

```

!module that defines the input variables without specifying the size, for using them all
the code along
module geometric_inputs

    integer,save::n_point, n_blades
    real, allocatable, save::geom(:, :)
    real :: hub_to_br, hub_height, tilt_angle
!validated
end module geometric_inputs
!-----
!module with the characteristics positions of the horseshoes and the control points
module horseshoe_locations
    integer,save::n_horseshoe
    real (kind=8),allocatable,save::horseshoe_loc(:, :),horseshoe_te_loc(:, :),
control_points_loc(:, :),area_distr(:),chord_distr(:),twist(:),thickness(:)
    real (kind=8),allocatable,save::hs_loc_pol(:, :),cp_loc_pol(:, :),
    real (kind=8),allocatable,save::temp_hs_loc(:, :, :),temp_cp_loc(:, :, :),temp_hs_te_loc(:,
:,:), cp_saved_pos(:, :, :),
    real (kind=8),allocatable,save::el_vect1(:, :, :),el_vect2(:, :, :),el_vect3(:, :, :)

end module horseshoe_locations
!-----
!module with the variables related to the time simulation
module clock
    integer,save::n_time_steps, time_step
    real,save::inc_time,time_sample
    real, allocatable, save:: time(:)

end module clock
!-----
!module containing some operating parameters of the turbine and external conditions
module operating_parameters
    real,save::omega, v0(3), density, upflow_angle, yaw_misalignment, wind_speed_module,
exp_a

end module operating_parameters
!-----
!module for the aerodynamic variables and coefficients
module aerodynamic_performance
    real,save::azimut,pitch
    real, save:: preazimut
    real (kind=8),allocatable,save::alpha(:, :),Cl(:, :),Cd(:, :),Cm(:, :),
    real, allocatable, save:: L(:, :),D(:, :),T(:),Q(:),P(:),Mz(:, :),Cp(:),Ct(:),MB_ip(:, :),
MB_op(:, :)

end module aerodynamic_performance
!-----
!Module dedicated to the aerodynamic coefficients of the introduced airfoils
module airfoil_polars
    integer,save:: n_data_points, n_airfoils
    real (kind=8),allocatable,save:: Cl_pol(:, :),Cd_pol(:, :),Cm_pol(:, :),
    real(kind=8), allocatable,save:: AOA(:, :),Cl_pol_aux(:, :),Cd_pol_aux(:, :),Cm_pol_aux(:,
:)
    real,allocatable,save:: airfoils_thickness(:)

end module airfoil_polars
!-----
!
module incident_speeds
    real (kind=8),allocatable,save:: wake_ind(:, :, :),wind(:, :, :),turning_ind(:, :),blade_ind
(:, :, :),total_wind(:, :, :),blade_ind2(:, :, :),
!test
    real (kind=8),allocatable,save:: old_wake_ind(:, :),test_blade_ind(:, :),wake_ind2(:, :, :),
wake_ind3(:, :, :),

end module incident_speeds
!-----
module vortex_intensities
    real (kind=8),allocatable,save:: pr_attached_hs_int(:, :), pr_wake_int(:, :),

end module vortex_intensities
!-----
module wake_sheet
    integer,save:: n_wake_lines !the total number of nodes in the wake sheet will be

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    n_wake_lines*n_horseshoe
    real (kind=8), save:: wake_x_spacing
    real (kind=8), allocatable, save:: pr_wake_nodes_loc(:,:,:)
    integer, allocatable, save:: wake_nodes_connec(:,:)
    real (kind=8), allocatable, save:: wake_segments_int_s(:,:), wake_segments_int_c(:,:)

end module wake_sheet
!-----
module tf_matrix
    real (kind=8), allocatable, save:: attached_s_matrix_u(:,:), attached_s_matrix_v(:,:),
    attached_s_matrix_w(:,:), attached_c_matrix_u(:,:), attached_c_matrix_v(:,:),
    attached_c_matrix_w(:,:)
    real (kind=8), allocatable, save:: wake_s_matrix_u(:,:,:), wake_s_matrix_v(:,:,:),
    wake_s_matrix_w(:,:,:), wake_c_matrix_u(:,:,:), wake_c_matrix_v(:,:,:), wake_c_matrix_w(:,
    :, :)
end module tf_matrix
!-----
module iteration_data
    real (kind=8), allocatable, save:: aux_attached_hs_int(:), pr_cycle_att_hs_int(:,:)
end module iteration_data
!-----
module simulation_data
    integer, save:: wake_int_init, wake_loc_init, b_wake_roll_up, b_allow_wakeevol,
    b_stationary_sim
    integer, allocatable, save:: blade_phase(:,:)
end module simulation_data
!-----
module lambda_variables
    real, allocatable:: lambda(:), cp_lambda(:), ct_lambda(:)
end module lambda_variables
!-----
module power_curve_variables
    real, allocatable:: pc_P(:), pc_T(:), pc_omega(:), pc_wspeedm(:)
    integer n_pc_steps
end module power_curve_variables
!-----
module file_names
    character(100) bladeplanform, airfoilpolars, inputfile
end module file_names
!-----
program hub

    use file_names
    implicit none
    real t1
    INTEGER :: count1, count_rate, count_max, count2
    integer n,m
    count_rate=1
    CALL SYSTEM_CLOCK(count1, count_rate, count_max)

    print*, "Please, write the name of the inputs file"
    read(*,*) inputfile
    print*, inputfile, "OK, simulation will begin"
    open(UNIT=7, FILE=inputfile, ACCESS='SEQUENTIAL')
    read(7,*) n
    read(7,*) bladeplanform
    read(7,*) airfoilpolars
    close(7)

    select case(n)
    case(1)
        call single_point_simulation
    case(2)
        call full_cycle_simulation
    case(3)
        call find_optimal_lambda
    case(4)
        call power_curve
    end select

    CALL SYSTEM_CLOCK(count2, count_rate, count_max)
    t1=(count2-count1)/count_rate
    open(UNIT=7, FILE='Results_Simulation_time.txt', ACCESS='SEQUENTIAL')

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        write(7,*) t1
    close(7)
    print*,"Simulation took ",t1, " seconds. Results are ready."

    pause

end program
!-----
!third version of the subroutine that calculates the induced speed at a certain point of
the 3D space by a vortex line segment defined by two points
subroutine VLtfnc(x,y,z,x1,y1,z1,x2,y2,z2,gam,u,v,w)
    implicit none
    real (kind=8),intent(in):: x,y,z,x1,y1,z1,x2,y2,z2,gam
    real (kind=8),intent(out):: u,v,w
    real (kind=8):: r1(3),r2(3),r0(3),magr1,magr2,magr0
    real (kind=8):: aux_cross(3),aux_dot,aux_sp(3)
    real,parameter:: dist=0.025, pi=4*atan(1.)

    r1(1)=x-x1
    r1(2)=y-y1
    r1(3)=z-z1
    r2(1)=x-x2
    r2(2)=y-y2
    r2(3)=z-z2
    r0(1)=x2-x1
    r0(2)=y2-y1
    r0(3)=z2-z1

    magr1=sqrt(dot_product(r1,r1))
    magr2=sqrt(dot_product(r2,r2))
    magr0=sqrt(dot_product(r0,r0))
    aux_dot=dot_product(r1,r2)
    call crossprod(r1,r2,aux_cross)

    aux_sp=(gam/(4*pi))*((magr1+magr2)*aux_cross)/(magr1*magr2*(magr1*magr2+aux_dot)+(dist
    *magr0)**2)

    u=aux_sp(1)
    v=aux_sp(2)
    w=aux_sp(3)

!created 12/01/2013
!this transfer function has been taken from the ECN report
!validated
end subroutine VLtfnc
!-----
--
!Alternative version, based on the theory presented in Low Speed Aerodynamics
subroutine VLtfnc_LSA(x,y,z,x1,y1,z1,x2,y2,z2,gam,u,v,w)
    !Low Speed Aerodynamics--> LSA
    implicit none
    real (kind=8),intent(in):: x,y,z,x1,y1,z1,x2,y2,z2,gam
    real (kind=8),intent(out):: u,v,w
    real (kind=8):: r1(3),r2(3),r0(3),magr1,magr2,magr0,magrx
    real (kind=8):: aux_cross(3),aux_dot,aux_sp(3),aux_dot1,aux_dot2
    real,parameter:: tol=0.0000003, pi=4*atan(1.)

    r1(1)=x-x1
    r1(2)=y-y1
    r1(3)=z-z1
    r2(1)=x-x2
    r2(2)=y-y2
    r2(3)=z-z2
    r0(1)=x2-x1
    r0(2)=y2-y1
    r0(3)=z2-z1

    magr1=sqrt(dot_product(r1,r1))
    magr2=sqrt(dot_product(r2,r2))
    magr0=sqrt(dot_product(r0,r0))
    call crossprod(r1,r2,aux_cross)
    aux_dot=dot_product(aux_cross,r0)
    aux_dot1=dot_product(r0,r1)
    aux_dot2=dot_product(r0,r2)

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magrx=dot_product(aux_cross,aux_cross)

if ((magr1.LT.tol).OR.(magr2.LT.tol).OR.(magrx.LT.tol)) then
  u=0.d0
  v=0.d0
  w=0.d0
else
  aux_sp=(gam/(4*pi))*aux_cross*((aux_dot1/magr1)-(aux_dot2/magr2))/magrx
  u=aux_sp(1)
  v=aux_sp(2)
  w=aux_sp(3)
end if

!created 20/04/2013
!this transfer function has been taken from Low Speed Aerodynamics book
!validated, not used
!Possible alternative
end subroutine VLtfnc_LSA
!-----
!subroutine that calculates the cross-product of two given vectors
subroutine crossprod(a,b,c)

  implicit none
  real (kind=8), intent(in)::a(3),b(3)
  real (kind=8),intent(out)::c(3)

  c(1)=a(2)*b(3)-a(3)*b(2)
  c(2)=a(3)*b(1)-a(1)*b(3)
  c(3)=a(1)*b(2)-a(2)*b(1)
!validated 3/11/2012
end subroutine crossprod
!-----
!subroutine that reads the input data (wing planform)
subroutine read_input()
  use geometric_inputs
  use airfoil_polars
  use horseshoe_locations
  use file_names
  implicit none
  integer i,j,max_n_data
  integer,allocatable ::n_data_input(:)

  real, parameter:: pi=4*atan(1.)
  real,allocatable::aux(:,:)
  real, allocatable:: aux_alpha(:,:),aux_cl(:,:),aux_cd(:,:),aux_cm(:,:)
  open(UNIT=5,FILE=bladeplanform,ACCESS='SEQUENTIAL')
  read(5,*) n_point
  allocate(geom(5,n_point))
  do i=1,n_point
    read(5,*) geom(:,i)
  end do
  close(5)
  !if twist has been introduced in degrees uncomment next line
  geom(4,:)=geom(4,:)*pi/180.
  open(UNIT=5,FILE=airfoilpolars,ACCESS='SEQUENTIAL')

  read(5,*) n_airfoils
  allocate(n_data_input(n_airfoils))
  max_n_data=1
  do i=1,n_airfoils
    read(5,*) n_data_input(i)
    if (n_data_input(i).GT.n_data_input(max_n_data)) then
      max_n_data=i
    end if
  end do

  allocate(airfoils_thickness(n_airfoils))
  allocate(aux_alpha(n_airfoils,n_data_input(max_n_data)))
  allocate(aux_cl(n_airfoils,n_data_input(max_n_data)))
  allocate(aux_cd(n_airfoils,n_data_input(max_n_data)))
  allocate(aux_cm(n_airfoils,n_data_input(max_n_data)))

  do i=1,n_airfoils

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    read(5,*) airfoils_thickness(i)
    do j=1,n_data_input(i)
        read(5,*) aux_alpha(i,j),aux_cl(i,j),aux_cd(i,j),aux_cm(i,j)
        !activate the next line if the angle of attack has been introduced in degrees
        aux_alpha(i,j)=pi*aux_alpha(i,j)/180
    end do
end do
close(5)

n_data_points=n_data_input(max_n_data)
allocate(AOA(n_airfoils,n_data_points))
allocate(Cl_pol_aux(n_airfoils,n_data_points))
allocate(Cd_pol_aux(n_airfoils,n_data_points))
allocate(Cm_pol_aux(n_airfoils,n_data_points))
allocate(Cl_pol(n_horseshoe,n_data_points))
allocate(Cd_pol(n_horseshoe,n_data_points))
allocate(Cm_pol(n_horseshoe,n_data_points))
allocate(aux(4,n_data_points))

do i=1,n_airfoils
    AOA(i,:)=aux_alpha(max_n_data,:)
    do j=1,n_data_points
        call spline(aux_alpha(i,1:n_data_input(i)),aux_cl(i,1:n_data_input(i)),AOA(i,
j),Cl_pol_aux(i,j),n_data_input(i),1)
        call spline(aux_alpha(i,1:n_data_input(i)),aux_cd(i,1:n_data_input(i)),AOA(i,
j),Cd_pol_aux(i,j),n_data_input(i),1)
        call spline(aux_alpha(i,1:n_data_input(i)),aux_cm(i,1:n_data_input(i)),AOA(i,
j),Cm_pol_aux(i,j),n_data_input(i),1)
    end do

end do

!validated 3/11/2012
!modified 13/04/2013
end subroutine read_input
!-----
!subroutine for calculating the components of the Hermite polynomial
subroutine hermite_poly(chi,poly)
    implicit none
    real, intent(in)::chi
    real, intent(out)::poly(4)
    poly(1)=(1+chi+chi)*(chi-1)*(chi-1)
    poly(2)=chi*(chi-1)*(chi-1)
    poly(3)=chi*chi*(3-2*chi)
    poly(4)=(chi-1)*chi*chi
!validated
end subroutine
!-----
!subroutine for calculating the first derivate values of a discrete defined function at
selected points
subroutine derivate(x,f,der,n)

    implicit none
    integer,intent(in)::n
    real,intent(in)::x(n),f(n)
    real (kind=8),intent(out)::der(n)
    real der_aux(n-1),aux1,aux2
    integer i

    do i=1,n-1
        der_aux(i)=(f(i+1)-f(i))/(x(i+1)-x(i))
    end do
    der(1)=der_aux(1)
    der(n)=der_aux(n-1)

    do i=2,n-1
        aux1=(x(i)+x(i-1))/2
        aux2=(x(i+1)+x(i))/2
        der(i)=der_aux(i-1)+(der_aux(i)-der_aux(i-1))*(x(i)-aux1)/(aux2-aux1)
    end do
!validated 3/11/2012
end subroutine derivate
!-----

```

!subroutine that interpolates the value of a discretely defined function, 2 points given
subroutine interpolate(x1,f1,deriv,x2,f2)

```

implicit none
real,intent(in)::x1(2),f1(2)
real (kind=8),intent(in)::x2,deriv(2)
real (kind=8),intent(out)::f2
real chi,poly(4)
chi=(x2-x1(1))/(x1(2)-x1(1))
call hermite_poly(chi,poly)
f2=poly(1)*f1(1)+poly(3)*f1(2)+(poly(2)*deriv(1)+poly(4)*deriv(2))*(x1(2)-x1(1))
!validated 3/11/2012
end subroutine interpolate
!-----

