

# Cutting edge control by monitoring the tapping torque of new and resharpened tapping tools in Inconel 718

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## Abstract

In aerospace industry, tapping is the prime choice for hole threading in jet aero-engine turbine components. Taps are not usually resharpened, unlike other tools which indeed are resharpened once they have accomplished a certain number of cycles of service. Instead, cutting taps are discarded when they reach a certain number of threaded holes. The waste of cutting tools and the timing for tool replacement is computed in the cost expenses of the end part. In order to reduce manufacturing waste, this work focuses on studying different tapping configurations, to prove the feasibility of resharpening in the interest of extending the tap's service life. Tapping experiments are performed in Inconel 718 under several conditions: (a) resharpened taps, (b) resharpened and recoated taps, (c) an additional 2° chamfer clearance angle taps and (d) as-received taps (baseline). The results show that the three-time resharpened taps increased their lifespan up to 200%. Tap's performance is analyzed considering several criteria: checking by Go/No-Go gauges, evaluating the flaking or notching of cutting edges, measuring axial forces and torques, and analyzing the surface integrity of the threaded holes. Therefore, monitoring the cutting forces and torque during tapping helps us to anticipate to the tool's fracture; and controlling the crest edge roundness progression is crucial for having an accurate thread tolerance and the desired surface integrity.

**Keywords** Inconel 718 · Tapping · Monitoring · Tapping torque · Surface integrity

## 1 Introduction

In the aeronautic industry, the production of threaded holes is a common operation which allows to assembly different components. Besides, tapping is considered a critical operation since it is one of the latest operations to perform in the production chain, which implies a high added value to the aeronautic component. For this reason, the selection of tapping conditions tends to be quite conservative, in order to avoid unacceptable threads or broken taps stuck in the holes, which would involve undesired procedures, such as tap removing

and reparation of surface damages, of a nearly finished component. Additionally, the surface integrity requirements in the aeronautics sector have extreme importance involving really tight tolerances compared to other industry sectors, like automotive or energy. It is a common practice to demand an exhaustive control of the tools which have been used along all the number of cycles of service, and it is not unusual to perform an in situ monitoring of the power consumption of the machine-tool interaction. Power consumption is a reliable indicator of the tool wear for drills, mills, taps, or reamers. Aero-engine materials are known to dispose high strength and work-hardening characteristics and, consequently, a low machinability. This affects the cutting tool efficiency on its service life. One of the most commonly used aeronautics materials is Inconel 718, which is a precipitation hardening nickel-based superalloy. This kind of alloy is used in the hot stages of jet engines and turbines of power plants, because of its high strength and corrosion resistance at extreme temperatures.

During machining operations, tool damage can be attributed either to a high spindle power consumption or thermal effects that originated a gradual failure of the cutting tool. Accordingly, the worn cutting tool must be resharpened to

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58 obtain new cutting edges, which allow us to perform another  
 59 cycle of service. In aerospace industry, tool resharpening pro-  
 60 cess is quite used in tools like drills [1], mills [2, 3], discs [4],  
 61 or reamers [5], but it is not so spread when it comes to cutting  
 62 taps. Cutting taps are discarded after performing several  
 63 rounds of threads, which depend on the diameter, material of  
 64 the workpiece, or even tap's material itself while considering a  
 65 conservative criterion. Unlike for machining, drills, or mills,  
 66 few works can be found involving tapping with resharpened  
 67 tools. One of these works was reported by Armarego et al. [6],  
 68 where tapping capabilities in carbon steels and stainless steels  
 69 were investigated. In addition, many of these cutting tools  
 70 present protective layers of coating to improve their mechan-  
 71 ical properties. In particular, tapping and other cutting tools,  
 72 like drills or mills, have a single or multiple coating layers to  
 73 reduce the friction forces and tool wear [7]. Physical vapor  
 74 deposition (PVD) coatings are widely spread, among them the  
 75 most used in industrial coatings are TiCN, TiN, or TiAlN. For  
 76 example, Reiter et al. [8] investigated the cutting capabilities  
 77 of these protective layers. The authors concluded that the ap-  
 78 plication of TiCN coating had an excellent abrasive and adhe-  
 79 sive wear resistance during blind hole tapping in austenitic  
 80 stainless steel.

81 In this investigation, tapping experiments are performed in  
 82 Inconel 718 with high speed steel (HSS) taps coated with a  
 83 single layer of TiCN. Afterwards, the cutting capabilities are  
 84 compared between as-received taps (reference), resharpened  
 85 taps, modified chamfer relief taps, and resharpened and  
 86 recoated taps. All these taps are right-hand spiral type with a  
 87 spiral point for threading blind holes. They also have a small  
 88 chamfer relief angle to shear off the chip root, rotating them in  
 89 reverse to take them out of the hole, although a greater cham-  
 90 fer relief angle was tested to determine its influence in the  
 91 cutting forces. Therefore, tool wear, cutting forces, tapping  
 92 torque, and surface integrity are analyzed after accomplishing  
 93 from 1 to 8 threaded holes in Inconel 718, in order to compare  
 94 tapping performance for all the aforementioned tap  
 95 configurations.

