the analysis of the passengers carried, Figure 1.4 shows the evolution of the freight transported in every European country. The behaviour of the graph is also lineal, but with a different slope. The one of the passengers is steeper than the one of freight.


Figure 1.4 freight and mail air traffic evolution by country

Looking again at the five leading countries, it is important to notice that the leading countries differ from ones of the passengers. The United Kingdom, Germany and France stay in the ranking, but Spain and Italy have been replaced by The Netherlands and Belgium as Figure 1.5 shows.


Figure 1.5 Five leading countries transporting freight and mail

Germany is leading the ranking for the transport of cargo and mail with a big difference. This is the result of their strong economy. The United Kingdom and France are respectively second and third.

The importance of the geographical location is a key factor for the transport of freight. That is the case for The Netherlands and Belgium. They are in a strategic position for transporting freight. The Netherlands has the biggest port in all of Europe, the port of Rotterdam, followed by Belgium with the second biggest, the port of Antwerp. Cargo and mail can be transported from anywhere in the world, and then it can be easily transported to the rest of Europe, either by sea or land.

It is important to recall the significance of the historical background in this case. Looking at the same period where there is a decrease of passengers carried, there is also a decrease of freight transport.

### 1.3. FUTURE EVOLUTION SCENARIOS

The evolution of air traffic during the past years, can help support long-term planning for aviation and take well-informed decisions. For that, EUROCONTROL Statistics and Forecast Service (STATFOR) developed the project "The Challenges of Growth" in 2017-2018. [3]

This report presents the forecast of annual numbers of instrument flight rules (IFR) movements in Europe up to 2040. Air traffic is in constant growth, but there are some factors and constraints that make the evolution volatile. The aspects to take into account are:

- Long-haul appearance
- New low-cost aircraft types and carriers
- Middle class growth in China
- Total network delay
- Alternatives to air transport
- Oil prices changes
- Climate changes

Furthermore, the assumption that the network will be constrained at airport level but not at airspace level is well studied.

Taking all of these into account, there are four possible scenarios:
> Global Growth: Globalization will increase and will lead to a strong global economic growth. Technology will be successful to mitigate the effects of sustainability challenges such as the environment or resource availability.
> Regulation and Growth: Moderate economic growth, with environmental and social regulations and economic demands to deal with the growing global sustainability concerns.
> Fragmenting World: Globalization will decrease and global tensions increase. This will rebound into more security threats, higher fuel prices, reduced trade and transport integration and knock-on effects of weaker economies.
> Happy Localism: Globalization will slightly decrease and Europe will look inwards and its fragility will increase. European economies will be exposed to shocks, pressure on costs will increase and stricter environmental constraints will take place.

Definitely, growth will not be uniform across Europe regardless the scenario. The states in Eastern Europe will grow quicker than the Western ones. The reason of such a clear statement is because the eastern European states are less developed than the western ones. Therefore, they have more chances to grow. Nevertheless, they growth will stay lower than the Western countries.

Figure 1.6 shows the evolution of the expected IFR movements in Europe up to 2040 with every possible scenario. Clearly a growth interval between $12 \%-84 \%$ is expected.


Figure 1.6 Expected IFR movements in Europe in the four scenarios by 2040 [15]

According to the situation of the past years, experts believed that the most likely scenario to happen was "Regulation and Growth". But, no one could ever expect or predict a world pandemic as Covid-19 in 2020. This new scenario changes all the prediction made up until the moment. Precisely, the "Fragmenting World" scenario looks like the most realistic at the moment. But, it is very important to keep this predictions in mind because the future is very unpredictable.

### 1.4. IMPACT OF COVID-19

The reality of the situation is very far from all the predictions. The arrival of the Covid19 virus resulted in a worldwide pandemic and impacted very hard on most countries in the world. For that, strong measurements as lockdowns and closing borders were implemented.

Since March 2019, Europe is involved in one of the biggest crises in history. The Schengen borders were close for external countries, and even the intra-Europe flow was restricted. On top, every country took its own restrictions. Figure 1.7 shows the traffic variation in Europe between 2019 and 2021 and a short prediction on what will happen in the second trimester of 2021. This figure shows that the impact of the pandemic it has been very strong and the air traffic has dropped immensely.

It is also important to recall the complexity and unknown of the situation. This can be seen in the difference of the top and bottom of figure 1.7. The top figure shows the actual traffic until November 2020, when the study was made, and the predictions until the moment. In April 2020, the first prediction was developed, but it was so optimistic that another one had to replace it. The second one, in September 2020 it got closer to reality.[4] But the situation is very volatile and it can change daily. For that, new predictions are constantly being studied.

$\rightarrow$ Actual $\rightarrow$ Previous scenario (24/04) $\rightarrow$ - Current status scenario (14/09)

Figure 1.7a Evolution traffic scenario


Figure 1.7b Evolution traffic scenario

EUROCONTROL published the latest STATFOR Forecast for the period 20202024.[5] This forecast shows the traffic outlook for the future. It takes into account the traffic trend and the economic growth. Also, it compares the situation with the evolution after other crises like 9/11 attack and the Great Financial crisis of 2008.

As this crisis has no precedents and the number of flights lost is greater than in the other crises, 3 scenarios are considered so far.
> Scenario 1: Considers that the vaccine will be effective and available for travellers or the pandemic will end by summer 2021. If this occurs, the recovery to 2019 levels will be in 2024.
> Scenario 2: Considers that the vaccine will be effective and available for travellers or the pandemic will end by summer 2022. If this occurs, the recovery to 2019 levels will be doubtfully in 2026.
$>$ Scenario 3: Considers that the vaccine will not be effective. This will carry into low passenger confidence and drop into propensity to fly. Recovery to 2019 levels will be in 2029.

Figure 1.8 shows the evolution of the traffic according to the 3 scenarios and compares it with the traffic of 2019.


Figure 1.85 -years forecast for Europe 2020-2024

## CHAPTER 2. AIRCRAFT TYPE VS ROUTE

Knowing that air traffic is constantly growing, now the interest goes into the existing type of routes and aircrafts used for each one. EUROCONTROL, with the R\&D data archive, provided a lot of data in order to make this study. [6] The data shows all the information of every flight for four different months a year between 2015 and 2019. As the big amount of data was difficult to analyse, the consideration of only using the data of one month was accepted. The chosen month is June 2015 as it would be representative enough because the period of the obtained data is not so long. Also, the program IBM SPSS Statistics has been used to do all the statistics results and graphs. [7]

The objective of this chapter is to study the type of aircraft used and its relation with the routes and range travelled. For that, and following the same structure until the moment, the study will be divided into the transport of passengers and freight. Furthermore, a division between traditional and low-cost carriers will be performed in the study of passenger transport. The importance of this differentiation is due to the origin of the kind of business.

The passengers transport has changed during all these years. The creation of the low-cost carriers and the evolution of the international market has been determining the way of travelling. Cargo has evolved due to the evolution of technology together with the deals between countries for the exchange of products, as for example the Schengen area.

Having these considerations into account, it is expected to find different outcomes, so it is important to analyse the routes and the kind of aircrafts used by these different markets.

Figure 2.1. shows the aircraft type used and a table of its usage frequency for traditional carriers, low-cost and cargo. In reality, there are more different types of aircraft used. But the study considers that planes with a usage frequency under 1\% are not influenceable.


| AC Type | AC Type |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | Frequency | Percentage | Valid percentage | Cumulative percentage |
| -A321 | Aircraft | A319 | 46433 | 16,0 | 16,0 | 16,0 |
| - $\begin{array}{r}\text { A332 } \\ \text { A388 }\end{array}$ | type | A320 | 76646 | 26,4 | 26,4 | 42,4 |
| - $\square^{\text {B738 }}$ |  | A321 | 35350 | 12,2 | 12,2 | 54,6 |
| -B752 |  | A332 | 11301 | 3,9 | 3,9 | 58,5 |
| B763 |  | A388 | 4078 | 1,4 | 1,4 | 59,9 |
| CRI9 |  | B738 | 45974 | 15,8 | 15,8 | 75,7 |
| E190 |  | B744 | 6186 | 2,1 | 2,1 | 77,8 |
|  |  | B752 | 4653 | 1,6 | 1,6 | 79,4 |
|  |  | B763 | 8519 | 2,9 | 2,9 | 82,4 |
|  |  | B77W | 12157 | 4,2 | 4,2 | 86,6 |
|  |  | CRJ9 | 10969 | 3,8 | 3,8 | 90,4 |
|  |  | E190 | 28011 | 9,6 | 9,6 | 100,0 |
|  |  | Total | 290277 | 100,0 | 100,0 |  |

Figure 2.1a Aircraft type used for traditional carriers


| AC Type |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | Frequency | Percentage | Valid percentage | Cumulative percentage |
| Aircraft type | A319 | 39866 | 17,4 | 17.4 | 17,4 |
|  | A320 | 64154 | 28,0 | 28,0 | 45,4 |
|  | A321 | 9818 | 4,3 | 4,3 | 49,7 |
|  | A332 | 1367 | , 6 | , 6 | 50,3 |
|  | B738 | 104749 | 45,7 | 45,7 | 96,1 |
|  | B752 | 2508 | 1,1 | 1,1 | 97,2 |
|  | B763 | 1006 | , 4 | . 4 | 97,6 |
|  | B772 | 13 | . 0 | . 0 | 97,6 |
|  | B788 | 1069 | . 5 | . 5 | 98,1 |
|  | CRJ9 | 3287 | 1,4 | 1.4 | 99,5 |
|  | E190 | 1135 | . 5 | . 5 | 100,0 |
|  | Total | 228972 | 100,0 | 100,0 |  |

Figure 2.1b Aircraft type used for low-cost carriers


| ACType | AC Type |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Frequency |  |  |  |  | Cumulative |
|  |  |  |  | Percentage | percenatge | percentage |
| -ATP | Aircraft type | A306 | 1908 | 9,5 | 9,5 | 9,5 |
| 日B463B733B734B744B752BB763B77LMD11SW4 |  | A332 | 652 | 3,2 | 3,2 | 12,8 |
|  |  | ATP | 1565 | 7,8 | 7,8 | 20,5 |
|  |  | B463 | 433 | 2,2 | 2,2 | 22,7 |
|  |  | B733 | 3115 | 15,5 | 15,5 | 38,2 |
|  |  | B734 | 2579 | 12,8 | 12,8 | 51,1 |
|  |  | B744 | 2290 | 11,4 | 11,4 | 62,5 |
|  |  | B752 | 2965 | 14,8 | 14,8 | 77,2 |
|  |  | B763 | 1166 | 5,8 | 5,8 | 83,0 |
|  |  | B77L | 1707 | 8,5 | 8,5 | 91,5 |
|  |  | MD11 | 849 | 4,2 | 4,2 | 95,8 |
|  |  | SW4 | 849 | 4,2 | 4,2 | 100,0 |
|  |  | Total | 20078 | 100,0 | 100,0 |  |

Figure 2.1c Aircraft type used for cargo

In the three markets, the utilization of narrow-body aircrafts prevails over the wide ones. This is not a surprise, as usually there are more short-haul flights than the long-haul ones and these are mainly operated by narrow-body aircraft whilst the long-haul flights are operated by wide-body aircrafts.

But what differs between these markets is the type of aircraft used and the percentage of usage of every type. Low-cost carriers mostly use narrow-body planes as they mainly have short-haul flights (except for a few airlines that operate both short- and long-haul), and they usually have the same kind of plane on their fleet. Traditional carriers operate both short- and long-haul flights and they have a diversity of aircrafts in their fleet.

For this, the percentage of usage of wide-body planes is higher than the one for lowcost carriers. The transport of cargo is either by short- and long-haul flights. This translates into a more equal percentage of plane usage.

Table 2.1. shows the usage percentage of wide- and narrow-body planes for the three different markets.

Table 2.1. Usage percentage of wide- and narrow-body planes

| AIRCRAFT TYPE | LOW-COST CARRIERS | TRADITIONAL CARRIERS | CARGO |
| :---: | :---: | :---: | :---: |
| NARROW-BODY | $98.4 \%$ | $85.4 \%$ | $61.5 \%$ |
| WIDE-BODY | $1.6 \%$ | $14.6 \%$ | $38.5 \%$ |

Figure 2.2 shows the aircraft type vs range on low-cost, traditional carriers, and cargo. A flight is considered long-haul if the distance travelled is higher than 2000 nm . So, this figure supports the analyses of the previous data obtained in this study.


Figure 2.2a Aircraft type vs Range on Traditional carriers


Figure 2.2b Aircraft type vs Range on Low-cost carriers


Figure 2.2c Aircraft type vs Range on Cargo

## CHAPTER 3. CASE OF STUDY

At this point, the situation is clear. The amount of air traffic is increasing very fast. There are more short- than long-haul flights and the narrow-body aircraft are the most used. If the situation keeps developing this way, soon there will not be physical space to keep carrying all the programmed flights, and the delays will increase very fast.

Besides, some extra investigation has been done. It has been found that trips between two specific cities are flown many times a day with a narrow-body aircraft. But with all the capacity problems that have been mentioned before, it is easy to ask what would happen if instead of flying so many times with a narrow-body, that specific route would be flown less frequently but with a wide-body plane instead.

This would mean a less saturated system. The number of planes taking off and landing, which is the constraint for capacity, would be reduced. Less delays due to capacity purposes and more environment friendly with less contamination and of course, less consumption of fuel.

In fact, the aim of this project is to study all the factors that determine whether it is better to use several narrow-body or fewer wide-body planes. Therefore, the route between Josep Tarradellas Barcelona-El Prat airport (BCN) and Gran Canaria airport (LPA) has been chosen. Even though there are many pairs of cities where there are constant planes transferring passengers, this one is perfect for this study due its tourism purposes. Other cities need more connectivity as on top of being touristic, many businesses passengers are travelling and business meetings can happen all along the day. But, in a touristic destination, the hour of arrival or departure it is not a big factor to consider.

To make the study case, the aircrafts to compare will be an A320 and an A330, carrying 180 and 300 passengers, respectively. The assumptions to take into account are that the planes will be completely full, carrying $45 \%$ of men, $45 \%$ of women and $10 \%$ of children and every adult will take a bag in the hold. Also, the study will consist in the evaluation of the turn flight including the turn round times in Josep Tarradellas Barcelona-EI Prat airport (BCN) and Gran Canaria airport.

The study case will be simplified as much as possible as the interest focuses on the comparison of the planes under the same circumstances. For that, no wind will be considered. It will be divided in different sections. First, the route will be stablished, then the weight and balance of both planes will be evaluated and finally the economic costs and environmental effects will be analysed.

### 3.1. ROUTE

The route will take place between Josep Tarradellas Barcelona-El Prat airport and Gran Canaria airport (BCN-LPA), which corresponds to 1191 nm or 2205.73km. The plane will depart from $B C N$ using the runway $25 R$ and will be following the procedure of the SID to arrive to the point LOTOS5D. [8] Then, the next route will be performed:

$$
\begin{aligned}
& \text { | }) \text { LOTOS B28 DIKUT } \rightarrow \text { LEVC } \rightarrow \text { SERRA B28 ASTRO B28 } \\
& \text { XEBAR UM985 MAMIS B28 BAZAS UM985 VIBAS UM985 } \\
& \text { MGA UN851 BRIKE UN851 VJF UL82 TAVSI UL82 IBALU } \\
& \text { A857 ERMED UN857 TOVRA A857 VEDOD UN857 TERTO } \\
& \text { A857 LZR UN871 GDV } \rightarrow
\end{aligned}
$$

Once the waypoint TERTO is reached, the approximation begins. The STAR TERTO7C procedure will be followed and the runway used to land will be 03L. [9] Figure 3.1 shows the route followed by the plane.


Figure 3.1 Route BCN-LPA [10]

In case of impossibility of landing in Gran Canaria airport, an alternate airport must be contemplated. In this case, Tenerife Norte airport has been chosen, being at a distance of 59 nm from Gran Canaria airport.

### 3.2. WEIGHT AND BALANCE

Following the steps mentioned before, the weight and balance of both planes will be calculated. This is very important as it has a direct effect on the stability and performance of the aircraft. Calculating all the weights and comparing them with the maximum weights given by the manufacturer is essential to assure that it is safe to fly in these conditions. Another very important factor is to know the needed fuel to reach the destination safely and the possible setbacks that can occur.

EASA normalized the mass of the passengers according to their gender to 88kg, 70 kg and 35 kg for men, women and children. Also, there is a normalized weight for the bags in the hold to 11 kg per suitcase. This information will help us calculate the payload of every plane. [11]

The Airbus Commercial Aircraft (ACAP) will be checked and the needed data will be collected. [12] Also, the consideration that the plane flies at FL350 in cruise from departure to destination and FL100 from destination to alternate will be taken into account.

### 3.2.1. WEIGHT AND BALANCE A320

For the calculations of the weight and balance for the A320 the tables of annex 3 will be needed.

