## Continuum and Fluid Mechanics

## CHAPTER 5: <br> Basic equations of Fluid Mech friicst

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## OUTLINE

1. Solids and fluids
2. Fluid Statics
3. Perfect gas
4. Newtonian fluids
5. Navier-Stokes equations
6. Rotating frames
7. Bernouilli equation

## 1. Solids and fluids

## Solid:

* has a preferred shape
* it takes another (constant) shape under the action of (constant) external forces
 * it relaxes to that shape when the external forces are removed


## Fluid:

* has not any preferred shape
* it changes shape continuously under the action of fixed external forces


This is so for shear stresses, but: $>$ for normal stresses fluids and solids behave similarly

normal stresses can only be a compression (usually)

## Fluids: $\left\{\begin{array}{l}\text { Gases: always tend to expand and occupy } \\ \text { the entire volume of any container } \\ \& \text { Liquids: the volume does not change very much }\end{array}\right.$

The distinction between solids and fluids apply well to many materials under normal conditions but:
$>$ solids under very strong shear stresses may behave partially as fluids (plasticity)
$>$ some fluids (like jelly, paint, polymer solutions, ...) may behave partially as solids and have a partial memory of a 'preferred shape' (viscoelasticity)

## 2. Fluid statics (or Hydrostatics)

## Fluid at rest:

It is empirically found that stresses are isotropic:
$\vec{g}=$ gravity acceleration

$$
\text { if } \mathrm{g}=\text { const. , }
$$

$$
\rho=\text { const. }
$$

## 3. Perfect gas

The density of fluids depends on pressure and temperature.

$$
\rho=\rho(p, T) \quad \text { is called the equation of state }
$$

most gasses in normal conditions obey the following equation of state:

$$
p=\rho R T
$$

$$
\begin{aligned}
& T=\text { absolute temperature } \\
& R=\frac{R_{U}}{m_{m}} \\
& R_{U}=\text { universal constant }=8.314 \mathrm{~J} \mathrm{~mol}^{-1} \mathrm{~K}^{-1} \\
& m_{m}=\text { molecular mass. For dry air }=28.97 \mathrm{Kg} / \mathrm{kmol}
\end{aligned}
$$

Perfect gas $=$ the limit case when this equation is verified exactly

## Thermodynamic properties of perfect gasses

Specific heat:

$$
\begin{array}{lr}
C_{V}=\frac{1}{m}\left(\frac{d Q}{d T}\right)_{V=\text { const. }} C_{p}=\frac{1}{m}\left(\frac{d Q}{d T}\right)_{p=\text { const. }} & C_{p}-C_{v}=R_{U} \\
\text { for air at ordinary temperatures: } & \gamma=\frac{C_{p}}{C_{V}}
\end{array}
$$

For an adiabatic (no heat transfer) process: $\frac{p}{\rho^{\gamma}}=$ const.
Thermal expansion coefficient $\quad \alpha=-\frac{1}{\rho}\left(\frac{\partial \rho}{\partial T}\right)_{p=\text { const. }}=\frac{1}{T}$

The specific internal energy is a function of the temperature only: $U=U(T)$

## 4. Newtonian fluids

## Problem of fluid motions

Equation of motion: Cauchy or momentum equations

$$
\rho \frac{d \vec{v}}{d t}=\nabla \cdot \boldsymbol{\tau}+\rho \vec{g}
$$

$$
(m \vec{a}=\vec{F})
$$

Usually, the body forces $\vec{g}$ are known
$>$ but, we need to know $\boldsymbol{\tau ( \vec { v } )}$
in order to solve for the motion

Fluid at rest:

$$
\begin{gathered}
\hline \boldsymbol{\tau}=-p 1 \\
p=\text { called } \\
\text { thermodynamic } \\
\text { pressure }
\end{gathered}
$$