96 **2 Methodology**

97 **2.1 Experimental setup**

98 A commercial Inconel 718 (50–55 wt% Ni, 17–21 wt% Cr,  
 99 4.75–5.5 wt% Nb, 2.8–3.3 wt% Mo, 0.65–1.15 wt% Ti, 0.2–  
 100 0.8 wt% Al and balance Fe), common in aircraft engines, was  
 101 used in this research. The main mechanical properties of  
 102 Inconel 718 are listed in Table 1. This alloy was characterized  
 103 by great mechanical performance and corrosion resistance,  
 104 which maintain its properties even at high operating tempera-  
 105 tures. It is also widely denominated as a hard-to-cut material,  
 106 due to its low machinability. Figure 1 shows the experimental

**Table 1** Overall mechanical properties and material hardness of tested Inconel 718

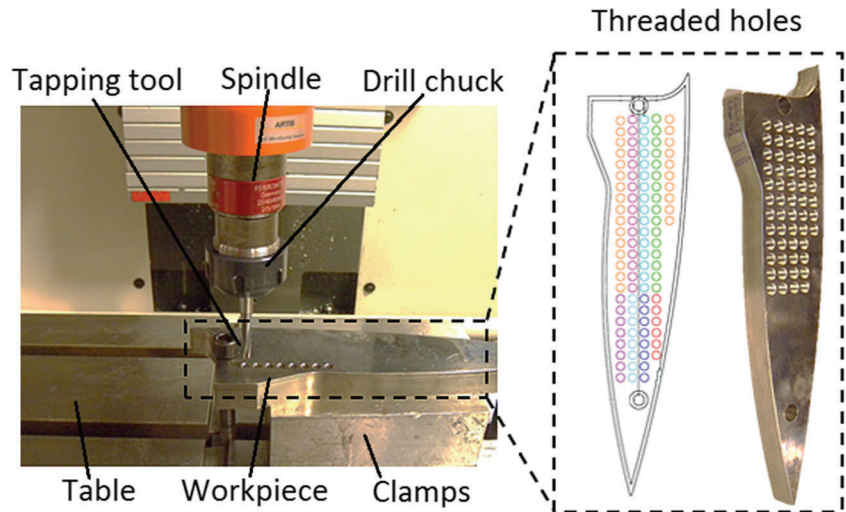
Inconel 718		t1.1
Inconel 718		t1.2
Density	8.19 g/cc	t1.3
Tensile strength (20 °C)	1375 MPa	t1.4
Tensile strength (650 °C)	1100 MPa	t1.5
Young's modulus	200 GPa	t1.6
Elongation (20 °C)	25%	t1.7
Elongation (650 °C)	18%	t1.8
Material hardness	42.25 HRC	t1.9

tests of drilling and tapping that were carried out in a 3-axis  
 moving table vertical machining center, model: Kondia A6.

The strategy which was adopted for making threads consisted of the following steps: drilling, countersinking, re-drilling, and, finally, tapping. In particular, the first drilling operation was performed with tungsten carbide drills up to a depth of cut of 14.2 mm and a diameter of 4.5 mm. External coolant system was used to fill blind holes with crude oil, and, after that, the holes were blown with compressed air to remove the metal debris from drilling. Then, a countersinking operation was performed with a tungsten carbide tool with a diameter of 6.1 mm and a chamfer angle of 120°. Later, a re-drilling operation was carried out with a tungsten carbide end-mill tool with a diameter of 5.1 mm to thread the optimal geometrical dimensions of the holes. A maximum of five holes were done with the end-mill tool to avoid wear. In this case, the fracture failures were reduced, and the thickness of the heat affected layer in the re-drilling was lower than in the first drilling operation, because of a lower friction at the material-tool interfaces. Consequently, this smaller friction coefficient affected the cutting force, tapping torque, and the surface integrity, as described in the result section. Finally, the last step in the process was the tapping operation with a plug tap reference EG UNF 10–32, Tolerance 3B, and a diameter of 0.2306 in (ref. ID 148008 at DC SWISS SA). The material of the tap contains 8% of cobalt, which enhances such mechanical properties, as strength, toughness, and wear resistance. Different tap configurations were tested in this superalloy, particularly geometrical modifications, coating layer, and resharpening operations were modified to analyze their impact on tapping. Table 2 exhibits the main characteristics of the tested taps and the number of tap configurations used. Table 3 presents the operational conditions for the threading process.

The cutting performance of the resharpened taps were evaluated by taking into account several aspects: checking of the threaded holes using the Go/No-Go gauge, recording the cutting forces, counting the number of broken cutting edges as a consequence of threading, and observing the surface integrity. Additionally, the following considerations were taken into

**Fig. 1** Tapping setup in a 3-axis machining center, hole strategy, and the final workpiece



146 account. The threaded holes were tested with the Go/No-Go  
 147 gauge, in order to assure that they were within the tolerance  
 148 range given by the ASME B1.1–2003 standard [9]. The tap-  
 149 ping axial force and torque were registered to study its evolu-  
 150 tion with respect to the number of threads deployed, as well as,  
 151 to compare the cutting performance of new and resharpened  
 152 taps. To do this, an Artis DDU4 system was used to record the  
 153 data. This system consisted in a DDU-Rotor, DDU-Stator, and  
 154 a post-processing software to handle the acquired data. The  
 155 communication between rotor and fixed stator was free of  
 156 contact (inductively) with a gap of 5 mm between them.  
 157 Finally, the wear progression of the cutting edges was moni-  
 158 tored through continuous observation of the surface profile,  
 159 notches, and material adhesion. Also, surface integrity was  
 160 analyzed to determine the level of damage in the different  
 161 cutting tool configurations with an optical surface profiler  
 162 (Alicona, InfiniteFocus).

t2.1 **Table 2** Main characteristics of the tapping tools and number of units tested

t2.2	Taps characteristic and number of taps used	
t2.3	Diameter of the tap	5.86 mm
t2.4	Type of tap	Plug tap
t2.5	Type of thread	UNF-3B
t2.6	Tap's material	HSS-E-PM
t2.7	Type of coating	TiCN
t2.8	Type of class	3B
t2.9	N° of as-received taps tested	2
t2.10	N° of once resharpened taps tested	14
t2.11	N° of three times resharpened taps tested	4
t2.12	N° of resharpened and recoated taps tested	1
t2.13	N° of additional 2 chamfer clearance angle taps tested	2
t2.14	Total of taps tested	23
t2.15	N° threaded holes performed per tap	8

