

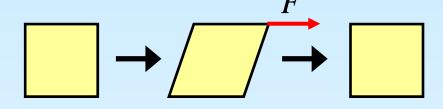
OUTLINE

- 1. Solids and fluids
- 2. Fluid Statics
- 3. Perfect gas
- 4. Newtonian fluids
- 5. Navier-Stokes equations
- 6. Rotating frames
- 7. Bernouilli equation

1. Solids and fluids

Solid:

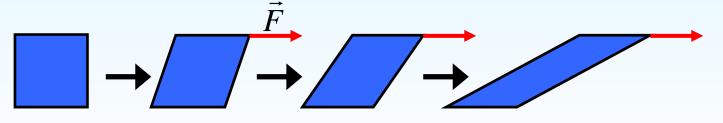
- has a preferred shape
- tit takes another (constant) shape under the action of (constant) external forces



* it relaxes to that shape when the external forces are removed

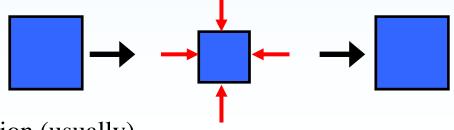
Fluid:

- ❖ has not any preferred shape
- * it changes shape continuously under the action of fixed external forces



This is so for shear stresses, but:

➤ for normal stresses fluids and solids behave similarly



* normal stresses can only be a compression (usually)

Fluids:

*Gases: always tend to expand and occupy the entire volume of any container

*Liquids: the volume does not change very much

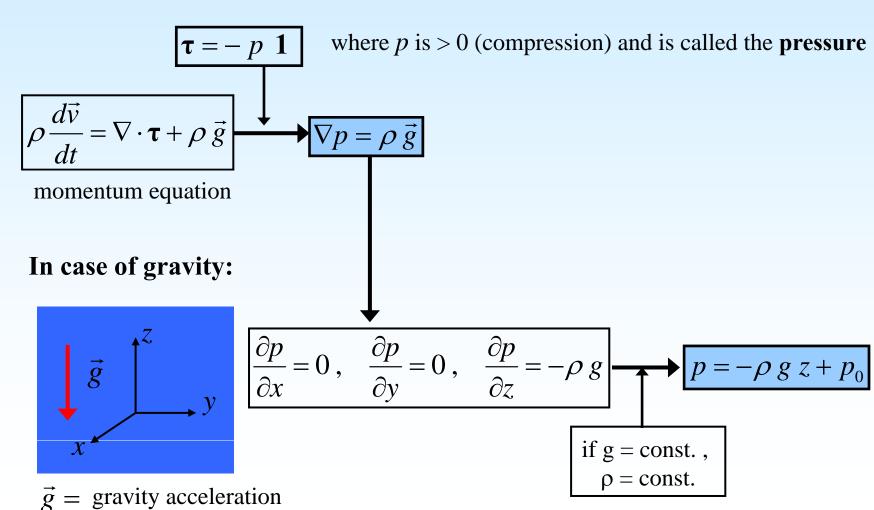
The distinction between solids and fluids apply well to many materials under normal conditions but:

- > solids under very strong shear stresses may behave partially as fluids (plasticity)
- > some fluids (like jelly, paint, polymer solutions, ...) may behave partially as solids and have a partial memory of a 'preferred shape' (viscoelasticity)

2. Fluid statics (or Hydrostatics)

Fluid at rest:

It is empirically found that stresses are isotropic:



3. Perfect gas

The density of fluids depends on pressure and temperature.