```

!subroutine for obtaining the values of a discretely defined function in all the points
 required
subroutine spline(x1,f1,x2,f2,n1,n2)

```

implicit none
integer,intent(in)::n1,n2
real,intent(in)::x1(n1),f1(n1)
real (kind=8),intent(in)::x2(n2)
real (kind=8),intent(out)::f2(n2)
integer i,j,point
real(kind=8) der(n1)

!first it is necessary to check if the inputs are correct
do i=1,n1-1
  if(x1(i).GT.x1(i+1)) then
    print*,'The points where the function is defined are not consecutive'
    print*, x1(i),x1(i+1)
    pause
    stop
  end if
end do
do i=1,n2-1
  if(x2(i).GT.x2(i+1)) then
    print*,'The points where the function is wanted to be defined are not
consecutive'
    pause
    stop
  end if
end do

if((x2(1).LT.x1(1)).OR.(x2(n2).GT.x1(n1))) then
  print*,'The new points are out of the defined range of points'
  print*,x2(1),x1(1),x1(n1),x2(n2)
  pause
  stop
end if
call derivate(x1,f1,der,n1)

point=1
do i=1,n2
  do j=point,n1-1
    if(x1(j+1).GT.x2(i)) exit
  end do
  point=j
  call interpolate(x1(j:j+1),f1(j:j+1),der(j:j+1),x2(i),f2(i))
end do
!validated 3/11/2012
end subroutine spline
!-----

```

subroutine place_elements

```

use horseshoe_locations
use geometric_inputs

```

implicit none

```

real (kind=8),parameter:: pi=4.*atan(1.)
real a,b,aux(n_horseshoe+1),auxd1,auxd2
real (kind=8):: aux_v1(n_horseshoe+1),aux_v2(n_horseshoe+1)
integer i,j

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```

integer, parameter:: k=40 !number of samples taken for calculating the area of each
strip
real (kind=8)::aux_area(k),aux_points(3,k)
auxd1=0.2
auxd2=0.2
a=minval(geom(1,:))
b=maxval(geom(1,:))
do i=1,n_horseshoe+1
    aux(i)=auxd1+(pi-auxd2-auxd1)/(n_horseshoe)*(i-1)
end do
do i=1,n_horseshoe+1
    !cosinus distribution of the points
    horseshoe_loc(1,i)=a+(cos(auxd1)-cos(aux(i)))/(cos(auxd1)-cos(pi-auxd2))*(b-a)
    !linear distribution of the points
    !horseshoe_loc(1,i)=a+(b-a)*(i-1)/(n_horseshoe)
end do
!the horseshoe nodes have been located following a cosine distribution in the spanwise
direction
horseshoe_te_loc(1,:)=horseshoe_loc(1,:)

!cosine distribution for the control points location in spanwise direction
do i=1,n_horseshoe
    aux(i)=auxd1+(pi-auxd2-auxd1)/(n_horseshoe)*(i-0.5)
    control_points_loc(1,i)=a+(cos(auxd1)-cos(aux(i)))/(cos(auxd1)-cos(pi-auxd2))*(b-
a)
    !control_points_loc(1,i)=(horseshoe_loc(1,i)+horseshoe_loc(1,i+1))/2.
end do

do i=1,n_horseshoe
    call spline(geom(1,:),geom(4,:),control_points_loc(1,i),twist(i),n_point,1)
    call spline(geom(1,:),geom(5,:),control_points_loc(1,i),thickness(i),n_point,1)
end do

do i=2,n_horseshoe
    call spline(geom(1,:),geom(2,:),horseshoe_loc(1,i),aux_v1(i),n_point,1)
    horseshoe_te_loc(2,i)=aux_v1(i)
    call spline(geom(1,:),geom(3,:),horseshoe_loc(1,i),aux_v2(i),n_point,1)
    horseshoe_loc(2,i)=aux_v2(i)+(aux_v1(i)-aux_v2(i))*0.25

end do
horseshoe_loc(1,1)=a
horseshoe_loc(1,n_horseshoe+1)=b
horseshoe_te_loc(1,1)=a
horseshoe_te_loc(1,n_horseshoe+1)=b
horseshoe_te_loc(2,1)=geom(2,1)
horseshoe_te_loc(2,n_horseshoe+1)=geom(2,n_point)
horseshoe_loc(2,1)=geom(3,1)+(horseshoe_te_loc(2,1)-geom(3,1))*0.25
horseshoe_loc(2,n_horseshoe+1)=geom(3,n_point)+(horseshoe_te_loc(2,n_horseshoe+1)-geom
(3,n_point))*0.25
!the horseshoe nodes have been located at the 25% of the local chord
do i=1,n_horseshoe
    call spline(geom(1,:),geom(2,:),control_points_loc(1,i),aux_v1(i),n_point,1)
    call spline(geom(1,:),geom(3,:),control_points_loc(1,i),aux_v2(i),n_point,1)
    control_points_loc(2,i)=aux_v2(i)+(aux_v1(i)-aux_v2(i))*0.25
    call spline(geom(1,:),geom(4,:),control_points_loc(1,i),twist(i),n_point,1)
    call spline(geom(1,:),geom(5,:),control_points_loc(1,i),thickness(i),n_point,1)

    chord_distr(i)=aux_v1(i)-aux_v2(i)
end do
!the control points have been located at the 25% of the local chord
!new way for calculating the area distribution
area_distr(:)=0.d0
do j=1,n_horseshoe
    a=horseshoe_loc(1,j)
    b=horseshoe_loc(1,j+1)
    do i=1,k
        aux_points(1,i)=a+(b-a)*(i-1)/(k-1)
        call spline(geom(1,:),geom(2,:),aux_points(1,i),aux_points(2,i),n_point,1)
        call spline(geom(1,:),geom(3,:),aux_points(1,i),aux_points(3,i),n_point,1)
    end do
    do i=1,k-1
        aux_area(i)=(aux_points(2,i)+aux_points(2,i+1)-aux_points(3,i)-aux_points(3,i+1)

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1))*(aux_points(1,i+1)-aux_points(1,i))/2.
    area_distr(j)=area_distr(j)+aux_area(i)
end do

end do

horseshoe_loc(1,:)=horseshoe_loc(1,:)+hub_to_br
horseshoe_te_loc(1,:)=horseshoe_te_loc(1,:)+hub_to_br
control_points_loc(1,:)=control_points_loc(1,:)+hub_to_br

!created 02/01/2013
!validated 02/01/2013
end subroutine place_elements
!-----
subroutine from_cartesian_to_polars
use horseshoe_locations
implicit none

real(kind=8),parameter:: pi=4.*atan(1.)
real aux1,aux2
integer i
allocate(hs_loc_pol(2,n_horseshoe+1))
allocate(cp_loc_pol(2,n_horseshoe))
do i=1,n_horseshoe
    aux1=sqrt((control_points_loc(1,i))**2)
    aux2=sqrt((control_points_loc(2,i))**2)
    cp_loc_pol(1,i)=sqrt(aux1**2+aux2**2)
    cp_loc_pol(2,i)=atan(aux2/aux1)
    if ((control_points_loc(1,i)).GE.0.) then
        if (control_points_loc(2,i).GE.0.) then
            cp_loc_pol(2,i)=cp_loc_pol(2,i)
        else
            cp_loc_pol(2,i)=2*pi-cp_loc_pol(2,i)
        end if
    else
        if (control_points_loc(2,i).GE.0.) then
            cp_loc_pol(2,i)=pi-cp_loc_pol(2,i)
        else
            cp_loc_pol(2,i)=cp_loc_pol(2,i)+pi
        end if
    end if

    aux1=sqrt((horseshoe_loc(1,i))**2)
    aux2=sqrt((horseshoe_loc(2,i))**2)
    hs_loc_pol(1,i)=sqrt(aux1**2+aux2**2)
    hs_loc_pol(2,i)=atan(aux2/aux1)
    if (horseshoe_loc(1,i).GE.0.) then
        if (horseshoe_loc(2,i).GE.0.) then
            hs_loc_pol(2,i)=hs_loc_pol(2,i)
        else
            hs_loc_pol(2,i)=2*pi-hs_loc_pol(2,i)
        end if
    else
        if (horseshoe_loc(2,i).GE.0.) then
            hs_loc_pol(2,i)=pi-hs_loc_pol(2,i)
        else
            hs_loc_pol(2,i)=hs_loc_pol(2,i)+pi
        end if
    end if

end do
aux1=sqrt((horseshoe_loc(1,n_horseshoe+1))**2)
aux2=sqrt((horseshoe_loc(2,n_horseshoe+1))**2)
hs_loc_pol(1,n_horseshoe+1)=sqrt(aux1**2+aux2**2)
hs_loc_pol(2,n_horseshoe+1)=atan(aux2/aux1)

if ((horseshoe_loc(1,n_horseshoe+1))/aux1.GE.0.) then
    if (horseshoe_loc(2,n_horseshoe+1).GE.0.) then
        hs_loc_pol(2,n_horseshoe+1)=hs_loc_pol(2,n_horseshoe+1)
    else
        hs_loc_pol(2,n_horseshoe+1)=hs_loc_pol(2,n_horseshoe+1)+3.*pi/2.
    end if
else
    if (horseshoe_loc(2,n_horseshoe+1).GE.0.) then

```



```

        hs_loc_pol(2,n_horseshoe+1)=hs_loc_pol(2,n_horseshoe+1)+pi/2.
    else
        hs_loc_pol(2,n_horseshoe+1)=hs_loc_pol(2,n_horseshoe+1)+pi
    end if
end if

!created 03/11/2012
!validated 03/11/2012 currently it is not used
end subroutine from_cartesian_to_polars
!-----
subroutine temp_location

    use horseshoe_locations
    use aerodynamic_performance
    use clock
    use geometric_inputs
    implicit none

    integer i,b
    real(kind=8),parameter::pi=4.*atan(1.)
    real mat_az(3,3),mat_pitch(3,3),mat_c(3,3),aux3(3),aux(3),aux_a, aux_mod
    real(kind=8):: aux_c(3),aux_s(3)

    do b=1,n_blades
        !for b=1-->the variable called azimuth is the corresponding azimuth of the first
        !blade, the other ones are evenly spaced
        aux_a=azimuth+(b-1)*2*pi/n_blades

        mat_az(1,1)=1.
        mat_az(2,1)=0.
        mat_az(3,1)=0.
        mat_az(1,2)=0.
        mat_az(2,2)=-sin(aux_a)
        mat_az(3,2)=cos(aux_a)
        mat_az(1,3)=0.
        mat_az(2,3)=-cos(aux_a)
        mat_az(3,3)=-sin(aux_a)

        mat_pitch(2,1)=0.
        mat_pitch(1,2)=0.
        mat_pitch(2,2)=1.
        mat_pitch(3,2)=0.
        mat_pitch(2,3)=0.

        mat_c(1,1)=1.
        mat_c(2,1)=0.
        mat_c(3,1)=0.
        mat_c(1,2)=0.
        mat_c(2,2)=1.
        mat_c(3,2)=0.
        mat_c(1,3)=0.
        mat_c(2,3)=0.
        mat_c(3,3)=-1.

    do i=1,n_horseshoe

        mat_pitch(1,1)=cos(pitch+twist(i))
        mat_pitch(3,1)=sin(pitch+twist(i))
        mat_pitch(1,3)=-sin(pitch+twist(i))
        mat_pitch(3,3)=cos(pitch+twist(i))
        aux3(1)=0.
        aux3(2)=control_points_loc(1,i)
        aux3(3)=control_points_loc(2,i)
        aux3=matmul(mat_c,aux3)
        aux3=matmul(mat_pitch,aux3)
        temp_cp_loc(b,:,i)=matmul(mat_az,aux3)
        aux3(1)=0.
        aux3(2)=horseshoe_loc(1,i)
        aux3(3)=horseshoe_loc(2,i)
        aux3=matmul(mat_c,aux3)
        aux3=matmul(mat_pitch,aux3)
        temp_hs_loc(b,:,i)=matmul(mat_az,aux3)

```

```

    aux3(1)=0.
    aux3(2)=horseshoe_te_loc(1,i)
    aux3(3)=horseshoe_te_loc(2,i)
    aux3=matmul(mat_c,aux3)
    aux3=matmul(mat_pitch,aux3)
    temp_hs_te_loc(b,:,i)=matmul(mat_az,aux3)

end do
aux3(1)=0.
aux3(2)=horseshoe_loc(1,n_horseshoe+1)
aux3(3)=horseshoe_loc(2,n_horseshoe+1)
aux3=matmul(mat_c,aux3)
aux3=matmul(mat_pitch,aux3)
temp_hs_loc(b,:,n_horseshoe+1)=matmul(mat_az,aux3)
aux3(1)=0.
aux3(2)=horseshoe_te_loc(1,n_horseshoe+1)
aux3(3)=horseshoe_te_loc(2,n_horseshoe+1)
aux3=matmul(mat_c,aux3)
aux3=matmul(mat_pitch,aux3)
temp_hs_te_loc(b,:,n_horseshoe+1)=matmul(mat_az,aux3)
end do !loop for b=1,n_blades--->for calculating the positions for all blades

do i=1,n_horseshoe
    el_vect1(time_step,1,i)=temp_hs_te_loc(1,1,i)+temp_hs_te_loc(1,1,i+1)-(temp_hs_loc(
(1,1,i)+temp_hs_loc(1,1,i+1))
    el_vect1(time_step,2,i)=temp_hs_te_loc(1,2,i)+temp_hs_te_loc(1,2,i+1)-(temp_hs_loc(
(1,2,i)+temp_hs_loc(1,2,i+1))
    el_vect1(time_step,3,i)=temp_hs_te_loc(1,3,i)+temp_hs_te_loc(1,3,i+1)-(temp_hs_loc(
(1,3,i)+temp_hs_loc(1,3,i+1))
    aux_mod=sqrt(el_vect1(time_step,1,i)**2+el_vect1(time_step,2,i)**2+el_vect1
(time_step,3,i)**2)
    el_vect1(time_step,:,i)=el_vect1(time_step,:,i)/aux_mod

    aux_c(1)=temp_hs_loc(1,1,i+1)-temp_hs_loc(1,1,i)
    aux_c(2)=temp_hs_loc(1,2,i+1)-temp_hs_loc(1,2,i)
    aux_c(3)=temp_hs_loc(1,3,i+1)-temp_hs_loc(1,3,i)

    call crossprod(el_vect1(time_step,:,i),aux_c,aux_s)

    aux_mod=sqrt(aux_s(1)**2+aux_s(2)**2+aux_s(3)**2)
    el_vect3(time_step,:,i)=aux_s(:)/aux_mod

    call crossprod(el_vect3(time_step,:,i),el_vect1(time_step,:,i),el_vect2(time_step,
:,i))
end do

!created 08/11/2012
!validated 09/11/2012
!modified 12/04/2013 el_vect1,2,3 definition included
end subroutine temp_location
!-----
!subroutine for calculating the induced speed on control points by blade rotation
subroutine movement_ind_speed

    use horseshoe_locations
    use operating_parameters
    use incident_speeds
    use aerodynamic_performance
    implicit none

    real vind_pol(2,n_horseshoe)
    integer i

    do i=1,n_horseshoe
        vind_pol(1,i)=omega*cp_loc_pol(1,i)*sin(cp_loc_pol(2,i))
        vind_pol(2,i)=omega*cp_loc_pol(1,i)*cos(cp_loc_pol(2,i))

        turning_ind(1,i)=vind_pol(1,i)*sin(azimut)+vind_pol(2,i)*cos(azimut)
        turning_ind(2,i)=-vind_pol(1,i)*cos(azimut)+vind_pol(2,i)*sin(azimut)
    end do
!validated