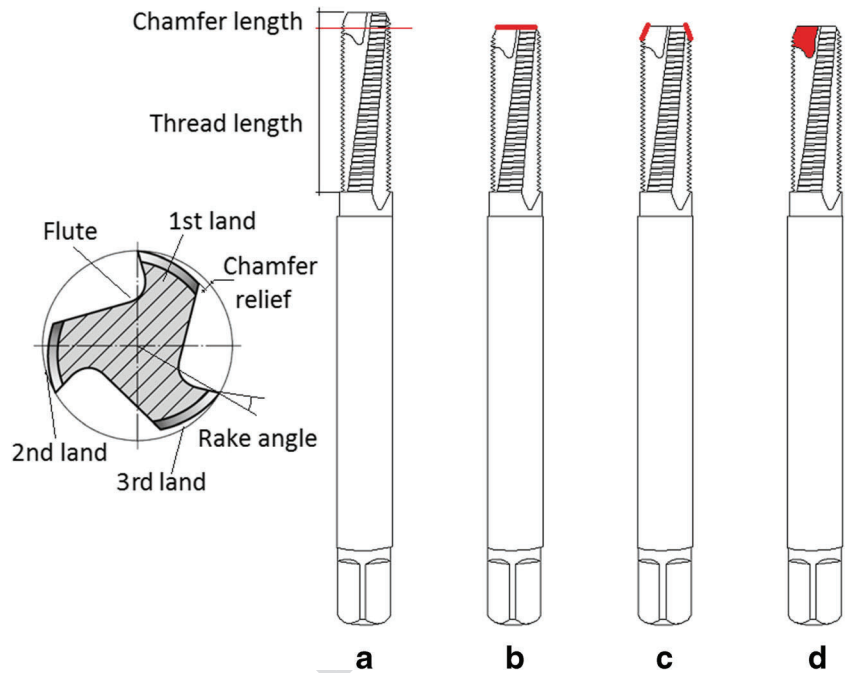
**2.2 Resharpening process**

The completion of resharpening depends on the employee's  
 experience, where the outer peak of the cutting edges and the  
 nominal flute diameter play a key role to perform an accurate  
 tapping operation. These cutting edges of the tap are respon-  
 sible for removing material, where the rake angle, chamfer  
 length, and relief angle, as well as the number of lands are  
 important parameters to take into account during tapping [6].  
 In this work, C spiral-flute taps with three cutting lands in-  
 volved in thread formation were investigated, as shown in  
 Fig. 2a. The tool wear was the main reason to carry out a  
 resharpened process of a worn cutting tap. The steps to  
 resharpening a worn tool are the following ones: the first step  
 consisted in cutting the initial part of the chamfer length, as  
 shown in Fig. 2b. The second step consisted in making a tap  
 chamfer regeneration, defining a longer and sharper chamfer  
 of the thread, as described in Fig. 2c. In this way, the tap can  
 gradually penetrate into the material [10]. This operation was  
 also very important to define the right chamfer clearance angle  
 of the tap that depends on the material to cut [11]. The chamfer  
 angle helps to shear off the chips when threading a hole, and  
 the chamfer regeneration was achieved with a helical grinding

**Table 3** Operational parameters used during tapping

Operational parameters		
Forming speed [m/min]	1.41	
Rotation speed [rpm]	77	
Feed rate [mm/rev]	0.794	
Feed speed [mm/min]	61	
Tool holder	Softsynchro #3 semirigid	
Collet type	ER32	
Clamping system	Cone ISO40	

**Fig. 2** Tap characteristics and different steps in a resharpening process: (a) as-received tap, (b) tap cut in length, (c) chamfer regeneration, (d) spiral point formation



185 operation, as described by Liao et al. [12]. Besides,  
 186 Karpuschewski et al. [13] described a method for searching  
 187 the wheel position in helical flute grinding. Finally, the third  
 188 step consisted in the cutting edge reinforcement operation, see  
 189 Fig. 2d, which was a grinding operation in the rake face, also  
 190 called spiral point. This resharpening operation brings two  
 191 aspects associated: (1) the tap exhibited a higher negative rake  
 192 angle which makes the cutting edges stronger; (2) the rein-  
 193 forcement operation helped to finishing the cutting edges by  
 194 creating new cutting edges which maximize the chamfer  
 195 length of the tap.

196 In order to confirm that the resharpening operation was cor-  
 197 rectly performed, a Leica DCM 3D laser scanning confocal mi-  
 198 croscopy was used. This confocal microscopy allowed to record  
 199 3D topographies and also 2D profiles to analyze tool's charac-  
 200 teristics in detail. In this sense, the as-received tap geometry was  
 201 compared with respect to resharpened taps. Afterwards, surface  
 202 topographies and profiles were acquired to study specific geo-  
 203 metrical parameters of the different tested tap configurations. In  
 204 particular, some of these analyzed parameters were rake angle,  
 205 chamfer relief angle, radius of tool tip, and chamfer length of the  
 206 tap. The measurements of the cutting edge angles were within the  
 207 range of 85°–89° for both as-received and resharpened taps. The  
 208 radius of the tool tip was around 20–22 μm, which according to  
 209 tool selection of the suppliers are appropriate for tapping with this  
 210 type of tapping tool.