$$\rho = \rho(p,T)$$
 is called the equation of state

most gasses in normal conditions obey the following equation of state:

$$T = \text{absolute temperature}$$

$$R = \frac{R_U}{m_m}$$

$$R_U = \text{universal constant} = 8.314 \text{ J mol}^{-1} \text{ K}^{-1}$$

$$m_m = \text{molecular mass. For dry air} = 28.97 \text{ Kg/kmol}$$

Perfect gas = the limit case when this equation is verified exactly

Thermodynamic properties of perfect gasses

Specific heat:

$$C_{V} = \frac{1}{m} \left(\frac{dQ}{dT} \right)_{V = \text{const.}} \qquad C_{p} = \frac{1}{m} \left(\frac{dQ}{dT} \right)_{p = \text{const.}} \qquad C_{p} - C_{v} = R_{U}$$

$$\gamma = \frac{C_{p}}{C}$$

for air at ordinary temperatures: $\gamma = 1.4$ and $C_p = 1005 \text{ J Kg}^{-1} \text{ K}^{-1}$

$$\gamma = \frac{C_p}{C}$$

For an adiabatic (no heat transfer) process:

$$\frac{p}{\rho^{\gamma}}$$
 = const.

Thermal expansion coefficient
$$\alpha = -\frac{1}{\rho} \left(\frac{\partial \rho}{\partial T} \right)_{p=\text{const.}} = \frac{1}{T}$$

The specific internal energy is a function of the temperature only: U = U(T)

4. Newtonian fluids

Problem of fluid motions

Equation of motion: Cauchy or momentum equations

$$\rho \frac{d\vec{v}}{dt} = \nabla \cdot \mathbf{\tau} + \rho \, \vec{g} \qquad \left(m\vec{a} = \vec{F} \right)$$

Usually, the body forces \vec{g} are known

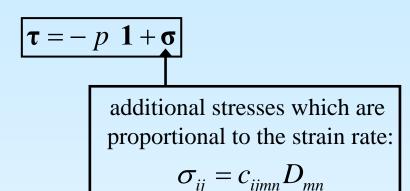
but, we need to know in order to solve for the motion constitutive equations

Fluid at rest:

$\tau = -p 1$

p = called
 thermodynamic
 pressure

Fluid in motion:



If the fluid is isotropic:

$$c_{ijmn} = \lambda \delta_{ij} \delta_{mn} + \mu \delta_{im} \delta_{jn} + \gamma \delta_{in} \delta_{jm}$$

$$\sigma_{ij} = \text{symmetric} \implies \qquad \mu - \gamma \qquad \implies \qquad \sigma_{ij} = 2\mu D_{ij} + \lambda D_{kk} \delta_{ij}$$

$$\tau_{ij} = -p\delta_{ij} + \lambda D_{kk}\delta_{ij} + 2\mu D_{ij}$$

But the pressure was already defined from the mean normal stress

$$\overline{p} = -\frac{1}{3}\tau_{ii}$$

this is called the **mean pressure** and it is different from the thermodynamic one

$$\boxed{\tau_{ij} = -p\delta_{ij} + \lambda D_{kk}\delta_{ij} + 2\mu D_{ij}} \implies \boxed{\overline{p} = p - \left(\lambda + \frac{2}{3}\mu\right)D_{ii}}$$

$$\overline{p} = p - \left(\lambda + \frac{2}{3}\mu\right)D_{ii}$$

$$p - \overline{p} = \left(\lambda + \frac{2}{3}\mu\right)\nabla \cdot \vec{v}$$

$$\nabla \cdot \vec{v} = 0$$

$$p = \overline{p}$$

For incompressible fluid:
$$\nabla \cdot \vec{v} = 0$$
 \Rightarrow $p = \overline{p}$ \Rightarrow $\tau_{ij} = -p\delta_{ij} + 2\mu D_{ij}$

In general, bulk viscosity: $\kappa = \lambda + \frac{2}{3}\mu$

$$\kappa = \lambda + \frac{2}{3}\mu$$

is found to be ≈ 0

$$\kappa = \lambda + \frac{2}{3}\mu = 0$$

Newtonian fluid:
$$\kappa = \lambda + \frac{2}{3}\mu = 0$$

$$\tau_{ij} = -\left(p + \frac{2}{3}\mu\nabla\cdot\vec{v}\right)\delta_{ij} + 2\mu D_{ij}$$