```

```

end subroutine movement_ind_speed
!-----
subroutine wind_speed
  use operating_parameters
  use horseshoe_locations
  use clock
  use incident_speeds
  use geometric_inputs
  implicit none

  integer i
  real,parameter:: pi=4*atan(1.)

  do i=1,n_horseshoe

    wind(time_step,1,i)=v0(1)*((temp_cp_loc(1,3,i)+hub_height)/hub_height)**exp_a
    wind(time_step,2,i)=v0(2)*((temp_cp_loc(1,3,i)+hub_height)/hub_height)**exp_a
    wind(time_step,3,i)=v0(3)*((temp_cp_loc(1,3,i)+hub_height)/hub_height)**exp_a

  end do

!validated 4/11/2012
!modified 17/11/2012 (dimensions of temp_hs_loc increased)
!validated if not time dependant 07/12/12
end subroutine wind_speed
!-----
subroutine wind_speed_for_wake(locations,speeds)
  use operating_parameters
  use horseshoe_locations
  use wake_sheet
  use clock
  use simulation_data
  use geometric_inputs
  implicit none

  integer i
  real (kind=8), intent(in):: locations(3,n_horseshoe+1)
  real (kind=8), intent(out):: speeds(3,n_horseshoe+1)

  do i=1,n_horseshoe+1

    speeds(1,i)=v0(1)*((locations(3,i)+hub_height)/hub_height)**exp_a
    speeds(2,i)=v0(2)*((locations(3,i)+hub_height)/hub_height)**exp_a
    speeds(3,i)=v0(3)*((locations(3,i)+hub_height)/hub_height)**exp_a

  end do

!created 17/11/2012
!validated 07/12/12
end subroutine wind_speed_for_wake
!-----
!subroutine that add the incident speeds on each cp at every loop iteration or time step
subroutine cp_incident_speeds

  use incident_speeds
  use aerodynamic_performance
  use horseshoe_locations
  use clock
  implicit none

  integer i,j1,j2
  real aux3(3), aux_alpha(n_horseshoe)

  do i=1,n_horseshoe
    aux3(1:3)=0.

    total_wind(time_step,1,i)=wind(time_step,1,i)+wake_ind(time_step,1,i)+blade_ind
(time_step,1,i)
    total_wind(time_step,2,i)=wind(time_step,2,i)+turning_ind(1,i)+wake_ind(time_step,
2,i)+blade_ind(time_step,2,i)
    total_wind(time_step,3,i)=wind(time_step,3,i)+turning_ind(2,i)+wake_ind(time_step,
3,i)+blade_ind(time_step,3,i)

    total_wind(time_step,4,i)=sqrt(dot_product(total_wind(time_step,1:3,i),total_wind

```

```

(time_step,1:3,i))) !all 3 components of the speed has been considered for incident
speed module calculation
end do
call check
do i=1,n_horseshoe
  alpha(time_step,i)=atan(dot_product(total_wind(time_step,1:3,i),el_vect3(time_step
, :,i))/dot_product(total_wind(time_step,1:3,i),el_vect1(time_step, :,i)))
end do

do i=1,n_horseshoe
  j1=max(1,i-2)
  j2=min(n_horseshoe,i+2)
  aux_alpha(i)=sum(alpha(time_step,j1:j2))/(j2-j1+1)
end do

!created 01/11/2012
!modified 25/11/2012
!validated 13/01/2013
end subroutine cp_incident_speeds
!-----
!subroutine just for giving the Cl of each strip for a given angle of attack
subroutine vortex_intensity_calculator !print lines to be removed

  use horseshoe_locations
  use aerodynamic_performance ! alpha, cl, cd cm tau, L D Q P
  use incident_speeds ! total_wind
  use airfoil_polars
  use vortex_intensities
  use iteration_data
  use clock
  implicit none
  integer i
  real (kind=8):: aux1, aux2(3),dl(3),relax,aux
  real (kind=8):: auxdot1,auxdot3,auxcros(3),denaux1,denaux3

  relax=0.1

  do i=1,n_horseshoe
    call spline(real(AOA(1, :)),real(Cl_pol(i, :)),alpha(time_step,i),Cl(time_step,i),
n_data_points,1)
  end do

  open(UNIT=6,FILE='Test_aux3_VorIntCal.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
  do i=1,n_horseshoe
    aux_attached_hs_int(i)=pr_attached_hs_int(time_step,i)

    dl(1)=temp_hs_loc(1,1,i+1)-temp_hs_loc(1,1,i)
    dl(2)=temp_hs_loc(1,2,i+1)-temp_hs_loc(1,2,i)
    dl(3)=temp_hs_loc(1,3,i+1)-temp_hs_loc(1,3,i)

    auxdot1=dot_product(total_wind(time_step,1:3,i),el_vect1(time_step, :,i))
    auxdot3=dot_product(total_wind(time_step,1:3,i),el_vect3(time_step, :,i))

    call crossprod(total_wind(time_step,1:3,i),dl,auxcros)

    denaux1=dot_product(auxcros,el_vect1(time_step, :,i))
    denaux3=dot_product(auxcros,el_vect3(time_step, :,i))

    pr_attached_hs_int(time_step,i)=(1.-relax)*pr_attached_hs_int(time_step,i)+relax*
(0.5*Cl(time_step,i)*(auxdot1**2+auxdot3**2)*area_distr(i)/sqrt(denaux1**2+denaux3**
2))

    call crossprod(total_wind(time_step,1:3,i),dl,aux2)

    aux1=sqrt(dot_product(aux2,aux2))

    write(6,*) pr_attached_hs_int(time_step,i)

  end do
  write(6,*) 'End of file'
  close (6)
!created 15/11/12
!validated 13/01/13

```

```

end subroutine vortex_intensity_calculator
!-----
!subroutine only with validation purposes, it will save some variables for test the code
!in different conditions before its release
subroutine check

  use horseshoe_locations
  use operating_parameters
  use incident_speeds
  use vortex_intensities
  use wake_sheet
  use aerodynamic_performance
  use tf_matrix
  use clock
  use simulation_data
  use airfoil_polars
  implicit none

  integer i,j,b_vortex_strength,b_wake_pos,b_hs_pos, b_cp_pos,b_AOA_dist,b_wake_int,
  b_tf_mat,b_blade_ind_speed,b_wake_ind_speed,b_cp_inc_speeds,b_cl,b_wind_speed,
  b_area_dist,b_chord_dist,b_hs_cp_2D_pos, b_wake_convec,b_sec_pol
  real (kind=8)::aux
  b_cp_pos=0
  b_hs_pos=0
  b_wake_pos=0
  b_vortex_strength=0
  b_AOA_dist=0
  b_wake_int=0
  b_tf_mat=0
  b_blade_ind_speed=0
  b_wake_ind_speed=0
  b_cp_inc_speeds=0
  b_wake_convec=0
  b_cl=0
  b_wind_speed=0
  b_area_dist=0
  b_chord_dist=0
  b_hs_cp_2D_pos=0
  b_sec_pol=0

  if (b_cp_pos.NE.0) then
    open(UNIT=6,FILE='Control_points_pos.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    write(6,*) 'Number of control points:', n_horseshoe
    write(6,*) 'Control points locations'
    write(6,*) 'x y z'
    do i=1,n_horseshoe
      write(6,*) temp_cp_loc(:, :,i)
    end do
    write(6,*) 'End of file'
    close (6)
  end if
  if (b_hs_pos.NE.0) then
    open(UNIT=6,FILE='Horseshoe_nodes_pos.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    write(6,*) 'Number of horseshoe nodes:', n_horseshoe
    write(6,*) 'Horseshoe locations'
    write(6,*) 'x y z'
    do i=1,n_horseshoe+1
      write(6,*) temp_hs_loc(1, :,i)
    end do
    write(6,*) 'End of file'
    close (6)
  end if
  if (b_vortex_strength.NE.0) then
    open(UNIT=6,FILE='Attached_vortex_strength_distribution.txt',ACCESS='SEQUENTIAL',
STATUS='REPLACE')
    write(6,*) 'Number of segments:', n_horseshoe
    write(6,*) 'Segments intensity'
    do j=1,n_time_steps
      do i=1,n_horseshoe
        write(6,*) pr_attached_hs_int(j,i)
      end do
    end do
    write(6,*) 'End of file'
    close (6)
  end if

```

```

end if
if (b_AOA_dist.NE.0) then
  open(UNIT=6,FILE='Angle_of_attack_distribution.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
  write(6,*) 'Number of control points:', n_horseshoe
  write(6,*) 'Angle of attack at each control point [rad]'
  do j=1,n_time_steps
    do i=1,n_horseshoe
      write(6,*) alpha(j,i)
    end do
  end do
  write(6,*) 'End of file'
  close (6)
end if
if (b_wake_pos.NE.0) then
  open(UNIT=6,FILE='Wake_nodes_pos.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
  write(6,*) 'Total number of nodes:', n_horseshoe*n_wake_lines
  write(6,*) 'Number of lines of nodes', n_wake_lines
  write(6,*) 'x y z'
  do i=1,(n_horseshoe+1)*n_wake_lines
    !write(6,*) pr_wake_nodes_loc(blade_phase(1,time_step),:,j+(i-1)*
(n_horseshoe+1))
    write(6,*) pr_wake_nodes_loc(1,:,i)
  end do
  write(6,*) 'End of file'
  close (6)
end if
if (b_wake_int.NE.0) then
  open(UNIT=6,FILE='Wake_vortex_int.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
  write(6,*) 'Total number of shed vortex:', n_horseshoe
  write(6,*) 'Intensities'
  do i=1,n_horseshoe*(n_wake_lines-1)
    !write(6,*) pr_wake_int(blade_phase(1,time_step),i)
    write(6,*) pr_wake_int(:,i)
  end do
  write(6,*) 'End of file'
  close (6)
end if
if (b_tf_mat.NE.0) then
  open(UNIT=6,FILE='Attached_s_matrix_u.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
  do i=1,n_horseshoe
    do j=1,n_horseshoe
      write(6,*) attached_s_matrix_u(i,j)
    end do
  end do
  write(6,*) 'End of document'
  close (6)
  open(UNIT=6,FILE='Attached_s_matrix_v.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
  do i=1,n_horseshoe
    do j=1,n_horseshoe
      write(6,*) attached_s_matrix_v(i,j)
    end do
  end do
  write(6,*) 'End of document'
  close (6)
  open(UNIT=6,FILE='Attached_s_matrix_w.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
  do i=1,n_horseshoe
    do j=1,n_horseshoe
      write(6,*) attached_s_matrix_w(i,j)
    end do
  end do
  write(6,*) 'End of document'
  close (6)
  open(UNIT=6,FILE='Attached_c_matrix_u.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
  do i=1,n_horseshoe
    do j=1,n_horseshoe+1
      write(6,*) attached_c_matrix_u(i,j)
    end do
  end do
  write(6,*) 'End of document'
  close (6)
  open(UNIT=6,FILE='Attached_c_matrix_v.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
  do i=1,n_horseshoe
    do j=1,n_horseshoe+1

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```

        write(6,*) attached_c_matrix_v(i,j)
    end do
end do
write(6,*) 'End of document'
close (6)
open(UNIT=6,FILE='Attached_c_matrix_w.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
do i=1,n_horseshoe
    do j=1,n_horseshoe+1
        write(6,*) attached_c_matrix_w(i,j)
    end do
end do
write(6,*) 'End of document'
close (6)
end if
if (b_blade_ind_speed.NE.0) then
    open(UNIT=6,FILE='Blade_induced_speeds_u.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE' )
)
do i=1,n_horseshoe
    write(6,*) blade_ind(1,:,i)
end do
write(6,*) 'End of document'
close (6)
end if
if (b_wake_ind_speed.NE.0) then
    open(UNIT=6,FILE='Wake_induced_speeds.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
do i=1,n_horseshoe
    write(6,*) wake_ind(1,:,i)
end do
write(6,*) 'End of document'
close (6)
open(UNIT=6,FILE='Wake2_induced_speeds.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
do i=1,n_horseshoe
    write(6,*) wake_ind2(1,:,i)
end do
write(6,*) 'End of document'
close (6)
end if
if (b_cp_inc_speeds.NE.0) then
    open(UNIT=6,FILE='CP_incident_speeds.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
do i=1,n_horseshoe
    write(6,*) total_wind(1,:,i)
end do
write(6,*) 'End of document'
close (6)
end if
if (b_cl.NE.0) then
    open(UNIT=6,FILE='Cl_distribution.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
do j=1,n_time_steps
    do i=1,n_horseshoe
        write(6,*) Cl(j,i)
    end do
end do
write(6,*) 'End of document'
close (6)
end if
if (b_wind_speed.NE.0) then
    open(UNIT=6,FILE='Wind_speed_incidence.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
do i=1,n_horseshoe
    write(6,*) wind(1,:,i)
end do
write(6,*) 'End of document'
close (6)
end if
if (b_area_dist.NE.0) then
    open(UNIT=6,FILE='Area_per_element.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
do i=1,n_horseshoe
    write(6,*) area_distr(i)
end do
write(6,*) 'End of document'
close (6)
end if
if (b_chord_dist.NE.0) then
    open(UNIT=6,FILE='Chord_distribution.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
do i=1,n_horseshoe

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```

        write(6,*) chord_distr(i)
    end do
    write(6,*) 'End of document'
    close (6)
end if
if (b_hs_cp_2D_pos.NE.0) then
    open(UNIT=6,FILE='CP_2D_locations.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    write(6,*) 'Number of control points:', n_horseshoe
    write(6,*) 'Control points locations'
    write(6,*) 'x y z'
    do i=1,n_horseshoe
        write(6,*) control_points_loc(:,i)
    end do
    write(6,*) 'End of file'
    close (6)
    open(UNIT=6,FILE='HS_2D_locations.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    write(6,*) 'Number of horseshoe nodes:', n_horseshoe+1
    write(6,*) 'HS nodes locations'
    write(6,*) 'x y z'
    do i=1,n_horseshoe+1
        write(6,*) horseshoe_loc(:,i)
    end do
    write(6,*) 'End of file'
    close (6)
    open(UNIT=6,FILE='HS_te_2D_locations.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    write(6,*) 'Number of horseshoe te nodes:', n_horseshoe+1
    write(6,*) 'HS te nodes locations'
    write(6,*) 'x y z'
    do i=1,n_horseshoe+1
        write(6,*) horseshoe_te_loc(:,i)
    end do
    write(6,*) 'End of file'
    close (6)
end if
if (b_wake_connec.NE.0) then
    open(UNIT=6,FILE='Wake_nodes_connectivity.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE'
    )
    do i=1,(n_wake_lines-1)*(n_horseshoe+1)
        write(6,*) wake_nodes_connec(:,i)
    end do
    write(6,*) 'End of file'
    close (6)
end if
if (b_sec_pol.NE.0) then
    open(UNIT=6,FILE='Section_polars.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    do i=1,n_horseshoe
        write(6,*) "new section"
        do j=1,n_data_points
            write(6,*) Cl_pol(i,j)
        end do
    end do
    write(6,*) "end of file"
    close(6)
end if
open(UNIT=6,FILE='Turning_induced_speed.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
do i=1,n_horseshoe
    write(6,*) turning_ind(:,i)
end do
close(6)

!Further writing codes can be add easily

!created 18/11/2012
!validated 07/12/12
end subroutine check
!-----
!subroutine for calculating the wake rings vortex intensities, from the vortex strength
!distribution along the blade
!this subroutine is just intended for stationary cases, since the wake strength is
!constant for each spanwise position
!it will be used aslo for the first approximation when building the wake
!for transient conditions it is necessary to use another subroutine
subroutine wake_vortex_intensities
    use horseshoe_locations