Operational Empty weight $($ OEW $)=40.500 \mathrm{~kg}[12]$
Payload $=$ PAX num $* \%$ males $*$ standard male weight + PAX num $* \%$ females

* standard female weight + PAX num $* \%$ children
* standard children weight + PAX num $* \%$ passengers with bags
* standard bag weight
$=180 * 0.45 * 88+180 * 0.45 * 70+180 * 0.10 * 35+180 * 0.90 * 11$
$=15.210 \mathrm{~kg}$
Zero Fuel Weight $=$ Operational Empty weight + payload $=40.500+15.210$
$=55.710 \mathrm{~kg}$
Block Fuel $=$ Taxi fuel + Trip fuel + Holding fuel + Alternate fuel + Reserve fuel
Taxi fuel $=120 \mathrm{~kg}$ (average quantity per 12 minutes of taxi)[13]
Holding fuel $=1.063 \mathrm{~kg}[13]$

The calculation of the alternate fuel is not simply checking a table and directly getting the desired value. In this case there are some steps to follow and some calculations if interpolation is needed. First, the fuel weight to travel from Gran Canaria airport to Tenerife Norte airport must be found in the correct table, given the distance of 59 nm between them [13]. Then a fuel burn correction regarding the estimated landing weight at the alternate airport and a reference weight ( 50.000 kg ) must be calculated. Finally, this extra burnt fuel will be added to the fuel weight needed between the two airports.

Alternate fuel (without considering the extra fuel burnt) $=611 \mathrm{~kg}$
Estimated landing weight at alternate(ELWalt) $=$ Zero fuel weight + holding fuel

$$
=55.710+1.063=56.773 \mathrm{~kg}
$$

Extra fuel burnt $=($ ELWalt - Reference weight $) *$ Fuel correction

$$
=(56,773-50) * 4=27,09 \mathrm{~kg}
$$

Alternate fuel $=611+27,09=638,09 \mathrm{~kg}$
Likewise, the trip weight will be calculated the same way as the alternate weight. But in this case, the pair of cities will be departure and destination (BCN-LPA).

Trip fuel (without considering the extra fuel burnt) $=6.237 \mathrm{~kg}$
Estimated landing weight at destination (ELWdest)
$=$ Estimated landing weight at alternate + trip destination to alternate $=56.773+631,32=57.404 \mathrm{~kg}$

Extra fuel burnt $=($ ELWdest - Reference weight $) *$ Fuel correction

$$
=(57,404-50) * 47=348 \mathrm{~kg}
$$

Trip fuel $=6.237+348=6.585 \mathrm{~kg}$
Reserve fuel $(5 \%$ trip fuel $)=6.585 * 0.05=329,25 \mathrm{~kg}$
Block fuel $=120+1.063+638,09+6.585+329,25=8.735 \mathrm{~kg}$
Takeoff weight $=$ OEW + Payload + Block Fuel - Taxi fuel

$$
=40.500+15.210+8.735-120=64.325 \mathrm{~kg}
$$

Landing weight $=$ Takeoff weight - Trip fuel $=64.325-6.585=57.740 \mathrm{~kg}$
Ramp weight $=$ OEW + Payload + Block Fuel $=40.500+15.210+8.735$

$$
=64.445 \mathrm{~kg}
$$

Table 3.1. A320 Weights summary.

| WEIGHTS | KG |
| :---: | :---: |
| OEW | 40.500 |
| PAYLOAD | 15.210 |
| BLOCK FUEL* | 8.735 |
| TAKE-OFF | 64.325 |
| LANDING | 57.740 |
| ZERO FUEL | 55.710 |
| RAMP FUEL | 64.445 |


| ${ }^{*}$ BLOCK FUEL | KG |
| :---: | :---: |
| TAXI | 120 |
| TRIP | 6.585 |
| HOLDING | 1.063 |
| ALTERNATE | 638,09 |
| RESERVE | 329,25 |
| TOTAL | $\mathbf{8 . 7 3 5}$ |

Next step, is to compare the maximum design weights with the ones obtained in this study and check if they are lower than the maximums.

Table 3.2. Comparison design weights. [12]

| WEIGHTS | MAXIMUM | CURRENT |
| :---: | :---: | :---: |
| TAKE-OFF | 70.000 kg | 64.325 kg |
| LANDING | 64.500 kg | 57.740 kg |
| ZERO FUEL | 60.500 kg | 55.710 kg |
| RAMP FUEL | 74.500 kg | 64.445 kg |

As table 3.2. shows, the calculated weights are never higher than the maximum design weights. Hence, flying in these conditions would be safe and it is possible to continue with the study.

### 3.2.2. WEIGHT AND BALANCE A330

The same procedure will be performed for the A330. This the tables needed will be found in annex 4.

Operational Empty weight $($ OEW $)=122.300 \mathrm{~kg}[14]$

$$
\begin{aligned}
\text { Payload }= & \text { PAX num } * \% \text { males } * \text { standard male weight }+ \text { PAX num } * \% \text { females } \\
& * \text { standard female weight }+ \text { PAX num } * \% \text { children } \\
& * \text { standard children weight + PAX num } * \% \text { passengers with bags } \\
& * \text { standard bag weight } \\
& =300 * 0.45 * 88+300 * 0.45 * 70+300 * 0.10 * 35+300 * 0.90 * 11 \\
& =25.350 \mathrm{~kg}
\end{aligned}
$$

Zero Fuel Weight $=$ Operational Empty weight + payload $=122.300+25.350$ $=147.650 \mathrm{~kg}$

Block Fuel $=$ Taxi fuel + Trip fuel + Holding fuel + Alternate fuel + Reserve fuel
Taxi fuel $=300 \mathrm{~kg}$ (average quantity per 12 minutes of taxi)[15]
Holding fuel $=1.100 \mathrm{~kg}[15]$
For the calculation of the alternate fuel, the reference weight, also found in the table corresponds to 140.000 kg .

Alternate fuel (without considering the extra fuel burnt) $=1.674 \mathrm{~kg}$
Estimated landing weight alternate(ELWalt) = Zero fuel weight + holding fuel

$$
=147.650+1.100=148.750 \mathrm{~kg}
$$

Extra fuel burnt $=($ ELWalt - Reference weight $) *$ Fuel correction

$$
=(148.750-140.000) * 2=17,5 \mathrm{~kg}
$$

Alternate fuel $=1.674+17,5=1691,5 \mathrm{~kg}$
Trip fuel (without considering the extra fuel burnt) $=13.930 \mathrm{~kg}$
Estimated landing weight destination (ELWdest)

$$
\begin{aligned}
& =\text { Estimated landing weight alternate }+ \text { trip destination to alternate } \\
& =148.750+13.930=162.680 \mathrm{~kg}
\end{aligned}
$$

Extra fuel burnt $=($ ELWdest - Reference weight $) *$ Fuel correction

$$
=(162.680-140.000) * 43=975,24 \mathrm{~kg}
$$

Trip fuel $=13.930+975,24=14.905 \mathrm{~kg}$
Reserve fuel $(5 \%$ trip fuel $)=14.905 * 0.05=745,26 \mathrm{~kg}$
Block fuel $=300+1.100+1.691,5+14.905+745,26=18.742 \mathrm{~kg}$
Takeoff weight $=$ OEW + Payload + Block Fuel - Taxi weight

$$
=122.300+25.350+18.742-300=166.092 \mathrm{~kg}
$$

Landing weight $=$ Takeoff weight - Trip fuel $=166.092-14.905=151.187 \mathrm{~kg}$
Ramp weight $=$ OEW + Payload + Block Fuel $=122.300+25.350+18.742$

$$
=166.392 \mathrm{~kg}
$$

Table 3.3. A330 Weights summary

| WEIGHTS | KG |
| :---: | :---: |
| OEW | 122.300 |
| PAYLOAD | 25.350 |
| BLOCK FUEL* | 18.742 |
| TAKE-OFF | 166.092 |
| LANDING | 151.187 |
| ZERO FUEL | 147.650 |
| RAMP FUEL | 166.392 |


| *BLOCK FUEL | KG |
| :---: | :---: |
| TAXI | 300 |
| TRIP | 14.905 |
| HOLDING | 1.100 |
| ALTERNATE | $1.691,5$ |
| RESERVE | 745,26 |
| TOTAL | $\mathbf{1 8 . 7 4 2}$ |

Next step, is to compare the maximum design weights with the ones obtained in this study and check if they are lower than the maximums.

Table 3.4. Comparison design weights. [14]

| WEIGHTS | MAXIMUM | CURRENT |
| :---: | :---: | :---: |
| TAKE-OFF | 184.000 kg | 166.092 kg |
| LANDING | 174.000 kg | 151.187 kg |
| ZERO FUEL | 164.000 kg | 147.650 kg |
| RAMP FUEL | 184.900 kg | 166.392 kg |

The same as happened with the A320, the calculated weights for the A330 are never higher than the maximum design weights. Hence, flying in these conditions would be safe and it is possible to continue with the study.

### 3.3. AIRCRAFT RANGE

Another important aspect is to know the maximum range that the aircrafts would be able to achieve with the current payload and compare if this is bigger than the distance needed to travel, including the travel to the alternate airport. In that case it would be 1250 nm .

The Payload/ Range diagram is the one that gives that information by illustrating the trade-off relationship between the payload and the range of one single aircraft.

### 3.3.1. A320 RANGE

Figure 3.2. shows the Payload/Range diagram for the A320.


Figure 3.2. A320 Payload/Range diagram

In this diagram, the purple line is the one that has to be analyzed as it is the closest to the MTOW of this study. The payload is 15.210 kg as the previous section shows. The maximum range for the A320 in these conditions is approximately 2.300 nm , which is higher than the 1.250 nm needed for a safe flight.

### 3.3.2. A330 RANGE

To know the range of the A330 the Payload/Range diagram for this aircraft is shown in figure 3.3.


Figure 3.3 A330 payload/range diagram

In this case the payload loaded in the plane is 25.350 kg and the maximum range is approximately 6.750 nm , which is also higher than the 1250 nm above mentioned.

### 3.3.3. RANGE COMPARISON

Figure 3.4. gives a better view of the range to achieve the journey BCN-LPA plus the alternate, the maximum range of the A320 and the maximum range of the A330.


Figure 3.4 Range for the study, A320 max range and A330 max range. [16]

The smaller one is the range of this study, the middle one is the maximum range of the A320 and the bigger one is the maximum of the A330. Any of the aircrafts would reach safely the destination.

On top, this figure shows the limitations of using an A320 for long range flights, their range is not big enough.

### 3.4. AIRPORT TAXES

The journey is travelled between two Spanish airports, which means that they are ruled by Spanish legislation. AENA is the society in charge of managing them. For that, some taxes, depending on the aircraft and the received services, must be paid to the airport manager, $A E N A$. These taxes can be found in the "Guia de tarifas de Aena 2020" on its website. [17]

For the calculation of the taxes that should be paid for the A320 and the A330, the assumption of travelling inside the airports timetable, travelling only with passengers without special needs and the aircrafts will park at the tube will be considered.

### 3.4.1. LANDING TAXES

The use of the runways to land and the derived services for its use is one of the taxes to pay. The price is calculated accordingly to the maximum take-off weight of the aircraft and it can vary depending on the type of flight and the acoustic rating. Depending on the acoustic rating, a percentage to pay might be added to the tax.

Looking at figure 3.2, both the A320 and the A330 have a cumulative noise margin slightly over-10EPNdb and are categorized as Chapter 4.


Figure 3.5 Aircraft cumulative noise margin

Knowing the chapter and looking at the table 3.5. the extra percentage to pay for these aircrafts is $0 \%$.

Table 3.5. Extra percentage according to the chapter of the aircraft.

| Acoustic <br> classification | 07:00-22:59 <br> (local time) | 23:00-06:59 <br> (local time) |
| :---: | :---: | :---: |
| Chapter 1 | $70 \%$ | $140 \%$ |
| Chapter 2 | $20 \%$ | $40 \%$ |
| Chapter 3 | $0 \%$ | $0 \%$ |
| Chapter 4 | $0 \%$ | $0 \%$ |

The last thing to consider is the type of flight and the airport where the landing is performed. For landing in Gran Canaria airport, the tax to pay is 6,910621 * MTOW(tones) and in Josep Tarradellas Barcelona-El Prat airport is 7,285036* MTOW (tones). Table 3.6. shows the total price for every plane.

Table 3.6. Landing taxes.

| AIRPORT | A320 | A330 |
| :---: | :---: | :---: |
| LPA | $483,74 €$ | $1.271,55 €$ |
| BCN | $509,95 €$ | $1.340,44 €$ |

### 3.4.2. PASSENGERS AND SECURITY

The tax for passengers allows them the access to the needed airport facilities with the purpose to reach the aircraft.

The security tax is for the inspection and control of the passengers and their suitcases as well as all the equipment and facilities for surveillance services in all the airport.

The taxes are applied to every departure passenger and it might differ at different airport.

In Gran Canaria airport is $5,03 €$ per passenger and 2.87 per passenger security and in Josep Tarradellas Barcelona-El Prat airport 13,25€ and 3.38 respectively.

The tax to pay is calculated as follows:

A320

$$
\begin{aligned}
& \text { Gran Canaria } \rightarrow(5,03+2,87) * 180=1.442 € \\
& \text { Barcelona } \rightarrow(13,25+3,38) * 180=2.993,40 €
\end{aligned}
$$

A330

$$
\begin{aligned}
& \text { Gran Canaria } \rightarrow(5,03+2.87) * 300=2.370 € \\
& \text { Barcelona } \rightarrow(13,25+3,38) * 300=4.989 €
\end{aligned}
$$

Table 3.7. Passengers and security taxes.

| AIRPORT | A320 | A330 |
| :---: | :---: | :---: |
| LPA | $1.442 €$ | $2.370 €$ |
| BCN | $2.993,40 €$ | $4.989 €$ |

### 3.4.3. USAGE OF PARKING BRIDGE

Parking in a position and using (or not) the bridge to move passengers corresponds a tax. It is calculated according to the time that it is used. For an A320 the turn round time is of 44 minutes and for an A330 59 minutes. [12] [14]

Figure 3.3 shows the corresponding TRT for both aircrafts.


TRT: 59 min


Figure 3.6. Turn Round Time A320 (left) and Turn Round Time A330 (right)

The tax is calculated following the equation:

$$
\mathrm{P}=(\mathrm{p} 1+\mathrm{p} 2 * \mathrm{Tm}) * \mathrm{Ft}
$$

where:
P: Tax to pay.
p1: Amount time staying in the tube.
p2: Amount per weight and time staying in the tube.
Tm: maximum take-off weight in tones.
Ft: Time staying in the tube in periods of 15 minutes.
The same as happens with the other taxes, the amount depends on the airport. For Gran Canaria airport p1 corresponds to $20,473658 €$ and p2 to $0 €$. In Josep Tarradellas Barcelona-El Prat airport is $23,280197 €$ and $0 €$ for p1 and p2. This makes the calculations as follows:

A320

$$
\begin{gathered}
\text { Gran Canaria } \rightarrow(20,473658+0) * \frac{44}{15}=60,05 € \\
\text { Barcelona } \rightarrow(23,280197+0) * \frac{44}{15}=68,29 €
\end{gathered}
$$

A330

$$
\begin{aligned}
& \text { Gran Canaria } \rightarrow(20,473658+0) * \frac{59}{15}=80,53 € \\
& \text { Barcelona } \rightarrow(23,280197+0) * \frac{59}{15}=91,56 €
\end{aligned}
$$

Table 3.8. Usage of the parking bridge.

| AIRPORT | A320 | A330 |
| :---: | :---: | :---: |
| LPA | $60,05 €$ | $80,53 €$ |
| BCN | $68,29 €$ | $91,56 €$ |

### 3.4.4. FUEL SUPPLY

This tax is charged for the use of the airport facilities to transport and supply the fuel. Unlike the other taxes, this is the same amount regardless the airport. The price is 0,003771€/Liter

For the calculation of this tax, the weight and balance have to be checked, specifically the block fuel. Then, the kilos have to be converted to liters and the calculation can easily be made. An average density of kerosene of $810 \mathrm{~kg} / \mathrm{m}^{3}$ has been chosen.

$$
\begin{aligned}
& \text { A320 Block fuel(conversion kg to l) }=\frac{8.735}{810} * 1000=10.776,54 \mathrm{l} \\
& \text { A320 } \rightarrow 10.776,54 * 0,003771=40,64 €
\end{aligned}
$$

A330 Block fuel $($ conversion kg to l$)=\frac{18.742}{810} * 1000=23.138,27 \mathrm{l}$

$$
\text { A330 } \rightarrow 23.138,27 * 0,003771=87,25 €
$$

Table 3.9. Tax to supply fuel.

| TAX | A320 | A330 |
| :---: | :---: | :---: |
| Fuel Supply | $40,64 €$ | $87,25 €$ |

### 3.4.5. AIRCRAFT GROUND ASSISTANCE

This tax is divided into the different ground services that can be given to the aircraft and it is charged for the use of the airport facilities to achieve this purpose. Most of the services taxes remain the same for every airport except for the catering service, whose price varies accordingly to the departure airport. In this study, it is assumed that the aircrafts only need to be filled in Josep Tarradellas Barcelona-El Prat airport with food and beverages.