## Fluid in motion:


additional stresses which are proportional to the strain rate:

$$
\sigma_{i j}=c_{i j m n} D_{m n}
$$

If the fluid is isotropic:

$$
\begin{gathered}
c_{i j m n}=\lambda \delta_{i j} \delta_{m n}+\mu \delta_{i m} \delta_{j n}+\gamma \delta_{i n} \delta_{j m} \\
\sigma_{\mathrm{ij}}=\text { symmetric } \Rightarrow \quad \mu=\gamma \Rightarrow \sigma_{i j}=2 \mu D_{i j}+\lambda D_{k k} \delta_{i j} \\
\tau_{i j}=-p \delta_{i j}+\lambda D_{k k} \delta_{i j}+2 \mu D_{i j}
\end{gathered}
$$

But the pressure was already defined from the mean normal stress $\quad \bar{p}=-\frac{1}{3} \tau_{\text {ii }}$
this is called the mean pressure and it is different from the thermodynamic one

$$
\begin{array}{r}
\overline{\tau_{i j}=-p \delta_{i j}+\lambda D_{k k} \delta_{i j}+2 \mu D_{i j}} \Rightarrow \begin{array}{|}
\bar{p}=p-\left(\lambda+\frac{2}{3} \mu\right) D_{i i} \\
p-\bar{p}=\left(\lambda+\frac{2}{3} \mu\right) \nabla \cdot \vec{v}
\end{array} \\
\end{array}
$$

For incompressible fluid: $\nabla \cdot \vec{v}=0 \Rightarrow p=\bar{p} \Rightarrow \tau_{i j}=-p \delta_{i j}+2 \mu D_{i j}$

In general, bulk viscosity: $\quad \kappa=\lambda+\frac{2}{3} \mu \quad$ is found to be $\approx 0$
Newtonian fluid: $\kappa=\lambda+\frac{2}{3} \mu=0 \longrightarrow \tau_{i j}=-\left(p+\frac{2}{3} \mu \nabla \cdot \vec{v}\right) \delta_{i j}+2 \mu D_{i j}$

Air and water obey very well the Newtonian fluid model

## Examples of non Newtonian fluids:

* solutions containing polymer molecules
$\star$ blood
* water with clay
$\rightarrow$ Stresses are nonlinear functions of strain rates
$\rightarrow$ Stresses depend not only on instantaneous values of strain rate but also on its history $\rightarrow$ material with memory $\rightarrow$ viscoelastic
$\mu=$ viscosity coefficient it depends on the thermodynamic properties ( $T, \rho$ )

Meaning: shear experiment


$$
\mu=\frac{F / A}{u / b}
$$

Meaning: the viscosity tends to smooth out the gradients in velocity


## Viscous dissipation

$$
\begin{array}{ccc|}
\hline \boldsymbol{\tau}: \mathbf{D} \quad= & -p \nabla \cdot \vec{v} & + \\
\begin{array}{c}
\boldsymbol{\tau}^{\prime}: \mathbf{D}^{\prime} \\
\hline \text { deformation } \\
\text { work }
\end{array} & \begin{array}{c}
\text { work } \\
\text { associated } \\
\text { to changes } \\
\text { in volume }
\end{array} & \\
& \text { shear } \\
& \text { work } \\
& & \\
\hline
\end{array}
$$