```



```

use wake_sheet
use vortex_intensities
use clock
use iteration_data
use simulation_data
implicit none

integer i,j,k

if (wake_int_init.EQ.1) then
  k=1
  do i=1,n_horseshoe
    do j=1,n_wake_lines-1
      pr_wake_int(:,(j-1)*n_horseshoe+i)=pr_attached_hs_int(blade_phase(1,
time_step+1),i)
    end do
  end do

else if (wake_int_init.EQ.2) then

  print*, "time(time_step) ", time(time_step)

  if (time_sample.NE.time(time_step)) then

    do i=1,n_horseshoe*(n_wake_lines-2)
      pr_wake_int(blade_phase(1,time_step+1),i+n_horseshoe)=pr_wake_int
(blade_phase(1,time_step),i)
    end do
    do i=1,n_horseshoe
      pr_wake_int(time_step,i)=pr_attached_hs_int(blade_phase(1,time_step),i)
    end do
    time_sample=time(time_step)
  end if
else
  print*,"There was an error. The value of wake_int_init must be contained within 1
and 2"
  pause
  stop
end if
!created 18/01/2013
!validated
end subroutine wake_vortex_intensities
!-----
!subroutine for calculating the speed induced by the wake
subroutine wake_induced_speed

use horseshoe_locations
use incident_speeds
use vortex_intensities
use wake_sheet
use clock
use geometric_inputs
use simulation_data
implicit none

integer i,j,k,b,q
real aux3(3)
real (kind=8):: auxpos(2,3),speeds(3)

do i=1,n_horseshoe
  aux3(1:3)=0.
  !test
  k=1
  q=0
  do b=1,n_blades
    do j=1,(n_wake_lines-1)*n_horseshoe

      auxpos(1,1)=pr_wake_nodes_loc(blade_phase(b,time_step+1),1,
wake_nodes_connec(2,j))
      auxpos(1,2)=pr_wake_nodes_loc(blade_phase(b,time_step+1),2,
wake_nodes_connec(2,j))
      auxpos(1,3)=pr_wake_nodes_loc(blade_phase(b,time_step+1),3,
wake_nodes_connec(2,j))

```

```

        auxpos(2,1)=pr_wake_nodes_loc(blade_phase(b,time_step+1),1,
wake_nodes_connec(3,j))
        auxpos(2,2)=pr_wake_nodes_loc(blade_phase(b,time_step+1),2,
wake_nodes_connec(3,j))
        auxpos(2,3)=pr_wake_nodes_loc(blade_phase(b,time_step+1),3,
wake_nodes_connec(3,j))
        !auxpos is the position of the nodes that delimits the vortex segment
        call VLtfunc(temp_cp_loc(1,1,i),temp_cp_loc(1,2,i),temp_cp_loc(1,3,i),
auxpos(1,1),auxpos(1,2),auxpos(1,3),auxpos(2,1),auxpos(2,2),auxpos(2,3),pr_wake_int
(blade_phase(b,time_step+1),wake_nodes_connec(1,j)),speeds(1),speeds(2),speeds(3))
        !test
        !call VLtfunc(temp_cp_loc(1,1,i),temp_cp_loc(1,2,i),temp_cp_loc(1,3,i),
auxpos(1,1),auxpos(1,2),auxpos(1,3),auxpos(2,1),auxpos(2,2),auxpos(2,3),1.d0,speeds(1)
,speeds(2),speeds(3))

        !if (j.GT.n_horseshoe) then
            aux3(1)=aux3(1)+speeds(1)
            aux3(2)=aux3(2)+speeds(2)
            aux3(3)=aux3(3)+speeds(3)
        !end if

        auxpos(1,1)=pr_wake_nodes_loc(blade_phase(b,time_step+1),1,
wake_nodes_connec(3,j))
        auxpos(1,2)=pr_wake_nodes_loc(blade_phase(b,time_step+1),2,
wake_nodes_connec(3,j))
        auxpos(1,3)=pr_wake_nodes_loc(blade_phase(b,time_step+1),3,
wake_nodes_connec(3,j))
        auxpos(2,1)=pr_wake_nodes_loc(blade_phase(b,time_step+1),1,
wake_nodes_connec(5,j))
        auxpos(2,2)=pr_wake_nodes_loc(blade_phase(b,time_step+1),2,
wake_nodes_connec(5,j))
        auxpos(2,3)=pr_wake_nodes_loc(blade_phase(b,time_step+1),3,
wake_nodes_connec(5,j))
        !auxpos is the position of the nodes that delimits the vortex segment
        call VLtfunc(temp_cp_loc(1,1,i),temp_cp_loc(1,2,i),temp_cp_loc(1,3,i),
auxpos(1,1),auxpos(1,2),auxpos(1,3),auxpos(2,1),auxpos(2,2),auxpos(2,3),pr_wake_int
(blade_phase(b,time_step+1),wake_nodes_connec(1,j)),speeds(1),speeds(2),speeds(3))
        !test
        !call VLtfunc(temp_cp_loc(1,1,i),temp_cp_loc(1,2,i),temp_cp_loc(1,3,i),
auxpos(1,1),auxpos(1,2),auxpos(1,3),auxpos(2,1),auxpos(2,2),auxpos(2,3),1.d0,speeds(1)
,speeds(2),speeds(3))
        !if (j.EQ.n_horseshoe*k) then
            ! k=k+1

            aux3(1)=aux3(1)+speeds(1)
            aux3(2)=aux3(2)+speeds(2)
            aux3(3)=aux3(3)+speeds(3)
        !end if

        auxpos(1,1)=pr_wake_nodes_loc(blade_phase(b,time_step+1),1,
wake_nodes_connec(5,j))
        auxpos(1,2)=pr_wake_nodes_loc(blade_phase(b,time_step+1),2,
wake_nodes_connec(5,j))
        auxpos(1,3)=pr_wake_nodes_loc(blade_phase(b,time_step+1),3,
wake_nodes_connec(5,j))
        auxpos(2,1)=pr_wake_nodes_loc(blade_phase(b,time_step+1),1,
wake_nodes_connec(4,j))
        auxpos(2,2)=pr_wake_nodes_loc(blade_phase(b,time_step+1),2,
wake_nodes_connec(4,j))
        auxpos(2,3)=pr_wake_nodes_loc(blade_phase(b,time_step+1),3,
wake_nodes_connec(4,j))
        !auxpos is the position of the nodes that delimits the vortex segment
        call VLtfunc(temp_cp_loc(1,1,i),temp_cp_loc(1,2,i),temp_cp_loc(1,3,i),
auxpos(1,1),auxpos(1,2),auxpos(1,3),auxpos(2,1),auxpos(2,2),auxpos(2,3),pr_wake_int
(blade_phase(b,time_step+1),wake_nodes_connec(1,j)),speeds(1),speeds(2),speeds(3))
        !test
        !call VLtfunc(temp_cp_loc(1,1,i),temp_cp_loc(1,2,i),temp_cp_loc(1,3,i),
auxpos(1,1),auxpos(1,2),auxpos(1,3),auxpos(2,1),auxpos(2,2),auxpos(2,3),1.d0,speeds(1)
,speeds(2),speeds(3))
        aux3(1)=aux3(1)+speeds(1)
        aux3(2)=aux3(2)+speeds(2)
        aux3(3)=aux3(3)+speeds(3)

        auxpos(1,1)=pr_wake_nodes_loc(blade_phase(b,time_step+1),1,

```

```

wake_nodes_connec(4,j)
    auxpos(1,2)=pr_wake_nodes_loc(blade_phase(b,time_step+1),2,
wake_nodes_connec(4,j)
    auxpos(1,3)=pr_wake_nodes_loc(blade_phase(b,time_step+1),3,
wake_nodes_connec(4,j)
    auxpos(2,1)=pr_wake_nodes_loc(blade_phase(b,time_step+1),1,
wake_nodes_connec(2,j)
    auxpos(2,2)=pr_wake_nodes_loc(blade_phase(b,time_step+1),2,
wake_nodes_connec(2,j)
    auxpos(2,3)=pr_wake_nodes_loc(blade_phase(b,time_step+1),3,
wake_nodes_connec(2,j)
    !auxpos is the position of the nodes that delimits the vortex segment
    call VLtfunc(temp_cp_loc(1,1,i),temp_cp_loc(1,2,i),temp_cp_loc(1,3,i),
auxpos(1,1),auxpos(1,2),auxpos(1,3),auxpos(2,1),auxpos(2,2),auxpos(2,3),pr_wake_int
(blade_phase(b,time_step+1),wake_nodes_connec(1,j)),speeds(1),speeds(2),speeds(3))
    !test
    !call VLtfunc(temp_cp_loc(1,1,i),temp_cp_loc(1,2,i),temp_cp_loc(1,3,i),
auxpos(1,1),auxpos(1,2),auxpos(1,3),auxpos(2,1),auxpos(2,2),auxpos(2,3),1.d0,speeds(1)
,speeds(2),speeds(3))
    !if (j-1.EQ.q*n_horseshoe) then
        !q=q+1
        aux3(1)=aux3(1)+speeds(1)
        aux3(2)=aux3(2)+speeds(2)
        aux3(3)=aux3(3)+speeds(3)
    !end if

    end do !loop for j
end do !loop for b
wake_ind(time_step,1,i)=aux3(1)
wake_ind(time_step,2,i)=aux3(2)
wake_ind(time_step,3,i)=aux3(3)
end do

!created 28/11/2012
!validated 06/12/12
end subroutine wake_induced_speed

!-----
!subroutine for calling the other subroutines in the proper order with the objective of
the final convergence of the vortex intensities
subroutine calculations

use iteration_data
use vortex_intensities
use horseshoe_locations
use clock
use operating_parameters
use aerodynamic_performance
use simulation_data

!test
use operating_parameters
implicit none

integer i,j,k,m
real convergence_t,convergence_i,precision_t,precision_i,aux
real (kind=8), parameter:: pi=4*atan(1.)

precision_t=0.01
precision_i=0.0001

pr_attached_hs_int(:,:)=0.d0
pr_cycle_att_hs_int(:,:)=1.d0
!
j=0
m=0
convergence_t=100
do while (convergence_t.GT.precision_t)
    j=j+1
    print*,"J",j
    do time_step=1,n_time_steps

        convergence_i=100
        azimuth=preazimut+(time_step-1)*omega*inc_time

```

```

if ((b_stationary_sim.EQ.1).AND.(b_wake_roll_up.EQ.1)) then
    m=m+1
    azimut=preazimut+(m-1)*omega*inc_time
end if

call temp_location
do i=1,n_horseshoe
    cp_saved_pos(time_step,:,i)=temp_cp_loc(1,:,i)
end do

call movement_ind_speed

call wind_speed

call place_wake_sheet

if (b_stationary_sim.NE.1) then

    call wake_vortex_intensities

    call wake_induced_speed
end if

k=0
do while (convergence_i.GT.precision_i)

    k=k+1
    print*, "K", k

    convergence_i=0
    if (b_stationary_sim.EQ.1) then

        call wake_vortex_intensities

        call wake_induced_speed
    end if

    call blade_direct_induced_speed

    call cp_incident_speeds
    call check
    call vortex_intensity_calculator

    do i=1,n_horseshoe
        aux=abs((pr_attached_hs_int(time_step,i)-aux_attached_hs_int(i))/(abs
        (pr_attached_hs_int(time_step,i))+0.1))
        aux_attached_hs_int(i)=pr_attached_hs_int(time_step,i)
        if (aux.GT.convergence_i) then
            convergence_i=aux
        end if
    end do
    print*, "convergence_i=",convergence_i
    print*, "sum of Cl",sum(Cl(time_step,:))
    call check
end do
if (wake_int_init.EQ.1.AND.b_stationary_sim.EQ.0) then
    wake_int_init=2
end if

end do

convergence_t=0
do time_step=1,n_time_steps
    do i=1,n_horseshoe
        aux=abs(pr_attached_hs_int(time_step,i)-pr_cycle_att_hs_int(time_step,i))/
        (abs(pr_attached_hs_int(time_step,i))+0.1)
        pr_cycle_att_hs_int(time_step,i)=pr_attached_hs_int(time_step,i)
        if (aux.GT.convergence_t) then
            convergence_t=aux
        end if
    end do
end do

```

```

        print*, "convergence_t=", convergence_t
    end do
    print*, "Simulation finalized"

end subroutine calculations
!-----
!subroutine for initializing all the needed variables, to be used after reading the inputs
subroutine initialize

    use horseshoe_locations
    use aerodynamic_performance
    use airfoil_polars
    use geometric_inputs
    use incident_speeds
    use vortex_intensities
    use wake_sheet
    use tf_matrix
    use iteration_data
    use operating_parameters
    use simulation_data
    use clock
    implicit none

    real (kind=8), parameter::pi=4.*atan(1.)
    integer i, aux

    allocate(horseshoe_loc(2,n_horseshoe+1))
    allocate(horseshoe_te_loc(2,n_horseshoe+1))
    allocate(control_points_loc(2,n_horseshoe))
    allocate(area_distr(n_horseshoe))
    allocate(chord_distr(n_horseshoe))
    allocate(twist(n_horseshoe))
    allocate(thickness(n_horseshoe))
    allocate(temp_hs_loc(n_blades,3,n_horseshoe+1))
    allocate(temp_hs_te_loc(n_blades,3,n_horseshoe+1))
    allocate(temp_cp_loc(n_blades,3,n_horseshoe))
    allocate(cp_saved_pos(n_time_steps,3,n_horseshoe))
    allocate(wake_ind(n_time_steps,3,n_horseshoe))

    allocate(el_vect1(n_time_steps,3,n_horseshoe))
    allocate(el_vect2(n_time_steps,3,n_horseshoe))
    allocate(el_vect3(n_time_steps,3,n_horseshoe))

    allocate(wind(n_time_steps,3,n_horseshoe))
    allocate(turning_ind(2,n_horseshoe))
    allocate(blade_ind(n_time_steps,3,n_horseshoe))
    allocate(blade_ind2(3,n_horseshoe))
    allocate(total_wind(n_time_steps,4,n_horseshoe))
    allocate(alpha(n_time_steps,n_horseshoe))
    allocate(Cl(n_time_steps,n_horseshoe))
    allocate(Cd(n_time_steps,n_horseshoe))
    allocate(Cm(n_time_steps,n_horseshoe))
    allocate(L(n_time_steps,n_horseshoe))
    allocate(D(n_time_steps,n_horseshoe))
    allocate(Mz(n_time_steps,n_horseshoe))
    allocate(MB_ip(n_time_steps,n_horseshoe))
    allocate(MB_op(n_time_steps,n_horseshoe))
    allocate(Q(n_time_steps))
    allocate(T(n_time_steps))
    allocate(P(n_time_steps))
    allocate(Cp(n_time_steps))
    allocate(Ct(n_time_steps))
    allocate(attached_s_matrix_u(n_horseshoe,n_horseshoe))
    allocate(attached_s_matrix_v(n_horseshoe,n_horseshoe))
    allocate(attached_s_matrix_w(n_horseshoe,n_horseshoe))
    allocate(attached_c_matrix_u(n_horseshoe,n_horseshoe+1))
    allocate(attached_c_matrix_v(n_horseshoe,n_horseshoe+1))
    allocate(attached_c_matrix_w(n_horseshoe,n_horseshoe+1))
    aux=n_horseshoe*n_wake_lines

    allocate(wake_s_matrix_u(n_blades,n_horseshoe,aux))
    allocate(wake_s_matrix_v(n_blades,n_horseshoe,aux))
    allocate(wake_s_matrix_w(n_blades,n_horseshoe,aux))
    allocate(wake_segments_int_s(n_blades,aux))