Additionally, the maximum take-off weight, divided into intervals, might also change the price of all the services taxes. The prices for the A320 and A330 are shown in table 3.10.

Table 3.10. Aircraft ground assistance taxes.

| SERVICE | A320 | A330 |
| :---: | :---: | :---: |
| Baggage assistance | $70,64 €$ | $126,07 €$ |
| Runway operations assistance | $22,44 €$ | $40,25 €$ |
| Cleaning services | $12,32 €$ | $22,09 €$ |
| Line maintenance services | $3,34 €$ | $5,99 €$ |
| Catering service | $20,22 €$ | $36,27 €$ |
| Total | $\mathbf{1 2 8 , 9 6 €}$ | $\mathbf{2 3 0 , 6 7} €$ |

### 3.4.6. METEOROLOGICAL SERVICES

Tax because of the meteorological services given to the airlines by the airport manager. This tax is not linked to the airport, but to the aircraft weight. The price is obtained by the calculation of 0,181781 $*$ TOW(tones).

Table 3.11. Meteorology services tax.

| TAX | A320 | A330 |
| :---: | :---: | :---: |
| Meteorological services | $11,65 €$ | $27,81 €$ |

### 3.4.7. GROUND POWER SERVICE

The usage of the equipment and the airport facilities to supply electrical energy transformed in 400 hertz to the aircrafts is the last tax to consider. The MTOW (in intervals) and the time of use of this service are the aspects to calculate the price. Every aircraft is included in a different interval, so the amount to pay for the A320 is $6,920039 €$ and for the A330 is $13,50 €$ per 15 minutes of usage. Remembering the turn round time of the aircrafts, the calculations are the following:

$$
\begin{aligned}
& \mathrm{A} 320 \rightarrow 6,920039 * \frac{44}{15}=20,30 € \\
& \mathrm{~A} 330 \rightarrow 13,5 * \frac{59}{15}=53,10 €
\end{aligned}
$$

Table 3.12. 400 Hz energy system service tax.

| TAX | A320 | A330 |
| :---: | :---: | :---: |
| 400 Hz energy system service | $20,30 €$ | $53,10 €$ |

### 3.4.8. OTHER SERVICES

There are other services, as checking desks, offices, machines and more that taxes are mandatory as well. But, for the aim of this project they can be neglected.

### 3.4.9. TOTAL TAXES PRICE

Once all the taxes are developed, it is possible to calculate the total taxes to pay to the airport manager if a plane wants to operate in their airports. Table 3.13. shows the total prices when the plane departs in J Josep Tarradellas Barcelona-El Prat airport and lands in Gran Canaria airport, the other way around and the total amount if the plane does both journeys.

Table 3.13. Total taxes prices.

| JOURNEY | A320 | A330 |
| :---: | :---: | :---: |
| BCN->LPA | $3.739,74 €$ | $6.739,91 €$ |
| LPA->BCN | $2.201,57 €$ | $4.164,56 €$ |
| Total | $\mathbf{5 . 9 4 1 , 3 1} €$ | $\mathbf{1 0 . 9 0 4 , 4 7 €}$ |

### 3.5. ATC TAXES

As AENA is the agency who manages the airports in the Spanish territory, ENAIRE is the one who manages the air navigation services. Thus, is who charges the taxes, which are divided into En-route charges and Terminal navigation charges. ENAIRE launched the "Guide to air navigation charges 2021" where these taxes are explained and can be calculates. [18]

### 3.5.1. EN-ROUTE CHARGES

This tax is related to the use of the en-route air navigation facilities and services. The tax is calculated accordingly to the next equation:

$$
r_{i}=t * N
$$

where:
$\mathbf{r}_{\mathrm{i}}$ : Total charge.
t: Spanish unit rate of the charge. This rate depends on the area of flying, differentiating between continental and Canary Island areas.

The flight of this study trespasses both areas. Therefore, it has been calculated the amount ok kilometres travelled in each area. This is 491 nm continental area and 700 nm in the Canary Island one.
$\mathbf{N}$ : Number of service units. This number is related to the distance travelled and the MTOW.

The calculation is $N=\frac{\text { Great circle distance travelled }(\mathrm{km})}{100} * \frac{\text { MTOW(tons) }}{50} * 0.5$
The total calculation according to this study is:

$$
\begin{aligned}
& \mathrm{A} 320=45,44 * \frac{491}{100} *\left(\frac{70}{50}\right)^{0.5}+40 * \frac{700}{100} *\left(\frac{70}{50}\right)^{0.5}=595,29 € \\
& \text { A330 }=45,44 * \frac{491}{100} *\left(\frac{184}{50}\right)^{0.5}+40 * \frac{700}{100} *\left(\frac{184}{50}\right)^{0.5}=965,13 €
\end{aligned}
$$

Table 3.14. En-route charges.

| TAX | A320 | A330 |
| :---: | :---: | :---: |
| En-route charges | $595,29 €$ | $965,13 €$ |

### 3.5.2. TERMINAL NAVIGATION CHARGES

The terminal navigation charges is associated to the use of air navigation services to ensure the safety and orderly flow of movements at the approach and take-off.

The formula for the charge levied is:

$$
R=\mathrm{t} *(\mathrm{P} / 50)^{\mathrm{n}}
$$

Where:
R: Total charge per operation.
t: Spanish unit rate of the charge. In this case, this rate depends on the departure airport and it is the same for Josep Tarradellas Barcelona-El Prat airport and Gran Canaria airport.

P: Licensed MTOW of the aircraft.
n: Weighting coefficient. Always 0,7 .
The total calculation in this case is:

$$
\begin{aligned}
& \mathrm{A} 320 \rightarrow 20,01 *\left(\frac{70}{50}\right)^{0.5}=25,32 € \\
& \mathrm{~A} 330 \rightarrow 20,01\left(\frac{184}{50}\right)^{0.5}=49,81 €
\end{aligned}
$$

Table 3.15. Terminal Navigation charges.

| TAX | A320 | A330 |
| :---: | :---: | :---: |
| Terminal navigation charges | $25,32 €$ | $49,81 €$ |

### 3.5.3. TOTAL ATC CHARGES

The total ATC charges are simply the addition of the En-route charges and the Terminal navigation charges. This is for one leg of the journey. As in this study we want to do a comparison when the plane does both legs, the only thing we would need is having to apply these taxes 2 times. Table 3.16 shows the total ATC charges.

Table 3.16. Total ATC charges.

| CHARGES | A320 | A330 |
| :---: | :---: | :---: |
| En-route | $595,29 €$ | $965,13 €$ |
| Terminal navigation | $25,32 €$ | $49,81 €$ |
| Total $(1$ leg $)$ | $620,61 €$ | $1.014,94 €$ |
| Total | $\mathbf{1 . 2 4 1 , 2 2} €$ | $\mathbf{2 . 0 2 9 , 8 8} €$ |

ENAIRE provides a document with all the related information and the prices needed to do all these calculations. These amounts are excluding taxes, so the real taxes to pay will be higher than the ones shown in this project. But once again, the focus of this project is to make a comparison of the two planes in the same conditions.

### 3.6. HANDLING COSTS

Until this moment, all the costs explained were for the payment of taxes derived of services. This section will talk about the costs of actual services on the plane.

When an aircraft is on ground it needs ground assistance. It either has to be prepared for departure or get unload if it just landed. These services are called handling, and they are usually done by external companies.

AENA, as the airport manager has to authorize these external companies to provide their services inside the airport facilities and also establishes a maximum rate that these companies can charge to offer their services. Although, under this maximum, the rate they charge is up to the company and the deals are different with every airline.

At Josep Tarradellas Barcelona-El Prat there are three authorized companies: Iberia, Globalia and Swissport. Instead, at Gran Canaria airport there are only two: Iberia and Globalia.

For this study, the handling services are considered to be carried by the same company in both airports and they charge the maximum rate price.

An aircraft classification is made to differentiate the handling prices. Table 3.17. shows the different aircraft classes with an example aircraft type. Then table 3.18. shows the maximum price for the handling services according to the aircraft class and the type of flight.

Table 3.17. Aircraft class with example of aircraft type.

| Aircraft <br> class | Aircraft type | Aircraft <br> class | Aircraft type |
| :---: | :--- | :---: | :--- |
| $\mathbf{4 A}$ | AEROSPATIALE AS350 Ecureuil <br> CESSNA TWIN PISTON | $\mathbf{7 1}$ | AIRBUS 320 |
| $\mathbf{4 B}$ | AEROSPATIALE SN365 <br> Dauphin <br> AVIOCAR CN212-200 | $\mathbf{7 2}$ | BOEING 727-200 |
| $\mathbf{4 C}$ | EMBRAER 120 BRASILIA <br> AVIOCAR CN235 | $\mathbf{8 1}$ | BOEING 757-300 <br> AIRBUS A310 |
| $\mathbf{3 1}$ | AEROSPATIALE ATR42 DE <br> HAVILLAND DHC-8 | $\mathbf{8 2}$ | AIRBUS A 300 B4/C4/F4 <br> BOEING 767-300 |
| $\mathbf{4 1}$ | AEROSPATIALE ATR 72 <br> CANADAIR REGIONAL JET 900 | $\mathbf{8 3}$ | AIRBUS 340-200 <br> BOEING 777-200 |
| $\mathbf{5 1}$ | BRITISH AEROSPACE 146-300 <br> McDONNELL DOUGLAS DC-9 | $\mathbf{9 1}$ | McDONNELL DOUGLAS MD-11 <br> BOEING 777-300 |
| $\mathbf{6 1}$ | BOEING 737 McDONNELL <br> DOUGLAS MD 83 | $\mathbf{9 3}$ | BOEING 747-200/400 <br> AIRBUS A340-600 |

For the A320 there is no doubt that the aircraft class is 71, as it is explicitly shown in the table. The problem comes with the A330. This plane is in the aircraft class 82 for its resemblances with the aircraft examples of this class.

Table 3.18. Maximum price for handling services.

| Maximum price for handling services |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| AIRCRAFT TYPE | AIRCRAFT CLASS | COMMERCIAL HANDLING |  |  | TECHNICAL HANDLING (PAX) |
|  |  | PAX | MIXED | FREIGHT |  |
|  |  | Euros | Euros | Euros | Euros |
| CESSNA SINGLE POSITION | 4A | 25,45 | 27,49 | 27,49 | 12,22 |
| CESSNA CITATION | 4B | 77,37 | 83,48 | 83,48 | 37,67 |
| EMBRAER 120 | 4 C | 153,73 | 167,98 | 165,95 | 75,34 |
| ATR 42 | 31 | 205,65 | 223,98 | 220,92 | 100,79 |
| ATR 72 | 41 | 328,84 | 357,35 | 354,29 | 160,86 |
| BOEING 717 (DC-9) | 51 | 798,17 | 869,44 | 984,48 | 391,96 |
| BOEING 737-400 | 61 | 1.025,21 | 1.116,83 | 1.264,46 | 502,93 |
| AIRBUS 320 | 71 | 1.233,91 | 1.343,87 | 1.521,01 | 604,74 |
| BOEING 727 | 72 | 1.387,64 | 1.510,83 | 1.710,37 | 680,08 |
| AIRBUS 310 | 81 | 1.541,37 | 1.677,80 | 1.899,74 | 755,42 |
| BOEING 767-300 | 82 | 1.839,67 | 2.003,58 | 2.267,26 | 902,02 |
| AIRBUS 340-200 | 83 | 2.076,88 | 2.261,16 | 2.559,45 | 1.018,08 |
| BOEING 777-300 | 91 | 2.715,22 | 2.957,52 | 3.346,43 | 1.330,63 |
| BOEING 747-400 | 92 | 3.226,30 | 3.514,41 | 3.976,62 | 1.581,08 |

Table 3.19 shows the total cost for the handling of both planes. The prices have to be multiplied by two to have the costs in both Josep Tarradellas Barcelona-El Prat airport and Gran Canaria airport.

Table 3.19. Total handling costs.

|  | A320 | A330 |
| :---: | :---: | :---: |
| Handling costs (per turn round) | $1.233,91 €$ | $1.839,67 €$ |
| Total | $\mathbf{2 . 4 6 7 , 8 2} €$ | $\mathbf{3 . 6 7 9 , 3 4} €$ |

### 3.7. AIRCRAFT OPERATING COSTS

Aircraft operating costs are the costs that a plane produces for being used. The type of aircraft used it is a relevant factor. Therefore, this is the most important cost to study as it includes many elements concerning an aircraft. The operating costs are usually expressed in cost per block hour which state the cost per one hour.

They can be divided into fixed and variable costs depending on the origin of the expense. The variable costs can fluctuate and fixed costs show little or no change.

### 3.7.1. VARIABLE COSTS

The variable costs depend on the fares or taxes of specific services and they can affect directly to aircraft operators and indirectly to the users of air service. These costs include fuel and oil, maintenance and crew salaries.

### 3.7.1.1. Fuel and Oil costs

This is the cost of the fuel and oil loaded in the plane. The price is based on the price of the fuel at the given moment and the fuel consumption for a given operation. This cost changes when the price of the fuel changes, which is constantly.

### 3.7.1.2. Maintenance costs

In order to meet the safety requirements at the airline industry, maintenance costs are unavoidable. Maintenance can be scheduled or unscheduled. The last one is what it makes this cost variable.

It includes maintenance labor; airframe, engine and avionics parts; APU, propeller and Thrust reverse overhaul and dynamic components cost.

### 3.7.1.3. Crew costs

It includes the crew salaries, including pilot, co-pilot, flight engineers and stewardesses. It also includes the trainers and instructors, the personnel expenses and employees benefit, which are an extra on their wage.

### 3.7.2. FIXED COSTS

The fixed cost does not oscillate and in short-term, they are independent of a change of activity. They include depreciation, rentals, insurance and others.

### 3.7.2.1. Depreciation

It is assumed that aircrafts lose a fixed percentage of their original purchase price each year, which converts into a cost and it is a significant component for fixed costs.

### 3.7.2.2. Rentals

The rentals of hangars and other facilities in order to maintain the activity is also an important fixed cost.

### 3.7.2.3. Insurance and others

The obligatory purchased insurances, sales costs, administration, accounts, general management and employment costs, among others are the rest of the fixed costs.

### 3.7.3. OPERATION COSTS CALCULATIONS

The operating costs of a plane depend on the air carrier who is using the aircraft, the fleet of the airline and the utilisation of the aircraft. Meaning that the operating costs are not equal for the same aircraft. The Federal Aviation Administration (FAA) released a document with standard values, which are the ones that will be used. [19] Once again, emphasizing the importance of the comparison in this project, these values are satisfactory. Table 3.20. shows in detail the variables and fixed costs per block hour of the different aircraft types.

Table 3.20. Aircraft type variable and fixed costs per block hour. [10]

| Aircraft Category | Cost per Block Hour |  |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Fuel and Oil | Maintenance | Crew | Total Variable | Depreciation | Rentals | Insurance | Other | Total Fixed | Total |
| Wide-body more than 300 seats | \$5,411 | \$1,331 | \$2,356 | \$9,097 | \$845 | \$406 | \$4 | \$1 | \$1,254 | \$10,351 |
| Wide-body 300 seats and below | \$4,080 | \$1,289 | \$1,857 | \$7,227 | \$685 | \$366 | \$4 | \$4 | \$1,058 | \$8,285 |
| Narrow-body more than 160 seats | \$2,054 | \$718 | \$1,152 | \$3,925 | \$355 | \$217 | \$3 | \$7 | \$582 | \$4,506 |
| Narrow-body 160 seats and below | \$1,741 | \$737 | \$1,034 | \$3,512 | \$306 | \$215 | \$5 | \$7 | \$533 | \$4,045 |
| RJ more than 60 seats | \$115 | \$431 | \$444 | \$991 | \$131 | \$252 | \$1 | \$13 | \$397 | \$1,388 |
| RJ 60 seats and below | \$92 | \$479 | \$470 | \$1,041 | \$58 | \$227 | \$1 | \$7 | \$293 | \$1,334 |
| Turboprop more than 60 seats | \$0 | \$880 | \$360 | \$1,241 | \$439 | \$103 | \$0 | \$2 | \$544 | \$1,785 |
| All Aircraft | \$1,681 | \$727 | \$1,012 | \$3,420 | \$314 | \$239 | \$4 | \$7 | \$564 | \$3,985 |

A cost per block hour is the total cost that a plane produces in a one-hour block. The block time is the time that the airplane needs for its operation, starting with the taxi at the departure airport until the end of the taxi at the arrival airport.