Air and water obey very well the Newtonian fluid model

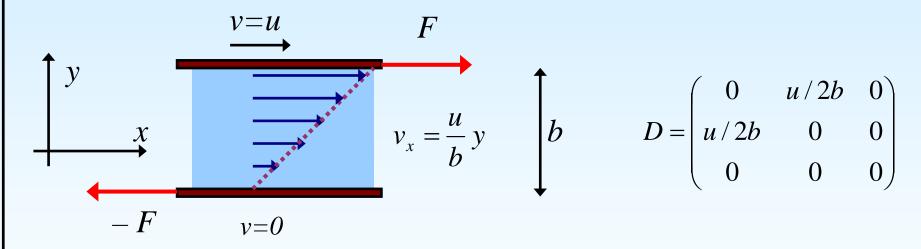
Examples of non Newtonian fluids:

- * solutions containing polymer molecules
- **❖** blood
- * water with clay
 - → Stresses are nonlinear functions of strain rates
 - → Stresses depend not only on instantaneous values of strain rate but also on its history → material with memory → viscoelastic

$\mu = viscosity coefficient$

it depends on the thermodynamic properties (T, ρ)

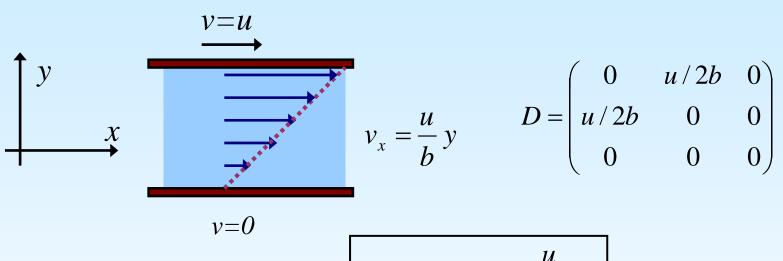
Meaning: shear experiment



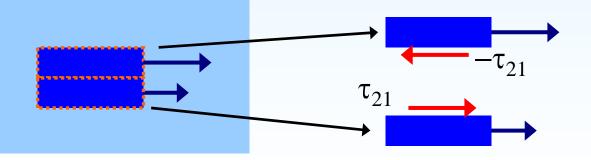
$$\tau_{ij} = -\left(p + \frac{2}{3}\mu\nabla\cdot\vec{v}\right)\delta_{ij} + 2\mu D_{ij} \implies \tau_{21} = 2\mu D_{21} \implies \frac{F}{A} = 2\mu \frac{u}{2b}$$