```

```

aux=(n_horseshoe+1)*(n_wake_lines-1)

allocate(wake_c_matrix_u(n_blades,n_horseshoe,aux))
allocate(wake_c_matrix_v(n_blades,n_horseshoe,aux))
allocate(wake_c_matrix_w(n_blades,n_horseshoe,aux))
allocate(wake_segments_int_c(n_blades,aux))
print*,n_horseshoe,n_wake_lines

allocate(wake_nodes_connec(5,n_horseshoe*(n_wake_lines-1)))
allocate(pr_attached_hs_int(n_time_steps,n_horseshoe))
allocate(pr_wake_int(n_time_steps,n_horseshoe*(n_wake_lines-1)))
allocate(pr_wake_nodes_loc(n_time_steps,3,(n_horseshoe+1)*n_wake_lines))
allocate(time(n_time_steps))
allocate(aux_attached_hs_int(n_horseshoe))
allocate(pr_cycle_att_hs_int(n_time_steps,n_horseshoe))

if (omega.EQ.0) then
  inc_time=0.1
else if (n_time_steps.GT.1) then
  inc_time=2*pi/(omega*n_time_steps)
else
  inc_time=0.1/omega
end if
do i=1,n_time_steps
  time(i)=inc_time*(i-1)
end do
time_sample=time(n_time_steps)

!created 08/12/12
!validated 09/12/12
end subroutine initialize
!-----
!subroutine for constructing the matrix that contains the temporal relations between the
  blades of the rotor and the previous positions
subroutine phase_between_blades

  use geometric_inputs
  use clock
  use simulation_data
  use horseshoe_locations
  implicit none

  integer i,b,aux(2*n_time_steps)

  do i=1,n_time_steps
    aux(i)=i
    aux(i+n_time_steps)=i
  end do

  allocate(blade_phase(n_blades,n_time_steps+1))

  do i=1,n_time_steps+1
    blade_phase(1,i)=aux(n_time_steps+i-1)
    do b=2,n_blades
      blade_phase(b,i)=aux(i+(b-1)*int(n_time_steps/n_blades)-1)
    end do
  end do
  print*, "Blade Phase", blade_phase
end subroutine phase_between_blades
!-----
!subroutine for interpolating the polar curves for each defined section
subroutine extrude_polars

  use horseshoe_locations
  use airfoil_polars

  implicit none

  integer i,j
  do i=1,n_horseshoe
    do j=1,n_data_points

```

```

        call spline(airfoils_thickness,real(Cl_pol_aux(:,j)),thickness(i),Cl_pol(i,j),
n_airfoils,1)
        call spline(airfoils_thickness,real(Cd_pol_aux(:,j)),thickness(i),Cd_pol(i,j),
n_airfoils,1)
        call spline(airfoils_thickness,real(Cm_pol_aux(:,j)),thickness(i),Cm_pol(i,j),
n_airfoils,1)
    end do
end do
open(UNIT=7,FILE='Results_polar_curves_sections.txt',ACCESS='SEQUENTIAL',STATUS=
'REPLACE')
    do i=1,n_horseshoe
        do j=1,n_data_points
            write(7,*) Cl_pol(i,j), Cd_pol(i,j), Cm_pol(i,j)
        end do
    end do
close(7)
!created 24/03/2013
!not validated
end subroutine extrude_polars
!-----
!subroutine for creating and filling the .txt files with the results of the simulations
subroutine write_outputs

    use horseshoe_locations
    use geometric_inputs
    use clock
    use operating_parameters
    use aerodynamic_performance
    use incident_speeds
    use wake_sheet
    use simulation_data
    use vortex_intensities
    use airfoil_polars

    implicit none

    integer i,j
    real, parameter:: pi=4*atan(1.)

    !lift coefficient cl, drag coefficient cd and aerodynamic moment cm
open(UNIT=7,FILE='Results_aerodynamic_coefficients.txt',ACCESS='SEQUENTIAL',STATUS=
'REPLACE')
do i=1,n_time_steps
    do j=1,n_horseshoe
        write(7,*) Cl(i,j) , Cd(i,j), Cm(i,j)
    end do
end do
close(7)

!angles of attack alpha
open(UNIT=7,FILE='Results_AoA.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
do i=1,n_time_steps
    do j=1,n_horseshoe
        write(7,*) alpha(i,j)
    end do
end do
close(7)

!position of the wake nodes
open(UNIT=7,FILE='Results_wake_nodes_pos.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
do i=1,n_time_steps
    do j=1,(n_horseshoe+1)*n_wake_lines
        write(7,*) pr_wake_nodes_loc(i,1,j), pr_wake_nodes_loc(i,2,j),
pr_wake_nodes_loc(i,3,j)
    end do
end do
close(7)

!wake rings intensity
open(UNIT=7,FILE='Results_wake_rings_intensity.txt',ACCESS='SEQUENTIAL',STATUS=
'REPLACE')
do i=1,n_time_steps
    do j=1,n_horseshoe*(n_wake_lines-1)
        write(7,*) pr_wake_int(i,j)
    end do
end do

```

```

end do

!some of the inputs of the calculation
open(UNIT=7,FILE='Results_Calculation_inputs.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE'
)
  write(7,*) n_horseshoe
  write(7,*) n_wake_lines
  write(7,*) n_time_steps
  write(7,*) omega
  write(7,*) v0(1)
  write(7,*) v0(2)
  write(7,*) v0(3)
  write(7,*) density
  write(7,*) pitch
  write(7,*) inc_time

close(7)

!position of the control points and the horseshoe nodes
open(UNIT=7,FILE='Results_cp_and_hs_positions.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE'
')
  do i=1,n_horseshoe
    write(7,*) temp_cp_loc(1,1,i), temp_cp_loc(1,2,i), temp_cp_loc(1,2,i)
  end do
  do i=1,n_horseshoe+1
    write(7,*) temp_hs_loc(1,1,i), temp_hs_loc(1,2,i), temp_hs_loc(1,3,i)
  end do
  do i=1,n_horseshoe+1
    write(7,*) temp_hs_te_loc(1,1,i), temp_hs_te_loc(1,2,i), temp_hs_te_loc(1,3,i)
  end do
close(7)

!2D position of the control points and the horseshoe nodes
open(UNIT=7,FILE='Results_2D_cp_and_hs_positions.txt',ACCESS='SEQUENTIAL',STATUS=
'REPLACE')
  do i=1,n_horseshoe
    write(7,*) control_points_loc(1,i), control_points_loc(2,i)
  end do
  do i=1,n_horseshoe+1
    write(7,*) horseshoe_loc(1,i),horseshoe_loc(2,i)
  end do
  do i=1,n_horseshoe+1
    write(7,*) horseshoe_te_loc(1,i),horseshoe_te_loc(2,i)
  end do
close(7)

!geometrical properties along the blade
open(UNIT=7,FILE='Results_geometrical_blade_properties.txt',ACCESS='SEQUENTIAL',STATUS
='REPLACE')
  do i=1,n_horseshoe
    write(7,*) chord_distr(i), thickness(i), twist(i)*180/pi, area_distr(i)
  end do

close(7)

!attached horseshoe intensity
open(UNIT=7,FILE='Results_attached_hs_vortex_intensity.txt',ACCESS='SEQUENTIAL',STATUS
='REPLACE')
  do i=1,n_time_steps
    do j=1,n_horseshoe
      write(7,*) pr_attached_hs_int(i,j)
    end do
  end do
close(7)

!element vectors
open(UNIT=7,FILE='Results_local_element_vectors.txt',ACCESS='SEQUENTIAL',STATUS=
'REPLACE')
  do i=1,n_time_steps
    do j=1,n_horseshoe
      write(7,*) el_vect1(i,:,j)
    end do
  end do
  do i=1,n_time_steps
    do j=1,n_horseshoe
      write(7,*) el_vect2(i,:,j)
    end do
  end do

```



```

    end do
    do i=1,n_time_steps
        do j=1,n_horseshoe
            write(7,*) el_vect3(i,:,j)
        end do
    end do
close(7)
!blade and wake induced speeds
open(UNIT=7,FILE='Results_blade_ind_speeds.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    do j=1,n_time_steps
        do i=1,n_horseshoe
            write(7,*) blade_ind(j,1,i),blade_ind(j,2,i),blade_ind(j,3,i)
        end do
    end do
close(7)
open(UNIT=7,FILE='Results_wake_ind_speeds.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    do j=1,n_time_steps
        do i=1,n_horseshoe
            write(7,*) wake_ind(j,1,i),wake_ind(j,2,i),wake_ind(j,3,i)
        end do
    end do
close(7)
!wind speed
open(UNIT=7,FILE='Results_unaffected_wind_speed.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    do i=1,n_time_steps
        do j=1,n_horseshoe
            write(7,*) wind(i,1,j), wind(i,2,j),wind(i,3,j)
        end do
    end do
close(7)
!total wind incidence
open(UNIT=7,FILE='Results_incident_speeds.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    do i=1,n_time_steps
        do j=1,n_horseshoe
            write(7,*) total_wind(i,1,j), total_wind(i,2,j), total_wind(i,3,j)
        end do
    end do
close(7)
!Forces and moments for the rotor
open(UNIT=7,FILE='Results_blade_dynamics.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    do i=1,n_time_steps
        do j=1,n_horseshoe
            write(7,*) L(i,j), D(i,j), Mz(i,j)
        end do
    end do
close(7)
open(UNIT=7,FILE='Results_rotor_dynamics.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    do i=1,n_time_steps
        write(7,*) Q(i),T(i),P(i)
    end do
close(7)
!Performance coefficients
open(UNIT=7,FILE='Results_rotor_coefficients.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
)
    do i=1,n_time_steps
        write(7,*) Cp(i),Ct(i)
    end do
close(7)
!Bending moments in blades
open(UNIT=7,FILE='Results_blade_bending_moments.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    do i=1,n_time_steps
        do j=1,n_horseshoe
            write(7,*) MB_ip(i,j), MB_op(i,j)
        end do
    end do
close(7)

!created 13/04/2013
!validated
end subroutine write_outputs
!-----
!subroutine for postprocessing the calculation and obtain the useful variables

```

```

subroutine postprocessing

  use horseshoe_locations
  use incident_speeds
  use aerodynamic_performance
  use operating_parameters
  use geometric_inputs
  use simulation_data
  use airfoil_polars
  use clock
  use incident_speeds

  implicit none

  integer i,j,b
  real(kind=8), parameter:: pi=4*atan(1.)
  real(kind=8):: aux_speed, vec_x(3),vec_y(3),vec_z(3), vec_L(n_time_steps,3,
n_horseshoe),vec_D(n_time_steps,3,n_horseshoe)
  real (kind=8):: aux3(3), aux_phi(n_horseshoe),aux4(3)
  real aux_wind

  vec_x(:)=0.
  vec_y(:)=0.
  vec_z(:)=0.
  vec_x(1)=1.
  vec_y(2)=1.
  vec_z(3)=1.
  Q(:)=0
  T(:)=0
  P(:)=0
  MB_ip(:,:)=0.
  MB_op(:,:)=0.
  do i=1,n_time_steps
    do j=1,n_horseshoe
      do b=1,n_blades
        call spline(real(AOA(1,:)),real(Cd_pol(j,:)),alpha(blade_phase(b,i+1),j),
Cd(blade_phase(b,i+1),j),n_data_points,1)
        call spline(real(AOA(1,:)),real(Cm_pol(j,:)),alpha(blade_phase(b,i+1),j),
Cm(blade_phase(b,i+1),j),n_data_points,1)
        end do

        aux_speed=dot_product(el_vect1(i,:,j),total_wind(i,1:3,j))*2+dot_product
(el_vect3(i,:,j),total_wind(i,1:3,j))*2
        L(i,j)=0.5*C1(i,j)*density*area_distr(j)*aux_speed
        D(i,j)=0.5*Cd(i,j)*density*area_distr(j)*aux_speed
        Mz(i,j)=0.5*Cm(i,j)*density*area_distr(j)*aux_speed

        call crossprod(total_wind(i,1:3,j),el_vect2(i,:,j),vec_L(i,:,j))
        vec_L(i,:,j)=vec_L(i,:,j)/sqrt(dot_product(vec_L(i,:,j),vec_L(i,:,j)))
        !aux3(:,j)=vec_L(i,:,j)
        vec_D(i,:,j)=total_wind(i,1:3,j)/sqrt(dot_product(total_wind(i,1:3,j),
total_wind(i,1:3,j)))
        !aux4(:,j)=vec_D(i,:,j)
      end do !j=1,n_horseshoe
    end do !i=1,n_time_steps !the loop must be split up, the L and D must be obtained
for all blades before calculating the Q,P and T

    open(UNIT=7,FILE="Cp_saved_pos.txt",ACCESS='SEQUENTIAL',STATUS='REPLACE')
    do i=1,n_horseshoe
      write(7,*) cp_saved_pos(1,:,i)
    end do
    close(7)

    do i=1,n_time_steps
      do j=1,n_horseshoe
        do b=1,n_blades
          Q(i)=Q(i)+L(blade_phase(b,i+1),j)*(dot_product(vec_L(i,:,j),vec_z)*
cp_saved_pos(i,2,j)-dot_product(vec_L(i,:,j),vec_y)*cp_saved_pos(i,3,j))
          Q(i)=Q(i)+D(blade_phase(b,i+1),j)*(dot_product(vec_D(i,:,j),vec_z)*
cp_saved_pos(i,2,j)-dot_product(vec_D(i,:,j),vec_y)*cp_saved_pos(i,3,j))

          T(i)=T(i)+D(blade_phase(b,i+1),j)*dot_product(vec_D(i,:,j),vec_x)
          T(i)=T(i)+L(blade_phase(b,i+1),j)*dot_product(vec_L(i,:,j),vec_x)

```

```

    end do

    do b=j+1,n_horseshoe
        call crossprod(cp_saved_pos(i,:,b)-cp_saved_pos(i,:,j),vec_L(i,:,b),aux3)
        call crossprod(cp_saved_pos(i,:,b)-cp_saved_pos(i,:,j),vec_D(i,:,b),aux4)
        MB_ip(i,j)=MB_ip(i,j)+L(i,b)*dot_product(vec_x,aux3)+D(i,b)*dot_product
    (vec_x,aux4)
        MB_op(i,j)=MB_op(i,j)+L(i,b)*dot_product(el_vect1(i,:,j),aux3)+D(i,b)*
dot_product(el_vect1(i,:,j),aux4)
    end do

    end do
    !aux_wind=sum(wind(i,1,:))/n_horseshoe
    P(i)=Q(i)*omega
    Cp(i)=P(i)/(0.5*density*pi*(control_points_loc(1,n_horseshoe))**2*
(wind_speed_module)**3)
    Ct(i)=T(i)/(0.5*density*pi*control_points_loc(1,n_horseshoe)**2*wind_speed_module*
*2)
end do

!created 21/04/2013
!modified 21/05/2013
!validated
end subroutine postprocessing
!-----
!subroutine with alternative version for calculating the blade self induced speed
subroutine blade_direct_induced_speed

    use horseshoe_locations
    use clock
    use incident_speeds
    use vortex_intensities
    use geometric_inputs
    use simulation_data

    implicit none

    integer i,j,b
    real(kind=8):: speeds(3),aux3(3),blade_ind_det(n_horseshoe,3,n_horseshoe)

    open(UNIT=7,FILE='Blade_ind_detail.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE')
    blade_ind(time_step,::)=0.d0
    do i=1,n_horseshoe

        do b=1,n_blades
            aux3(:)=0.d0
            do j=1,n_horseshoe

                call VLtfnc(temp_cp_loc(1,1,i),temp_cp_loc(1,2,i),temp_cp_loc(1,3,i),
temp_hs_loc(b,1,j),temp_hs_loc(b,2,j),temp_hs_loc(b,3,j),temp_hs_loc(b,1,j+1),
temp_hs_loc(b,2,j+1),temp_hs_loc(b,3,j+1),pr_attached_hs_int(blade_phase(b,time_step+
1),j),speeds(1),speeds(2),speeds(3))

                aux3(1)=aux3(1)+speeds(1)
                aux3(2)=aux3(2)+speeds(2)
                aux3(3)=aux3(3)+speeds(3)

                blade_ind_det(i,1,j)=speeds(1)
                blade_ind_det(i,2,j)=speeds(2)
                blade_ind_det(i,3,j)=speeds(3)

                call VLtfnc(temp_cp_loc(1,1,i),temp_cp_loc(1,2,i),temp_cp_loc(1,3,i),
temp_hs_loc(b,1,j+1),temp_hs_loc(b,2,j+1),temp_hs_loc(b,3,j+1),temp_hs_te_loc(b,1,j+1)
,temp_hs_te_loc(b,2,j+1),temp_hs_te_loc(b,3,j+1),pr_attached_hs_int(blade_phase(b,
time_step+1),j),speeds(1),speeds(2),speeds(3))

                aux3(1)=aux3(1)+speeds(1)
                aux3(2)=aux3(2)+speeds(2)
                aux3(3)=aux3(3)+speeds(3)

                blade_ind_det(i,1,j)=blade_ind_det(i,1,j)+speeds(1)
                blade_ind_det(i,2,j)=blade_ind_det(i,2,j)+speeds(2)