Afterwards, the total block time for this study needs to be calculated. For the taxi time, 12 minutes average time is assumed either at Josep Tarradellas Barcelona-El Prat airport and at Gran Canaria airport. This means 12 minutes per 4 taxis is a total of 48 minutes in taxi. The total flying time can be extracted from the tables of the Flight Crew Operating Manual (FCOM). (Annex 3 and Annex 4)

The total block time for the A320 and A330 is the following:

$$
\begin{gathered}
\text { A320 } \rightarrow \text { taxi }(\mathrm{BCN})+\text { flying time }+ \text { taxi }(\mathrm{LPA})+\text { taxi }(\mathrm{LPA})+\text { flying time }+ \text { taxi }(\mathrm{BCN}) \\
=12+171+12+12+171+12=389 \text { minutes }=6 \mathrm{~h} 29 \mathrm{~min}
\end{gathered} \quad \begin{gathered}
\mathrm{A} 330 \rightarrow \operatorname{taxi}(\mathrm{BCN})+\text { flying time }+ \text { taxi }(\mathrm{LPA})+\text { taxi }(\mathrm{LPA})+\text { flying time }+ \text { taxi }(\mathrm{BCN}) \\
=12+168+12+12+168+12=384 \text { minutes }=6 \mathrm{~h} 24 \mathrm{~min}
\end{gathered}
$$

Considering that the block times can be counted per period of times of 15 minutes, both aircrafts would need a block time of 6 h and 30 min . This means that the results in the table have to be multiplied by 6.5 in order to get the total operating costs. It is important to mention that the prices of the table are in dollars, so a conversion to euros will be needed first.

Table 3.21. shows the operation cost per block hour and the total operating costs of the A320 and A330 to perform the journey that is being studied.

Table 3.21. Total aircraft operating cost.

|  | A320 | A330 |
| :---: | :---: | :---: |
| Operating cost per block hour | $3.696,29 €$ | $6.796,23 €$ |
| Total aircraft operation cost | $24.025,89 €$ | $44.175,50 €$ |

### 3.8. COST RESULTS

Once all the costs have been calculated, the results will be compiled. Table 3.22 shows a global overview of all the costs of a return flight from Josep Tarradellas Barcelona-El Prat airport and Gran Canaria airport.

Table 3.22. Costs to perform a return flight BCN-LPA.

|  | A 320 | $\mathbf{A} 330$ |
| :--- | :--- | :--- |
| AENA TAXES | $5.941,74 €$ | $10.904,47 €$ |
| ATC TAXES | $1.241,22 €$ | $2.029,88 €$ |
| HANDLING COSTS | $2.467,82 €$ | $3.679,34 €$ |
| OPERATING COSTS | $24.025,89 €$ | $44.175,50 €$ |
| TOTAL | $\mathbf{3 3 . 6 7 6 , 6 7} €$ | $\mathbf{6 0 . 7 8 9 , 1 9 €}$ |

The operating costs is the parameter that determines the total costs, as they are surprisingly higher than the rest. This is due to the fact that the operating costs include several costs related to the motion of the aircraft, meanwhile the others are taxes related or only indirect related to this motion.

The total price of performing the return flight and the big differences between planes is surprising.

Another way of interpreting these results is to observe the price per passenger. This number is obtained dividing the total cost by the number of passengers that every plane carries. As the total costs obtained are for a return flight, the passengers are counted two times.

$$
\begin{aligned}
& \text { Price per passenger } \mathrm{A} 320=\frac{33.676,67}{360}=93,55 \frac{€}{\text { pax }} \\
& \text { Price per passenger } \mathrm{A} 330=\frac{60.789,19}{600}=101,32 \frac{€}{\mathrm{pax}}
\end{aligned}
$$

Besides, there are two other interesting cases to know the price per passenger. The first one is the case where the price is higher for the A320 than for the A330. This happens when 27 seats of the A320s are empty, which corresponds to $8 \%$ of the total.

$$
\text { Price per passenger } A 320(8 \% \text { empty seats })=\frac{33.676,67}{332}=101,43 \frac{€}{\operatorname{pax}}
$$

The other one is when in both aircrafts are carried the same number of passengers, 300. This happens when the A330 is full and when in the A320s 60 seats, or $17 \%$ of the total are empty.

$$
\text { Price per passenger } \mathrm{A} 320(17 \% \text { empty seats })=\frac{33.676,67}{300}=112,26 \frac{€}{\text { pax }}
$$

### 3.9. ENVIRONMENTAL IMPACT

The last topic to be studied is the environmental impact of flying. Aircraft engines produce emissions and those are harmful for the environment.

Studies estimate that the total $\mathrm{CO}_{2}$ aviation emissions are approximately $3 \%$ of the Global Greenhouse emissions, a very low percentage.

The Greenhouse gases extracted from airplanes are C02, H 2 O , sulphites, soot, and nitrogen oxide ( NOx ). The emission index for nitrogen oxide varies with altitude, Mach number, and fuel-flow rate, but it can be estimated with a fuel-flow-correlation model.

The aim of this project is not to calculate all these values, so they have been extracted from another study. [20] Table 3.23 shows a comparison of emission per passenger for a 1500 nm flight. The data extracted is a bit higher than the one it would be for this study, but once again the importance is to do the calculations for both planes under the same basis in order to be able to get a reliable comparison.

Table 3.23. Comparison of emission per passenger for the A320 and A330 over a 1.500 nm flight.

|  | A320 | A330 |
| :---: | :---: | :---: |
| PAX number | 150 | 293 |
| Fuel burnt | 8.797 kg | 18.749 kg |
| Emissions $(\mathrm{kg} / \mathrm{pax})$ |  |  |
| $\mathrm{CO}_{2}$ | 185 | 202 |
| $\mathrm{H}_{2} \mathrm{O}$ | 72,1 | 78,7 |
| Sulphites | 0,0117 | 0,0128 |
| Soot | 0,00235 | 0,00256 |
| $\mathrm{NO}_{\mathrm{x}}$ | 0,785 | 1,08 |

The emission shown in table 3.23 are the kg per passenger. But the interest is in the total emission of the plane, which can be calculated for the specifications of the referenced study.

### 3.9.1. A320 emission

$$
\begin{gathered}
\mathrm{CO2}=185 * 150=27.750 \mathrm{~kg} \\
\mathrm{H} 20=72,1 * 150=10.815 \mathrm{~kg} \\
\text { Sulphites }=0,0117 * 150=1,775 \mathrm{~kg} \\
\text { Soot }=0,00235 * 150=0,3525 \mathrm{~kg} \\
\text { NOx }=0,785 * 150=117,75 \mathrm{~kg}
\end{gathered}
$$

### 3.9.2. A330 emission

$$
\begin{gathered}
\mathrm{C} 02=202 * 293=59.186 \mathrm{~kg} \\
\mathrm{H} 20=78,7 * 293=23.059,1 \mathrm{~kg} \\
\text { Sulphites }=0,0128 * 293=3,7504 \mathrm{~kg} \\
\text { Soot }=0,00256 * 293=0,75 \mathrm{~kg} \\
\text { NOx }=1,08 * 293=316,44 \mathrm{~kg}
\end{gathered}
$$

### 3.9.3. Emission comparison

After the different emission are calculated for both planes, table 3.24. shows the emission per a travel of 1500 nm of the A320, emission if the A320 would be used two times and emission of the A330.

Table 3.24. Comparison of aircraft total emission over a 1.500 nm flight.

|  | A320 | $2 \times$ A320 | A330 |
| :---: | :---: | :---: | :---: |
| $\mathrm{CO}_{2}$ | 27.750 kg | 55.500 kg | 59.186 kg |
| $\mathrm{H}_{2} 0$ | 10.815 kg | 21.630 kg | $23.059,10 \mathrm{~kg}$ |
| Sulfites | $1,775 \mathrm{~kg}$ | $3,55 \mathrm{~kg}$ | $3,7504 \mathrm{~kg}$ |
| Soot | $0,3525 \mathrm{~kg}$ | $0,705 \mathrm{~kg}$ | $0,75 \mathrm{~kg}$ |
| $\mathrm{NO}_{\mathrm{x}}$ | $117,75 \mathrm{~kg}$ | $235,5 \mathrm{~kg}$ | $316,44 \mathrm{~kg}$ |

## 4. RESULTS

This section shows and comments on all the results of this study. Table 4.1 shows the total costs to operate a return flight from Josep Tarradellas Barcelona-El Prat airport and Gran Canaria airport with an A320 and an A330.

Table 4.1. Costs to perform a return flight BCN-LPA.

|  | A320 | A330 |
| :---: | :---: | :---: |
| TOTAL COSTS | $33.676,67 €$ | $60.789,19 €$ |

From table 4.1. a big cost difference between operating an A320 and an A330 can be seen. This study considered that the planes fly completely full all the time, which means that the A320 transports 180 passengers and the A330 transports 300 passengers.

In these conditions, it is more profitable to use two times an A320 than one A330, as the relationship between economical cost and number of passengers transported is more efficient with the A320. The A330 can carry 66\% more passengers than the A320 (300 passengers compared to 180), but the cost to use an A330 increases by 80\%.

But the airplanes are not always full, and the occupancy rate is not always $100 \%$. Then, the results change. In the case that two A320 and only one A330 are used, if the number of empty seats is equal or higher than 17\% of the capacity of the A320, in other words, 60 seats are empty, then the option to use the A330 is better. Figure 4.1. shows a representation and a further explanation will be after.


Figure 4.1 Aircraft type vs number of passengers transported

When $17 \%$ of the seats of the A320 are empty, the total number of passengers transported are 300, which is the maximum number of passengers that fit in just one A330. This would mean the possibility of unifying the two A320 into an A330 and the outcome is a reduction of costs for transporting the same number of passengers.
Another factor to consider is that the prices shown above might be modified when airlines get better deals with the airport managers or the external companies that give services. But in this project the maximum prices are the ones that have been used.

A different way of interpreting the results is to observe the price per passenger. Figure 4.2. show the total cost depending on the amount of empty seats.


Figure 4.2. aircraft type vs passenger cost according to the empty seats

Following the same structure as before, if the planes occupancy rate is $100 \%$, the cost per passenger for an A330 is higher than for the A320 and if 17\% of the seats are empty, the situation reverts. But, the difference of using this way of interpretation is the point when the price per passenger for using the A320 becomes higher than for the A330. This occurs when $8 \%$ or 27 of the seats are empty. At this point, with two A320 the number of passengers transported would be 333, which would not fit in an A330. This is the trade-off point in what the decision to transport more passengers at a higher price and in two times or at a lower price and just in one time has to be decided.

Hence, the results depending on the number of empty seats, would be a good seasonal study for airlines to take into account when planning future flights. In summer, the occupancy of the planes is always higher than in winter.
Environmentally speaking, there is an impacting result. Figure 4.3. shows the emissions in kg per aircraft.


Figure 1.3. Aircraft emissions in kg per a 1.500 nm flight

Using an A320 contaminates less than an A330. But the surprising result is that using two A320 contaminates less than a single A330. So, for the environment is better to keep using narrow- above wide-body aircrafts.

Lastly, another concept seems relevant to be mentioned. Nowadays, wide-body aircrafts are designed to fly long- instead of short-haul. Therefore, they are equipped and certified with the necessary instruments.

Perhaps, the possibility of adapting and manufacturing the wide-body aircraft for a short-haul travel, the so-called LASR (Large Aircraft for Short Range) could be the new way of operating. Removing the unnecessary instruments, like aid navigation to cross the Atlantic, and reducing the avionics of the plane removing extra wiring inside the plane, like the one used for the passenger's screens, would be some examples of this adjustment.

The outcome will affect the operational empty weight (OEW), which will be reduced. Therefore, the maximum take-off weight (MTOW) and the fuel consumption will be lower. Consequently, the operating costs and taxes, like landing taxes or meteorology taxes among others will be reduced. On top, the environmental impact would also be reduced.

Next step, knowing the new weights of the LASR aircraft, this study should be repeated to evaluate which option would be better.

## 5.CONCLUSIONS

The conclusions of this project can be divided basically into three general points. Economically, it is better to use narrow-body aircrafts above wide-body for shorthaul flights if the planes are full. Instead, if the planes have a lower occupancy rate, then a wide-body aircraft is better.

Environmentally, using narrow-body aircrafts results the best option.
Finally, the idea of the LASR seems interesting as it could save costs and they could be better for the environment. But it also has negative aspects like the reduction of choices of schedule for the passengers. Although for touristic travels as the one studied in this project, the schedule choice would not have such impact and it would be a good way to start implementing this type of aircrafts.

## BIBLIOGRAPHY

[1] AENA. http://www.aena.es/es/corporativa/aeropuerto-josep-tarradellas-barcelona-el-prat-supera-355-millones-pasajeros-en-ocho-primeros-meses-
2019.html? $p=1237548067436$
[2] Eurostat. https://ec.europa.eu/eurostat/web/transport/data/database
[3] EUROCONTROL, "European Aviation in 2040. Challenges of growth".
[4] EUROCONTROL. https://www.eurocontrol.int/covid19
[5] EUROCONTROL, "Five-Year Forecast 2020-2024", STATFOR - November 2020.
[6] EUROCONTROL R\&D data archive. https://www.eurocontrol.int/dashboard/rnd-data-archive
[7] IBM SPSS Statistics program (TRIAL). https://www.ibm.com/analytics/spss-statistics-software
[8] ENAIRE. "LE_AD_2_LEBL_SID_5_en.pdf", AIP ESPAÑA
[9] ENAIRE. "LE_AD_2_GCLP_STAR_1_en.pdf", AIP ESPAÑA
[10] SkyVector. https://skyvector.com
[11]EASA. "Anexo VI al proyecto de Reglamento de la Comisión sobre "Operaciones aéreas OPS"", https://www.easa.europa.eu/sites/default/files/dfu/EASA 201200020002 ES TR A.pdf
[12] AIRBUS. A320 AIRCRAFT CHARACTERISTICS AIRPORT AND MAINTENANCE PLANNING (2005).
[13] AIRBUS. A318/A319/A320/A321 FLIGHT CREW OPERATING MANUAL (2005).
[14] AIRBUS A330 AIRCRAFT CHARACTERISTICS AIRPORT AND MAINTENANCE PLANNING (2005).
[15] AIRBUS. A330 FLIGHT CREW OPERATING MANUAL.
[16] Calcmaps. https://www.calcmaps.com/es/map-radius/
[17]ía de tarifas de Aena 2020, noviembre 2020
[18] ENAIRE. Guide to air navigation charges 2021
[19] FAA. "faa-Direct-operating-cost.pdf"
[20] Gaetan Kenway, Jason Hicken, David Zingg, Joaquim R. R. A. Martins . "Reducing Aviation's Environmental Impact Through Large Aircraft For Short Ranges", (2010)

## ANNEX：

ANNEX 1．AIR PASSENGER TRANSPORT BY REPORTING COUNTRY

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| 990 88t＇し | 070 ${ }^{\circ} \mathrm{CL} 8^{\prime}$ ！ | $100.990 \cdot 1$ |  |  |  |  |  |  |  |  |  |  |  | е！пıา |
| 201ヵかぐ9 |  | 861＇Lてt＇9 | ع0t＇LLO＇9 |  | 99 ${ }^{\prime} 688^{\prime} 9$ |  |  |  |  |  |  |  |  | snudरo |
| こ8どヤ16＇s6 | 8\＆t＇906 ${ }^{\text {L8 }}$ | $0 \varepsilon 8 \cdot \mathrm{S08}$＇08 | 己20＇LL6＇$¢$ | 089＇$\angle 2 \chi^{\prime}$＇9 | 0¢0＇682＇0ヵ | 乙6๕＇乙¢8＇¢ |  | 乙68 290 ¢ $¢$ |  | 8L1 898.8 t | OtG＇ 298 ＇tt | 881 ${ }^{\circ} 08 L^{\prime} 92$ | \＆¢1．089＇๕ | K184 |
|  |  |  |  |  |  |  |  |  |  |  |  |  |  | ¢！⿺𠃊od |
|  | 90＜＇996＜01 | Sz6＇てro＇80 | 169＇S6て＇96 |  | \＆ $2 L^{\circ} 068{ }^{\prime}$＇6 | عเ6：998：96 | 9\＆t＇6zく＇06 | 906 ＇29＇$\downarrow 8$ | Z6て＇\＆と¢ 62 | 962＇926＇8t | 888 ＇ 299 ＇t | 292＇tel＇t＇ | 681．ctiot | ә）${ }^{\text {arex }}$ |
| $16 \chi^{6665}$ OS | 128629 \％t | 8L8＇1LL＇621 | 8＜8＇88て＇021 | ャ¢L＇LEでてい | ع09986＇て1 | 1S1＇LL6＇601 | $0<\varepsilon^{\prime} 98 \varepsilon^{\prime} 101$ | 806＇29L＇89 | عモて＇¢LL＇ 29 | LSc $2500^{\circ} \mathrm{LS}$ | t68＇G20＇cs | 乙८z＇ $2 ¢ 9$＇OS | て6S＇tLて＇St | u！eds |
| ャ62＇089＇$¢$ | ZS9＇ZLS＇ 1 ¢ | ¢\＆8＇ャ｜と＇0¢ | Lてt＇s91＇82 | S9t＇SZく＇8Z | ¢9t＇GZL＇6Z | S9t＇SZL＇08 | LL6＇869＇LZ |  | 296 $1+9$ 8 | 002＇60 21 | ¢\＆1－ELE＇6 | 0くざとで＇OZ |  | әэә๐๐ |
| \＆と1＇8sc＇LZ | 86て＇ヤ乌て＇ヤて | \＆¢ $\varepsilon^{\prime}$ เ¢ $8^{\circ} 0$ |  |  | เટ1＇91E＇く | LLL＇989＇91 | เ上ย＇¢とて＇91 | と¢8＇ヤLて＇غا | ع60＇\＆L\＆＇z1 | $08<$＇Gเ6\％ | 0t0＇ $88{ }^{\prime} 6$ | L6て＇1018 | ＋16＇tscs | pue｜an |
|  | S01＇$\varepsilon 68$＇। | 9＜L＇066 |  |  |  |  |  |  |  |  |  |  |  | e！uotsy |
| LL0＇060＇ tS ， | 6Lt＇zol＇9t1 | ＋98＇606＇¢8 | S0t＇181＇Lて | £¢て＇888＇ロー | 998．98181। | 010＇ $699^{\circ} \mathrm{OL}$ | 868＇161＇zı | 296＇\＆\＆＇$\bigcirc 0 \downarrow$ | $16 L^{\prime} 826.86$ | ＋09＇ $106^{\prime} \mathrm{L8}$ | 698 ＇ 666 ＇＞8 | เع¢＇scs＇$\dagger 9$ | $690 \cdot \mathrm{St8} 8 \mathrm{8}$ | Киешнәэ |
| 099 L96＇zz | 0LL＇EL1＇Z乙 | $880 \cdot 900 \cdot 1<$ | 9t6 ${ }^{\text {1．98 }}$ |  |  |  |  |  |  |  |  | L86．909 | 162＇0LZ | угешиәа |
|  | ＋92＇99て＇ト | ャレع096．6 | 892＇t9L＇L | 08t $68 s^{\prime} 9$ |  |  |  |  |  |  |  |  |  | е！чэәzう |
|  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| \＆ $29 . \mathrm{tST} \cdot 6 \mathrm{~L}$ |  | 809 $89 \square^{\circ} \mathrm{L}$ | $628.960 \cdot 91$ | 296：899＇\＆ا | てヤ8＇ヶ69＇81 | てヤ8＇ャ69＇レて | 682＇z0000 | $008 \cdot 18{ }^{\prime} 81$ | $010 \cdot 8 z 6$＇st | 201＇ナ¢¢＇$¢$ | LIて＇90Gて＇ |  | 269＇820 0 | un！ ¢ $_{\text {｜eg }}$ |
|  |  |  |  |  |  |  |  |  |  |  |  |  |  | uoun ueadome |
| 9002 | s00z | t002 | 8002 | 2002 | 1002 | 0002 | 6661 | 8661 | 2661 | 9661 | ¢661 | ヶ661 | 866 | WWIL |