$$\mu = \frac{F/A}{u/b}$$

Meaning: the viscosity tends to smooth out the gradients in velocity



$$\tau_{21} = 2\mu D_{21} = \mu \frac{u}{b} > 0$$



faster particles are slowed down

slower particles are accelerated

Viscous dissipation

$$\mathbf{\tau}:\mathbf{D} = -p\nabla\cdot\vec{v} + \mathbf{\tau}':\mathbf{D}'$$

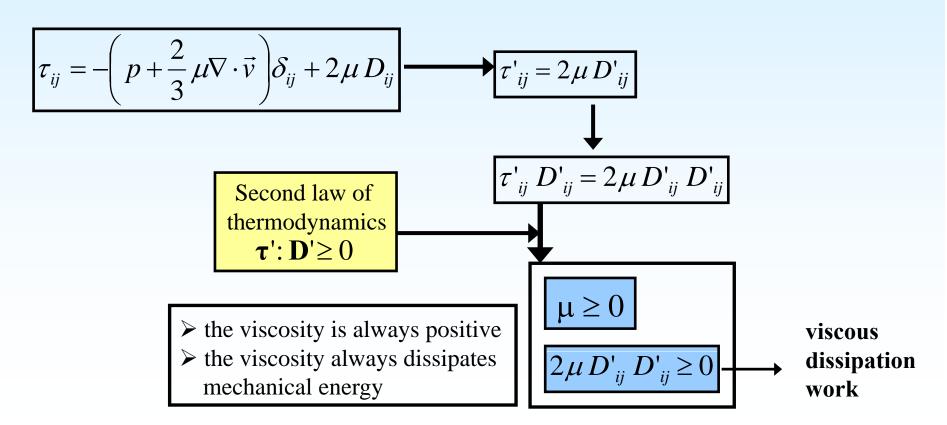
deformation work

work

shear work

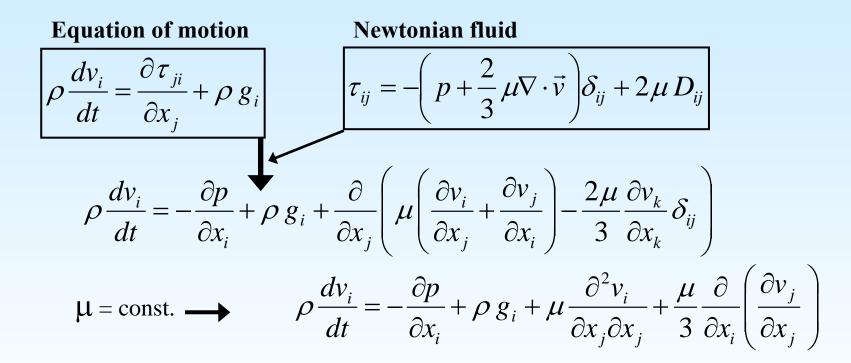
associated to changes

in volume



5. Navier-Stokes equations

Problem of fluid motions



Navier-Stokes equations

$$\frac{dv_i}{dt} = \frac{\partial v_i}{\partial t} + v_j \frac{\partial v_i}{\partial x_i}$$

$$\rho \frac{d\vec{v}}{dt} = -\nabla p + \rho \vec{g} + \mu \nabla^2 \vec{v} + \frac{\mu}{3} \nabla (\nabla \cdot \vec{v})$$

but the pressure and the density changes are unknown $\rightarrow \rho$ and p are also variables



continuity equation and state equation are involved:

$$\frac{d\rho}{dt} + \rho \, \nabla \cdot \vec{v} = 0$$

$$\rho = \rho(p, T)$$

 \Rightarrow T is also a variable!

equation for the temperature is needed: first law of Thermodynamics

$$\rho \frac{du}{dt} = \mathbf{\tau} : \mathbf{D} - \nabla \cdot \vec{q} + \rho \, r$$

$$u = u(T, \rho)$$

Therefore, the dynamical problem involves at least (in general):

6 variables: velocity, pressure, density, temperature:

$$\vec{v}(\vec{x},t), p(\vec{x},t), \rho(\vec{x},t), T(\vec{x},t)$$

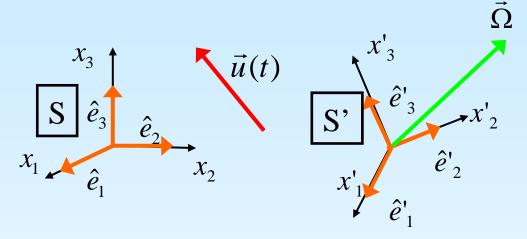
6 equations:

$$\begin{cases}
\rho \frac{d\vec{v}}{dt} = -\nabla p + \rho \, \vec{g} + \mu \nabla^2 \vec{v} + \frac{\mu}{3} \nabla (\nabla \cdot \vec{v}) \\
\frac{d\rho}{dt} + \rho \, \nabla \cdot \vec{v} = 0 \\
\rho = \rho(p, T) \\
\rho \frac{du}{dt} = \mathbf{\tau} : \mathbf{D} - \nabla \cdot \vec{q} + \rho \, r \qquad u = u(T, \rho)
\end{cases}$$

If viscosity can be neglected:

$$\frac{d\vec{v}}{dt} = -\nabla p + \rho \, \vec{g}$$
Euler equations
$$\frac{\partial v_i}{\partial t} + v_j \frac{\partial v_i}{\partial x_j} = -\frac{1}{\rho} \frac{\partial p}{\partial x_i} + g_i$$