```

```

blade_ind_det(i,3,j)=blade_ind_det(i,3,j)+speeds(3)

      call VLtfnc(temp_cp_loc(1,1,i),temp_cp_loc(1,2,i),temp_cp_loc(1,3,i),
temp_hs_te_loc(b,1,j+1),temp_hs_te_loc(b,2,j+1),temp_hs_te_loc(b,3,j+1),temp_hs_te_loc
(b,1,j),temp_hs_te_loc(b,2,j),temp_hs_te_loc(b,3,j),pr_attached_hs_int(blade_phase(b,
time_step+1),j),speeds(1),speeds(2),speeds(3))

      aux3(1)=aux3(1)+speeds(1)
      aux3(2)=aux3(2)+speeds(2)
      aux3(3)=aux3(3)+speeds(3)

      blade_ind_det(i,1,j)=blade_ind_det(i,1,j)+speeds(1)
      blade_ind_det(i,2,j)=blade_ind_det(i,2,j)+speeds(2)
      blade_ind_det(i,3,j)=blade_ind_det(i,3,j)+speeds(3)

      call VLtfnc(temp_cp_loc(1,1,i),temp_cp_loc(1,2,i),temp_cp_loc(1,3,i),
temp_hs_te_loc(b,1,j),temp_hs_te_loc(b,2,j),temp_hs_te_loc(b,3,j),temp_hs_loc(b,1,j),
temp_hs_loc(b,2,j),temp_hs_loc(b,3,j),pr_attached_hs_int(blade_phase(b,time_step+1),j)
,speeds(1),speeds(2),speeds(3))

      aux3(1)=aux3(1)+speeds(1)
      aux3(2)=aux3(2)+speeds(2)
      aux3(3)=aux3(3)+speeds(3)

      blade_ind_det(i,1,j)=blade_ind_det(i,1,j)+speeds(1)
      blade_ind_det(i,2,j)=blade_ind_det(i,2,j)+speeds(2)
      blade_ind_det(i,3,j)=blade_ind_det(i,3,j)+speeds(3)

    end do
    do j=1,n_horseshoe
      write(7,*) blade_ind_det(i,1,j),blade_ind_det(i,2,j),blade_ind_det(i,3,j)
    end do

    blade_ind(time_step,1,i)=blade_ind(time_step,1,i)+aux3(1)
    blade_ind(time_step,2,i)=blade_ind(time_step,2,i)+aux3(2)
    blade_ind(time_step,3,i)=blade_ind(time_step,3,i)+aux3(3)
  end do

end do

close(7)

end subroutine blade_direct_induced_speed
!-----
!subroutine to run for single point simulations
subroutine single_point_simulation

  use simulation_data

  use horseshoe_locations
  use operating_parameters
  use geometric_inputs
  use wake_sheet
  use clock
  use aerodynamic_performance
  use file_names

  implicit none

  real, parameter:: pi=4.*atan(1.)

  open(UNIT=7,FILE=inputfile,ACCESS='SEQUENTIAL')
  read(7,*)
  read(7,*)
  read(7,*)
  read(7,*) n_blades
  read(7,*) hub_to_br
  read(7,*) pitch
  pitch=pitch*pi/180
  read(7,*) omega
  read(7,*) wind_speed_module
  read(7,*) density

```

```

read(7,*) n_horseshoe
read(7,*) n_wake_lines
read(7,*) b_wake_roll_up

v0(1)=wind_speed_module
v0(2:3)=0.
n_time_steps=1
upflow_angle=0.
yaw_misalignment=0.
hub_height=50.
exp_a=0.
b_stationary_sim=1
azimut=0.
preazimut=azimut
wake_loc_init=1
wake_int_init=1

call initialize

call read_input

call phase_between_blades

call place_elements

call extrude_polars

call from_cartesian_to_polars

call calculations

call check

call postprocessing

call write_outputs

!created 25/04/2013
!validated 01/05/2013
end subroutine single_point_simulation
!-----
!subroutine for simulating transient conditions
subroutine full_cycle_simulation

use simulation_data

use horseshoe_locations
use wake_sheet
use operating_parameters
use aerodynamic_performance
use clock
use geometric_inputs
use file_names

implicit none

real, parameter:: pi=4.*atan(1.)

open(UNIT=7,FILE=inputfile,ACCESS='SEQUENTIAL')
read(7,*)
read(7,*)
read(7,*)
read(7,*) n_blades
read(7,*) hub_to_br
read(7,*) pitch
pitch=pitch*pi/180
read(7,*) hub_height
read(7,*) exp_a
read(7,*) tilt_angle
read(7,*) upflow_angle
read(7,*) yaw_misalignment
read(7,*) omega

```

```

read(7,*) wind_speed_module
read(7,*) density
read(7,*) n_horseshoe
read(7,*) n_wake_lines
read(7,*) n_time_steps
read(7,*) b_wake_roll_up
!-----
azimut=0.
preazimut=azimut
b_stationary_sim=0
wake_int_init=1
wake_loc_init=1
upflow_angle=upflow_angle+tilt_angle
upflow_angle=upflow_angle*pi/180.
yaw_misalignment=yaw_misalignment*pi/180
v0(1)=wind_speed_module*cos(upflow_angle)*cos(yaw_misalignment)
v0(2)=wind_speed_module*sin(yaw_misalignment)
v0(3)=wind_speed_module*sin(upflow_angle)

call initialize

call read_input

call phase_between_blades

call place_elements

call extrude_polars

call from_cartesian_to_polars

call calculations

call check

call postprocessing

call write_outputs

!created 01/05/2013
!validated
end subroutine full_cycle_simulation
!-----
!subroutine for obtaining the optimum lambda for the specified geometry
subroutine find_optimal_lambda

use horseshoe_locations
use incident_speeds
use wake_sheet
use operating_parameters
use aerodynamic_performance
use clock
use geometric_inputs
use simulation_data
use lambda_variables
use file_names

implicit none

real R, min_lambda, max_lambda, lambda_step
integer n_lambda
real(kind=8), parameter:: pi=4*atan(1.)

open(UNIT=7, FILE=inputfile, ACCESS='SEQUENTIAL')
read(7,*)
read(7,*)
read(7,*)
read(7,*) n_blades
read(7,*) hub_to_br
read(7,*) pitch
pitch=pitch*pi/180
read(7,*) hub_height

```

```

read(7,*) exp_a
read(7,*) tilt_angle
read(7,*) upflow_angle
read(7,*) yaw_misalignment
read(7,*) omega
read(7,*) density
read(7,*) n_horseshoe
read(7,*) n_wake_lines
read(7,*) n_time_steps
read(7,*) b_wake_roll_up
read(7,*) min_lambda
read(7,*) max_lambda
read(7,*) n_lambda
!-----
azimut=0.
preazimut=azimut
b_stationary_sim=0
if ((upflow_angle.EQ.0).AND.(yaw_misalignment.EQ.0).AND.(exp_a.EQ.0)) then
    b_stationary_sim=1
    n_time_steps=1
end if
print*,b_stationary_sim,"b_stationary_sim"
wake_int_init=1
wake_loc_init=1
upflow_angle=upflow_angle*pi/180.
yaw_misalignment=yaw_misalignment*pi/180
!-----
call initialize
call read_input
call phase_between_blades
call place_elements
call extrude_polars
call from_cartesian_to_polars
allocate(lambda(n_lambda))
allocate(cp_lambda(n_lambda))
allocate(ct_lambda(n_lambda))
R=hs_loc_pol(1,n_horseshoe+1)

do lambda_step=1,n_lambda
    lambda(lambda_step)=min_lambda+(lambda_step-1)*(max_lambda-min_lambda)/(n_lambda-
1)
    wind_speed_module=omega*R/lambda(lambda_step)
    v0(1)=wind_speed_module*cos(upflow_angle)*cos(yaw_misalignment)
    v0(2)=wind_speed_module*sin(yaw_misalignment)
    v0(3)=wind_speed_module*sin(upflow_angle)

    call calculations
    call postprocessing
    cp_lambda(lambda_step)=sum(Cp(:))/n_time_steps
    ct_lambda(lambda_step)=sum(Ct(:))/n_time_steps

end do

open(UNIT=7,FILE='Results_cp_and_ct_lambda_curve.txt',ACCESS='SEQUENTIAL',STATUS=
'REPLACE')
do lambda_step=1,n_lambda
    write(7,*) lambda(lambda_step),cp_lambda(lambda_step),ct_lambda(lambda_step)
end do
close(7)

call write_outputs

end subroutine find_optimal_lambda
!-----
!simulation mode for obtaining the power curve for a fixed lambda
subroutine power_curve

use horseshoe_locations
use incident_speeds
use wake_sheet
use operating_parameters
use aerodynamic_performance
use clock

```

```

use geometric_inputs
use simulation_data
use power_curve_variables
use file_names

implicit none

real R, omega_min, omega_max, P_max, lambda, inc_pc_step
real(kind=8),parameter:: pi=4*atan(1.)
integer i

open(UNIT=7,FILE=inputfile,ACCESS='SEQUENTIAL')
read(7,*)
read(7,*)
read(7,*)
read(7,*) n_blades
read(7,*) hub_to_br
read(7,*) pitch
pitch=pitch*pi/180
read(7,*) hub_height
read(7,*) exp_a
read(7,*) tilt_angle
read(7,*) upflow_angle
read(7,*) yaw_misalignment
read(7,*) density
read(7,*) n_horseshoe
read(7,*) n_wake_lines
read(7,*) n_time_steps
read(7,*) b_wake_roll_up
read(7,*) lambda
read(7,*) omega_min
read(7,*) omega_max
read(7,*) P_max
read(7,*) n_pc_steps

!-----
if ((exp_a.EQ.0).AND.(upflow_angle.EQ.0).AND.(yaw_misalignment.EQ.0)) then
    b_stationary_sim=1
    n_time_steps=1
else
    upflow_angle=upflow_angle*pi/180
    yaw_misalignment=yaw_misalignment*pi/180
end if
wake_int_init=1
wake_loc_init=1
azimut=0
preazimut=0

call initialize
call read_input
call phase_between_blades
call place_elements
call extrude_polars
call from_cartesian_to_polars

inc_pc_step=(omega_max-omega_min)/n_pc_steps
allocate(pc_P(n_pc_steps+1))
allocate(pc_T(n_pc_steps+1))
allocate(pc_omega(n_pc_steps+1))
allocate(pc_wsppedm(n_pc_steps+1))
R=hs_loc_pol(1,n_horseshoe+1)
i=1
pc_P(:)=0.
do while ((pc_P(i).LT.P_max).AND.(i.LT.n_pc_steps+1))
    omega=omega_min+(i-1)*inc_pc_step
    wind_speed_module=omega*R/lambda

    v0(1)=wind_speed_module*cos(upflow_angle)*cos(yaw_misalignment)
    v0(2)=wind_speed_module*sin(yaw_misalignment)
    v0(3)=wind_speed_module*sin(upflow_angle)

    call calculations
    call postprocessing

```



```

    pc_P(i)=sum(P(:))/n_time_steps
    pc_T(i)=sum(T(:))/n_time_steps
    pc_omega(i)=omega
    pc_wspeedm(i)=wind_speed_module

    i=i+1
end do

open(UNIT=7,FILE='Results_Power_Curve_performance.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE'
)
do i=1,n_pc_steps
    write(7,*) pc_P(i), pc_T(i)
end do
close(7)
open(UNIT=7,FILE='Results_Power_Curve_speeds.txt',ACCESS='SEQUENTIAL',STATUS='REPLACE'
)
do i=1,n_pc_steps
    write(7,*) pc_omega(i), pc_wspeedm(i)
end do
close(7)

end subroutine power_curve
!-----
subroutine place_wake_sheet !former full_detail

use horseshoe_locations
use wake_sheet
use incident_speeds
use vortex_intensities
use aerodynamic_performance
use operating_parameters
use clock
use simulation_data
use geometric_inputs
implicit none

integer i,j,k,b,m, aux_wl,aux_ts
real azimuth_blade,azimut_wake,mat_az(3,3),mat_pitch(3,3),mat_c(3,3)
real (kind=8):: aux_te_loc(3,n_horseshoe+1)
real (kind=8):: aux_nodes_loc(3,(n_horseshoe+1)*n_wake_lines), aux_nodes_speeds(3,
(n_horseshoe+1)*n_wake_lines)
real (kind=8):: aux3(3), aux_pos(3), aux_pos_w1(3),aux_pos_w2(3),u,v,w
real (kind=8):: aux_x0_pos(n_time_steps,3,n_horseshoe+1), aux_speeds(3,n_horseshoe+1)

if (wake_loc_init.EQ.1) then
    !this routine is only done if any wake model hasn't been built before during the
    current simulation
    !pause
    do i=1,n_time_steps
        azimuth_blade=preazimut+omega*inc_time*(i-1)
        do j=1,n_wake_lines
            azimut_wake=azimut_blade-omega*inc_time*(j-1)

            mat_az(1,1)=1.
            mat_az(2,1)=0.
            mat_az(3,1)=0.
            mat_az(1,2)=0.
            mat_az(2,2)=-sin(azimut_wake)
            mat_az(3,2)=cos(azimut_wake)
            mat_az(1,3)=0.
            mat_az(2,3)=-cos(azimut_wake)
            mat_az(3,3)=-sin(azimut_wake)

            mat_pitch(2,1)=0.
            mat_pitch(1,2)=0.
            mat_pitch(2,2)=1.
            mat_pitch(3,2)=0.
            mat_pitch(2,3)=0.

            mat_c(1,1)=1.
            mat_c(2,1)=0.
            mat_c(3,1)=0.
            mat_c(1,2)=0.