|  |  |  |  |  |  |  |  |  |  |  |  | 869 289 ¢ |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| L8t＇8t\％ 0 | Sol 108000 t | 8LL＇68L＇88 | 9tG＇LZL＇LE | z50＇ 809 L 8 | S61－809 $\llcorner\varepsilon$ | t9 9 ＇989＇98 | ¢2Z 269 ＇$\downarrow$ ¢ | ટて9＇80†てを | 6tt＇81962 | 619＇L9＇Lz | 991－LLくLz | 92 | － |
| L61＇t8S＇ | 988 | ャ¢く8188 | ＋18．08＇9 | 88でく48＇t | ＋198 | 992＇6 | $16900 \ll 2$ | －68 294 C | 998\％980＇己 | 991 Le | 8660ヶて＇ | とトて9tて |  |
| 088てzt＇LLて | ¢S1061てLz | ＋5t＇629＇t92 | cl88988t | ＜£＇0Lでて | で， ZzO | 08689 | ¢10 20 | ¢glegt | c288 | 2L6¢9 |  | 098＇ 6 CLLI |  |
| 6¢て＇tic | 960 | 10 | 895 | £9 | \＆t0＇99\ll $<8$ | ¢zでもt | $6 \pm 8098$ | くちでて¢L | t929t9 | ＋8＜＇81で | 0¢8 218 Lz | £g＇ 296.92 |  |
| 626 $28 \times 8$ ¢ | 08 | L66 t 50002 | D96 660 ＇81 | 9で＇6L゙く | $1 \varepsilon$ | 16 | 9188St＇9 | 868＇$+\angle$ ¢＇91 | 98902\％＇t | 218888¢ | 2890098 t |  |  |
| $281.688 \%$ | ャ60＇t6L＇z | 159 zot ＇z | 1928912 | 999 \＆\％6 ${ }^{\text {b }}$ | 062＇ $129 \cdot 1$ |  | L6，E99＇ | L81 8088 |  | $198886{ }^{\prime}$ | ＋88969\％ |  | yenols |
| $68061{ }^{\prime} 1$ | L990018＇ | ع¢ا $289{ }^{\circ} \mathrm{L}$ | 2s＇tot＇t | ع00＇98t＇ | 821－208＇ | 992＇G92＇1 | L28 $29 \mathrm{l}^{\prime}$ | 264＇88¢＇ | เ¢ 288 | 168 Eとて | L268t9 | 9th＇tos | บขกо |
| t0て＇9ts＇t | 2796086 | ＋LL＇¢6＇L | 6LC＇\＆s＇g | HL＇08S | L8t $206^{\circ} \mathrm{O}$ | ع86910 | 9z2＇ヤ29＇6 | 9st 2896 | 6568888 | LSO＇ 286 | L92＇180＇8 | 6698069 |  |
| ${ }^{6} 68200$ | 8698800＇ts | Ls0＇E29 $2 t$ | ＋20 0860 | จ18＇500＇98 | 129099 28 | 9 ¢1 +696 | ャ¢ 9 ＇9818 | L01－629 L | 009 ＇s8く＇s | عاغ9け＇t | 106＇z8＇ç | 88て＇928 | e6nuod |
| 99／668．9t | 8tG $29 / 8$ \＆t | 899 ＇89 28 | 198992＇ | $68 \square^{2} 06$ |  |  | 992008 | L9 | St－E888 | 668960 | G98 LzL | 820 O2＇L |  |
| 881 | L｜＇881＇ L | $6 \angle$ LLz¢ 82 | 119＇181 Lz | L00＇tgL．92 | 92988892 | ャてL＇6t＜＇S | L26＇9969 | 2l9 21 | ¢St ze＇\＆ | L92＇L8＇1 | ＋89 668 $\varepsilon$ ¢ | pot 926 |  |
| LO9＇261＇ 18 | ع91＇tt962 | to80tて＇9 | 966－L1802 | 02 | ¢9 | LZ | 602089＇s | 262＇968＇8 | 6＜88198 | Lとと＇6L＇9t | 10966itos | SLC＇zos os | N |
| Lร¢ 818 L | 4 | $18<200 \cdot 9$ | $9+50^{1080} 0^{\circ}$ | L99619 | 280 06 ＇t $^{\circ}$ | 620 ＇80＇t | 8 Os | ャ18＇909＇$\varepsilon$ | 8t9＇862＇ | 929816 亿 | 668＇601＇ 8 | 898＇126 |  |
| 91 | 86t＇9 | 620 Og¢ $\varepsilon 1$ | 998099 1 | Zsc：8z＇0 | 858＇t50＇6 | 618 | عt8＇ | L88＇8888 | OLStくて＇8 | L90＇ | 280＇62t＇8 | $197^{\circ} 089^{\circ}$ | תe6unh |
| 699 998 $\dagger$ | ＋08 | 0¢L＇ts ¢ $\varepsilon$ | ででヤ86て | 19L＇1．19\％ | 99688t＇ | Lz86912 | $166^{868}{ }^{\prime}$ | 082 $9888^{\prime \prime}$ | 969\％19 1 | 192＇989＇ | Lle¢İ। | ¢9\％＇te9 1 | noquexn7 |
| 589 º＇s 9 | l＇t | て＇9 | ＇ 82 ＇t | $688^{\prime 2} \mathrm{Lz't}$ | 011－861＇ | 898 288 ¢ | 829991＇8 | $166.169 \%$ | ¢88 28 ＇て | $16{ }^{2} 298{ }^{\prime}$ | －20＇zs ${ }^{\text {a }}$ | 6＇56 | 417 |
| 92L＇98L＇L | $0<0 \cdot \angle 80 \cdot L$ | t98 $120 \cdot 9$ | 091＇¢88＇ 9 | 998＇Stl＇s | $288208{ }^{\text {¢ }}$ | LSて $281{ }^{\text {¢ }}$ | Oes＇rg＇t | 098 $8600^{\circ} \mathrm{S}$ | 868 ¢99＇t | ＋02 $290{ }^{\circ} \mathrm{t}$ | $62 \varepsilon<898$ | 1＜L＇9s＇$¢$ | M127 |
|  | LOPL | $\varepsilon$ \＆ | L28＇196＇8 | 281069 2 | 9t¢ 888 L | LEt＇LO＇L | $0088888^{2}$ | L8806 $01 /$ | $887^{866} 9$ | L81＇6z＇＇9 | ャ80 $81 \mathrm{I}^{\text {c }}$ L | ¢IE ${ }^{\circ} 00^{\circ} \mathrm{L}$ |  |
|  | bt | sz8 | 18LLLせtE1 | $12 z$ |  |  | 888620＇9 |  |  |  |  |  | ｜en |
| $68\ulcorner$ ¢ $¢ 90$ | เ6て＇18L＇6 | ع90¢888 | ع9t | $869 \%$ LS | 26 | Lt⿺辶で＇s | こと9 zt ＇s | $9816866^{\prime}$ |  | $80 \chi^{\prime 988}$ | L29＇tos＇t |  |  |
| 2866248 | 62＇ $166{ }^{\text {c }} 191$ | ¢8t＇960＇tg | 209082＇st | 69\％ 2980 | 12909898 | ¢18729＇zr | 29t＇99 ${ }^{\prime \prime 6}$ | Lszel0＇9a | 202＇1698 | 688989 | ع＇096zz |  | эıuers $^{\text {a }}$ |
| 210．681／2zz | 6で＇H902 | 680＇288602 | 180＇z1886 | 809 $299 \%$＇t | 288＇¢¢¢ 99 | EL6＇ $181 \cdot 151$ | 192＇12L＇6s | ¢¢と＇gS | to 2888 ¢ | 362818 | 56 $000{ }^{\text {c }} 19$ | 0．Ez9 | upds |
| $9^{\prime} 880$ | 928889＇ts | 82LOLL ${ }^{\circ} \mathrm{OS}$ |  | 20t＇960 ${ }^{\prime \prime}$ | ¢๕8 21 | เ\＆6\％¢ ${ }^{\prime}$ | 988280 | 681.024 | 2L0＇889 | $692^{\circ}$ | 0869 | 2＇29 | ¢0ar |
| 019 LL6＇ 28 | s00 | LLL＇LZ＇̇¢ | 60／ | 020＇st9＇62 | 9880189 | 0ヶ9＇809＇tz | $680 \cdot 66982$ | 688 298๕ | $012+60^{\prime}$ \＆ | L88892＇92 | ＜8＇810＇0 | 020＇0086 |  |
| ع00＇8g＇$\varepsilon$ | ；66＇i | ＇s¢9 | 686 | 8260912 | $908610 \%$ | 999 $896 \cdot$ | LでてOでて | $699206 \cdot$ | 290 $188^{\prime}$＇ | t62＇ | 0¢t＇08＇ | Soc＇zzL＇ | uois |
| 980＇991．922 | 198 でも 2 ¢ | ¢8688 212 | 662 189002 | 0¢t＇96686 | ャ18＇Str＇98 | 88 E 8208 | 801 169884 | 92＇918＇9 | ¢880¢199 | 628881：891 | 978898＇991 | 868 88 88 | ueu |
| 1081＇1建 | 681 $100^{\prime \prime} \mathrm{E}$ | ャレて＇197＇¢ | 9：z8 | 909 $560^{\circ} \mathrm{O}$ | $0 \cdot 6$ | 296st＇LL | \＆くて8¢9 | z¢808＇s | $\angle 89$ ¢ ¢ ¢ | ¿てGLLZ | 2L0＇699＇って | Ll＇\＆00 | ешиәа |
| z88 | 12 C 88 | ts9 $9 t<' 91$ | 298て2981 | to0 2 | ع $188^{6} 20 \cdot \mathrm{Zt}$ | 218． $688^{\prime 2}$ | てSčでく | て89099 てt | 988てえでて1 | 29わく 298 て1 |  | ｜tr 860 ＇$\%$ |  |
| 890＇\＆LL |  | 199 $\mathrm{ZbO}^{\circ} \mathrm{LL}$ | ＇ャを＇6 | 666019 L | L69009＇L | $262^{\prime} 620^{\circ} \mathrm{L}$ | ع01＇618＇9 | 200＇zc9＇9 | ع88691＇9 | 528008＇9 | 266\％LL＇9 | ＇z |  |
| 881 ＇988＇ |  | ع6t＇09\％＇ 8 | 2889 $\mathrm{LH}^{\circ} \mathrm{O}$ | 88 | 892＇9Lく＇82 | $\angle 266889$ | ¢1966＇s |  | ＋91 2699 Zz | 669 | ع89 $886^{\circ} \mathrm{L}$ | Sst＇s08 02 | un！｜¢өg |
| －602＇Et｜＇1 | 650 288801 ＇ | tet 8880001 | 20て¢¢¢0L6 | 9LL88t966 | 917801＇S28 | 99s tsc Le8 | 888＇60＇ 8 ¢8 | ع̌＇9¢1818 | を¢t＇sscrul | เ6¢＇998＇GL | 69t＇cz9 882 | ¢z1＇s09z18 | upu |
| 6102 | 8102 | LO2 | 9108 | S102 |  |  |  |  |  |  |  |  |  |