211 **2.3 Recoating process**

212 Some tapping tools were resharpened to extend their tool life  
 213 and save expenses of tool replacement. Consequently, these

214 tools had also been recoated with the original coating treat-  
 215 ment that they had in their as-received state. The aim was to  
 216 study the influence of the coating when comparing the tool  
 217 wear with respect to similar uncoated tools of this protective  
 218 layer. Then, a commercial TiCN coating treatment was de-  
 219 ployed by using the PVD method. Confocal microscopy was  
 220 used to ensure a reduced and homogeneous concentration of  
 221 droplets, an edge radius of around 20–25 μm, and a layer of  
 222 coating thickness of about 4 μm. TiCN coating was chosen  
 223 because of being the original coating that these tools had in  
 224 their as-received state. This kind of coated had being widely  
 225 used in machining operations to enhance the cutting tool life  
 226 [14].

227 **2.4 Test stop criterion**

228 The test stop criterion and, subsequently, tool replacement was  
 229 defined for three different circumstances. The first stop crite-  
 230 rion is defined by the quality of the thread, which is the most  
 231 common stop criterion in industry. This quality is tested by  
 232 using Go/No-Go gauges. The second stop criterion was due to  
 233 high values of tapping torque, particularly higher values than  
 234 12 Nm. Accordingly, high values of torque are associated with  
 235 a high level of tool wear; breakages or notching in several  
 236 edges of the tap; or the catastrophic failure that sometimes  
 237 could happen, due to the high demands in the machining of  
 238 Inconel 718. For these worn cutting edges with low cutting  
 239 capabilities, undesired findings during tapping can be found,  
 240 such as unremoved metal debris and notches. Finally, the third  
 241 stop criterion is to analyze the wear degradation of the tapping  
 242 tool in a microscope. In particular, notches of about 0.25 mm

243 in a cutting edge of the tap are considered as a stop criterion  
 244 for replacing the tapping tool.

245 **3 Results**

246 **3.1 Tool wear**

247 The evaluation of the cutting performance of the different tap  
 248 configurations presents some difficulties, because of the low  
 249 number of threaded holes that each tap can perform in this  
 250 superalloy before their failure. During the tapping operation  
 251 in this hard-to-cut alloy, it is common to have broken or flaked  
 252 edges, which sometimes even happens during the first  
 253 threaded hole. Determining the wear evolution of the cutting  
 254 edges is a complex task to assess, where notches and worn  
 255 edges are limiting factors in the service life and cutting capa-  
 256 bilities of these cutting tools. Consequently, surface topogra-  
 257 phies and profiles of the different tap configurations are repet-  
 258 itively measured to analyze the wear evolution and surface  
 259 damages. Figure 3 exhibits the cutting edges topographies  
 260 and the profiles of the tool tip radius for an as-received tap  
 261 and a tap resharpener once.

262 The surface topographies in both taps present similar geo-  
 263 metrical dimensions, although the angle between the relief  
 264 face and the rake angle changes a few degrees depending on  
 265 the tap configurations. The resharpener taps present a sharper  
 266 cutting edge compared with the as-received ones, as expected

from the removal material process of tool resharpener. 267  
 Additionally, the tool tip radius is also different between both 268  
 taps. The as-received tap exhibits a radius of 20.5°, while the 269  
 radius of the resharpener tap is about 25.5°. These few de- 270  
 grees will make a great difference during tapping, due to a 271  
 higher contact area between tool-workpiece interfaces. A big- 272  
 ger geometrical difference is observed when comparing these 273  
 two taps with a worn tap after its service life. The radius value 274  
 of the worn tool tip is noticeably blunt than the as-received 275  
 tap. Besides, notches and material adhesion are commonly 276  
 found at the worn cutting edges. These higher values of tool 277  
 tip radius and flaked cutting edges have been proved to affect 278  
 the thrust force and torque during tapping [15, 16]. Figure 4 279  
 shows an as-received tool at the end of its service life with a 280  
 tool tip radius of 41° and a notch in the first cylinder teeth after 281  
 the chamfer clearance. Note that the tip radius is twice the 282  
 value of the radius of an as-received tap. This higher value 283  
 of tool tip radius increases the contact area between the inter- 284  
 faces of the tapping tool and the material to cut. As a conse- 285  
 quent, higher friction coefficient is expected in tapping. This 286  
 higher friction coefficient will bring higher values of cutting 287  
 force and tapping torque, greater wear of the cutting tool, and 288  
 poorer quality of the thread of the final workpiece. 289

290 **3.2 Cutting force and tapping torque**

In this section, axial forces and the tapping torques are ana- 291  
 lyzed for the different number of threaded holes performed in 292