```

```

mat_c(2,2)=1.
mat_c(3,2)=0.
mat_c(1,3)=0.
mat_c(2,3)=0.
mat_c(3,3)=-1.

mat_pitch(1,1)=cos(pitch+twist(1))
mat_pitch(3,1)=sin(pitch+twist(1))
mat_pitch(1,3)=-sin(pitch+twist(1))
mat_pitch(3,3)=cos(pitch+twist(1))

aux_pos(1)=0.
aux_pos(2)=horseshoe_te_loc(1,1)
aux_pos(3)=horseshoe_te_loc(2,1)
aux_pos=matmul(mat_c,aux_pos)
aux_pos=matmul(mat_pitch,aux_pos)
pr_wake_nodes_loc(i,:,1+(j-1)*(n_horseshoe+1))=matmul(mat_az,aux_pos)
pr_wake_nodes_loc(i,1,1+(j-1)*(n_horseshoe+1))=pr_wake_nodes_loc(i,1,1+(j-
1)*(n_horseshoe+1))+(j-1)*inc_time*v0(1)
pr_wake_nodes_loc(i,2,1+(j-1)*(n_horseshoe+1))=pr_wake_nodes_loc(i,2,1+(j-
1)*(n_horseshoe+1))+(j-1)*inc_time*v0(2)
pr_wake_nodes_loc(i,3,1+(j-1)*(n_horseshoe+1))=pr_wake_nodes_loc(i,3,1+(j-
1)*(n_horseshoe+1))+(j-1)*inc_time*v0(3)

mat_pitch(1,1)=cos(pitch+twist(n_horseshoe))
mat_pitch(3,1)=sin(pitch+twist(n_horseshoe))
mat_pitch(1,3)=-sin(pitch+twist(n_horseshoe))
mat_pitch(3,3)=cos(pitch+twist(n_horseshoe))

aux_pos(1)=0.
aux_pos(2)=horseshoe_te_loc(1,n_horseshoe+1)
aux_pos(3)=horseshoe_te_loc(2,n_horseshoe+1)
aux_pos=matmul(mat_c,aux_pos)
aux_pos=matmul(mat_pitch,aux_pos)
!print*,aux_pos,"pre"
pr_wake_nodes_loc(i,:,j*(n_horseshoe+1))=matmul(mat_az,aux_pos)
!print*,pr_wake_nodes_loc(i,:,j*(n_horseshoe+1)),"post"
pr_wake_nodes_loc(i,1,j*(n_horseshoe+1))=pr_wake_nodes_loc(i,1,j*
(n_horseshoe+1))+(j-1)*inc_time*v0(1)
pr_wake_nodes_loc(i,2,j*(n_horseshoe+1))=pr_wake_nodes_loc(i,2,j*
(n_horseshoe+1))+(j-1)*inc_time*v0(2)
pr_wake_nodes_loc(i,3,j*(n_horseshoe+1))=pr_wake_nodes_loc(i,3,j*
(n_horseshoe+1))+(j-1)*inc_time*v0(3)

do k=2,n_horseshoe
mat_pitch(1,1)=cos(pitch+0.5*(twist(k-1)+twist(k)))
mat_pitch(3,1)=sin(pitch+0.5*(twist(k-1)+twist(k)))
mat_pitch(1,3)=-sin(pitch+0.5*(twist(k-1)+twist(k)))
mat_pitch(3,3)=cos(pitch+0.5*(twist(k-1)+twist(k)))

aux_pos(1)=0.
aux_pos(2)=horseshoe_te_loc(1,k)
aux_pos(3)=horseshoe_te_loc(2,k)
aux_pos=matmul(mat_c,aux_pos)
aux_pos=matmul(mat_pitch,aux_pos)
pr_wake_nodes_loc(i,:,k+(j-1)*(n_horseshoe+1))=matmul(mat_az,aux_pos)
pr_wake_nodes_loc(i,1,k+(j-1)*(n_horseshoe+1))=pr_wake_nodes_loc(i,1,k
+(j-1)*(n_horseshoe+1))+(j-1)*inc_time*v0(1)
pr_wake_nodes_loc(i,2,k+(j-1)*(n_horseshoe+1))=pr_wake_nodes_loc(i,2,k
+(j-1)*(n_horseshoe+1))+(j-1)*inc_time*v0(2)
pr_wake_nodes_loc(i,3,k+(j-1)*(n_horseshoe+1))=pr_wake_nodes_loc(i,3,k
+(j-1)*(n_horseshoe+1))+(j-1)*inc_time*v0(3)
end do

end do
end do
!connectivity of the nodes in the wake for defining the rings
do i=1,n_horseshoe*(n_wake_lines-1)
wake_nodes_connec(1,i)=i
wake_nodes_connec(2,i)=i+int((i-0.5)/n_horseshoe)
wake_nodes_connec(3,i)=i+int((i-0.5)/n_horseshoe)+1
wake_nodes_connec(4,i)=i+int((i-0.5)/n_horseshoe)+1+n_horseshoe

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```

wake_nodes_connec(5,i)=i+int((i-0.5)/n_horseshoe)+2+n_horseshoe
end do

wake_loc_init=b_wake_roll_up+wake_loc_init
else if (wake_loc_init.EQ.2) then
!this calculation will be done only when the simulation time increases, since the
position of the wake is defined by its previous positions and the vorticity of the
blade and th
!and the wake itself at the previous instant

do i=1,n_wake_lines
!free stream contribution
call wind_speed_for_wake(pr_wake_nodes_loc(blade_phase(1,time_step+
b_stationary_sim),:(i-1)*(n_horseshoe+1)+1:i*(n_horseshoe+1)),aux_speeds)
do j=1,n_horseshoe+1
aux_nodes_speeds(1,(i-1)*(n_horseshoe+1)+j)=aux_speeds(1,j)
aux_nodes_speeds(2,(i-1)*(n_horseshoe+1)+j)=aux_speeds(2,j)
aux_nodes_speeds(3,(i-1)*(n_horseshoe+1)+j)=aux_speeds(3,j)
end do

!blade induced speed
do b=1,n_blades
do j=1,n_horseshoe+1
aux3(:)=0
aux_pos(1)=pr_wake_nodes_loc(blade_phase(b,time_step+b_stationary_sim)
,1,(i-1)*(n_horseshoe+1)+j)
aux_pos(2)=pr_wake_nodes_loc(blade_phase(b,time_step+b_stationary_sim)
,2,(i-1)*(n_horseshoe+1)+j)
aux_pos(3)=pr_wake_nodes_loc(blade_phase(b,time_step+b_stationary_sim)
,3,(i-1)*(n_horseshoe+1)+j)
do k=1,n_horseshoe
call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),temp_hs_te_loc(b,1,k)
,temp_hs_te_loc(b,2,k),temp_hs_te_loc(b,3,k),temp_hs_loc(b,1,k),temp_hs_loc(b,2,k),
temp_hs_loc(b,3,k),pr_attached_hs_int(blade_phase(b,time_step+b_stationary_sim),k),u,v
,w)

aux3(1)=aux3(1)+u
aux3(2)=aux3(2)+v
aux3(3)=aux3(3)+w
call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),temp_hs_loc(b,1,k),
temp_hs_loc(b,2,k),temp_hs_loc(b,3,k),temp_hs_loc(b,1,k+1),temp_hs_loc(b,2,k+1),
temp_hs_loc(b,3,k+1),pr_attached_hs_int(blade_phase(b,time_step+b_stationary_sim),k),u
,v,w)

aux3(1)=aux3(1)+u
aux3(2)=aux3(2)+v
aux3(3)=aux3(3)+w
call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),temp_hs_loc(b,1,k+1),
temp_hs_loc(b,2,k+1),temp_hs_loc(b,3,k+1),temp_hs_te_loc(b,1,k+1),temp_hs_te_loc(b,2,k
+1),temp_hs_te_loc(b,3,k+1),pr_attached_hs_int(blade_phase(b,time_step+
b_stationary_sim),k),u,v,w)

aux3(1)=aux3(1)+u
aux3(2)=aux3(2)+v
aux3(3)=aux3(3)+w
call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),temp_hs_te_loc(b,1,k+
1),temp_hs_te_loc(b,2,k+1),temp_hs_te_loc(b,3,k+1),temp_hs_te_loc(b,1,k),
temp_hs_te_loc(b,2,k),temp_hs_te_loc(b,3,k),pr_attached_hs_int(blade_phase(b,time_step
+b_stationary_sim),k),u,v,w)
!
!
!
aux3(1)=aux3(1)+u
aux3(2)=aux3(2)+v
aux3(3)=aux3(3)+w
end do ! loop for k--> from 1 to n_horseshoe contribution of each
attached ring
aux_nodes_speeds(1,(i-1)*(n_horseshoe+1)+j)=aux_nodes_speeds(1,(i-1)*
(n_horseshoe+1)+j)+aux3(1)
aux_nodes_speeds(2,(i-1)*(n_horseshoe+1)+j)=aux_nodes_speeds(2,(i-1)*
(n_horseshoe+1)+j)+aux3(2)
aux_nodes_speeds(3,(i-1)*(n_horseshoe+1)+j)=aux_nodes_speeds(3,(i-1)*
(n_horseshoe+1)+j)+aux3(3)
end do ! loop for j--> from 1 to n_horseshoe+1 effect on each node along
the i spanwise wake line
end do !loop for b=1,n_blades--> for adding the effect of all the blades

!wake self induced speed
do b=1,n_blades

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do j=1,n_horseshoe+1
  aux3(:)=0
  aux_pos(1)=pr_wake_nodes_loc(blade_phase(1,time_step+b_stationary_sim)
,1,(i-1)*(n_horseshoe+1)+j)
  aux_pos(2)=pr_wake_nodes_loc(blade_phase(1,time_step+b_stationary_sim)
,2,(i-1)*(n_horseshoe+1)+j)
  aux_pos(3)=pr_wake_nodes_loc(blade_phase(1,time_step+b_stationary_sim)
,3,(i-1)*(n_horseshoe+1)+j)
  do k=1,n_horseshoe*(n_wake_lines-1)
    aux_pos_w1(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(2,k))
    aux_pos_w1(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(2,k))
    aux_pos_w1(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(2,k))
    aux_pos_w2(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(3,k))
    aux_pos_w2(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(3,k))
    aux_pos_w2(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(3,k))
    call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),aux_pos_w1(1),
aux_pos_w1(2),aux_pos_w1(3),aux_pos_w2(1),aux_pos_w2(2),aux_pos_w2(3),pr_wake_int
(blade_phase(b,time_step+b_stationary_sim),k),u,v,w)
    aux3(1)=aux3(1)+u
    aux3(2)=aux3(2)+v
    aux3(3)=aux3(3)+w

    aux_pos_w1(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(3,k))
    aux_pos_w1(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(3,k))
    aux_pos_w1(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(3,k))
    aux_pos_w2(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(5,k))
    aux_pos_w2(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(5,k))
    aux_pos_w2(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(5,k))
    call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),aux_pos_w1(1),
aux_pos_w1(2),aux_pos_w1(3),aux_pos_w2(1),aux_pos_w2(2),aux_pos_w2(3),pr_wake_int
(blade_phase(b,time_step+b_stationary_sim),k),u,v,w)
    aux3(1)=aux3(1)+u
    aux3(2)=aux3(2)+v
    aux3(3)=aux3(3)+w

    aux_pos_w1(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(5,k))
    aux_pos_w1(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(5,k))
    aux_pos_w1(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(5,k))
    aux_pos_w2(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(4,k))
    aux_pos_w2(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(4,k))
    aux_pos_w2(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(4,k))
    call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),aux_pos_w1(1),
aux_pos_w1(2),aux_pos_w1(3),aux_pos_w2(1),aux_pos_w2(2),aux_pos_w2(3),pr_wake_int
(blade_phase(b,time_step+b_stationary_sim),k),u,v,w)
    aux3(1)=aux3(1)+u
    aux3(2)=aux3(2)+v
    aux3(3)=aux3(3)+w

    aux_pos_w1(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(4,k))
    aux_pos_w1(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(4,k))
    aux_pos_w1(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(4,k))
    aux_pos_w2(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(2,k))

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        aux_pos_w2(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(2,k))
        aux_pos_w2(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(2,k))
        call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),aux_pos_w1(1),
aux_pos_w1(2),aux_pos_w1(3),aux_pos_w2(1),aux_pos_w2(2),aux_pos_w2(3),pr_wake_int
(blade_phase(b,time_step+b_stationary_sim),k),u,v,w)
        aux3(1)=aux3(1)+u
        aux3(2)=aux3(2)+v
        aux3(3)=aux3(3)+w

        end do !loop for k=1,n_horseshoe*(n_wake_lines-1)--> for adding the
effect of all the wake rings
        aux_nodes_speeds(1,(i-1)*(n_horseshoe+1)+j)=aux_nodes_speeds(1,(i-1)*
(n_horseshoe+1)+j)+aux3(1)
        aux_nodes_speeds(2,(i-1)*(n_horseshoe+1)+j)=aux_nodes_speeds(2,(i-1)*
(n_horseshoe+1)+j)+aux3(2)
        aux_nodes_speeds(3,(i-1)*(n_horseshoe+1)+j)=aux_nodes_speeds(3,(i-1)*
(n_horseshoe+1)+j)+aux3(3)
        end do !loop for j=1,n_horseshoe+1--> for calculating the velocities on
each wake node along a spanwise wake line
        end do !loop fo b=1,n_blades--> for adding the effect of the wakes of each
blade
        end do !loop for i --> from 1 to n_wake_lines for calculating the effect on each
wake line

        do i=(n_horseshoe+1)*n_wake_lines,n_horseshoe+2,-1
            pr_wake_nodes_loc(time_step,1,i)=pr_wake_nodes_loc(blade_phase(1,time_step+
b_stationary_sim),1,i-(n_horseshoe+1))+inc_time*aux_nodes_speeds(1,i)
            pr_wake_nodes_loc(time_step,2,i)=pr_wake_nodes_loc(blade_phase(1,time_step+
b_stationary_sim),2,i-(n_horseshoe+1))+inc_time*aux_nodes_speeds(2,i)
            pr_wake_nodes_loc(time_step,3,i)=pr_wake_nodes_loc(blade_phase(1,time_step+
b_stationary_sim),3,i-(n_horseshoe+1))+inc_time*aux_nodes_speeds(3,i)
        end do !loop for i=identification number of the first node of the second line to
the last of the last line, in order to update the position

        do i=1,n_horseshoe+1
            pr_wake_nodes_loc(time_step,1,i)=temp_hs_te_loc(1,1,i)
            pr_wake_nodes_loc(time_step,2,i)=temp_hs_te_loc(1,2,i)
            pr_wake_nodes_loc(time_step,3,i)=temp_hs_te_loc(1,3,i)
        end do

    end if

!created 19/01/2013
!not validated
end subroutine place_wake_sheet !former full_detail
!-----
!this subroutine is an attemp to reduce the simulation time needed to build the wake
geometry
subroutine place_wake_sheet_reduced

    use horseshoe_locations
    use wake_sheet
    use incident_speeds
    use vortex_intensities
    use aerodynamic_performance
    use operating_parameters
    use clock
    use simulation_data
    use geometric_inputs
    implicit none

    integer i,j,k,b,m,n, aux_wl,aux_ts, lim1,lim2
    real azimuth_blade,azimut_wake,mat_az(3,3),mat_pitch(3,3),mat_c(3,3)
    real (kind=8):: aux_te_loc(3,n_horseshoe+1)
    real (kind=8):: aux_nodes_loc(3,(n_horseshoe+1)*n_wake_lines), aux_nodes_speeds(3,
(n_horseshoe+1)*n_wake_lines)
    real (kind=8):: aux3(3), aux_pos(3), aux_pos_w1(3),aux_pos_w2(3),u,v,w
    real (kind=8):: aux_x0_pos(n_time_steps,3,n_horseshoe+1), aux_speeds(3,n_horseshoe+1)

    if (wake_loc_init.EQ.1) then

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```

!this routine is only done if any wake model hasn't been built before during the
current simulation
!pause
do i=1,n_time_steps
  azimuth_blade=preazimuth+omega*inc_time*(i-1)
  do j=1,n_wake_lines
    azimuth_wake=azimuth_blade-omega*inc_time*(j-1)

    mat_az(1,1)=1.
    mat_az(2,1)=0.
    mat_az(3,1)=0.
    mat_az(1,2)=0.
    mat_az(2,2)=-sin(azimuth_wake)
    mat_az(3,2)=cos(azimuth_wake)
    mat_az(1,3)=0.
    mat_az(2,3)=-cos(azimuth_wake)
    mat_az(3,3)=-sin(azimuth_wake)

    mat_pitch(2,1)=0.
    mat_pitch(1,2)=0.
    mat_pitch(2,2)=1.
    mat_pitch(3,2)=0.
    mat_pitch(2,3)=0.

    mat_c(1,1)=1.
    mat_c(2,1)=0.
    mat_c(3,1)=0.
    mat_c(1,2)=0.
    mat_c(2,2)=1.
    mat_c(3,2)=0.
    mat_c(1,3)=0.
    mat_c(2,3)=0.
    mat_c(3,3)=-1.

    mat_pitch(1,1)=cos(pitch+twist(1))
    mat_pitch(3,1)=sin(pitch+twist(1))
    mat_pitch(1,3)=-sin(pitch+twist(1))
    mat_pitch(3,3)=cos(pitch+twist(1))

    aux_pos(1)=0.
    aux_pos(2)=horseshoe_te_loc(1,1)
    aux_pos(3)=horseshoe_te_loc(2,1)
    aux_pos=matmul(mat_c,aux_pos)
    aux_pos=matmul(mat_pitch,aux_pos)
    pr_wake_nodes_loc(i,:,1+(j-1)*(n_horseshoe+1))=matmul(mat_az,aux_pos)
    pr_wake_nodes_loc(i,1,1+(j-1)*(n_horseshoe+1))=pr_wake_nodes_loc(i,1,1+(j-
1)*(n_horseshoe+1))+(j-1)*inc_time*v0(1)
    pr_wake_nodes_loc(i,2,1+(j-1)*(n_horseshoe+1))=pr_wake_nodes_loc(i,2,1+(j-
1)*(n_horseshoe+1))+(j-1)*inc_time*v0(2)
    pr_wake_nodes_loc(i,3,1+(j-1)*(n_horseshoe+1))=pr_wake_nodes_loc(i,3,1+(j-
1)*(n_horseshoe+1))+(j-1)*inc_time*v0(3)

    mat_pitch(1,1)=cos(pitch+twist(n_horseshoe))
    mat_pitch(3,1)=sin(pitch+twist(n_horseshoe))
    mat_pitch(1,3)=-sin(pitch+twist(n_horseshoe))
    mat_pitch(3,3)=cos(pitch+twist(n_horseshoe))

    aux_pos(1)=0.
    aux_pos(2)=horseshoe_te_loc(1,n_horseshoe+1)
    aux_pos(3)=horseshoe_te_loc(2,n_horseshoe+1)
    aux_pos=matmul(mat_c,aux_pos)
    aux_pos=matmul(mat_pitch,aux_pos)
    !print*,aux_pos,"pre"
    pr_wake_nodes_loc(i,:,j*(n_horseshoe+1))=matmul(mat_az,aux_pos)
    !print*,pr_wake_nodes_loc(i,:,j*(n_horseshoe+1)),"post"
    pr_wake_nodes_loc(i,1,j*(n_horseshoe+1))=pr_wake_nodes_loc(i,1,j*
(n_horseshoe+1))+(j-1)*inc_time*v0(1)
    pr_wake_nodes_loc(i,2,j*(n_horseshoe+1))=pr_wake_nodes_loc(i,2,j*
(n_horseshoe+1))+(j-1)*inc_time*v0(2)
    pr_wake_nodes_loc(i,3,j*(n_horseshoe+1))=pr_wake_nodes_loc(i,3,j*
(n_horseshoe+1))+(j-1)*inc_time*v0(3)

    do k=2,n_horseshoe