viscous
dissipation work

## 5. Navier-Stokes equations

## Problem of fluid motions

$$
\begin{aligned}
& \text { Equation of motion } \\
& \left.\left.\begin{array}{l}
\text { Newtonian fluid } \\
\rho \frac{d v_{i}}{d t}=\frac{\partial \tau_{j i}}{\partial x_{j}}+\rho g_{i} \\
\rho \frac{d v_{i}}{d t}=-\frac{\partial p}{\partial x_{i}}+\rho g_{i}+\frac{\partial}{\partial x_{j}}\left(\mu+\frac{2}{3} \mu \nabla \cdot \vec{v}\right) \delta_{i j}+2 \mu D_{i j} \\
\hline x_{j}
\end{array}+\frac{\partial v_{j}}{\partial x_{i}}\right)-\frac{2 \mu}{3} \frac{\partial v_{k}}{\partial x_{k}} \delta_{i j}\right) \\
& \mu=\text { const. } \rightarrow \quad \rho \frac{d v_{i}}{d t}=-\frac{\partial p}{\partial x_{i}}+\rho g_{i}+\mu \frac{\partial^{2} v_{i}}{\partial x_{j} \partial x_{j}}+\frac{\mu}{3} \frac{\partial}{\partial x_{i}}\left(\frac{\partial v_{j}}{\partial x_{j}}\right)
\end{aligned}
$$

Navier-Stokes equations

$$
\frac{d v_{i}}{d t}=\frac{\partial v_{i}}{\partial t}+v_{j} \frac{\partial v_{i}}{\partial x_{j}}
$$

$$
\rho \frac{d \vec{v}}{d t}=-\nabla p+\rho \vec{g}+\mu \nabla^{2} \vec{v}+\frac{\mu}{3} \nabla(\nabla \cdot \vec{v})
$$

but the pressure and the density changes are unknown $\rightarrow \rho$ and $p$ are also variables

continuity equation and state equation are involved:

$$
\begin{aligned}
& \frac{d \rho}{d t}+\rho \nabla \cdot \vec{v}=0 \\
& \rho=\rho(p, T)
\end{aligned}
$$

$\Rightarrow T$ is also a variable!
equation for the temperature is needed: first law of Thermodynamics

$$
\rho \frac{d u}{d t}=\boldsymbol{\tau}: \mathbf{D}-\nabla \cdot \vec{q}+\rho r
$$

$$
u=u(T, \rho)
$$

Therefore, the dynamical problem involves at least (in general):
6 variables: velocity, pressure, density, temperature:

$$
\vec{v}(\vec{x}, t), p(\vec{x}, t), \rho(\vec{x}, t), T(\vec{x}, t)
$$

## 6 equations:

$$
\left\{\begin{array}{l}
\rho \frac{d \vec{v}}{d t}=-\nabla p+\rho \vec{g}+\mu \nabla^{2} \vec{v}+\frac{\mu}{3} \nabla(\nabla \cdot \vec{v}) \\
\frac{d \rho}{d t}+\rho \nabla \cdot \vec{v}=0 \\
\rho=\rho(p, T) \\
\rho \frac{d u}{d t}=\boldsymbol{\tau}: \mathbf{D}-\nabla \cdot \vec{q}+\rho r \quad u=u(T, \rho)
\end{array}\right.
$$

If viscosity can be neglected:

6. Rotating frames
time derivative in S

$$
\left(\frac{d \vec{u}}{d t}\right)_{S}=\frac{d u_{i}}{d t} \hat{e}_{i}
$$

$$
\begin{gathered}
x_{3} \uparrow \\
\vec{u}=u_{i} \hat{e}_{i}=u_{i}^{\prime} \hat{e}_{i}^{\prime} \Rightarrow\left(\frac{d u}{d t}\right)_{S}=\frac{d u_{i}^{\prime}}{d t} \hat{e}_{i}^{\prime}+u_{i}^{\prime}\left(\frac{d \hat{e}_{i}^{\prime}}{d t}\right)_{S}=\left(\frac{d \vec{u}}{d t}\right)_{S^{\prime}}+u_{i}^{\prime}\left(\frac{d \hat{e}_{i}^{\prime}}{d t}\right)_{S}=\frac{d u_{i}}{d t} \hat{e}_{i} \\
\left(\frac{d \hat{e}_{i}^{\prime}}{d t}\right)_{S}=? \\
\left.\hat{e}_{i}^{\prime}=Q_{j i} \hat{e}_{j} \Rightarrow \frac{d \vec{u}}{d t}\right)_{S^{\prime}}=\frac{d u_{i}^{\prime}}{d t} \hat{e}_{i}^{\prime}
\end{gathered}
$$