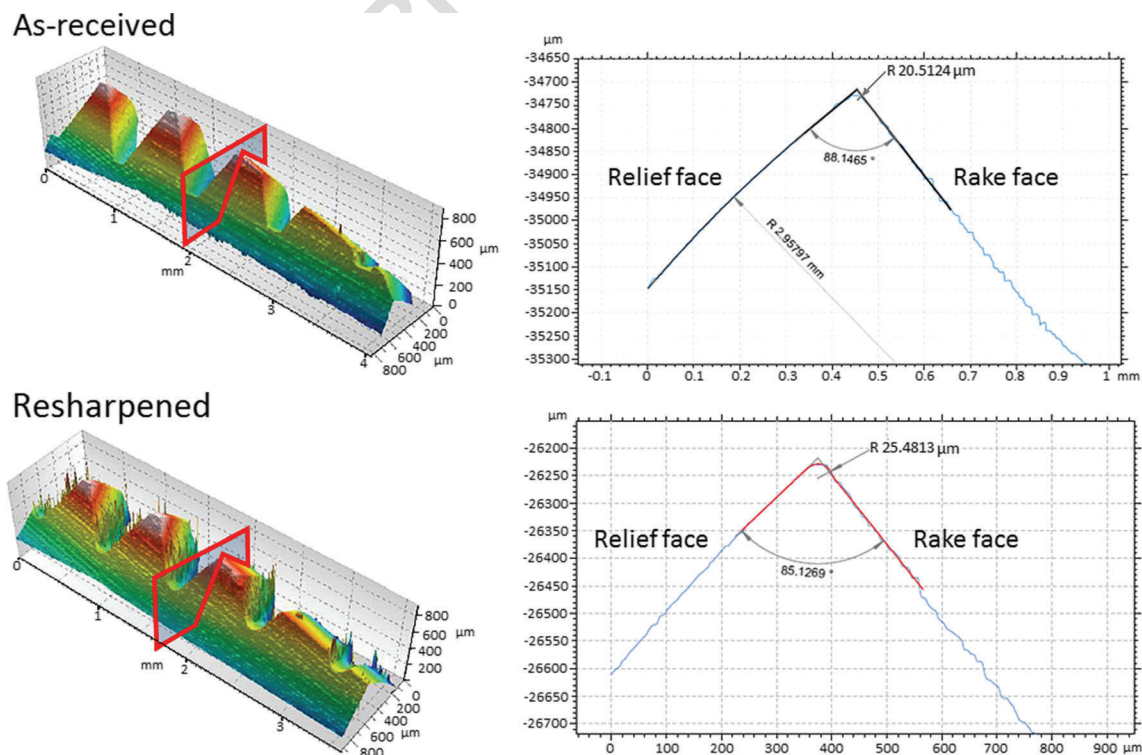


Fig. 3 Topographies of the cutting edges and tool tip radius characteristics for an as-received tap and a tap resharpener once

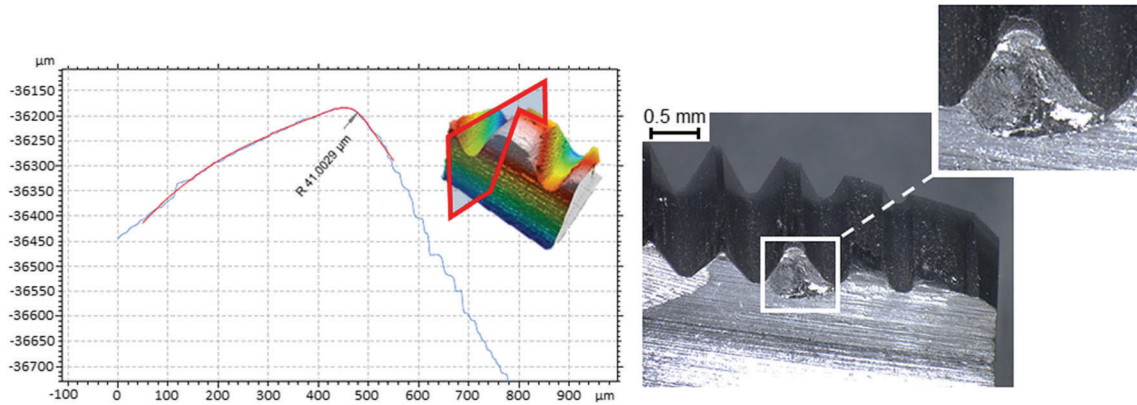


Fig. 4 Typical value of a worn cutting edge and a notch commonly found in these cutting tools at the end of its service life

293 Inconel 718. The magnitudes of the axial force and torque are  
 294 measured to describe the cutting capabilities of each tap con-  
 295 figurations. Thus, Fig. 5 exhibits the axial force and torque of  
 296 the eight threads performed in Inconel 718 with an as-received  
 297 tap, the same tap resharpened once and resharpened three  
 298 times.

299 Firstly, it is denoted that the axial force increases with  
 300 the number of threads. The 1st thread exhibits the lowest  
 301 cutting forces, and the 8th thread needs a higher cutting  
 302 force for tapping, independently of the tap configurations.  
 303 The maximum values of axial force in 8th thread were  
 304 about -350 N, -300 N, and -220 N for an as-received  
 305 tap, a tap resharpened once, and a tap resharpened 3 times,  
 306 respectively. Regarding the tapping torque, the maximum  
 307 values of torque 8th thread were about 11.6 Nm, 12 Nm,  
 308 and 10 Nm for an as-received tap, a resharpened tap, and a  
 309 3 times resharpened tap, respectively. Subsequently, the

resharpened cutting tools decrease the required axial force  
 and torque in tapping, which seems to be associated with  
 the lower angle between the relief and rake faces. This  
 lower angle decreases the contact area between tool and  
 workpiece and, consequently, promotes a decrease of friction  
 between the interfaces of the tap and the material to be  
 threaded. Figure 6 shows the required tapping torque ver-  
 sus the number of threads performed in Inconel 718 with  
 an as-received tap, a tap resharpened once, a tap  
 resharpened and recoated, and a tap with 2° extra of cham-  
 fer clearance angle. In this sense, wear capabilities of these  
 tap configurations are analyzed according to the number of  
 threads achieved following the same procedure. The average  
 value of torque per thread is calculated with the area  
 under the curves divided by the total time taken to tap the  
 hole, rotate the tap in reverse to take them out of the hole,  
 and clean the metal shavings from the threaded hole. Each