6. Rotating frames



time derivative in S

$$\left(\frac{d\vec{u}}{dt}\right)_{S} = \frac{du_{i}}{dt}\hat{e}_{i}$$

time derivative in S'

$$\left(\frac{d\vec{u}}{dt}\right)_{S'} = \frac{du'_{i}}{dt}\hat{e}'_{i}$$

$$\vec{u} = u_i \hat{e}_i = u'_i \hat{e}'_i \implies \left(\frac{d\vec{u}}{dt}\right)_S = \frac{du'_i}{dt} \hat{e}'_i + u'_i \left(\frac{d\hat{e}'_i}{dt}\right)_S = \left(\frac{d\vec{u}}{dt}\right)_{S'} + u'_i \left(\frac{d\hat{e}'_i}{dt}\right)_S$$

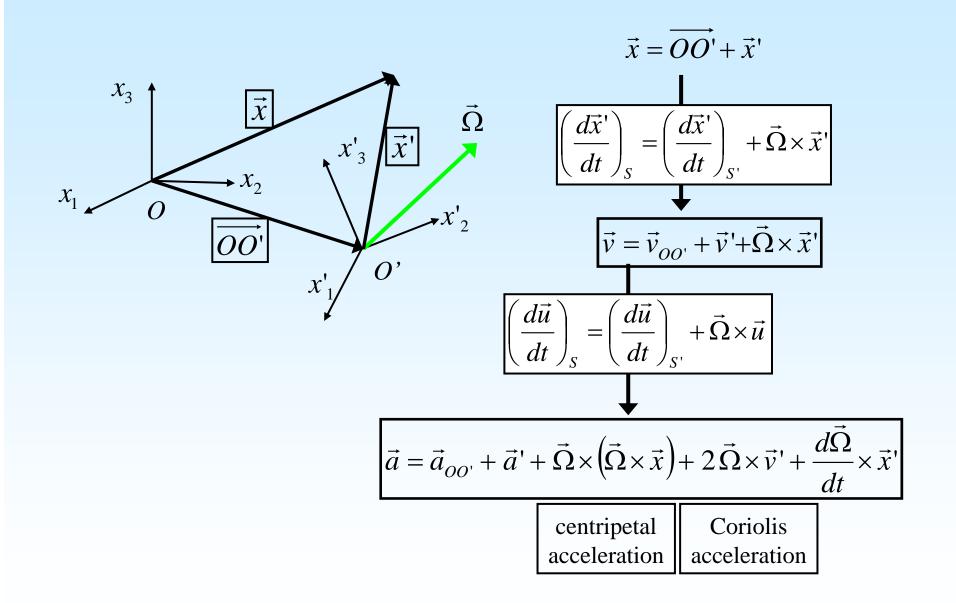
$$\left(\frac{d\hat{e}'_{i}}{dt}\right)_{S} = ? \qquad \hat{e}'_{i} = Q_{ji}\hat{e}_{j} \implies \frac{d\hat{e}'_{i}}{dt} = \frac{dQ_{ji}}{dt}\hat{e}_{j} = \frac{dQ_{ji}}{dt}Q_{jk}\hat{e}'_{k} = W_{ki}\hat{e}'_{k}$$
antisymmetric

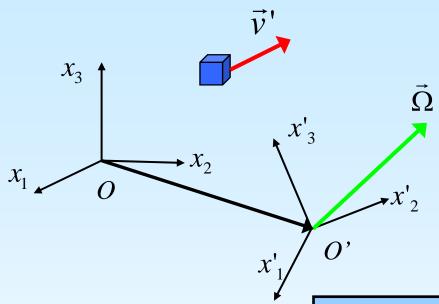
$$\Omega_{i} = -\frac{1}{2} \varepsilon_{ijk} W_{jk} \longrightarrow \left(\frac{d\hat{e}'_{i}}{dt} \right)_{S} = \vec{\Omega} \times \hat{e}'_{i}$$