$$
\begin{aligned}
& \text { time derivative in } \mathrm{S}^{\prime} \\
& \left(\frac{d \vec{u}}{d t}\right)_{S^{\prime}}=\frac{d u_{i}^{\prime}}{d t} \hat{e}_{i}^{\prime}
\end{aligned}
$$

antisymmetric

Euler equations
$\vec{\Omega}=$ angular velocity of frame $S$, relative to frame $S$

$$
\left(\frac{d \vec{u}}{d t}\right)_{S}=\left(\frac{d \vec{u}}{d t}\right)_{S^{\prime}}+\vec{\Omega} \times \vec{u}
$$




## 7. Bernouilli equation

Simplified expression of the Euler equations (equations of motion for inviscid fluid, i.e., viscosity is neglected)

## Assumptions:

$>$ no viscosity, $\mu=0$
$>$ barotropic flow, i.e., $\rho=\rho(p)$
$>$ body force is gravity (= const.)


Euler equations:

$$
\begin{aligned}
& \frac{\partial \vec{v}}{\partial t}+\vec{v} \cdot \nabla \vec{v}=-\frac{1}{\rho} \nabla p+\vec{g} \\
& >\vec{g}=-\nabla\left(g x_{3}\right) \\
& >\vec{v} \cdot \nabla \vec{v}=\vec{\omega} \times \vec{v}+\nabla\left(\frac{1}{2} v^{2}\right) \quad \text { vector identity already proven } \\
& >F(p) \equiv \int \frac{d p}{\rho(p)} \Rightarrow \nabla F=\frac{d F}{d p} \nabla p=\frac{1}{\rho} \nabla p
\end{aligned}
$$



This equation is specially useful in two particular cases:

1. Steady flow
2. Irrotational flow

## Bernouilli equation for steady flow



The constant may be different for different streamlines
$\vec{\omega}=0 \Rightarrow$ the constant is the same everywhere in the flow

## Bernouilli equation for irrotational flow

Irrotational flow: $\vec{\omega}=\nabla \times \vec{v}=0 \quad \Leftrightarrow \quad \exists \phi \mid \vec{v}=\nabla \phi$
potential flow

$\frac{\partial \phi}{\partial t}+\frac{1}{2} v^{2}+g z+\int \frac{d p}{\rho(p)}=F(t)$

## Application of Bernouilli: Pitot tube

device to measure a flow velocity
$>$ steady flow
$>$ viscosity is neglected
$>\rho=$ const.

$$
\frac{1}{2} v_{1}^{2}+g z_{1}+\frac{p_{1}}{\rho}=\frac{1}{2} v_{2}^{2}+g z_{2}+\frac{p_{2}}{\rho}
$$



$$
\begin{aligned}
& \text { inside the vertical tubes there is hydrostatic balance: } \\
& p_{1}=p_{a t m}+\rho g h_{1}, p_{2}=p_{a t m}+\rho g h_{2}
\end{aligned}
$$

## Application of Bernouilli: orifice in a tank

$>$ orifice small $\Rightarrow$ flow approximately steady
$>$ viscosity is neglected
> $\rho=$ const.

$$
\frac{1}{2} v_{1}^{2}+g z_{1}+\frac{p_{1}}{\rho}=\frac{1}{2} v_{2}^{2}+g z_{2}+\frac{p_{2}}{\rho}
$$



Actually, the function

$$
\Theta=\frac{1}{2} v^{2}+g z+\frac{p}{\rho}
$$

has the same value everywhere since it has the same value at any point on the free surface $\Rightarrow$ flow is irrotational

$$
\begin{aligned}
& \frac{\partial \vec{v}}{\partial t}+\nabla \Theta=\vec{v} \times \vec{\omega} \\
& \vec{v} \times \vec{\omega}=0 \quad \Rightarrow \quad \vec{\omega}=0
\end{aligned}
$$