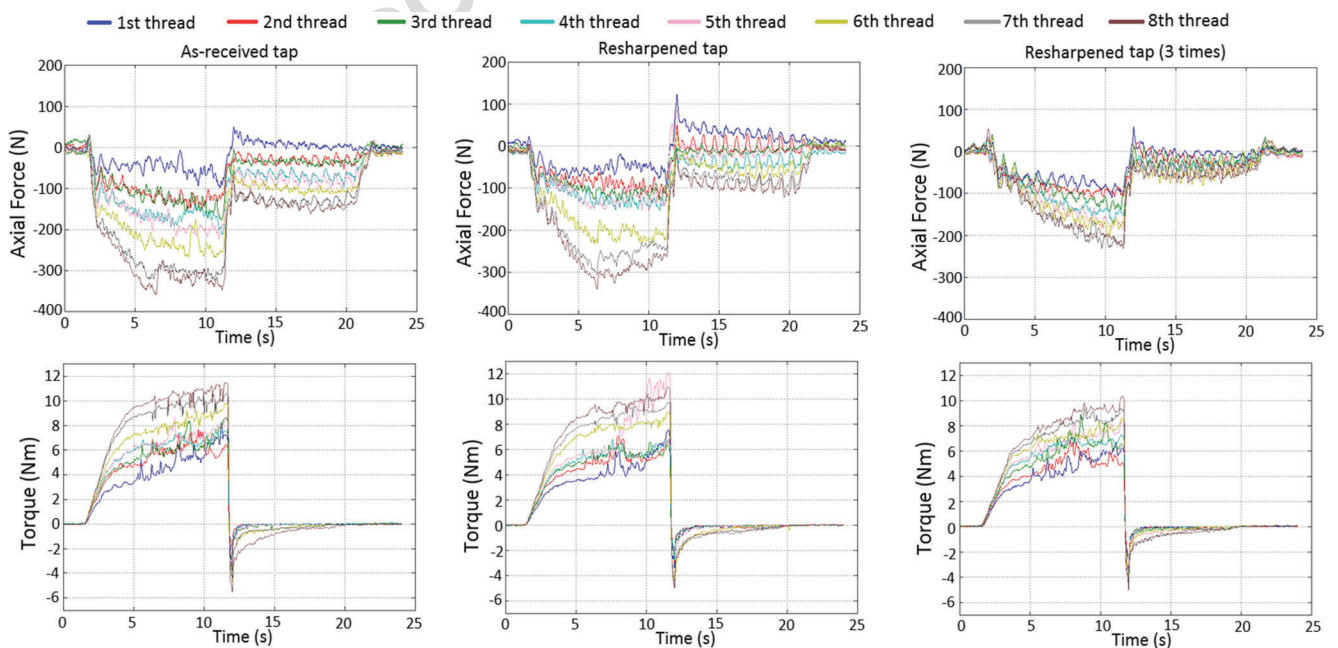
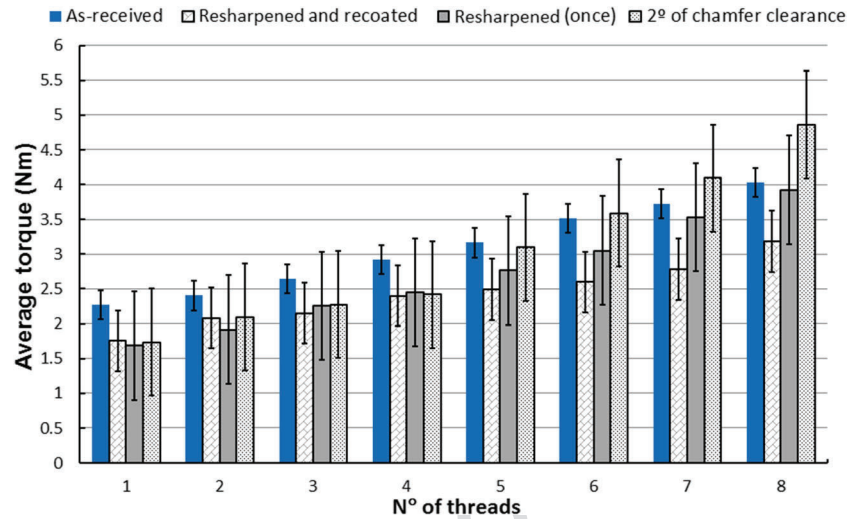


Fig. 5 Axial force and tapping torque of the different threads performed in Inconel 718 with an as-received tap, a tap resharpened once, and a tap resharpened three times

**Fig. 6** Evolution of the average torque for the tested tap configurations as a function of the number of threaded holes. Note that the average torque is represented by the average of three measurements and the error bars correspond to a standard deviation



327 experimental value is the average of three measurements  
 328 and the error bars correspond to the standard deviation.  
 329 The tapping torque tends to increase with the number of  
 330 threads, due to the wear of the tool that affects the cutting  
 331 capabilities of the taps [17]. Accordingly, the as-received  
 332 taps show the highest torque values of 2.2 Nm and 3.2 Nm  
 333 in average for the 1st and the 5th thread, respectively. For  
 334 the last threads, the cutting tool with 2° extra of chamfer  
 335 clearance angle exhibits the highest torque values of  
 336 3.6 Nm and 4.8 Nm in average for the 6th and the 8th  
 337 thread, respectively. On the contrary, the tap resharpended  
 338 once and recoated exhibits the lowest values of torque of  
 339 about 1.7 Nm, 2.5 Nm, and 3.2 Nm in average for the 1st,  
 340 the 5th, and the 8th thread, respectively. Regarding the  
 341 resharpended taps with and without recoating, they present  
 342 similar torque values than the tap with 2° extra of chamfer  
 343 clearance angle up to the 5th threaded hole. Similar results  
 344 were reported by Lorenz et al. [18], where a reduction of  
 345 torque is found for 7/16–20 UNF taps with a greater cham-  
 346 fer relief. Oezkaya and Biermann [19] also compared nu-  
 347 merically and experimentally uncoated and coated tapping  
 348 tools. They found lower average torque values for the coat-  
 349 ed tools independently of the metric thread. When compar-  
 350 ing the results from the 5th to the 8th threaded holes, no-  
 351 ticeable torque differences are found between the  
 352 resharpended taps with respect to the resharpended and  
 353 recoated taps. In particular, lower torque values are ob-  
 354 served for the recoated taps, due to the excellent wear ca-  
 355 pabilities of TiCN which helps to enhance the service life  
 356 by reducing tool wear. Finally, the worst scenario is found  
 357 for the tap with 2° extra of chamfer clearance with values  
 358 of torque of 4.8 Nm, when performing the 8th threaded  
 359 hole. This high value of torque is likely attributed to the  
 360 higher surface degradation observed in the first cylinder  
 361 teeth after the chamfer clearance modification, as can be  
 362 seen in the surface integrity subsection.