## ANNEX 2．FREIGHT AND MAIL TRANSPORT BY REPORTING COUNTRY

| ८є¢ ¢¢ |  | ع80＇618 | 069 ＇zع | 688 ＇ 98 |  |  | 286＇ $2 \downarrow$ | 9LL＇10t | 928＇90t | 628.68 | 29t＇868 | 926006 | $8 \downarrow<$ ¢ $¢$ ¢ | рuepraz！Ms |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| OtL＇9 | ع¢96 | $\angle$ | L99 | ＋20＇t |  |  |  |  |  |  |  |  |  | Kemion |
| ＋8L＇19 | 09969 | 188＇99 | 268で |  |  |  |  |  |  |  |  |  |  | риерәэ |
| 209068 2 | 62609t＇z | こせでしくせでて | 209＇982＇兀 | 88て＇9かして | L90＇660＇z | LSO＇ヤくて＇て | 868062 ＇ | 80t＇88して | 189988＇1 | 60t $\angle 99{ }^{\prime}$ | ＋s0＇889＇t | 016＇s8t＇t | tLt＇str＇t | шорби！у рө！̣迷 |
| 0 | 0 | Ls6．OSt |  |  |  |  |  |  |  |  |  |  | 0 | иәремs |
| 888821 | 69964 | ع6t＇$¢ \downarrow$ | 869 68 | 20＇68 |  | £で＇10t | $910^{\circ} 96$ | 689 zor | 6Stて6 |  |  |  |  | puelu！ |
| $9<8{ }^{\circ}$ | $850 \%$ | 2618 | くヵ8 て1 | Lsc9 9 | 929＊ |  |  |  |  |  |  |  |  | elyenols |
| 8659 | 6 ts ＇t | \＆86＇$\downarrow$ |  |  |  |  |  |  |  |  |  |  |  | घ！บขก이 |
| 88L02 | 086.4 | $68 \% 61$ | ¢8\％＇st | $88 \mathrm{t}^{\prime} 91$ | ャて9＇9 |  |  |  |  |  |  |  |  | ع！uешоу |
| 乙¢¢＇9¢। | 9z¢ 621 | 608 ¢ 1 | 96¢¢ 81 | 628821 | くゅ゙を¢ | 806\％ LL | 028＇991 | 028＇991 | 0＜8＇st | 028＇sz | $980 \cdot \mathrm{Zz}$ | ¢99 ¢6 | 8Lt＇98 | 186пи\％d |
| $0<6 \cdot 68$ | $0 \varepsilon \vdash \cdot ¢$ | \＆で＇¢ |  |  |  |  |  |  |  |  |  |  |  | puelod |
| 989：20 | ¢¢¢＇81 | ¢99691 | 88L＇821 | 688＇s21 |  |  |  |  |  |  |  |  |  | elusnv |
| $699^{\prime \prime 2} \mathrm{LC}$＇t | $98.099^{\prime}+$ | LS6＇tL＇t | trs ${ }^{\circ 888^{\circ} \text {＇}}$ |  | Stticlz＇t | عz9 $292 \cdot$ | L88 $281^{\prime}$ | $609^{\circ} 81 \cdot$ | $80089 \cdot 1$ | $90 \mathrm{~S}^{\circ} 880^{\circ}$ | Lt¢ 286 | ع८8＇ $1+8$ | 8897LL | spueparion |
| 190＇81 | 16891 | ＜L＇S＇ | てt¢9 91 | 889 21 |  |  |  |  |  |  |  |  |  | Enew |
| $288 \times 9$ | $\varepsilon<t ' g s$ | －1゙09 | ¢zs os | 20＇9t |  |  |  |  |  |  |  |  |  | $\wedge^{\text {＾1ebunn }}$ |
| く८L＇$¢ 89$ | ع08 ヶて¢ | 889999 |  |  |  |  |  |  |  |  |  |  |  | 6．noquexn7 |
| cl9 zt | $089^{6}$ | غ8＇s | $\angle\llcorner ' 9$ |  |  |  |  |  |  |  |  |  |  | в！иепч！ |
| stc＇r | 82t＇G1 | $928 \%$ |  |  |  |  |  |  |  |  |  |  |  | епиет |
|  | 己¢＇6¢ | 06128 | 918 เع |  | 乙 |  |  |  |  |  |  |  |  | snidko |
| 9troor8 | 208tg | ع69014 | ¢98 ¢89 | ＋01－869 | ع＜L＇902 |  |  | 6z\％＇spt | ャ08＇¢\＆ | \＆ 1 ＇ts 9 | 80＜＇ヤ\＆s | tol $88 \pm$ | Los 200 | K12］ |
|  |  |  |  |  |  |  |  |  |  |  |  |  |  | Plear |
| 081 $\mathrm{t} 6 \mathrm{~S}^{\prime} \mathrm{L}$ | ャ89 \ll＋ | \＆t＜${ }^{\text {cost＇t }}$ | L96＇915＇ | 26898t＇t | LS66z8＇ | 626024＇1 |  | \＆oて＇zo＇t | $680^{\circ} \mathrm{Sz} 0^{\circ} \downarrow$ | t8 $8^{\circ} \mathrm{LSO} 0^{\circ} \mathrm{L}$ | $088^{\circ} \mathrm{E} 0^{\prime}$ | O8¢ ${ }^{\circ} 900^{\circ}$ | 609 298 | әouras |
| E92＇0， | Sz1＇9zs | ع090\％s | 16668 68 | s90．${ }^{\circ}$ ¢ | too＇tst | E¢68Lt | 9¢t¢¢t | LDて＇608 | 988808 | 978082 | 988 ¢ ¢ | O¢¢＇tL | 9686く | u！eds |
| 156．601 | 228201 | 608 PL | 9t0＇ 28 |  |  | غ¢¢ $\downarrow$ ¢ | ¢¢9．9\％ | 868＇ç1 | ャレて＇901 | 099ZL |  | о¢て＇て8 | ¢2908 | ขอәมง |
| z90＇z\＆ | 99868 | ع91－29 |  | $98 \gtrless^{\circ}$ けャ | 08t＇69 | 89でくL | \＆6\％ $1 / 2$ | 98989 | 88892 | 0でで | 01609 | 80＜ $2+$ | $61^{\circ} \mathrm{O}$ | pue｜eal |
| \＆90 01 | 681＇6 | $866{ }^{\circ}$ |  |  |  |  |  |  |  |  |  |  |  | E！uoiss |
| 60て＇69て＇\＆ | 29t＇900 8 | 00て88くて | $190{ }^{\circ} \mathrm{Lt} \mathrm{C}^{\prime} \mathrm{C}$ |  | 2s．692＇z | 091＇ャ98 ${ }^{\circ}$ |  |  |  |  |  |  |  | Кишшәэ |
| 919\％ | 988 L | 826 L | Lᄂ |  |  |  |  |  |  |  |  |  |  | янешиа |
| £と¢ 69 | 692＇99 | くıg $<9$ | ટટて¢я | 999¢t |  |  |  |  |  |  |  |  |  | е！чขәг） |
|  |  |  |  |  |  |  |  |  |  |  |  |  |  | عب．1e6．na |
| $\angle ¢ \varepsilon^{\circ} \mathrm{LE} 0^{\circ} \mathrm{L}$ | ع ¢＇$^{\prime} 69$ | เ\＆t＇099 | tos＇909 | $18 \chi^{\prime} 66$ | てヤ¢＇¢Z¢ | てヤ¢＇¢t¢ | てセ¢ 999 | 2૪¢＇989 | 600＇815 | เ\＆¢ 6 t | عغ̇＇9zt | $918 ¢ \downarrow \varepsilon$ | ع8＇908 | un！6｜ə |
| 9008 | S002 | ¢00z | 8002 | 2008 | 1002 | 0002 | 666｜ | 866｜ | 2661 | 9661 | 966। | ヶ66｜ | 866 | awn |



## ANNEX 3. A320 WEIGHT AND BALANCE CALCULATION TABLES

\begin{tabular}{|c|c|c|c|c|c|c|c|c|c|}
\hline \multicolumn{10}{|c|}{FLIGHT PLANNING FROM BRAKE RELEASE TO LANDING
CLIMB \(: 250 \mathrm{KT} / 300 \mathrm{KT} / \mathrm{M} .78-\) CRUISE \(:\) M. 78 - DESCENT \(:\) M. \(78 / 300 \mathrm{KT} / 250 \mathrm{KT}\) IMC PROCEDURE: 110 KG (6MIN)} \\
\hline \multicolumn{4}{|l|}{REF. LANDING WEIGHT \(=50000 \mathrm{KG}\) NORMAL AIR CONDITIONING ANTI-LCING OFF} \& \multicolumn{2}{|l|}{\[
\begin{aligned}
\& \text { ISA } \\
\& C G=33.0 \%
\end{aligned}
\]} \& \multicolumn{4}{|c|}{FUEL CONSUMED (KG)
\(\qquad\)} \\
\hline \multirow[t]{2}{*}{} \& \multicolumn{6}{|l|}{FUGHT Level} \& \multicolumn{3}{|l|}{CORREELIION ON
FUEE
(KONS 1000 MPG )} \\
\hline \& \multirow[t]{2}{*}{\[
\frac{290}{5089}
\]} \& 310 \& \[
330
\] \& \& 370 \& 390 \& \[
\begin{aligned}
\& \text { FL290 } \\
\& \text { FL310 }
\end{aligned}
\] \& \[
\begin{aligned}
\& \text { FLL300 } \\
\& \text { FL350 } \\
\& \hline
\end{aligned}
\] \& \multirow[t]{2}{*}{\[
\begin{array}{r}
\text { FL370 } \\
\text { FL390 } \\
\hline 46
\end{array}
\]} \\
\hline 825 \& \& \& 4304
407
207 \& \& \begin{tabular}{l} 
3 \\
\(\begin{array}{l}433 \\
202\end{array}\) \\
\hline
\end{tabular} \& \begin{tabular}{|}
3839 \\
\hline 202 \\
\hline
\end{tabular} \& \& \& \\
\hline 850 \& \begin{tabular}{|c}
5232 \\
203 \\
\hline 1
\end{tabular} \&  \& 4774
204 \& \begin{tabular}{c}
4592 \\
\hline 205
\end{tabular} \& \begin{tabular}{l}
4447 \\
205 \\
\hline
\end{tabular} \& \begin{tabular}{|c}
4350 \\
205 \\
\hline
\end{tabular} \& \({ }^{27}\) \& 35 \& 47 \\
\hline 875 \& 206 \& \begin{tabular}{l}
5120 \\
2.07 \\
\hline
\end{tabular} \& \(\begin{array}{r}4900 \\ 207 \\ \hline 207\end{array}\) \& \({ }_{2}^{4711}\) \& \begin{tabular}{c}
4562 \\
209 \\
\hline 20
\end{tabular} \& \begin{tabular}{c}
4461 \\
209 \\
\hline 20
\end{tabular} \& 27 \& 35 \& 49 \\
\hline 900 \& 5518
209
209 \& 5253
\(\begin{aligned} \& 510 \\ \& 210\end{aligned}\) \& 2027 \& 48312
212 \& \({ }_{\substack{467 \\ 212}}^{\substack{4 \\ 2}}\) \& \(\begin{array}{r}4537 \\ \hline 12 \\ \\ \hline 12\end{array}\) \& \({ }^{28}\) \& \({ }^{36}\) \& 50 \\
\hline 925 \&  \& ( \& S1,54
2 \&  \& 272
2.15
2.15 \&  \& \({ }^{28}\) \& \({ }^{37}\) \& 51 \\
\hline 950 \& \begin{tabular}{l}
5805 \\
216 \\
\hline 15
\end{tabular} \& 5324
216
216 \& \(\begin{array}{r}5281 \\ \hline 2.17\end{array}\) \& (1818 \& 4906
219
219 \& 4795
219 \& 29 \& \({ }^{38}\) \& 53 \\
\hline 975 \& \begin{tabular}{|c}
5948 \\
219 \\
\hline 19
\end{tabular} \& \begin{tabular}{l}
5658 \\
220 \\
\hline 20
\end{tabular} \& ( 2408 \& \(\begin{array}{r}5192 \\ \hline 222 \\ \hline 1\end{array}\) \& \begin{tabular}{|c}
5021 \\
202 \\
\hline
\end{tabular} \& \(\begin{array}{r}4996 \\ 222 \\ \hline 2\end{array}\) \& \({ }^{30}\) \& 39 \& 54 \\
\hline 1000 \& \begin{tabular}{l}
8091 \\
202 \\
\hline
\end{tabular} \& \begin{tabular}{l}
519 \\
\(\begin{array}{l}573 \\
223\end{array}\) \\
\hline
\end{tabular} \&  \&  \& \begin{tabular}{|c}
5137 \\
225 \\
\hline 28
\end{tabular} \& \begin{tabular}{l}
5018 \\
\hline 225 \\
\hline 20
\end{tabular} \& 30 \& 40 \& 56 \\
\hline 1025 \& \({ }_{2234}^{623}\) \& - \& \begin{tabular}{l}
5662 \\
\hline 227
\end{tabular} \& (2433 \&  \& \begin{tabular}{l}
5130 \\
239 \\
\hline
\end{tabular} \& 31 \& 41 \& 57 \\
\hline 1050 \& \begin{tabular}{l}
6378 \\
\hline 298 \\
\hline 1
\end{tabular} \& ( \(\begin{array}{r}6062 \\ 230\end{array}\) \& \begin{tabular}{l} 
5730 \\
\hline 23
\end{tabular} \& \begin{tabular}{l}
5554 \\
232 \\
\hline
\end{tabular} \& \begin{tabular}{|c}
5368 \\
232 \\
\hline 23
\end{tabular} \& \(\underset{232}{524}\) \& 32 \& \({ }^{41}\) \& 58 \\
\hline 1075 \& \begin{tabular}{l} 
6521 \\
\\
232 \\
\hline
\end{tabular} \&  \& \begin{tabular}{c}
593 \\
294 \\
\hline 23
\end{tabular} \& \(\begin{array}{r}5635 \\ \hline 235 \\ \hline\end{array}\) \& \begin{tabular}{|c}
5483 \\
\(\begin{array}{c}536 \\
236\end{array}\) \\
\hline
\end{tabular} \& \begin{tabular}{l}
5335 \\
\(\substack{535 \\
236 \\
\hline}\)
\end{tabular} \& \({ }^{32}\) \& 42 \& 60 \\
\hline 1100 \& \(\stackrel{6565}{635}\) \& \begin{tabular}{c} 
6332 \\
\hline
\end{tabular} \& \begin{tabular}{l} 
3043 \\
\\
23 \\
\hline
\end{tabular} \& ¢ \& (239 \& \(\begin{array}{r}545 \\ \hline 239 \\ \hline 29\end{array}\) \& \({ }^{33}\) \& \({ }^{43}\) \& 61 \\
\hline 1125 \& \begin{tabular}{l}
6839 \\
\hline 289
\end{tabular} \& \begin{tabular}{r}
636 \\
\hline 639 \\
239
\end{tabular} \& \begin{tabular}{l}
6172 \\
241 \\
\hline 18
\end{tabular} \& \begin{tabular}{l}
59217 \\
\hline 262
\end{tabular} \&  \&  \& \({ }^{34}\) \& 44 \& \({ }^{63}\) \\
\hline 1150 \& \(\stackrel{6952}{242}\) \& \begin{tabular}{l}
6603 \\
\hline 243
\end{tabular} \& \begin{tabular}{l}
6300 \\
244 \\
\hline
\end{tabular} \& 6038
245 \&  \& \(\mathrm{O}^{515} 5\) \& \({ }^{34}\) \& 45 \& \({ }^{64}\) \\
\hline 1175 \& \begin{tabular}{l}
7096 \\
\hline 245 \\
\hline
\end{tabular} \&  \& \begin{tabular}{l}
8488 \\
\hline 84
\end{tabular} \& 6160
248 \& \begin{tabular}{l}
5948 \\
249 \\
\\
\hline
\end{tabular} \& \(\begin{array}{r}5806 \\ 249 \\ \hline 29\end{array}\) \& \({ }^{35}\) \& 48 \& 65 \\
\hline 1200 \& \begin{tabular}{l}
1240 \\
248 \\
\hline
\end{tabular} \& \(\begin{array}{r}683 \\ \hline 249 \\ \hline 18\end{array}\) \&  \& \(\begin{array}{r}6281 \\ \\ \\ \hline 252\end{array}\) \& (2604 \& \(\stackrel{5920}{525}\) \& \({ }^{36}\) \& 47 \& \({ }^{67}\) \\
\hline 1225 \& 1384
285 \& \begin{tabular}{l}
1709 \\
253 \\
\hline 10
\end{tabular} \& \begin{tabular}{l}
6684 \\
\hline 2.54 \\
\hline
\end{tabular} \& \begin{tabular}{l}
6453 \\
\hline 25 \\
\hline
\end{tabular} \& \begin{tabular}{l}
6380 \\
\hline 256 \\
\hline 56
\end{tabular} \& \(\begin{array}{r}6633 \\ \hline 256 \\ \hline 26\end{array}\) \& \({ }^{36}\) \& \({ }^{48}\) \& \({ }^{68}\) \\
\hline 1250 \& 255 \& Th44
256 \& 685

257 \&  \& | 2397 |
| :--- |
| 259 | \& 6159

269 \& 37 \& ${ }^{48}$ \& 70 <br>

\hline 1275 \& | 759 |
| :--- |
| 268 |
| 28 | \& | 7280 |
| :--- |
| 259 |
| 29 | \& 6941

3 \& \begin{tabular}{l}
6646 <br>
302 <br>
\hline

 \& 

644 <br>
302 <br>
<br>
\hline
\end{tabular} \& 6261

302 \& ${ }^{38}$ \& 49 \& 1 <br>
\hline 1300 \& 7817
301
300 \& 7815
302

302 \& l \& | 6768 |
| :--- |
| 3 |
| 05 | \&  \& 6335

306
3 \& ${ }^{38}$ \& 50 \& ${ }^{73}$ <br>

\hline 1325 \& | 7961 |
| :--- |
| 304 |
| 04 | \& 3.06 \& 719

307 \& ${ }_{3680}$ \& 3668
309
309 \& ${ }^{6499}$ \& 39 \& 51 \& 74 <br>
\hline 1350 \& 8105
308
308 \&  \& 7326
3.10 \& 7012
3012 \& ¢ \& $\underset{\substack{6603 \\ 3.13}}{ }$ \& 40 \& 52 \& 76 <br>

\hline 1375 \&  \& cisiz \& | 73.45 |
| :--- |
| 3 |
|  |
|  |
| 14 | \& 7134

3
3 \& ${ }_{3116}^{6832}$ \& ${ }_{3}^{5116}$ \& 40 \& 53 \& 78 <br>
\hline 1400 \& ${ }_{3}^{8394}$ \& 7939

3.16 \& | 7388 |
| :--- |
| 3.17 |
|  | \&  \& ¢ ${ }^{6999}$ \& ¢ 319 \& 41 \& 54 \& 79 <br>

\hline 1425 \& 3.17 \& ${ }_{319}$ \& ¢ \& 739
3
372 \& 717
323 \& 6967
373 \& 42 \& 55 \& 81 <br>
\hline 1450 \& 8684 \& ${ }_{323}^{823}$ \& $\begin{array}{r}7241 \\ 324 \\ \hline\end{array}$ \& 7501
325 \& ${ }_{3}^{1236}$ \& 7062
326 \& ${ }^{42}$ \& 56 \& 82 <br>

\hline \multicolumn{3}{|l|}{$$
\begin{aligned}
& \text { LOW AIR CONOITIONING } \\
& \triangle \text { FUEL }=-0.5 \%
\end{aligned}
$$} \& \multicolumn{3}{|l|}{ENGINE ANTI ICE ON $\triangle$ FUEL $=+2 \%$} \& \multicolumn{4}{|c|}{\[

$$
\begin{aligned}
& \text { Total ANTIICE ON } \\
& \Delta \text { AUEL }=+45 \%
\end{aligned}
$$
\]} <br>