```

```

mat_pitch(1,1)=cos(pitch+0.5*(twist(k-1)+twist(k)))
mat_pitch(3,1)=sin(pitch+0.5*(twist(k-1)+twist(k)))
mat_pitch(1,3)=-sin(pitch+0.5*(twist(k-1)+twist(k)))
mat_pitch(3,3)=cos(pitch+0.5*(twist(k-1)+twist(k)))

aux_pos(1)=0.
aux_pos(2)=horseshoe_te_loc(1,k)
aux_pos(3)=horseshoe_te_loc(2,k)
aux_pos=matmul(mat_c,aux_pos)
aux_pos=matmul(mat_pitch,aux_pos)
pr_wake_nodes_loc(i,:,k+(j-1)*(n_horseshoe+1))=matmul(mat_az,aux_pos)
pr_wake_nodes_loc(i,1,k+(j-1)*(n_horseshoe+1))=pr_wake_nodes_loc(i,1,k
+(j-1)*(n_horseshoe+1))+(j-1)*inc_time*v0(1)
pr_wake_nodes_loc(i,2,k+(j-1)*(n_horseshoe+1))=pr_wake_nodes_loc(i,2,k
+(j-1)*(n_horseshoe+1))+(j-1)*inc_time*v0(2)
pr_wake_nodes_loc(i,3,k+(j-1)*(n_horseshoe+1))=pr_wake_nodes_loc(i,3,k
+(j-1)*(n_horseshoe+1))+(j-1)*inc_time*v0(3)
end do

end do
end do
!print*,pr_wake_nodes_loc(1,:,:)
!pause
!connectivity of the nodes in the wake for defining the rings
do i=1,n_horseshoe*(n_wake_lines-1)
wake_nodes_convec(1,i)=i
wake_nodes_convec(2,i)=i+int((i-0.5)/n_horseshoe)
wake_nodes_convec(3,i)=i+int((i-0.5)/n_horseshoe)+1
wake_nodes_convec(4,i)=i+int((i-0.5)/n_horseshoe)+1+n_horseshoe
wake_nodes_convec(5,i)=i+int((i-0.5)/n_horseshoe)+2+n_horseshoe
end do

wake_loc_init=b_wake_roll_up+wake_loc_init
else if (wake_loc_init.EQ.2) then
!pause
!this calculation will be done only when the simulation time increases, since the
position of the wake is defined by its previous positions and the vorticity of the
blade and th
!and the wake itself at the previous instant

do i=1,n_wake_lines
!free stream contribution
call wind_speed_for_wake(pr_wake_nodes_loc(blade_phase(1,time_step+
b_stationary_sim),:,(i-1)*(n_horseshoe+1)+1:i*(n_horseshoe+1)),aux_speeds)
do j=1,n_horseshoe+1
aux_nodes_speeds(1,(i-1)*(n_horseshoe+1)+j)=aux_speeds(1,j)
aux_nodes_speeds(2,(i-1)*(n_horseshoe+1)+j)=aux_speeds(2,j)
aux_nodes_speeds(3,(i-1)*(n_horseshoe+1)+j)=aux_speeds(3,j)
end do

!blade induced speed
do b=1,n_blades
do j=1,n_horseshoe+1
aux3(:)=0
aux_pos(1)=pr_wake_nodes_loc(blade_phase(b,time_step+b_stationary_sim)
,1,(i-1)*(n_horseshoe+1)+j)
aux_pos(2)=pr_wake_nodes_loc(blade_phase(b,time_step+b_stationary_sim)
,2,(i-1)*(n_horseshoe+1)+j)
aux_pos(3)=pr_wake_nodes_loc(blade_phase(b,time_step+b_stationary_sim)
,3,(i-1)*(n_horseshoe+1)+j)
do k=1,n_horseshoe
call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),temp_hs_te_loc(b,1,k)
,temp_hs_te_loc(b,2,k),temp_hs_te_loc(b,3,k),temp_hs_loc(b,1,k),temp_hs_loc(b,2,k),
temp_hs_loc(b,3,k),pr_attached_hs_int(blade_phase(b,time_step+b_stationary_sim),k),u,v
,w)

aux3(1)=aux3(1)+u
aux3(2)=aux3(2)+v
aux3(3)=aux3(3)+w
call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),temp_hs_loc(b,1,k),
temp_hs_loc(b,2,k),temp_hs_loc(b,3,k),temp_hs_loc(b,1,k+1),temp_hs_loc(b,2,k+1),
temp_hs_loc(b,3,k+1),pr_attached_hs_int(blade_phase(b,time_step+b_stationary_sim),k),u

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,v,w)
        aux3(1)=aux3(1)+u
        aux3(2)=aux3(2)+v
        aux3(3)=aux3(3)+w
        call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),temp_hs_loc(b,1,k+1),
temp_hs_loc(b,2,k+1),temp_hs_loc(b,3,k+1),temp_hs_te_loc(b,1,k+1),temp_hs_te_loc(b,2,k
+1),temp_hs_te_loc(b,3,k+1),pr_attached_hs_int(blade_phase(b,time_step+
b_stationary_sim),k),u,v,w)
        aux3(1)=aux3(1)+u
        aux3(2)=aux3(2)+v
        aux3(3)=aux3(3)+w
!        call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),temp_hs_te_loc(b,1,k+
1),temp_hs_te_loc(b,2,k+1),temp_hs_te_loc(b,3,k+1),temp_hs_te_loc(b,1,k),
temp_hs_te_loc(b,2,k),temp_hs_te_loc(b,3,k),pr_attached_hs_int(blade_phase(b,time_step
+b_stationary_sim),k),u,v,w)
!
!        aux3(1)=aux3(1)+u
!        aux3(2)=aux3(2)+v
!        aux3(3)=aux3(3)+w
        end do ! loop for k--> from 1 to n_horseshoe contribution of each
attached ring
        aux_nodes_speeds(1,(i-1)*(n_horseshoe+1)+j)=aux_nodes_speeds(1,(i-1)*
(n_horseshoe+1)+j)+aux3(1)
        aux_nodes_speeds(2,(i-1)*(n_horseshoe+1)+j)=aux_nodes_speeds(2,(i-1)*
(n_horseshoe+1)+j)+aux3(2)
        aux_nodes_speeds(3,(i-1)*(n_horseshoe+1)+j)=aux_nodes_speeds(3,(i-1)*
(n_horseshoe+1)+j)+aux3(3)
        end do ! loop for j--> from 1 to n_horseshoe+1 effect on each node along
the i spanwise wake line
        end do !loop for b=1,n_blades--> for adding the effect of all the blades

!wake self induced speed
!do b=1,n_blades
b=1
        do j=1,n_horseshoe+1
            aux3(:)=0
            aux_pos(1)=pr_wake_nodes_loc(blade_phase(1,time_step+b_stationary_sim)
,1,(i-1)*(n_horseshoe+1)+j)
            aux_pos(2)=pr_wake_nodes_loc(blade_phase(1,time_step+b_stationary_sim)
,2,(i-1)*(n_horseshoe+1)+j)
            aux_pos(3)=pr_wake_nodes_loc(blade_phase(1,time_step+b_stationary_sim)
,3,(i-1)*(n_horseshoe+1)+j)
            do n=-min(i-1,10),min(n_wake_lines-i,10)
                lim1=max(i-1+n,0)*(n_horseshoe+1)+max(j-10,1)
                lim2=max(i-1+n,0)*(n_horseshoe+1)+min(j+10,n_horseshoe+1)
                do k=1,n_horseshoe*(n_wake_lines-1)
                    aux_pos_w1(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(2,k))
                    aux_pos_w1(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(2,k))
                    aux_pos_w1(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(2,k))
                    aux_pos_w2(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(3,k))
                    aux_pos_w2(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(3,k))
                    aux_pos_w2(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(3,k))
                    call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),aux_pos_w1(1),
aux_pos_w1(2),aux_pos_w1(3),aux_pos_w2(1),aux_pos_w2(2),aux_pos_w2(3),pr_wake_int
(blade_phase(b,time_step+b_stationary_sim),k),u,v,w)
                    aux3(1)=aux3(1)+u
                    aux3(2)=aux3(2)+v
                    aux3(3)=aux3(3)+w

                    aux_pos_w1(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(3,k))
                    aux_pos_w1(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(3,k))
                    aux_pos_w1(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(3,k))
                    aux_pos_w2(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(5,k))
                    aux_pos_w2(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(5,k))

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        aux_pos_w2(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(5,k))
        call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),aux_pos_w1(1),
aux_pos_w1(2),aux_pos_w1(3),aux_pos_w2(1),aux_pos_w2(2),aux_pos_w2(3),pr_wake_int
(blade_phase(b,time_step+b_stationary_sim),k),u,v,w)
        aux3(1)=aux3(1)+u
        aux3(2)=aux3(2)+v
        aux3(3)=aux3(3)+w

        aux_pos_w1(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(5,k))
        aux_pos_w1(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(5,k))
        aux_pos_w1(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(5,k))
        aux_pos_w2(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(4,k))
        aux_pos_w2(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(4,k))
        aux_pos_w2(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(4,k))
        call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),aux_pos_w1(1),
aux_pos_w1(2),aux_pos_w1(3),aux_pos_w2(1),aux_pos_w2(2),aux_pos_w2(3),pr_wake_int
(blade_phase(b,time_step+b_stationary_sim),k),u,v,w)
        aux3(1)=aux3(1)+u
        aux3(2)=aux3(2)+v
        aux3(3)=aux3(3)+w

        aux_pos_w1(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(4,k))
        aux_pos_w1(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(4,k))
        aux_pos_w1(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(4,k))
        aux_pos_w2(1)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),1,wake_nodes_connec(2,k))
        aux_pos_w2(2)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),2,wake_nodes_connec(2,k))
        aux_pos_w2(3)=pr_wake_nodes_loc(blade_phase(b,time_step+
b_stationary_sim),3,wake_nodes_connec(2,k))
        call VLtfnc(aux_pos(1),aux_pos(2),aux_pos(3),aux_pos_w1(1),
aux_pos_w1(2),aux_pos_w1(3),aux_pos_w2(1),aux_pos_w2(2),aux_pos_w2(3),pr_wake_int
(blade_phase(b,time_step+b_stationary_sim),k),u,v,w)
        aux3(1)=aux3(1)+u
        aux3(2)=aux3(2)+v
        aux3(3)=aux3(3)+w

        end do !loop for k=1,n_horseshoe*(n_wake_lines-1)--> for adding
the effect of all the wake rings
        end do !loop for n ---> reducing the number of influencing nodes
        aux_nodes_speeds(1,(i-1)*(n_horseshoe+1)+j)=aux_nodes_speeds(1,(i-1)*
(n_horseshoe+1)+j)+aux3(1)
        aux_nodes_speeds(2,(i-1)*(n_horseshoe+1)+j)=aux_nodes_speeds(2,(i-1)*
(n_horseshoe+1)+j)+aux3(2)
        aux_nodes_speeds(3,(i-1)*(n_horseshoe+1)+j)=aux_nodes_speeds(3,(i-1)*
(n_horseshoe+1)+j)+aux3(3)
        end do !loop for j=1,n_horseshoe+1--> for calculating the velocities on
each wake node along a spanwise wake line
        !end do !loop fo b=1,n_blades--> for adding the effect of the wakes of each
blade
        end do !loop for i --> from 1 to n_wake_lines for calculating the effect on each
wake line

        do i=(n_horseshoe+1)*n_wake_lines,n_horseshoe+2,-1
        pr_wake_nodes_loc(time_step,1,i)=pr_wake_nodes_loc(blade_phase(1,time_step+
b_stationary_sim),1,i-(n_horseshoe+1))+inc_time*aux_nodes_speeds(1,i)
        pr_wake_nodes_loc(time_step,2,i)=pr_wake_nodes_loc(blade_phase(1,time_step+
b_stationary_sim),2,i-(n_horseshoe+1))+inc_time*aux_nodes_speeds(2,i)
        pr_wake_nodes_loc(time_step,3,i)=pr_wake_nodes_loc(blade_phase(1,time_step+
b_stationary_sim),3,i-(n_horseshoe+1))+inc_time*aux_nodes_speeds(3,i)
        end do !loop for i=identification number of the first node of the second line to
the last of the last line, in order to update the position

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```
do i=1,n_horseshoe+1
  pr_wake_nodes_loc(time_step,1,i)=temp_hs_te_loc(1,1,i)
  pr_wake_nodes_loc(time_step,2,i)=temp_hs_te_loc(1,2,i)
  pr_wake_nodes_loc(time_step,3,i)=temp_hs_te_loc(1,3,i)
end do

end if

!created 19/01/2013
!not validated
end subroutine place_wake_sheet_reduced
```