 $\vec{\Omega} = \text{angular velocity of frame S'}$ relative to frame S

Euler equations

$$\left(\frac{d\vec{u}}{dt}\right)_{S} = \left(\frac{d\vec{u}}{dt}\right)_{S'} + \vec{\Omega} \times \vec{u}$$





Navier Stokes equations in the rotating frame

$$\rho \frac{d\vec{v}'}{dt} = -\nabla p + \rho \, \vec{g}' + \mu \nabla^2 \vec{v} + \frac{\mu}{3} \nabla (\nabla \cdot \vec{v}) - 2\rho \, \vec{\Omega} \times \vec{v}$$

with

Coriolis force

$$\vec{g}' = \vec{g} - \vec{\Omega} \times (\vec{\Omega} \times \vec{x}') - \vec{a}_{O'O} - \frac{d\vec{\Omega}}{dt} \times \vec{x}'$$
centrifugal

force / p

7. Bernouilli equation

Simplified expression of the Euler equations (equations of motion for inviscid fluid, i.e., viscosity is neglected)

Assumptions:

- \triangleright no viscosity, $\mu = 0$
- \triangleright barotropic flow, i.e., $\rho = \rho(p)$
- ➤ body force is gravity (= const.)

$$\vec{g} = -g\hat{e}_3$$

$$x_1$$

Euler equations:

$$\frac{\partial \vec{v}}{\partial t} + \vec{v} \cdot \nabla \vec{v} = -\frac{1}{\rho} \nabla p + \vec{g}$$

$$\Rightarrow \quad \vec{g} = -\nabla(gx_3)$$

$$\vec{g} = -\nabla(gx_3)$$

$$\vec{v} \cdot \nabla \vec{v} = \vec{\omega} \times \vec{v} + \nabla\left(\frac{1}{2}v^2\right) \text{ vector identity already proven}$$

$$F(p) \equiv \int \frac{dp}{\rho(p)} \implies \nabla F = \frac{dF}{dp} \nabla p = \frac{1}{\rho} \nabla p$$

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$$\Rightarrow$$
 $\vec{g} = -\nabla(gx_3) \equiv -\nabla(gz)$

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$$\Rightarrow F(p) \equiv \int \frac{dp}{\rho(p)} \Rightarrow \nabla F = \frac{dF}{dp} \nabla p = \frac{1}{\rho} \nabla p$$

$$\frac{\partial \vec{v}}{\partial t} + \nabla \Theta = \vec{v} \times \vec{\omega}$$

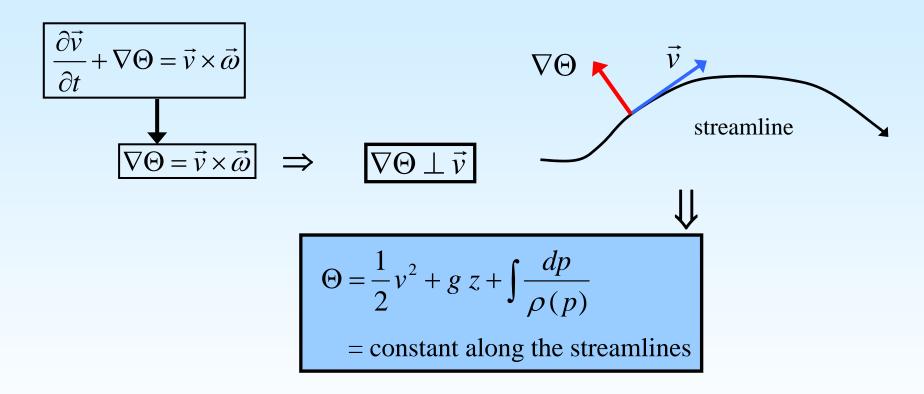
$$\frac{\partial \vec{v}}{\partial t} + \nabla \Theta = \vec{v} \times \vec{\omega} \qquad \text{with} \qquad \Theta = \frac{1}{2} v^2 + g z + \int \frac{dp}{\rho(p)}$$

Bernouilli **function**

This equation is specially useful in two particular cases:

- Steady flow
- Irrotational flow

Bernouilli equation for steady flow

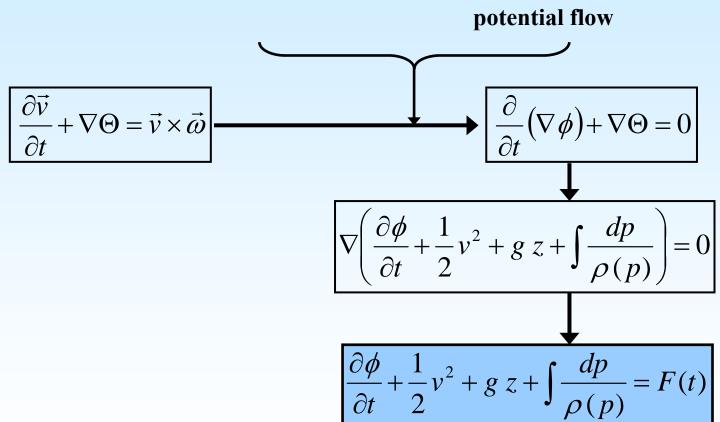


The constant may be different for different streamlines

 $\vec{\omega} = 0 \implies$ the constant is the same everywhere in the flow

Bernouilli equation for irrotational flow

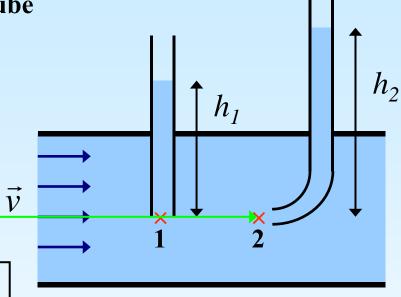
Irrotational flow:
$$\vec{\omega} = \nabla \times \vec{v} = 0$$
 \Leftrightarrow $\exists \phi \mid \vec{v} = \nabla \phi$ potential flow



Application of Bernouilli: Pitot tube

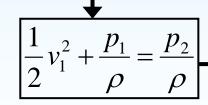
device to measure a flow velocity

- > steady flow
- > viscosity is neglected
- $\triangleright \rho = const.$



$$\frac{1}{2}v_1^2 + gz_1 + \frac{p_1}{\rho} = \frac{1}{2}v_2^2 + gz_2 + \frac{p_2}{\rho}$$

streamline



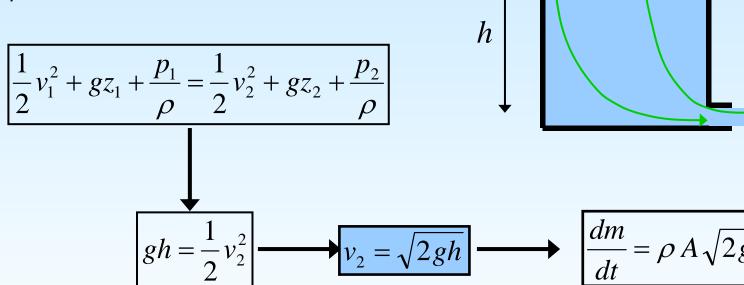
$$v_1 = \sqrt{2g(h_2 - h_1)}$$

inside the vertical tubes there is hydrostatic balance:

$$p_1 = p_{atm} + \rho g h_1$$
 , $p_2 = p_{atm} + \rho g h_2$

Application of Bernouilli: orifice in a tank

- \triangleright orifice small \Rightarrow flow approximately steady
- ➤ viscosity is neglected
- $\triangleright \rho = \text{const.}$



Actually, the function

$$\Theta = \frac{1}{2}v^2 + gz + \frac{p}{\rho}$$

has the same value everywhere since it has the same value at any point on the free surface \Rightarrow flow is irrotational

$$\frac{\partial \vec{v}}{\partial t} + \nabla \Theta = \vec{v} \times \vec{\omega}$$

$$\vec{v} \times \vec{\omega} = 0 \implies \vec{\omega} = 0$$

streamline

 p_{atm}