\hline
\end{tabular}

| ALTERNATE PLANNING FROM DESTINATION TO ALTERNATE AIRPORT G0-AROUND : 100 KG - CLIMB : $250 \mathrm{KT} / 300 \mathrm{KT} / \mathrm{M} .78$ - CRUISE : LONG RANGE DESCENT : M. $78 / 300 \mathrm{KT} / 250 \mathrm{KT}$ - VMC PROCEDURE : 60 KG (4MIN) |  |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| REF. LDG. WT AT ALTERNATE $=50000 \mathrm{KG}$ NORMAL AIR CONDITIONING ANTL-ICING OFF |  |  |  | $\begin{gathered} \text { ISA } \\ \mathrm{CG}=33.0 \% \end{gathered}$ |  | FUEL CONSUMED (KG) TIME (H.MIN) |  |  |  |
| AIRDIST.(NM) | FLIGHT LEVEL |  |  |  |  |  | CORRECTION ONFUEL CONSUMPIION (KG/1000KG) |  |  |
|  | 100 | 120 | 140 | 160 | 180 | 200 | $\begin{aligned} & \text { FL100 } \\ & \text { FL120 } \end{aligned}$ | $\begin{aligned} & \text { FL140 } \\ & \text { FL160 } \end{aligned}$ | $\begin{aligned} & \text { FL180 } \\ & \text { FL200 } \end{aligned}$ |
| 20 |  |  |  |  |  |  |  |  |  |
| 40 | ${ }_{4}^{4717}$ | Interp | late |  |  |  | 2 |  |  |
| 60 | 618 0.17 | 602 0.16 | 603 0.16 | 608 0.16 |  |  | 4 | 3 |  |
| 80 | 765 0.21 | 745 0.20 | 740 0.20 | 738 0.19 | 740 0.19 | 745 0.19 | 5 | 4 | 5 |
| 100 | 913 0.25 | 887 0.24 | 877 0.23 | 868 0.23 | 884 8.23 | 884 0.22 | 6 | 5 | 6 |
| 120 | 1061 0.30 | 1030 0.28 | 1024 0.27 | 999 0.27 | 989 0.26 | 983 0.26 | 7 | 6 | 6 |
| 140 | 1209 0.34 | 1172 0.32 | 1151 0.31 | 1130 0.30 | 1114 0.30 | 1102 <br> 0.29 <br> 1 | ${ }^{9}$ | 7 | 7 |
| 160 | 1358 0.38 0. | 1315 0.36 0 | 1288 0.34 0. | 1260 0.34 0. | 1238 0.33 | 1221 0.33 | 10 | 8 | 8 |
| 180 | 1506 0.43 | 1458 0.40 | $\begin{array}{r}1425 \\ 0.38 \\ \hline\end{array}$ | 1391 0.37 | 1363 0.36 | 1340 0.36 | 11 | 9 | 9 |
| 200 | 1655 0.47 | 1602 0.44 0 | 1562 0.42 | 1522 0.41 | 1489 0.40 | 1459 0.40 | 13 | 10 | 10 |
| 220 | 1804 0.51 | 1745 0.48 | 1700 0.46 | 1653 0.44 | 1614 0.43 | 1579 0.43 | 14 | 11 | 11 |
| 240 | 1953 0.55 0.5 | 1889 0.52 | 1837 0.49 | 1785 0.48 | 1739 0.47 | 1698 0.47 | 15 | 12 | 12 |
| 260 | 2103 <br> 1.00 <br> 100 | 2033 0.56 | 1975 0.53 | 1916 0.52 | 1855 0.50 | 1818 0.50 0.5 | 16 | 13 | 13 |
| 280 | 2252 <br> 1.04 | 2177 1.00 | 2113 0.57 | 2048 0.55 | 1990 <br> 0.54 | $\begin{array}{r}1938 \\ 0.53 \\ \hline\end{array}$ | 18 | 14 | 14 |
| 300 | 2402 <br> 1.08 <br> 1 | 2321 1.04 | 2251 1.00 | 2179 0.59 | 2116 0.57 | 2057 0.57 | 19 | 15 | 15 |
| 320 | 2552 1,13 | 2486 1.07 | 2389 1.04 | 2311 1.02 | 2242 1.01 | 2177 1.00 | 20 | 16 | 15 |
| 340 | 2702 1.17 | 2611 | 2528 <br> 1.08 | 2443 1.06 | 2368 <br> 1.04 | 2297 1.04 | 21 | 17 | 16 |
| 360 | 2853 1.21 | 2755 1.15 | $\underset{\substack{2866 \\ 1.11}}{ }$ | 2575 1.09 | 2434 <br> 1.08 | 2417 1.07 | 23 | 18 | 17 |
| 380 | 3004 <br> 1.25 | 2901 1.19 | 2805 1.15 | 2708 1.13 | ${ }_{1}^{2620}$ | 2537 111 | 24 | 19 | 18 |
| 400 | $\begin{array}{r}3155 \\ 1.30 \\ \hline\end{array}$ | 3046 1.22 | 2943 1.18 | 2840 1.16 | 2746 1.15 | 2657 114 1 | 25 | 20 | 19 |
| 420 | 3306 <br> 1134 <br> 1 | 3191 1.26 | $\begin{array}{r}3082 \\ 1.22 \\ \hline\end{array}$ | 2972 <br> 1.20 <br> 1 | 2873 1.18 | 2778 1.18 | 25 | 21 | 20 |
| 440 | 3457 <br> 138 <br> 18 | 3337 <br> 1.30 | 3221 1.26 | 3105 <br> 1.23 | 2999 1.22 | ${ }_{121}^{2898}$ | 28 | 22 | 21 |
| 460 | 3608 1.42 | 3483 133 | 3360 1.29 | $\begin{array}{r}3238 \\ 127 \\ \hline\end{array}$ | 3126 1.25 | 3019 125 | 29 | 23 | 22 |
| 480 | 3760 <br> 1.47 | 3629 1 137 | 3500 <br> 1.33 | $\begin{array}{r}3370 \\ 1.30 \\ \hline\end{array}$ | 3253 <br> 1.28 | 3139 <br> 128 <br> 18 | 30 | 24 | 23 |
| 500 | 3912 1.51 | 3775 1.41 | 3639 1.37 | 3503 1.34 | 3379 132 | 3260 132 | 31 | 25 | 23 |
| LOW AIR CONDITIONING$\triangle F U E L=-0.5 \%$ |  |  | ENGINE ANTI ICE ON$\triangle$ FUFL $=+4 \%$ |  |  | TOTAL ANTIICE ON $\triangle$ FUEL $=+7 \%$ |  |  |  |


| RACE TRACK HOLDING PATTERN - GREEN DOT SPEED |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| MAX. CRUISE THRUST LIMITS CLEAN CONFIGURATION NORMAL AIR CONDITIONING ANTI-ICING OFF |  |  |  |  | $\begin{gathered} \text { ISA } \\ \text { CG }=33.0 \% \end{gathered}$ |  | $\begin{gathered} \text { N1 (\%) } \\ \text { FF (KG/H/ENG) } \end{gathered}$ |  |
| WEIGHT (1000KG) | FL 15 | FL 50 | FL100 | FL140 | FL180 | FL200 | FL220 | FL250 |
| 44 | $\begin{array}{r} 44.7 \\ 854 \end{array}$ | $\begin{array}{r} 46.8 \\ 836 \end{array}$ | $\begin{array}{r} 50.2 \\ 806 \end{array}$ | $\begin{array}{r} 52.9 \\ 781 \end{array}$ | $\begin{aligned} & 56.3 \\ & 760 \end{aligned}$ | $\begin{aligned} & 57.8 \\ & 753 \end{aligned}$ | $\begin{array}{r} 59.5 \\ 750 \end{array}$ | 62.2 749 |
| 46 | $\begin{array}{r}45.6 \\ 888 \\ \hline\end{array}$ | 47.8 871 | 51.1 837 | $\begin{array}{r} 54.0 \\ 811 \end{array}$ | 57.4 792 | 58.9 787 | 60.6 785 | 63.5 783 |
| 48 | 46.5 923 | $\begin{array}{r} 48.8 \\ 906 \end{array}$ | $\begin{array}{r} 52.0 \\ 868 \end{array}$ | $55.1$ | $58.4$ $826$ | 59.9 821 | 61.7 819 | 64.7 816 |
| 50 | 47.3 959 | $\begin{array}{r} 390 \\ \hline 49.8 \\ 938 \end{array}$ | $\begin{gathered} 500 \\ 52.9 \\ 898 \end{gathered}$ | $\begin{array}{r} 542.1 \\ 5674 \\ \hline 8 \end{array}$ | $59.3$ | 60.9 856 | 62.8 853 | $\begin{array}{r}65.7 \\ 848 \\ \hline\end{array}$ |
|  | 48.2 | 50.5 | 53.9 | 57.3 | 60.3 | 61.9 | 63.9 | 66.6 |
| 52 | 994 | 968 | 929 | 906 | 894 | 890 | 887 | 880 |
| 54 | 49.1 1030 | $\begin{aligned} & 51.3 \\ & 1000 \end{aligned}$ | $\begin{gathered} 54.8 \\ 960 \end{gathered}$ | $\begin{array}{r} \hline 58.2 \\ 939 \end{array}$ | $\begin{aligned} & \hline 61.2 \\ & 929 \end{aligned}$ | 63.0 923 | 65.0 921 | 67.5 912 |
|  | 50.0 | 52.1 | 55.8 | 59.0 | 62.2 | 64.0 | 66.0 | 68.5 |
| 56 | 1063 | 1031 | 992 | 972 | 961 | 957 | 952 | 944 |
| 58 | $\begin{array}{r} 50.8 \\ 1094 \end{array}$ | $\begin{aligned} & 52.9 \\ & 1061 \end{aligned}$ | $\begin{aligned} & 56.7 \\ & 1024 \end{aligned}$ | $\begin{gathered} 59.9 \\ 1006 \end{gathered}$ | $\begin{aligned} & 63.1 \\ & 995 \end{aligned}$ | $\begin{aligned} & 65.0 \\ & 992 \end{aligned}$ | $\begin{array}{r}66.8 \\ 984 \\ \hline\end{array}$ | 69.4 976 |
|  | 51.5 | 53.7 | 57.7 | 60.7 | 64.1 | 66.0 | 67.7 | 70.2 |
| 60 | 1125 | 1091 | 1057 | 1041 | 1029 | 1023 | 1016 | 1008 |
| 62 | $\begin{array}{r} 52.2 \\ 1155 \\ \hline \end{array}$ | $\begin{array}{r} 54.5 \\ 1122 \\ \hline \end{array}$ | $\begin{array}{r} 58.7 \\ 1090 \\ \hline \end{array}$ | $\begin{aligned} & 61.5 \\ & 1075 \\ & \hline \end{aligned}$ | $\begin{array}{r} 65.0 \\ 1063 \\ \hline \end{array}$ | $\begin{aligned} & 66.9 \\ & 1055 \\ & \hline \end{aligned}$ | $\begin{aligned} & 68.5 \\ & 1048 \\ & \hline \end{aligned}$ | $\begin{array}{r} 71.0 \\ 1041 \\ \hline \end{array}$ |
|  | 52.9 | 55.3 | 59.4 | 62.4 | 66.0 | 67.6 | 69.3 | 71.8 |
| 64 | 1186 | 1154 | 1123 | 1108 | 1095 | 1087 | 1081 | 1075 |
| 66 | $\begin{aligned} & 53.6 \\ & 1217 \end{aligned}$ | $\begin{aligned} & 56.1 \\ & 1186 \end{aligned}$ | $\begin{aligned} & 60.1 \\ & 1157 \\ & \hline \end{aligned}$ | $\begin{array}{r} 63.2 \\ 1141 \\ \hline \end{array}$ | $\begin{array}{r} \hline 66.9 \\ 1125 \\ \hline \end{array}$ | $\begin{array}{r} 68.4 \\ 1119 \\ \hline \end{array}$ | $\begin{array}{r} 70.1 \\ 1113 \\ \hline \end{array}$ | 72.7 1109 |
| 68 | 54.3 1247 | $56.9$ | $60.8$ | $64.0$ | 117.6 1158 | 69.2 1151 | 70.9 1146 | 73.5 1144 |
| 70 | $\begin{array}{r} 55.0 \\ 1279 \end{array}$ | $\begin{aligned} & 57.7 \\ & 1251 \end{aligned}$ | $\begin{aligned} & 61.6 \\ & 1225 \end{aligned}$ | $\begin{aligned} & 64.9 \\ & 1208 \end{aligned}$ | $\begin{gathered} 68.4 \\ \hline 1190 \end{gathered}$ | $\begin{array}{r} 69.9 \\ 1184 \end{array}$ | $\begin{array}{r} 71.6 \\ 1179 \end{array}$ | $\begin{gathered} 74.3 \\ 1180 \end{gathered}$ |
| 72 | 55.7 1311 | 125.6 1285 | $\begin{array}{r} 62.3 \\ \hline 62.3 \end{array}$ | $\begin{array}{r} 65.7 \\ 1241 \end{array}$ | 69.1 1223 | 70.7 1216 | 172.3 1212 | 75.1 1217 |
|  | 56.5 | 59.4 | 63.1 | 66.6 | 69.8 | 71.4 | 73.1 | 75.9 |
| 74 | 1344 | 1319 | 1292 | 1272 | 1255 | 1249 | 1247 | 1256 |
| 76 | $\begin{aligned} & 57.2 \\ & 1377 \end{aligned}$ | $\begin{array}{r} 60.2 \\ 1352 \\ \hline \end{array}$ | $\begin{array}{r} 63.8 \\ 1325 \end{array}$ | $\begin{array}{r} 67.4 \\ 1303 \\ \hline \end{array}$ | $\begin{array}{r} 70.5 \\ 1288 \\ \hline \end{array}$ | $\begin{aligned} & 72.1 \\ & 1283 \\ & \hline \end{aligned}$ | $\begin{aligned} & 73.8 \\ & 1282 \end{aligned}$ | 76.6 1295 |
| LOW AIR CONDITIONING $\Delta F F=-0.3 \%$ | EN <br> ANTI <br> FF | E ON $5 \%$ | ThTI <br> AFF | $\begin{aligned} & \text { E ON } \\ & +9 \% \end{aligned}$ | PER $1^{\circ}$ | VE ISA $0.3 \%$ | STRAI $\triangle F F$ | T LINE $-5 \%$ |

## ANNEX 4. A330 WEIGHT AND BALANCE CALCULATION TABLES

\begin{tabular}{|c|c|c|c|c|c|c|c|c|c|}
\hline \multicolumn{10}{|c|}{FLIGHT PLANNING FROM BRAKE RELEASE TO LANDING CLIMB : 250KT/300KT/M. 80 - CRUISE : M. 80 - DESCENT : M.80/300KT/250KT IMC PROCEDURE : 240 KG ( 6 MIN )} \\
\hline \multicolumn{4}{|l|}{REF. LANDING WEIGHT \(=140000 \mathrm{KG}\) NORMAL AIR CONDITIONING ANTI ICING OFF} \& \multicolumn{2}{|l|}{\[
\begin{gathered}
\text { ISA } \\
\text { CG }=37.0 \%
\end{gathered}
\]} \& \multicolumn{4}{|c|}{FUEL CONSUMED (KG) TIME (H.MIN)} \\
\hline \multirow[t]{2}{*}{\begin{tabular}{l}
AIR DIST. \\
(NM)
\end{tabular}} \& \multicolumn{6}{|c|}{FLIGHT LEVEL} \& \multicolumn{3}{|r|}{CORRECTION ON
FUEL CONSUMPTION (KG/1000KG)} \\
\hline \& 310 \& 330 \& 350 \& 370 \& 390 \& 410 \& \[
\begin{aligned}
\& \text { FL310 } \\
\& \text { FL330 } \\
\& \hline
\end{aligned}
\] \& \[
\begin{aligned}
\& \text { FL350 } \\
\& \text { FL370 }
\end{aligned}
\] \& \[
\begin{aligned}
\& \text { FL390 } \\
\& \text { FL410 } \\
\& \hline
\end{aligned}
\] \\
\hline 200 \& 3483
0.39 \& 3477
0.39 \& \& \& \& \& 11 \& \& \\
\hline 300 \& \begin{tabular}{l}
4654 \\
0.52 \\
\hline
\end{tabular} \& 4580
0.52 \& 4523
0.52 \& 4485
0.52 \& 4465
0.52 \& \begin{tabular}{l}
4460 \\
0.52 \\
\hline
\end{tabular} \& 14 \& 16 \& 18 \\
\hline 400 \& \(\begin{array}{r}5827 \\ 1.05 \\ \hline\end{array}\) \& \(\begin{array}{r}5685 \\ 1.05 \\ \hline\end{array}\) \& \(\begin{array}{r}5568 \\ 1.05 \\ \hline\end{array}\) \& \(\begin{array}{r}5478 \\ 1.05 \\ \hline\end{array}\) \& 5414
1.05 \& \begin{tabular}{l}
5375 \\
1.05 \\
\hline
\end{tabular} \& 16 \& 19 \& 22 \\
\hline 500 \& 7002
1.17 \& 6793
1.18 \& \begin{tabular}{l}
6615 \\
1.18 \\
\hline 18
\end{tabular} \& ¢473 \& \begin{tabular}{l}
6366 \\
1.18 \\
\hline 1
\end{tabular} \& 6393
1.18
1.18 \& 18 \& 22 \& 26 \\
\hline 600 \& \(\begin{array}{r}8181 \\ 1.30 \\ \hline\end{array}\) \& 7904
1.31 \& 7665