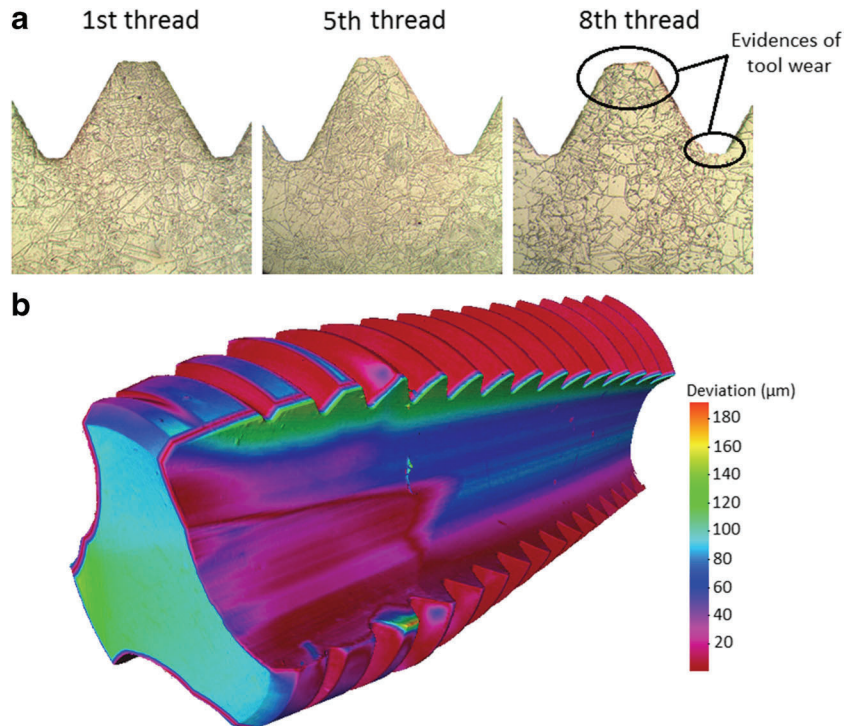
**3.3 Surface integrity**

363

364 The surface analysis of the different tested taps is carried out  
 365 after performing different number of threaded holes (from one  
 366 to eight threads). For this purpose, the threaded samples were  
 367 cut from the end part and encapsulated in Bakelite. After that,  
 368 they were grinded with a series of grits in a rotating wheel,  
 369 polished by using several cloths with diamond suspensions  
 370 spray up to 1 μm and etched with Kalling’s reagent n° 2  
 371 (5 g CuCl<sub>2</sub>, 100 ml HCl, 100 ml ethanol). Then, the material’s  
 372 microstructure was analyzed as recommended by the ASTM  
 373 standard E407 [20]. Following this procedure, the edges of the  
 374 cutting tools are observed with the aim to describe surface  
 375 damage, material adhesion, microcracks, laminations’ effects,  
 376 and worn of the tool tip radius. These undesired findings are  
 377 aggravated right at the end of the chamfer length, which de-  
 378 fines the finished size of the desired thread. All these surface  
 379 defects will affect the quality and lifespan of these tapping  
 380 tools. Sometimes microcracks, material adhesion and lamina-  
 381 tion can be found in the first threaded hole, even for threads  
 382 generated with a brand new cutting tap. Figure 7a shows the  
 383 surface integrity of the specimens threaded with an as-  
 384 received tap in the 1st, 5th, and 8th threaded hole. Additionally,  
 385 Fig. 7b exhibits the surface deviation after performing the 8th  
 386 threaded hole compared with respect to the as-received tap.  
 387 An evident worn progression of 180 μm is denoted at the tool  
 388 crest after performing the 8th threaded hole.

390 No noticeable surface differences are found at the tool  
 391 crests and valleys of the taps between the 1st and 5th thread.  
 392 Similar results were found when comparing equivalent  
 393 threads for as-received and resharpended taps. Evidences of  
 394 worn of tool tip radius or the threaded material are denoted  
 395 in the 8th threaded hole. In particular, flatted edges with higher  
 396 radius are found, but not micro cracks, material adhesion, or  
 397 deformed grains. Higher tool tip degradation and cutting

**Fig. 7** (a) Surface integrity of a tapping hole with an as-received tap in the 1st, 5th, and 8th threaded hole. (b) Worn progression tool tip radius and surface damage at the valley after performing the 8th tapping hole with respect to the as-received tap