1.31 \& 7472
1.31 \& 7321

1.31 \& | 7214 |
| :--- |
| 1.31 |
| 1 | \& 21 \& 25 \& 29 <br>

\hline 700 \& $\begin{array}{r}9362 \\ 1.43 \\ \hline 1\end{array}$ \& 9017

1.44 \& | 8718 |
| :--- |
| 1.44 | \& 8474

1.45 \& 8280
1.45 \& 8140
1.45 \& 23 \& 28 \& 33 <br>
\hline 800 \& 10545
1.56
1 \& 10133
1.57
1.5 \& 9774
1.57 \& 9478
1.58 \& 1.92
1.58

1.58 \& | 9059 |
| :--- |
| 1.58 | \& 26 \& 31 \& 37 <br>

\hline 900 \& 11731
2.09 \& 11252

1209 \& 10833
2.10 \& 10486
2.11 \& 10206
2.11 \& 10002
2.11 \& 28 \& 34 \& 41 <br>
\hline 1000 \& 12920
2.21
1 \& 12373

2.22 \& 18994
2
2 \& |riticicl \& Pd12 ${ }^{2}$ \& 10939
2.24
1.20 \& 31 \& 37 \& 45 <br>

\hline 1100 \& | 14111 |
| :---: |
| 2.34 |
| 1 | \& $\begin{array}{r}13497 \\ \\ \\ \hline 235\end{array}$ \& $\begin{array}{r}12958 \\ \hline 2.36 \\ \hline 1\end{array}$ \& 12510

2.37
1.3 \& 12145
2.37
1 \& 11879
2.37
102 \& 33 \& 40 \& 49 <br>

\hline 1200 \& $\begin{array}{r}15305 \\ 2.47 \\ \hline 1\end{array}$ \& | 14624 |
| :--- |
|  |
|  |
| 2.48 | \& $\begin{array}{r}14026 \\ \hline 2.49 \\ \hline 1\end{array}$ \& $\begin{array}{r}13526 \\ 2.50 \\ \\ \hline\end{array}$ \& $\begin{array}{r}13119 \\ 2.50 \\ \hline\end{array}$ \& $\begin{array}{r}12823 \\ 2.50 \\ \hline 1\end{array}$ \& 36 \& 43 \& 53 <br>

\hline 1300 \& 16502
3.00 \& 15754
3.01
1 \& $\begin{array}{r}15096 \\ 3.02 \\ \hline\end{array}$ \& 14547
3.03
15 \& $\begin{array}{r}14096 \\ 3.03 \\ \hline 15\end{array}$ \& 13770
3.03 \& 38 \& 46 \& 57 <br>
\hline 1400 \& 17702
3.13 \& 16887
3.14 \& $\begin{array}{r}16170 \\ 3.15 \\ \hline 1.2\end{array}$ \& 15570
3.16 \& $\begin{array}{r}15077 \\ 3.16 \\ \hline 1\end{array}$ \& $\begin{array}{r}14423 \\ 3.16 \\ \hline\end{array}$ \& 41 \& 49 \& 61 <br>
\hline 1500 \& 18904
3.26 \& 18023
3.27 \& $\begin{array}{r}17247 \\ 3.28 \\ \hline 1\end{array}$ \& 16596
3.29
11 \& $\begin{array}{r}16062 \\ 3.29 \\ \hline\end{array}$ \& $\begin{array}{r}15679 \\ 3.29 \\ \hline 1\end{array}$ \& 44 \& 53 \& 65 <br>
\hline 1600 \& $\begin{array}{r}20109 \\ 3.38 \\ \hline\end{array}$ \& 19161
3.40 \& 18326

3.42 \& $\begin{array}{r}17625 \\ 3.42 \\ \hline\end{array}$ \& $\begin{array}{r}17050 \\ 3.42 \\ \hline\end{array}$ \& | 16639 |
| ---: |
| 3.42 |
| 1 | \& 46 \& 56 \& 70 <br>

\hline 1700 \& | 21317 |
| :---: |
| 3.51 | \& 20303

3.53 \& $\begin{array}{r}19409 \\ 3.55 \\ \hline\end{array}$ \& $\begin{array}{r}18657 \\ 3.55 \\ \hline\end{array}$ \& $\begin{array}{r}18041 \\ 3.55 \\ \hline\end{array}$ \& $\begin{array}{r}17602 \\ 3.55 \\ \hline\end{array}$ \& 49 \& 59 \& 74 <br>
\hline 1800 \& 22529
4.04 \& 21448
4.06 \& 20495
4.08 \& 19692
4.09 \& 19036
4.09 \& 18570
4.09 \& 52 \& 63 \& 79 <br>
\hline 1900 \& 23143
4.17 \& 22597
4.19 \& 21585
4.21 \& 20731
4.22 \& 20034
4.22 \& 19542
4.22 \& 55 \& 66 \& 83 <br>
\hline 2000 \& 24960
4.30 \& 23748
4.32 \& 22678
4.34 \& 21773
4.35 \& 21036
4.35 \& 20518
4.35 \& 58 \& 70 \& 88 <br>
\hline 2100 \& 26179
4.42 \& 24903
4.45 \& 23774
4.47 \& 22819
4.48 \& 22043
4.48 \& 21499
4.48 \& 61 \& 73 \& 94 <br>
\hline 2200 \& 27402
4.55 \& 26060
4.57 \& 24873
5.00 \& $\begin{array}{r}23868 \\ 5.01 \\ \hline\end{array}$ \& 23053
5.01 \& 22502
5.01 \& 64 \& 77 \& 99 <br>

\hline 2300 \& | 28628 |
| :---: |
| 5.08 | \& 27220

5.10 \& 25975
5.13 \& 24920
5.14 \& $\begin{array}{r}24066 \\ 5.14 \\ \hline\end{array}$ \& $\begin{array}{r}23494 \\ 5.14 \\ \hline\end{array}$ \& 67 \& 80 \& 104 <br>
\hline 2400 \& 29857
5.21 \& 28384
5
5.23 \& 27081
5.26 \& 25975
5.27 \& 25083
5.27 \& 24491
5.27 \& 70 \& 84 \& 110 <br>
\hline 2500 \& $\begin{array}{r}31089 \\ 5.34 \\ \hline\end{array}$ \& 29551
5.36 \& 28190
5.39 \& 27034
5.40 \& 26103
5.40 \& 25493
5.40 \& 73 \& 88 \& 115 <br>
\hline 2600 \& 32325
5
5.46 \& 30721
5.49 \& $\begin{array}{r}29303 \\ 5.52 \\ \hline\end{array}$ \& $\begin{array}{r}28097 \\ 5.53 \\ \hline\end{array}$ \& $\begin{array}{r}27128 \\ 5.53 \\ \hline\end{array}$ \& 26499
5.53 \& 76 \& 91 \& 121 <br>
\hline 2700 \& 33563
559
5 \& 31895
602 \& 30419
605 \& 29.63
6.06 \& 28183
606 \& 27510
6.06 \& 79 \& 95 \& 126 <br>

\hline $$
\begin{aligned}
& \text { PACK FLO } \\
& \triangle \text { FUEL }=
\end{aligned}
$$ \& \& \multicolumn{3}{|l|}{\[

$$
\begin{aligned}
& \text { PACK FLOW HI OR/ } \\
& \text { AND CARGO COOL ON } \\
& \triangle F U E L=+1 \% \\
& \hline
\end{aligned}
$$
\]} \& \multicolumn{2}{|l|}{ENGINE ANTICE ON

$$
\triangle \text { FUEL }=+1.5 \%
$$} \& \multicolumn{3}{|c|}{TOTAL ANTIICE ON

$$
\triangle \text { FUEL }=+7 \%
$$} <br>

\hline
\end{tabular}

\begin{tabular}{|c|c|c|c|c|c|c|c|c|c|}
\hline \multicolumn{10}{|c|}{ALTERNATE PLANNING FROM DESTINATION TO ALTERNATE AIRPORT GO-AROUND : 500 KG - CLIMB : \(250 \mathrm{KT} / 300 \mathrm{KT} / \mathrm{M} .80\) - CRUISE : LONG RANGE DESCENT : M. \(80 / 300 \mathrm{KT} / 250 \mathrm{KT}\) - VMC PROCEDURE : 160 KG (4MIN)} \\
\hline \multicolumn{4}{|l|}{REF. LDG WT AT ALTERNATE \(=140000 \mathrm{KG}\) NORMAL AIR CONDITIONING ANTI ICING OFF} \& \multicolumn{2}{|l|}{\[
\begin{gathered}
\text { ISA } \\
\mathrm{CG}=30.0 \%
\end{gathered}
\]} \& \multicolumn{4}{|c|}{FUEL CONSUMED (KG) TIME (H.MIN)} \\
\hline \multirow[t]{2}{*}{\begin{tabular}{l}
AIR \\
DIST. \\
(NM)
\end{tabular}} \& \multicolumn{6}{|c|}{FUGHT LEVEL} \& \multicolumn{3}{|l|}{CORRECTION ON
FUEE CONSUMPTION
(KG/1000KG)} \\
\hline \& 100 \& 120 \& 140 \& 160 \& 180 \& 200 \& \[
\begin{aligned}
\& \text { FL100 } \\
\& \text { FL120 }
\end{aligned}
\] \& \[
\begin{aligned}
\& \text { FL140 } \\
\& \text { FL160 }
\end{aligned}
\] \& \[
\begin{aligned}
\& \hline \text { FL180 } \\
\& \text { FL200 } \\
\& \hline
\end{aligned}
\] \\
\hline 50 \& 1532
0.14 \& Inter \& olate \& \& \& \& 2 \& \& \\
\hline 100 \& 2324
0.24 \& 2242
0.25 \& 2228
0.24 \& 2225
0.24 \& 2230
0.23 \& 2239
0.23 \& 4 \& 5 \& 5 \\
\hline 150 \& 3118
0.34 \& 3004
0.35 \& 2960
0.34 \& 2933
0.34 \& 2917
0.32 \& 2902
0.31 \& 7 \& 7 \& 7 \\
\hline 200 \& 3913
0.45 \& 3768
0.45 \& 3694
0.45 \& 3642
0.44 \& 3606
0.41 \& 3567
0.40 \& 9 \& 10 \& 10 \\
\hline 250 \& 4711
0.55 \& 4533
0.55 \& 4430
0.55 \& \begin{tabular}{l}
4354 \\
0.54 \\
\hline
\end{tabular} \& 4298
0.50 \& \begin{tabular}{l}
4233 \\
0.48 \\
\hline
\end{tabular} \& 11 \& 12 \& 12 \\
\hline 300 \& \(\begin{array}{r}5510 \\ 1.05 \\ \hline\end{array}\) \& 5300
1.05 \& \(\begin{array}{r}5167 \\ 1.05 \\ \hline\end{array}\) \& \begin{tabular}{l}
5068 \\
1.04 \\
\hline
\end{tabular} \& 4991
0.59 \& 4901
0.57 \& 13 \& 15 \& 14 \\
\hline 350 \& 6311
1.15 \& 6069
1.15 \& 5906
1.15 \& 5784
1.13 \& \(\begin{array}{r}5686 \\ 1.08 \\ \hline\end{array}\) \& 5569
1.05 \& 16 \& 18 \& 17 \\
\hline 400 \& 7113
1.25 \& \begin{tabular}{l}
1839 \\
1.26 \\
\hline 1
\end{tabular} \& 6647
1.25 \& 6502
1.23 \& 1384
1.16 \& 6239
1.14 \& 18 \& 20 \& 19 \\
\hline 450 \& 7918
1.35 \& 7611
1.36
1 \& 7390
1.35 \& 7221
1.33 \& 7083
1.25 \& 6911
1.22 \& 20 \& 23 \& 21 \\
\hline 500 \& 8724
1.45 \& \begin{tabular}{l}
8384 \\
1.46 \\
\hline 1
\end{tabular} \& 8134
1.45 \& 7942
1.43 \& 1784
1.34 \& 7583
1.31 \& 22 \& 26 \& 24 \\
\hline 550 \& \(\begin{array}{r}9532 \\ 1.56 \\ \hline\end{array}\) \& 9160
1.56 \& 8880
1.55 \& 8666
1.53 \& 8487
1.42 \& 8257
1.39 \& 25 \& 28 \& 26 \\
\hline 600 \& 10342
2.06
1 \& \begin{tabular}{l}
9937 \\
2.06 \\
\hline 106
\end{tabular} \& \({ }_{2628}^{9.05}\) \& 9391
2.02 \& 9193
1.51 \& 8933
1.48
1 \& 27 \& 31 \& 28 \\
\hline 650 \& 11154
2.16 \& 10715
2.16 \& 10378

2.15 \& 10119
2.12 \& 1990
1.59 \& 9609
1.56 \& 29 \& 34 \& 31 <br>
\hline 700 \& 11967
2.26 \& 11496
2.26
1029 \& 11130
2.25 \& 10848
2.22
102 \& 10609

2.08
108 \& 10287
2.04
1 \& 32 \& 37 \& 33 <br>
\hline 750 \& 12782
2.36
1 \& $\begin{array}{r}12278 \\ \hline 2.36 \\ \hline 108\end{array}$ \& 11883
2.35
1 \& 11579
2.32
1 \& 11321
2.16 \& 10966
2.13 \& 34 \& 40 \& 35 <br>
\hline 800 \& 13599
2.46 \& $\begin{array}{r}13061 \\ 2.47 \\ \hline 1\end{array}$ \& 12639
2.45 \& $\begin{array}{r}12313 \\ 2.41 \\ \hline 15\end{array}$ \& $\begin{array}{r}12034 \\ 2.25 \\ \hline\end{array}$ \& 11647
2.21
1029 \& 36 \& 42 \& 37 <br>
\hline 850 \& 14418
2.56 \& $\begin{array}{r}13847 \\ 2.57 \\ \hline 1\end{array}$ \& 13396

2.55 \& 13048

1.51 \& 12750
2.33 \& 12329
2.29
102 \& 39 \& 45 \& 39 <br>
\hline 900 \& 15238
3.06 \& $\begin{array}{r}14634 \\ 3.07 \\ \hline\end{array}$ \& $\begin{array}{r}14.155 \\ 3.05 \\ \hline\end{array}$ \& 13785
3.01 \& 13468

1.41 \& 13012
2.38
18 \& 41 \& 48 \& 42 <br>
\hline 950 \& 16059
3.16 \& $\begin{array}{r}15423 \\ 3.17 \\ \hline 1\end{array}$ \& 14916
3.15 \& $\begin{array}{r}14525 \\ 3.10 \\ \hline 15\end{array}$ \& 14187
2.49 \& 13697
2.46
1 \& 43 \& 51 \& 44 <br>
\hline 1000 \& 16883
3.26 \& 16214
3.27 \& 15679
3.25 \& $\begin{array}{r}15266 \\ 3.20 \\ \hline\end{array}$ \& 14909
2.57 \& 14383
2.54
1 \& 46 \& 54 \& 46 <br>
\hline 1050 \& 17708
3.37 \& $\begin{array}{r}17007 \\ 3.37 \\ \hline\end{array}$ \& $\begin{array}{r}16444 \\ 3.35 \\ \hline 17\end{array}$ \& $\begin{array}{r}16009 \\ 3.29 \\ \hline 1\end{array}$ \& 15628
3.06 \& 15071
3.03
15 \& 48 \& 57 \& 48 <br>
\hline 1100 \& $\begin{array}{r}18534 \\ 3.47 \\ \hline\end{array}$ \& 17801
3.47 \& 17212
3.45 \& $\begin{array}{r}16754 \\ 3.38 \\ \hline 15\end{array}$ \& 16346
3.14
1 \& 15760
3.11 \& 50 \& 60 \& 50 <br>
\hline 1150 \& $\begin{array}{r}19363 \\ 3.57 \\ \hline\end{array}$ \& $\begin{array}{r}18597 \\ 3.57 \\ \hline\end{array}$ \& $\begin{array}{r}17981 \\ 3.55 \\ \hline\end{array}$ \& 17501
3.47 \& $\begin{array}{r}17066 \\ 3.23 \\ \hline 178\end{array}$ \& 16451
3.19 \& 53 \& 63 \& 52 <br>
\hline 1200 \& 20193
4.07 \& 19395

4.07 \& $$
\begin{array}{r}
18752 \\
\hline 4.05 \\
\hline
\end{array}
$$ \& 18250

3 \& 17787
3

3.31 \& $$
\begin{array}{r}
17143 \\
3.27 \\
\hline
\end{array}
$$ \& 55 \& 66 \& 54 <br>

\hline  \& VL0 \& \multicolumn{3}{|l|}{$\qquad$} \& \multicolumn{2}{|l|}{ENGINE
ANTI ICE ON
$\triangle$ FUEL $=+1.5$} \& \multicolumn{3}{|c|}{TOTAL ANTI ICE ON $\triangle$ FUEL $=+6 \%$} <br>
\hline
\end{tabular}