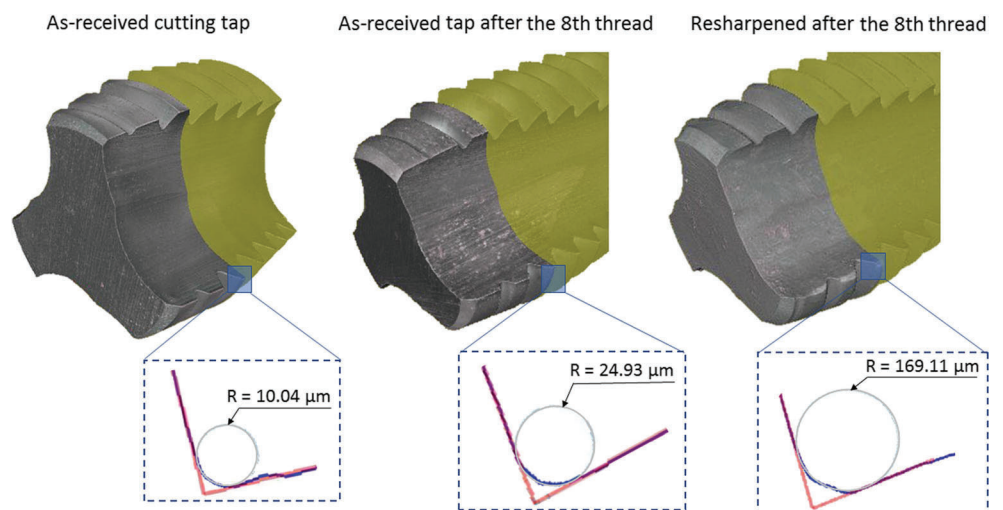


398 forces were found in the crest of the first full thread of the  
 399 cutting tool, right after the chamfer length, which is referred to  
 400 a maximum load distribution across the cutting teeth and chip  
 401 formation. Similar results were described by Axinte et al. [21]  
 402 when investigating the surface integrity of different polished  
 403 methods in Ti6AL4V. Figure 8 shows the profiles of the cutting  
 404 edges for an as-received tap, an as-received tap in the 8th  
 405 thread, and a resharpened tap without coating in the 8th  
 406 thread. The profile measurement is performed at the first full  
 407 thread, where higher tool degradation is observed. Resulting  
 408 that the radius of the tool tip for an as-received tap is around  
 409 2.5 time sharper than the same tool after performing 8th  
 410 threaded holes. Also, the radius of the tool tip for an as-

received tap is around 7 times sharper than a resharpened tool,  
 both after achieving the 8th thread. Note that the resharpening  
 process is performed to the worn cutting edge to achieve similar  
 tool tip radius and rake angle than the as-received tap (see  
 Fig. 3). In this sense, it is expected to weaken the cutting  
 edges due to the rake angle decrease with the resharpening  
 process.

Despite of the roundness differences at the tool tip radius,  
 all the threads were correctly assessed without deformed  
 grains, cracks, or material adhesion in the threaded holes,  
 independently of the tap configurations. Regarding the surface  
 deviation of the cutting tool when performing the 8th threaded  
 hole, crest edge smoothness evidences up to 25 µm are denoted  
 right at the end of the chamber length, particularly at the

**Fig. 8** 2D profiles of the cutting edge for an as-received tap, an as-received tap after the 8th thread, and a resharpened tap after the 8th thread





424 first full-thread teeth. In this first full thread is expected to bear  
 425 the higher cutting force [22]. The resharpened and uncoated  
 426 taps showed higher radius of the tool tip after the tapping  
 427 process, which is associated to higher cutting force deviation  
 428 and worse quality of the surface integrity. Furthermore, the  
 429 surface deviation in average is found to be 180 μm along the  
 430 tap's flute (see Fig. 7b), due to the interaction between the  
 431 cutting tool with the metal shavings. Similar values of wear  
 432 of the as-received tap are found for resharpened and recoated  
 433 taps. This is because recoating the helical flute facilitate firstly  
 434 to remove the metal chips and secondly to reach the lubricant  
 435 the cutting teeth. Therefore, resharpening and recoating taps  
 436 have shown the best machining performance in tapping  
 437 Inconel 718, because of reducing adhesion which was the  
 438 main wear mechanism. Hence, resharpening and recoating  
 439 taps could be the right choice for improving the tapping oper-  
 440 ation and saving expenses of tool replacement and  
 441 manufacturing downtime.

442 **4 Conclusions**

443 In this research, tapping capabilities are explored through  
 444 monitoring different tap configurations, according to the num-  
 445 ber of threaded holes in a common aeronautical alloy.  
 446 Therefore, the main conclusions can be summarized as the  
 447 following:

- 448 • The cutting performance of as-received, resharpened,  
 449 recoated, and modified chamfer clearance taps are studied  
 450 for threading holes in Inconel 718. Accordingly,  
 451 resharpened taps have been denoted to be an alternative  
 452 way instead of using new taps in order to achieve suitable  
 453 threads and to save costs. Particularly, lower axial force,  
 454 tapping torque, and wear rate are examined when the  
 455 resharpened taps are also recoated compared to the as-  
 456 received taps.
- 457 • Monitoring the cutting forces and torque during the tap-  
 458 ping process help us to detect the failure of the tap and,  
 459 consequently, avoid the extraction a broken tap from a  
 460 high added value end part. In the 1st threaded hole, the  
 461 as-received tap exhibited the highest value of torque, com-  
 462 pared to the rest of tap configurations. On the contrary, the  
 463 tap with a chamfer clearance modification exhibited the  
 464 highest value of torque for the 8th threaded hole. The  
 465 resharpened and recoated tap showed the lowest values  
 466 of torque for the 1st and the 8th threaded hole.  
 467 Consequently, resharpening is an effective operation in  
 468 spite of extending the service life of the tapping tools.
- 469 • Tap configuration and coating layer play a key role in  
 470 metal cutting of this nickel-based superalloy with low ma-  
 471 chinability. Greater tool degradation is denoted in the first  
 472 threaded hole, where metal chipping, notches, or worn

cutting edges are examined independently of the tap con- 473  
 figuration. Therefore, monitoring the tapping torque and 474  
 the edge roundness progression is crucial in order to have 475  
 an accurate thread tolerance, achieve the desired surface 476  
 integrity, and prevent tool fracture. 477

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